

# **The Shape Synthesis of Antenna Decoupling Surfaces**

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## ABSTRACT

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Although multi-element antenna (MEA) systems are already used in some modern wireless communication systems, the issue of mutual coupling between elements remains a challenge during MEA system design. Indeed, communications engineers continue to bemoan the fact that the antenna elements are often still designed with such coupling as an afterthought. Thus, some authors have used the decoupling surface (DS) idea, whereby a separate DS is added to the MEA systems to reduce the above coupling. Whereas a DS may indeed lower the coupling levels between the elements of a given MEA system, it usually changes the other performance parameters as well, and in an undesirable way. Thus, this design route is a complicated one that is not easily effected. In this thesis we propose, for the first time, a new design process for MEA systems based on shape synthesis. The MEA system performance indicators are combined into an objective that sets the goal of the shape synthesis procedure. The application of the proposed design process is illustrated for three different geometrical arrangements of patch antennas and decoupling surfaces. This confirms the efficacy of the new design method.

**Keywords:** multi-element antennas, MIMO, mutual coupling, decoupling, isolation, total active reflection coefficient (TARC), envelope correlation coefficient (ECC).

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# Chapter 1 Introduction

## 1.1 Introductory Comments

Multiple-input multiple-output (MIMO) systems<sup>1</sup> have been a topic of great interest in the field of wireless communications. It has been shown, theoretically and through testing, that MIMO systems make it possible to exceed the theoretical limit of channel capacity predicted by the conventional form of the Shannon - Hartley formula. The point of reference here is the communication channel, when referring to MIMO we refer to multiple signals entering the communication channel and multiple signals *output* at the receiver end. The key advantage offered by such systems is that the increase in capacity that they offer does not come at the expense of increased consumption of resources such as power and bandwidth. Wireless communication channels have many impairments. Line-of-sight (LOS) propagation channels suffer from path loss and interference. With portable wireless user equipment being widely available, robust and reliable communication in non line-of-sight (NLOS) environments is needed. In addition to suffering from the LOS impairments mentioned, NLOS communications also suffer from significant fluctuations of the received signal. In NLOS environments propagating signals are obstructed by manmade or natural obstacles. This causes the signal to suffer reflection, refraction and diffraction by material objects (eg. buildings), processes collectively referred to as scattering. The scattering phenomena cause multiple copies of the same signal to arrive at the receiver; these interfere with each other constructively or destructively in a random manner [FRAN 08]. Consequently, the power level of the received signal fluctuates at the receiver and in some instances the received signal becomes too weak to be retrieved. The received signal is said to be experiencing fading.

Prior to the invention of MIMO, most researchers focused on ways to combat fading in propagation environments, as multipath fading was the main culprit in causing degradation of the performance of traditional single input single output (SISO) systems. However, this view of multipath fading changed in 1996 when it was shown that the use of multi-element antenna (MEA) systems using multiple element antennas (MEA) for transmission and reception in a Rayleigh fading environment

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<sup>1</sup> Although such ideas have become familiar, some technical detail on the topic of MIMO systems is provided for convenience in Appendix A.

can result in a linear increase in channel capacity [FOSC 96]. The work presented in [FOSC 96], combined with the powerful signal processing circuitry available today, made it possible to implement MIMO.

The use of antenna diversity at the receiver end to decrease the probability of fading had been known for a long time. Already in the 1930's, Beverage combined the output of three receiving antennas in order to increase the signal to noise ratio (SNR) by mitigating the effects of fading [BEVE 31]. Modern diversity schemes for SISO systems include time diversity and frequency diversity. In time diversity, the same signal is sent in different time slots, thus reducing the probability of fading as the channel behavior is expected to change over time. Frequency diversity is effectively sending the same signal using different frequencies; this is based on the idea that different frequencies have different scattering characteristics. Modern MIMO systems make use of fading in order to increase the channel capacity by exciting different independent sub-channels in the communication channel. Nonetheless, if the communication sub-channels are not sufficiently uncorrelated or independent, the MIMO system will not function properly as signal loss could occur.

The channel capacity of a SISO system is mainly affected by the antenna gain. However, that is not the case when one is designing antennas for MIMO since the system's capacity is also dependent on the way the antennas excite the propagation channel [JENS 04]. As such, the cost of achieving this increased capacity is the necessary development and engineering of more complex systems. In addition to accounting for "classical" antenna performance parameters such as radiation patterns, radiation efficiency, and bandwidth, the antenna designer has to design for MIMO specific performance indicators such as low correlation<sup>2</sup>. As will be shown in this thesis, all such requirements can be met simultaneously using shape synthesis.

## 1.2 Multi-Element Antenna Systems

The IEEE defines an array antenna as "an antenna comprised of a number of radiating elements the inputs (or outputs) of which are combined" [IEEE 13]. Although this definition can be considered to incorporate what is meant by a multi-element antenna (MEA) system, it will prevent confusion if we clarify the difference between a "conventional" (multi-element) array antenna and

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<sup>2</sup> More will be said about this in Section 1.4, and then in much more detail in Section 2.3.

what is usually meant by a multi-element antenna system. In an array antenna the elements typically used in concert, through combination of the element ports via a beamforming network (BFN), to obtain a single-port antenna with a gain higher than that offered by individual elements. The entire array is used to receive or transmit the same signal. However, this is not the case in MEA systems where the elements are designed to operate on independent signals [VAUG 03]. In an MEA system each element is physically separated from the other elements and has a dedicated port. Each element can even have distinct characteristics (polarizations, radiation patterns, and so on).

In addition to being strong candidates for 5G communication applications, MEA systems have been a topic of interest among radar design engineers. MIMO radars use MEA to transmit and receive independent waveforms thereby exploiting the system's diversity. It has been shown that MIMO radars demonstrate superior performance in terms of resolution, and radar cross section fluctuation [FISH 04] [XU 07]. Often, antennas meant for radar applications are well spaced, therefore mitigating the coupling effect does not pose a challenge. Therefore, this thesis will only focus on the decoupling of the elements in MEA systems used for wireless communications.

### **1.3 Coupling in Multi-Element Antenna Systems**

Consider a two-element antenna system (consisting of antennas "1" and "2"), where the elements are placed near to each other. When antenna "1" transmits a signal, assuming a "good" antenna design with reasonable port-to-port isolation, most of the signal will be radiated into the environment. However, some of the signal will be incident on the adjacent transmit antenna. A portion of this incident field will make its way into the transmission line feed of antenna "2", and affect its performance metrics. Similarly, some of this incident field on "2" will be re-scattered off the structure of antenna "2" and some of this re-scattered field will be incident back on antenna "1" even though most will be transmitted into the environment. This unintentional scattering results in mutual coupling between the antennas [ALLE 66] [CHEN 18]. If we extend the above to an N-element antenna system, we can see that the unwanted coupling into any one element will be stronger as the signals would scatter from N-1 adjacent antennas. In other words, the mutual coupling would occur between antennas 1 and 2, 1 and 3, 2 and 3, and so on. The scaling for mutual coupling in N-element MEA systems is true for both transmitting and receiving modes. Generally, mutual coupling in MEA systems is a function of the relative spacing of the antennas

and the number of antenna elements. Additional factors which affect the mutual coupling include the relative arrangements of the antennas as well as the relative orientation of the elements [BALA 16].

In this thesis we will consider MEA systems whose elements are microstrip patches. Coupling between microstrip patch elements occurs due to the excitation of space waves and surface waves [BALA 16]. Space waves are waves which are radiated into free space (become detached from the substrate) [BHAT 90] and have an inverse distance decay rate [BALA 16]. The other source of coupling in microstrip patch elements are surface waves, which are spawned as a result of the excitation of guided waves [JACK 93] that remain “attached” to the substrate. Surface waves typically have a decay rate of the inverse square root of the distance [BHAT 90]. Due to the different decay variation rates, mutual coupling is dominated by space waves for small inter-patch spacing, whereas surface waves are more dominant at larger separation distances [BALA 16]. A measurement campaign conducted in [JEDI 81] for L-band microstrip antennas shows that space wave coupling is more dominant in electrically thin substrates.

## 1.4 Multi-Element Antenna System Design

The need for MEA systems, by which is meant the use of a structure that has multiple beamports, has been adequately motivated in [FRAN 08]. Their use in MIMO applications is currently the prevalent interest. MEA system designs have several performance metrics. Traditional metrics include, radiation efficiency, and port-to-port isolation. Non-traditional metrics<sup>3</sup> such as more severe space constraints often dictate that the antennas be very closely spaced. The need for design compactness is driven by the ever-increasing portability and increasing computational ability<sup>4</sup> of user mobile equipment. From a design perspective, the compactness of the design is measured in the literature as the center to center separation between the adjacent antenna elements and it is quoted in terms of wavelength  $\lambda$ . Ease of manufacturing and scalability to massive MIMO are other characteristics by which a design is judged. Some traditional metrics such as the scattering parameter  $S_{11}$  at element ports are still crucial in the evaluation process, but do not fully capture the performance of the antenna elements within a MIMO implementation, for instance. Other

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<sup>3</sup> To be discussed in Chapter 2.

<sup>4</sup> That permits complex signal processing to be used to affect the communication link.

metrics such as the total active reflection coefficient (TARC) are calculated instead, but TARC calculations are still based on the element  $S_{11}$  values in combination with the other scattering parameters. The properties of the antennas, the physical channel, and the complex signal processing needed, determine the eventual capacity of the MIMO communications link. One thing that is agreed upon, is that the coupling between the beamports of an MEA system should be low. Antenna designers have consequently devised many ingenious ways of achieving this, as will be discussed in Section 2.4. However, to the best of the author's knowledge, MEA design for MIMO applications that use shape synthesis (as opposed to the optimisation of a set of dimensions on a pre-selected geometry) has not been cited in the literature.

## 1.5 Overview of the Thesis

The research described in this thesis investigates the use of shape synthesis for the design of MEA antennas (the elements proper, plus decoupling surfaces) with decreased coupling.

Chapter 2 assembles the key technical concepts that are needed to conduct the research described in the thesis, and upon which it builds. Section 2.2 clarifies what we mean by “shape synthesis” in the context of this thesis. Section 2.3 covers the performance metrics which will be needed to evaluate the suitability of a structure as a MIMO MEA system, and identifies the technical background behind the need for low mutual coupling between the elements in such MEA system designs. Section 2.4 summarizes the different approaches that have been utilized by other researchers to reduce such coupling. Section 2.5 identifies the computational tool to be used in the present research. The chapter is concluded in Section 2.6.

The principal contributions of the present work are contained in Chapters 3 and 4. Each describes a different class of decoupling shape synthesis problem. Chapter 3 considers an MEA system consisting of two different arrangements of two patch elements, with the decoupling surface inserted between the two patch elements on the same plane as the elements. Chapter 4 extends the shape synthesis to the case of a 4-element MEA system, also consisting of patch elements, but where the decoupling surface is placed on a plane above (albeit parallel to) that in which the four elements lie. We successfully show that, in order to control the input reflection coefficient, TARC,

coupling, envelope correlation coefficient and radiation pattern shape properties of the elements in the resulting MEA system it is necessary to shape both the decoupling surface and the elements.

Chapter 5 summarises the contributions of the thesis work.

# Chapter 2 Review of Key Background Concepts & Techniques

## 2.1 Introduction

We are interested in the shape synthesis of multi-element antenna systems with decoupling surfaces. In order to place the work of this thesis in the context of previous work on the topic, and so to demonstrate that it has made a recognizable contribution to the field, a review of approaches that have been used by others to decrease coupling in multi-element antenna systems is provided in Section 2.4. Before this can be done with clarity, and also as background to the work described in Chapter 3 and Chapter 4, Section 2.3 defines the essential performance measures that will be used in the thesis. Section 2.2 briefly clarifies more precisely what is meant by the term “shape synthesis” in this thesis. Section 2.5 specifies the computational tool that will be harnessed in the thesis to achieve the desired shape synthesis. Section 2.6 concludes the chapter.

## 2.2 Clarification of the Meaning of “Shape Synthesis”

The most general synthesis methods, if they existed, would take the information on the functionality desired for an antenna and provide us with the shape of the antenna and its material properties. Practicalities would mandate that it does so subject to volume/area (“footprint”) constraints, restrictions on the types of materials actually available, and no over-sensitivity to small material-property and dimension changes. Stutzman and Licul [STUT 08] correctly observe that, in fact, such general antenna synthesis methods do not exist. Nevertheless, less ambitious but eminently useful synthesis techniques are already in use or under development.

As outlined in [ETHI 14, Sect. I], these include excitation optimization, size optimization (which includes traditional design methods), shape optimization [JOHN 99] (sometimes called topology optimization or inverse design in other engineering disciplines), and material optimization. By shape optimization is meant the shaping of the actual antenna geometry, and not merely the adjustment of the values of a defined set of geometrical features on a pre-selected shape. The shape optimization of individual microstrip antennas, with the conductor geometry undergoing shaping through the removal or retention of conducting pixels/cells into which some starting shape is divided, has allowed both miniaturization and bandwidth increases.

## 2.3 Multi-Element Antenna System Figures of Merit

### 2.3.1 Introductory Comments

In this section we cover the most important antenna figures of merit which we will need to build and evaluate the shape synthesis process as a means of designing so-called decoupling surfaces to decrease the coupling between antennas in an MEA System. Many of these figures of merit are inherited from single antenna applications, whereas others are specific to MEAs, including MEAs used in MIMO applications. We begin by establishing some of the fundamental antenna performance indicators. We then derive, comment on, and validate the performance metrics used in the literature for evaluating MEAs.

### 2.3.2 Far-Zone Fields & Radiation Patterns

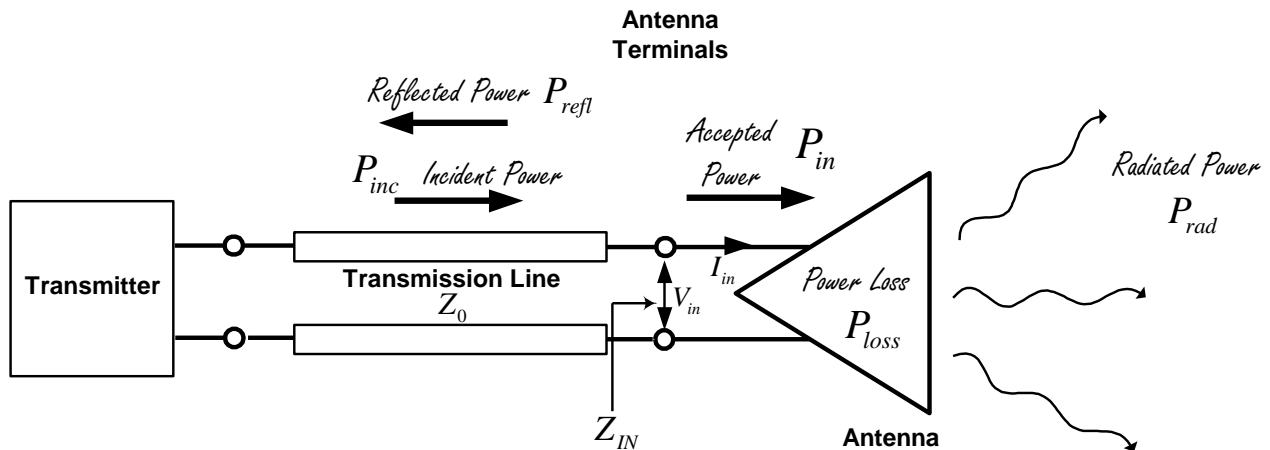


Figure 2.3-1: Systems view of an antenna (Diagram courtesy of D.A.McNamara)

Any electromagnetic field, and hence the electromagnetic field of any antenna, can be written (with respect to the chosen coordinate origin) as

$$\vec{E}(r, \theta, \phi) = E_{\theta}(r, \theta, \phi)\hat{\theta} + E_{\phi}(r, \theta, \phi)\hat{\phi} + E_r(r, \theta, \phi)\hat{r} \quad (2.3-1)$$

and,

$$\bar{H}(r, \theta, \phi) = H_\theta(r, \theta, \phi)\hat{\theta} + H_\phi(r, \theta, \phi)\hat{\phi} + H_r(r, \theta, \phi)\hat{r} \quad (2.3-2)$$

where  $(r, \theta, \phi)$  specifies an observation point in the three-dimensional space surrounding the antenna. A useful viewpoint is to think of a sphere of radius  $r$  surrounding the antenna, with observation point  $(r, \theta, \phi)$  a point on this sphere, as depicted in Figure 2.3-2. The pair  $(\theta, \phi)$  gives the direction of the observation point, and  $r$  its distance from the origin of the coordinate system. The first two components in each of the above expressions are the *transverse* components of the antenna's electromagnetic field. The last component in each of (2.3-1) and (2.3-2), namely  $E_r(r, \theta, \phi)$  and  $H_r(r, \theta, \phi)$ , are the *radial* components of the electromagnetic field. The magnetic field  $\bar{H}(r, \theta, \phi)$  can be found from  $\bar{E}(r, \theta, \phi)$  using Maxwell's equations, and vice versa. At each such observation point the magnitude of the electric field is

$$\begin{aligned} |\bar{E}(r, \theta, \phi)| &= \sqrt{\bar{E}(r, \theta, \phi) \cdot \bar{E}^*(r, \theta, \phi)} \\ &= \sqrt{|E_\theta(r, \theta, \phi)|^2 + |E_\phi(r, \theta, \phi)|^2 + |E_r(r, \theta, \phi)|^2} \end{aligned} \quad (2.3-3)$$

The time-averaged power density at any point  $(r, \theta, \phi)$  at which the antenna fields are  $\bar{E}(r, \theta, \phi)$  and  $\bar{H}(r, \theta, \phi)$  is given by the Poynting vector<sup>5</sup>

$$\bar{S}(r, \theta, \phi) = \frac{1}{2} \text{Re}\{\bar{E}(r, \theta, \phi) \times \bar{H}^*(r, \theta, \phi)\} \quad (2.3-4)$$

and is measured in Watts/metre<sup>2</sup>. Now suppose we denote the imaginary spherical surface which encloses the antenna entirely, as in Figure 2.3-1, by  $S_{ant}$ . The time-averaged power (in Watts)

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<sup>5</sup> Note that we use the symbol  $\bar{S}(r, \theta, \phi)$  for the time-averaged version.

passing through this surface  $S_{ant}$  is given from Poynting's theorem as the integral of  $\bar{S}(r, \theta, \phi)$  over the closed surface  $S_{ant}$ , namely<sup>6</sup>

$$\begin{aligned}
 P_{rad} &= \oiint_{S_{ANT}} \bar{S}(r, \theta, \phi) \cdot \bar{dS} = \oiint_{S_{ANT}} \bar{S}(r, \theta, \phi) \cdot \hat{r} dS \\
 &= \int_0^{2\pi} \int_0^\pi \bar{S}(r, \theta, \phi) \cdot \hat{r} r^2 \sin \theta d\theta d\phi \quad (2.3-5)
 \end{aligned}$$

The physical space around an antenna (in fact this is all of space) is considered to consist of three different regions: the reactive near-zone region, the radiating near-zone region (also called the Fresnel region), and the radiating far-zone region (also called the Fraunhofer region). The boundaries of these regions cannot be precisely delineated, but the general properties of the electromagnetic fields in each zone can be established. The reactive near-zone is defined as that region of the field immediately surrounding the antenna wherein the reactive near-field dominates. If we consider the "centre of mass" of the antenna to be located at the coordinate origin as shown in Figure 2.3-2, then  $r$  can be considered to be a measure of distance from the antenna. In the reactive near-zone, the reactive fields predominate and fall off at rates  $1/r^2$  or  $1/r^3$ . The Fresnel zone is defined as that region of the field of an antenna between the reactive near-zone region and the far-zone region where the radiation fields predominate but wherein the angular field distribution (that is, the dependence on  $\theta$  and  $\phi$ ) is dependent on the distance  $r$  from the antenna. Both the reactive near-zone and Fresnel zone are usually simply referred to as the *near-zone*. Finally, the *far-zone* is the region where the angular field distribution is essentially<sup>7</sup> independent of the distance  $r$  from the antenna. In this region the fields decay as  $1/r$ .

The far-zone fields – namely those at observation points  $(r, \theta, \phi)$  that are distant from the antenna – are best considered in terms of their transverse components. These are also those most easily

---

<sup>6</sup> Note that  $P_{RAD}$  is the power actually radiated by the antenna. It is not the power incident on the antenna input terminals, or the power accepted into the antenna input terminals.

<sup>7</sup> We could for all practical purposes say "within measurement uncertainty" instead of "essentially".

connected to measurements that are made of the radiated electromagnetic fields of antennas. If an antenna is of finite size (a condition met in all practical circumstances!) then it is found that:

- The *radial* components of the fields (namely  $E_r$  and  $H_r$ ) are negligible compared to the transverse components in the far-zone region of the antenna, so that we can say

$$E_r(r, \theta, \phi) = \hat{r} \cdot \bar{E}(r, \theta, \phi) \approx 0 \quad (2.3-6)$$

$$H_r(r, \theta, \phi) = \hat{r} \cdot \bar{H}(r, \theta, \phi) \approx 0 \quad (2.3-7)$$

- The components of the *transverse* fields take the form

$$E_\theta(r, \theta, \phi) = F_\theta(\theta, \phi) \frac{e^{-jkr}}{r} \quad (2.3-8)$$

$$E_\phi(r, \theta, \phi) = F_\phi(\theta, \phi) \frac{e^{-jkr}}{r} \quad (2.3-9)$$

So that the transverse electric field is

$$\bar{E}(r, \theta, \phi) = \bar{F}(\theta, \phi) \frac{e^{-jkr}}{r} \quad (2.3-10)$$

With

$$\bar{F}(\theta, \phi) = F_\theta(\theta, \phi) \hat{\theta} + F_\phi(\theta, \phi) \hat{\phi} \quad (2.3-11)$$

If we set  $E_r$  and  $H_r$  to zero then from the Maxwell curl equations, and the expression for the curl operation in spherical coordinates, we find that the electric and magnetic fields *in the far-zone of an antenna* (and *only* in the far-zone) are related through

$$\bar{H}(r, \theta, \phi) = \frac{1}{\eta} \hat{r} \times \bar{E}(r, \theta, \phi) = \frac{1}{\eta} \hat{r} \times \bar{F}(\theta, \phi) \frac{e^{-jkr}}{r} \quad (2.3-12)$$

The structure of the electromagnetic fields in the near-zone of an antenna is complicated and sensitive to the geometrical details of its structure.

The radiation intensity  $U(\theta, \phi)$  of an antenna, in a specific direction  $(\theta, \phi)$  is defined as the power radiated from the antenna per unit solid angle in that direction. Radiation intensity is a far-zone quantity, a fact emphasised by purposefully denoting it as a function of the angular coordinates  $\theta$  and  $\phi$  only. The relation between the radiation intensity and the far-zone power density is simply

$$U(\theta, \phi) = r^2 S(r, \theta, \phi) \quad (2.3-13)$$

Note that although  $S(r, \theta, \phi)$  is a function of  $r$ , in the far-zone its  $r$ -dependence is always the  $1/r^2$  decay, and hence the radiation intensity  $U(\theta, \phi)$  is indeed independent of  $r$ , being

$$U(\theta, \phi) = \frac{|\bar{F}(\theta, \phi)|^2}{2\eta} \quad (2.3-14)$$

Graphical representations of  $U(\theta, \phi)$  are represented in either polar or rectangular form. The radiation intensity is usually a maximum in some direction, say  $(\theta_o, \phi_o)$ , and the radiation pattern is most often plotted in dB in normalised form, namely

$$U_{dB}(\theta, \phi) = 10 \log \left\{ \frac{|\bar{F}(\theta, \phi)|^2}{|\bar{F}(\theta_o, \phi_o)|^2} \right\} = 20 \log \left\{ \frac{|\bar{F}(\theta, \phi)|}{|\bar{F}(\theta_o, \phi_o)|} \right\} = 20 \log\{|\bar{F}_N(\theta, \phi)|\} = |\bar{F}_N(\theta, \phi)|_{dB} \quad (2.3-15)$$

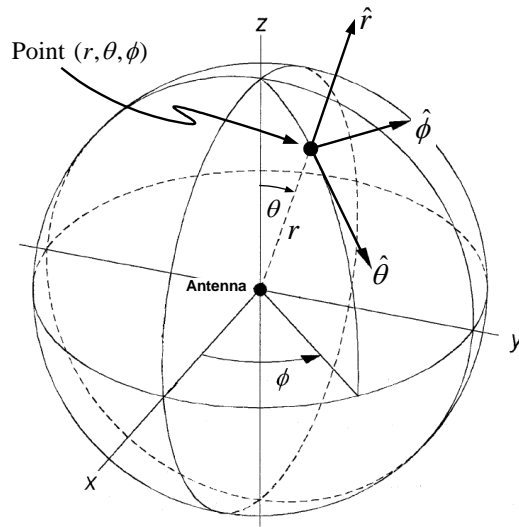


Figure 2.3-2: Spherical Coordinates with antenna at the center (Diagram courtesy of D.A.McNamara)

It is often the purpose of an antenna to increase the amount of radiation in particular directions and to minimise it in others. A measure of the ability of an antenna to do this is its *directivity*. The directivity  $D(\theta, \phi)$  of an antenna in direction  $(\theta, \phi)$  is defined as

$$D(\theta, \phi) = \frac{\text{Radiation Intensity in Direction } (\theta, \phi)}{\text{Spatially Averaged Radiation Intensity}} \quad (2.3-16)$$

If  $P_{rad}$  is the total power *radiated* by the antenna, it follows that we can express directivity as

$$D(\theta, \phi) = \frac{4\pi U(\theta, \phi)}{P_{rad}} \quad (2.3-17)$$

where

$$P_{rad} = \frac{1}{2\eta} \int_0^{2\pi} \int_0^\pi U(\theta, \phi) r^2 \sin \theta \, d\theta d\phi \quad (2.3-18)$$

It will come as no surprise that engineers express directivity in dB as

$$D^{dB}(\theta, \phi) = 10 \log\{D(\theta, \phi)\} \quad (2.3-19)$$

### 2.3.3 Fractional Bandwidth

The bandwidth of an antenna is defined as the range of frequencies within which the performance of the antenna, with respect to some characteristic, conforms to a specified standard. The bandwidth can be considered to be the range of frequencies between some lower frequency  $f_L$  to some upper frequency  $f_U$ , namely

$$f_L \leq f \leq f_U \quad (2.3-20)$$

on either side of some center frequency  $f_o$ , over which the required antenna characteristics are within acceptable values. The various characteristics of an antenna will not necessarily have the same frequency dependence. For instance, the input reflection coefficient may be within acceptable values over a smaller frequency range than are the sidelobe levels. Thus, there is no unique definition for bandwidth. We can talk about the impedance bandwidth (also referred to as the reflection coefficient bandwidth), the pattern bandwidth, and so forth.

Suppose that an antenna satisfies a required set of specifications over the frequency band

$$\Delta f = f_U - f_L \quad (2.3-21)$$

The quantity  $\Delta f$  is the absolute bandwidth. This is the bandwidth specification usually of interest to the designer of the system in which the antenna is to be used. Given the above absolute bandwidth, the *center-frequency* is

$$f_o = f_L + \frac{\Delta f}{2} = \frac{1}{2}(f_U + f_L) \quad (2.3-22)$$

All the above quantities are of course measured in Hz. The *fractional bandwidth* of the antenna is defined as

$$\text{Fractional Bandwidth} = \frac{\Delta f}{f_o} \quad (2.3-23)$$

and is dimensionless. It is usually expressed as a percentage, namely

$$\text{Fractional Bandwidth as a Percentage} = 100 \left( \frac{\Delta f}{f_o} \right) \% \quad (2.3-24)$$

and so is sometimes referred to as the *percentage bandwidth*.

### 2.3.4 Input Reflection Coefficient of an Isolated Antenna

Consider the general set-up shown in Figure 2.3-1, which shows an antenna of input impedance  $Z_{IN}$  fed by a transmission line of characteristic impedance  $Z_o$ . The reflection coefficient at the input terminals of the antenna is

$$\Gamma_{IN} = \frac{Z_{IN} - Z_o}{Z_{IN} + Z_o} \quad (2.3-25)$$

We will denote the power *incident* at the antenna terminals by  $P_{inc}$ . The power actually *accepted* by the antenna is

$$P_{in} = \{1 - |\Gamma_{IN}|^2\}P_{inc} \quad (2.3-26)$$

As in previous sections,  $P_{rad}$  represents the total power actually radiated by the antenna. A portion of the power accepted by the antenna is not radiated due to losses<sup>8</sup> on the antenna structure. We will denote this lost power as  $P_{loss}$ . Hence, we have

$$P_{rad} = P_{in} - P_{loss} \quad (2.3-27)$$

At any frequency  $\omega=2\pi f$ , we have an input impedance

$$Z_{IN}(\omega) = R_{IN}(\omega) + jX_{IN}(\omega) \quad (2.3-28)$$

As with any electrical network, the resistance  $R_{IN}(\omega)$  models the energy that is input from the source and is not returned to the source. In the case of the antenna it thus represents two physical effects, namely power that is radiated away from the antenna and power that is dissipated on the antenna structure itself. We can write

$$R_{IN}(\omega) = R_{RAD}(\omega) + R_{LOSS}(\omega) \quad (2.3-29)$$

---

<sup>8</sup> For example, dissipative losses (i.e. energy converted to a form of energy other than electromagnetic) in the antenna materials, which will usually consist of both conductors and dielectrics.

$R_{RAD}(\omega)$  accounts for the radiated power<sup>9</sup>, and  $R_{LOSS}(\omega)$  for the power dissipated (e.g. Ohmic loss in conducting material; dielectric loss in printed circuit substrates, and so on). We have purposefully shown the frequency dependence explicitly to emphasise the fact that the values of the equivalent network parameters may not be the same at all frequencies of interest<sup>10</sup>.

### 2.3.5 Radiation Efficiency

The *radiation efficiency* of an antenna is defined as the ratio of the total power radiated by the antenna to the net power accepted by the antenna from the connected transmission line, namely

$$e_{\text{rad}} = \frac{P_{\text{rad}}}{P_{\text{in}}} \quad (2.3-30)$$

which can be re-written as,

$$e_{\text{rad}} = \frac{P_{\text{rad}}}{P_{\text{rad}} + P_{\text{loss}}} \quad (2.3-31)$$

and hence

$$e_{\text{rad}} = \frac{R_{\text{rad}}}{R_{\text{rad}} + R_{\text{loss}}} \quad (2.3-32)$$

### 2.3.6 Total Active Reflection Coefficient (TARC)

The total active<sup>11</sup> reflection coefficient (TARC) was first introduced in [MANT 03]. The motivation for introducing this parameter is that the conventional return loss figure of merit does not account for the constructive/destructive interaction between the signals of an antenna in an MEA system. It is noted in [SHAR 13] and [CHAE 07] that TARC not only captures the total reflection of power in an MEA, but also takes the different phase of each excitation signal into consideration. As will be seen below, the value of TARC will always be between 1 and 0 (like the “ordinary” reflection coefficient in Section 2.3.4); if the TARC of an element in an MEA system

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<sup>9</sup> Radiation resistance is often described as a “fictitious resistance” that represents power flowing out of the antenna. We of course rather call it an equivalent resistance since it models the power that has left the antenna never to return.

<sup>10</sup> In what follows we will not always show this frequency dependence explicitly, but it should always be kept in mind.

<sup>11</sup> The word “active” does not imply the presence of biased active devices (eg. transistors), but uses the word in the manner done by the term “active element pattern” in array antenna work.

is equal to unity then no signal is able to enter that element and so it is not radiating. On the other hand, a TARC value of zero means that there is no input reflection experienced by the particular element when operating in the MEA system environment. In summary, TARC for a MEA system is the same as the reflection coefficient for a single antenna [BROW 06], and so it is a key parameter for evaluating MEA system performance. Apparently, for adequate MIMO performance, each antenna should not exceed a TARC value of -6 dB [SHAR 13b].

The general expression for the TARC s of an N-element MEA is

$$\Gamma_a^t = \sqrt{\frac{\sum_{i=1}^N |b_i|^2}{\sum_{i=1}^N |a_i|^2}} \quad (2.3-33)$$

This is based on considering the MEA as a multiport network where  $a_i$  is the excitation signal at port  $i$ . Similarly,  $b_i$  is the signal reflected at port  $i$ . In the case of a 2-port MEA system we can of course characterize the port-behaviour in terms of a scattering matrix as

$$\begin{bmatrix} b_1 \\ b_2 \end{bmatrix} = \begin{bmatrix} S_{11} & S_{12} \\ S_{21} & S_{22} \end{bmatrix} \begin{bmatrix} a_1 \\ a_2 \end{bmatrix} \quad (2.3-34)$$

that, multiplied out, is of course simply

$$b_1 = a_1 S_{11} + a_2 S_{12} \quad (2.3-35)$$

$$b_2 = a_1 S_{21} + a_2 S_{22} \quad (2.3-36)$$

We can define the excitation vector as

$$a_1 = a_0 e^{j\psi_1} \quad (2.3-37)$$

$$a_2 = a_0 e^{j\psi_2} \quad (2.3-38)$$

where,  $a_0$  is an arbitrary amplitude (usually set to unity), and  $\psi_{1,2}$  is the phase of the signal entering port#1 (of the one MEA system antenna) and port#2 (of the other antenna in the MEA system).

The relative phase is

$$\psi = \psi_2 - \psi_1 \quad (2.3-39)$$

Substituting this into (2.3-35) and (2.3-36) we obtain

$$b_1 = a_0 e^{j\psi_1} S_{11} + a_0 e^{j\psi_2} S_{12} \quad (2.3-40)$$

and hence

$$b_1 = \underbrace{a_0 e^{j\psi_1}}_{a_1} (S_{11} + e^{j\psi} S_{12}) \quad (2.3-41)$$

Similarly,

$$b_2 = \underbrace{a_0 e^{j\psi_1}}_{a_1} (S_{21} + e^{j\psi} S_{22}) \quad (2.3-42)$$

This can be inserted back into the general mathematical definition of TARC given by (2.3-33) to yield (for the case  $N = 2$ )

$$\begin{aligned} \Gamma_a^t &= \sqrt{\frac{\sum_{i=1}^N |b_i|^2}{\sum_{i=1}^N |a_i|^2}} = \sqrt{\frac{|b_1|^2 + |b_2|^2}{|a_1|^2 + |a_2|^2}} = \sqrt{\frac{|a_1|^2 [(S_{11} + S_{12} e^{j\psi})^2 + (S_{21} + S_{22} e^{j\psi})^2]}{|a_1|^2 + |a_1|^2}} \\ \Gamma_a^t &= \frac{\sqrt{[(S_{11} + S_{12} e^{j\psi})^2 + (S_{21} + S_{22} e^{j\psi})^2]}}{\sqrt{2}} \end{aligned} \quad (2.3-43)$$

TARC can then be plotted using the equation above by fixing the amplitude of the excitation signals to unity and varying the relative phase  $\psi \in [0, \pi]$  between the two signals. Since the phase of incoming signals is considered random, plotting the TARC curves for different values of  $\psi$  at the frequency band of interest will yield multiple plots. The designer can then use these plots to determine the “worst-case” reflection coefficient for the MEA system as a whole.

$$|\Gamma_a^t|_{dB} = 10 \log(\Gamma_a^t) \quad (2.3-44)$$

There is a factor of ten in front of the logarithm because in the above operations we are combining incoherent signals and so are adding powers and not voltages [SMIT 07].

### 2.3.7 Envelope Correlation Coefficient

The envelope correlation coefficient (ECC) is a metric which provides a measure of the correlation between the signals exiting the ports of the antennas in an MEA system, and so assists engineers in assessing the spatial-correlation/diversity-ensuring roles of the MEA system in communications systems. ECC is used to evaluate the diversity performance of systems other than MIMO as well [VAUG 03]. As will be seen once it is defined below, the ECC has a value between

0 and 1; it approaches 1 in the presence of poor multipath propagation environment when the different antenna elements start behaving like a single antenna [VAUG 87] [SHAR 13b]. On the other hand, the MEA's are said to 'excite' isolated wireless channels if the ECC value is equal to zero [SHAR 17]. We can conclude that in order to obtain a high degree of diversity, and achieve the improvements offered by MIMO (for example), one should aim to have zero or very low ECC values for the individual antennas in an MEA system.

#### A. Basic Definition of ECC in Terms of Far-Zone Radiation Patterns

A truly rigorous derivation of the expression given below is very involved<sup>12</sup>, and so will not be considered here. The envelope correlation coefficient between the  $i$ -th and  $j$ -th antennas in the MEA system can be expressed, with integration over all solid angles (and an asterisk denoting complex conjugation), as

$$\rho_{ij} = \left| \frac{\iint \text{XPD} \times E_i^\theta(\Omega) E_j^{\theta*}(\Omega) P^\theta(\Omega) + E_i^\phi(\Omega) E_j^{\phi*}(\Omega) P^\phi(\Omega) d\Omega}{\sqrt{\iint \text{XPD} \times G_i^\theta(\Omega) P^\theta(\Omega) + G_i^\phi(\Omega) P^\phi(\Omega) d\Omega} \sqrt{\iint \text{XPD} \times G_j^\theta(\Omega) P^\theta(\Omega) + G_j^\phi(\Omega) P^\phi(\Omega) d\Omega}} \right|^2 \quad (2.3-45)$$

where

$$G_i^\theta(\Omega) = E_i^\theta(\Omega) E_i^{\theta*}(\Omega) \quad (2.3-46)$$

and  $\Omega$  denotes solid angle. XPD is the cross-polarization discrimination factor of the propagation environment in which the MEA operates (and usually needs to be determined experimentally). Functions  $P^\theta$  and  $P^\phi$  are the statistical distributions (one for each polarisation) describing the probability of the incident field approaching from a specific direction. In a so-called isotropic propagation environment  $\text{XPD} = 1$  and  $P^\theta = P^\phi = \pi/4$  [VAUG 03]. In an isotropic propagation environment expression (2.3-45) can be re-written as

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<sup>12</sup> R.Vaughan, Department of Engineering Science, Simon Fraser University, Canada. Private communication.

$$\rho_{i,j} = \frac{|\iint [\vec{E}_i(\Omega) \cdot \vec{E}_j^*(\Omega)] d\Omega|^2}{\iint \vec{E}_i(\Omega) \cdot \vec{E}_i^*(\Omega) d\Omega \iint \vec{E}_j(\Omega) \cdot \vec{E}_j^*(\Omega) d\Omega} \quad (2.3-47)$$

The electric field of the  $n$ -th antenna can be written

$$\vec{E}_n(\theta, \phi) = E_n^\theta(\theta, \phi) \hat{\theta} + E_n^\phi(\theta, \phi) \hat{\phi} \quad (2.3-48)$$

Substituting this expression into the ECC definition (2.3-45), we obtain

$$\rho_{i,j} = \frac{\left| \int_{\phi=0}^{2\pi} \int_{\theta=0}^{\pi} \{E_i^\theta(\theta, \phi) E_j^{\theta*}(\theta, \phi) + E_i^\phi(\theta, \phi) E_j^{\phi*}(\theta, \phi)\} \sin \theta d\theta d\phi \right|^2}{\int_{\phi=0}^{2\pi} \int_{\theta=0}^{\pi} \{E_i^\theta(\theta, \phi) E_i^{\theta*}(\theta, \phi) + E_i^\phi(\theta, \phi) E_i^{\phi*}(\theta, \phi)\} \sin \theta d\theta d\phi \int_{\phi=0}^{2\pi} \int_{\theta=0}^{\pi} \{E_j^\theta(\theta, \phi) E_j^{\theta*}(\theta, \phi) + E_j^\phi(\theta, \phi) E_j^{\phi*}(\theta, \phi)\} \sin \theta d\theta d\phi} \quad (2.3-49)$$

The surface integrals in (2.3-49) can be converted into discrete summations for computation purposes, a process that yields

$$\rho_{i,j} = \frac{|F_{ij}(\theta, \phi)|^2}{F_{ii}(\theta, \phi) F_{jj}(\theta, \phi)} \quad (2.3-50)$$

where<sup>13</sup>,

$$F_{ij}(\theta, \phi) = \sum_{m=0}^M \sum_{n=0}^N \left[ E_i^\theta(\theta_n, \phi_m) E_j^{\theta*}(\theta_n, \phi_m) + E_i^\phi(\theta_n, \phi_m) E_j^{\phi*}(\theta_n, \phi_m) \right] \Delta\theta \Delta\phi \sin \theta_n \quad (2.3-51)$$

$$F_{ii}(\theta, \phi) = \sum_{m=0}^M \sum_{n=0}^N \{E_i^\theta(\theta_n, \phi_m) E_i^{\theta*}(\theta_n, \phi_m) + E_i^\phi(\theta_n, \phi_m) E_i^{\phi*}(\theta_n, \phi_m)\} \Delta\theta \Delta\phi \sin \theta_n \quad (2.3-52)$$

$$F_{jj}(\theta, \phi) = \sum_{m=0}^M \sum_{n=0}^N \{E_j^\theta(\theta_n, \phi_m) E_j^{\theta*}(\theta_n, \phi_m) + E_j^\phi(\theta_n, \phi_m) E_j^{\phi*}(\theta_n, \phi_m)\} \Delta\theta \Delta\phi \sin \theta_n \quad (2.3-53)$$

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<sup>13</sup> In Chapters 3 and 4, where these terms will be evaluated, the individual antenna patterns are sufficiently broad (and the angular variations of the fields thus sufficiently 'slow') that use of  $\Delta\theta = \Delta\phi = 5^\circ$  provides accurate evaluation of the integrals.

The above expressions for computing ECC provide accurate results. However, a major drawback to using this method is the requirement to measure or compute the 3D complex<sup>14</sup> far-field patterns, at every frequency point within the desired bandwidth of the MEA system. Therefore a simpler, but approximate expression, has been derived in the literature for ECC calculation, as we will next discuss.

### ***B. Calculation in Terms of Multi-Port Scattering Parameters***

Driven by the copious measurements and simulations which are involved in obtaining the complex far-field 3D patterns, as well as the lengthy computation time required to calculate the ECC via expression (2.3-50), researchers developed a simplified expression for the ECC [BLAN 03], namely

$$\rho_{i,j} = \frac{|S_{ii}^* S_{ij} + S_{ji}^* S_{jj}|^2}{(1 - |S_{ii}|^2 - |S_{ji}|^2)(1 - |S_{jj}|^2 - |S_{ij}|^2)} \quad (2.3-54)$$

Expression (2.3-54) makes the assumption that antennas are lossless<sup>15</sup>, so that their radiation efficiencies are unity ( $e_i^{rad} = e_j^{rad} = 1$ ). Moreover, it should be highlighted that as the antenna losses increase (efficiency decreases), the discrepancy between the correct values obtained from (2.3-50) and those found using (2.3-54) becomes more significant [HALL 05]. One must be careful when using the S-parameter approach as it often produces unrealistically low (optimistic) ECC values which might lead the designer down the “*garden path*” [BHAT 17]. A good way of understanding why finding the ECC using S-parameters is not accurate for lossy antennas is this<sup>16</sup>: “Imagine going to the limit of extremely high loss. Then [S] just gives us the noise, which is uncorrelated!”. Finally, it should be noted that enhancing the isolation between the elements does not guarantee a low ECC. This is because by definition the ECC is a measure of the signal correlations and thus depends on the extent to which the radiated fields overlap [ASHR 21].

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<sup>14</sup> Amplitude and phase.

<sup>15</sup> It is important to realise that this assumption is needed in order to arrive at (2.3-54) from the fundamental definition of ECC. One should not assume that by using, in (2.3-54), the S-parameters for the actual lossy antennas that this assumption is somehow removed.

<sup>16</sup> R.Vaughan, Department of Engineering Science, Simon Fraser University, Canada. Private communication.

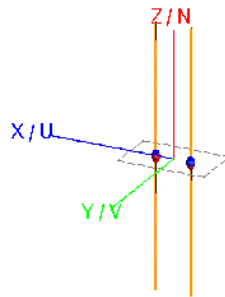
### C. ECC Requirements for MIMO Antennas

Typically, an ECC value of less than 0.7 is sufficiently low for MEA at base stations. This requirement drops to less than 0.5 for mobile terminals in order to achieve good diversity [VAUG 03]. Due to the presence of more elements in massive MIMO, signal correlation between non-adjacent elements must be taken into account, and thus the ECC requirement in M-MIMO is typically more stringent.

### D. ECC Calculation Examples

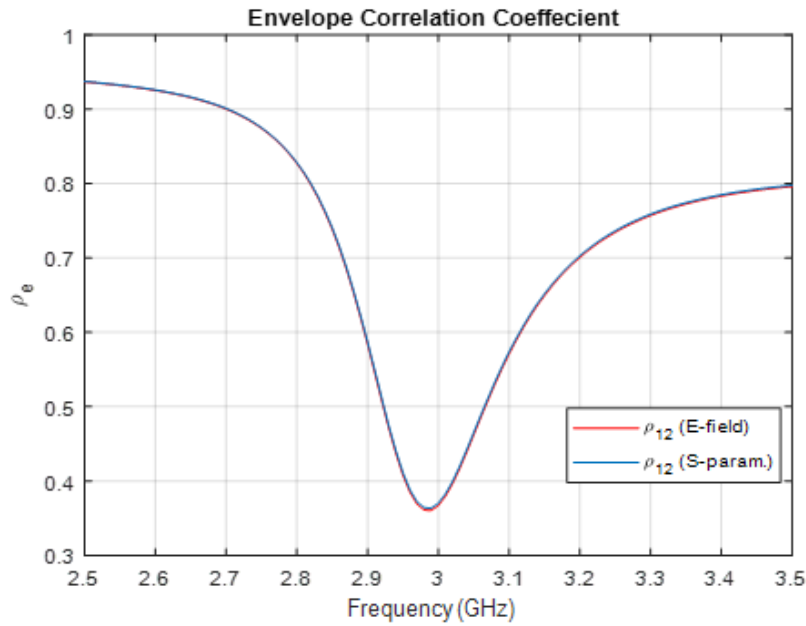
In this section the expression (2.3-50) that we will call the far-field method, and expression (2.3-54) that will be called the S-parameter method, will be used to compute the ECC for three different examples. The resonances of three example antennas were chosen to be within the ISM band (2.45 GHz - 5.8 GHz) to ensure that no discrepancies occur between the models studied due to large frequency differences. The first deals with a lossless case, with two dipoles in free space within proximity of each other. The second is like the first (that is, lossless) but with a loosely coupled additional wire object present. The third case will be for two microstrip patch antennas printed on a lossy substrate.

Figure 2.3-3 shows the two dipole antennas parallel to each other and separated along the x-axis, in free space. The dipoles are identical, and each has a dedicated port. This two-dipole MEA was modelled as consisting of thin wires using the full-wave electromagnetic simulation software based on the method of moments [FEKO 18], and the antennas' scattering parameters and far-field patterns computed.



**Figure 2.3-3: Two closely spaced dipole antennas**

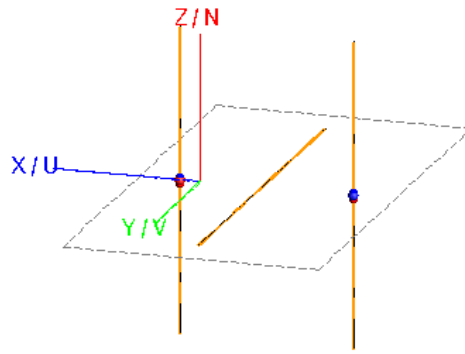
The simulation data from FEKO was then run through a script<sup>17</sup> to calculate the ECC. Figure 2.3-4 shows the computed ECC for the two-dipole example using the far-field method and the S-parameter method. The two sets of results are identical, as might be expected because the dipoles are lossless (perfect conductors).



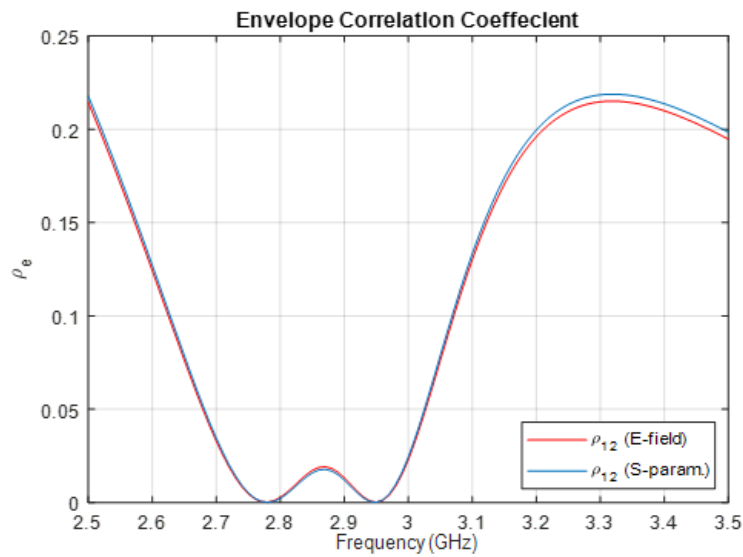
**Figure 2.3-4: ECC plots for dipole antennas**

Since Figure 2.3-4 shows identical results using either method, we decided to try the slightly altered arrangement shown in Figure 2.3-5, in which a loosely-coupled additional wire has been introduced at the origin and lies along the y-axis. The passive wire introduces some slight discrepancy between the two ECC curves, as seen in Figure 2.3-6, but are still practically identical. We can conclude that the S-parameter approach for computing the ECC is a very good approximation for lossless antennas. The comparison also serves to validate the script that implements expression (2.3-50) to find the ECC based on the definition of this quantity.

<sup>17</sup> Developed by the author in MATLAB.



**Figure 2.3-5: Two dipole antenna and a passive wire.**



**Figure 2.3-6: ECC results for two dipole antennas and a passive wire**

We now compare the outcomes of the two expressions for the ECC when the antennas in the MEA are lossy. Figure 2.3-7 shows a simple symmetrical design of two pin-fed patch antennas etched on a substrate with  $\epsilon_r = 4$  and  $\tan \delta = 0.005$  is simulated. Figure 2.3-8 reveals a slight discrepancy between the computed ECC values which is roughly about 1%. Moreover, Figure 2.3-9 shows how the S-parameter approach can lead to inaccurately low ECC values.

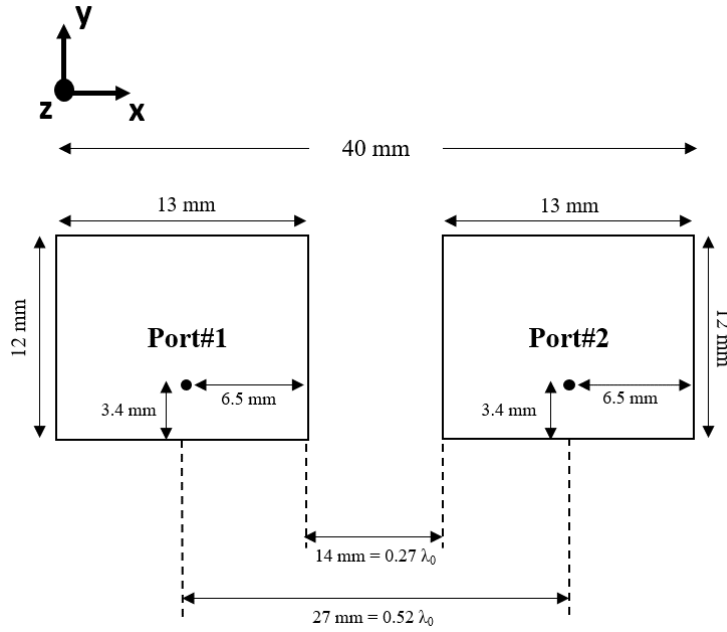


Figure 2.3-7: Two patch antennas on a lossy substrate with  $\epsilon_r = 0.04$  and  $\tan(\delta) = 0.005$ .

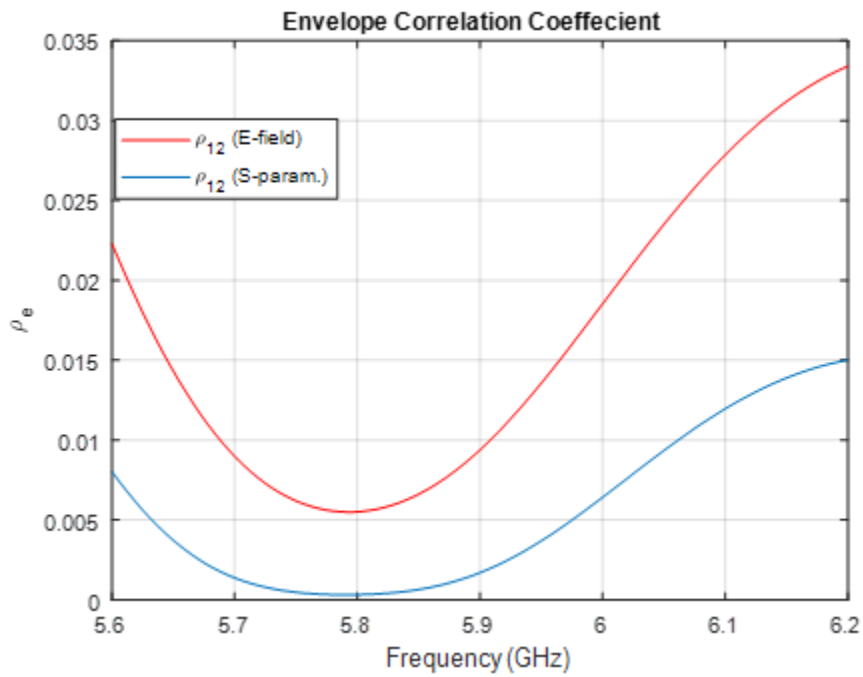


Figure 2.3-8: ECC curves for the two patch antennas shown in Figure 2.3-7.

To develop a sense of how the lossy-ness affects the ECC calculation, we recalculated the ECC for the same two-patch MEA design, but with the loss tangent of the substrate increased to  $\tan \delta = 0.01$  and then  $\tan \delta = 0.025$ , the results being provided in Figure 2.3-9 and Figure 2.3-10, respectively. These show that as the amount of loss increases the ECC computed using (2.3-50) - the correct result - stays largely the same but that the result found using the S-parameters moves further from the correct result to increasingly lower (over optimistic) values. This serves to highlight the importance of using complex 3D far-field patterns in the calculation of the ECC.

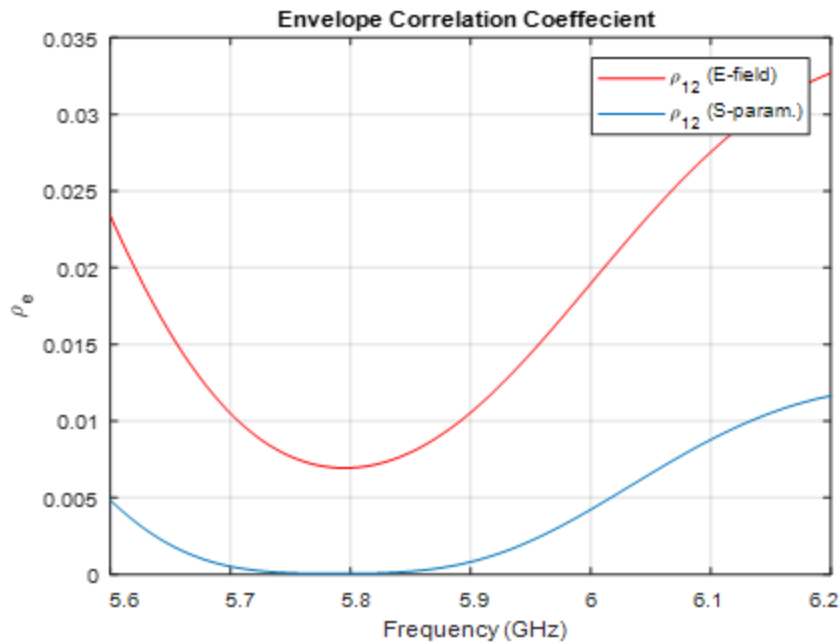


Figure 2.3-9: ECC curves for the two patch antennas shown in Figure 2.3-7 but with  $\tan \delta = 0.01$ .

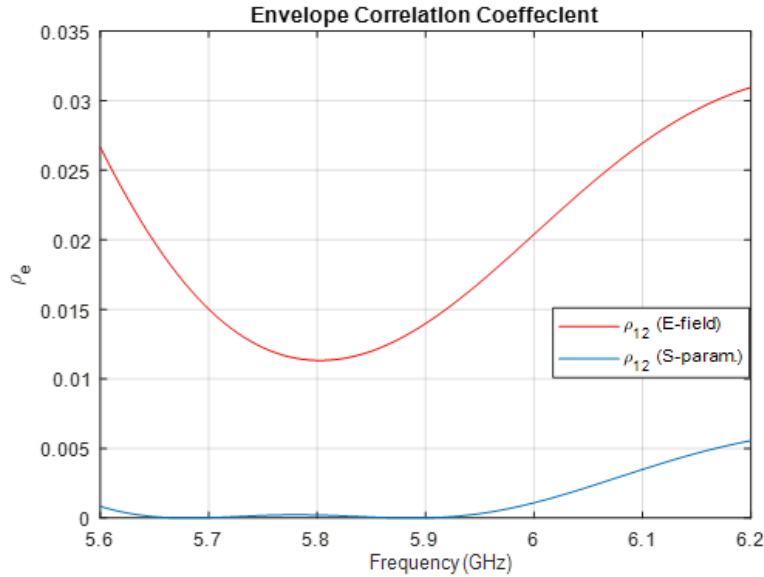


Figure 2.3-10: ECC curves for the two patch antennas shown in Figure 2.3-7 but with  $\tan\delta = 0.025$

### 2.3.8 The Need for Low Mutual Coupling Between MIMO Antennas

It is widely accepted<sup>18</sup> among MIMO antenna designers that low mutual coupling is required in order to achieve good diversity. Numerous studies have shown that mutual coupling can have a negative effect on the correlation between channels, and thus ultimately a deleterious effect on the capacity of the MIMO system. In cases where the inter-element separation distance is less than  $0.5\lambda$ , strong mutual coupling exists and it can cause a decrease in the channel capacity, especially so as the number of antenna elements increases [JANA 02], [ABOU 06]. The conclusion is that mutual coupling effects are detrimental to MIMO performance. We would like it to be as low as possible.

As stated in [OZDE 03], arguments both for and against mutual coupling in MIMO are correct for different scenarios. In environments with poor scattering (highly correlated environment), if the mutual coupling effect is allowed to deform<sup>19</sup> the far-field radiation pattern, this might serve as a decorrelation factor. However, if the environment is rich in multipath scattering, and the separation

<sup>18</sup> There are some works which suggest that mutual coupling can in fact serve as a decorrelation factor (and thus be advantageous), thereby increasing the channel capacity [SVAN 01] [CLER 03] [BORJ 03], but this does not at present to be the majority opinion of wireless communication theory experts.

<sup>19</sup> Which might be undesirable for other reasons.

distance between adjacent elements is less than  $0.5\lambda$ , mutual coupling between the antenna elements is non-negligible and it increases the correlation.

It should be noted that even for signals propagating in a poor environment, the decorrelation effect due to mutual coupling is only significant for very small separation distances ( $0.1\lambda - 0.2\lambda$ ) between the MIMO antennas.

MIMO was developed to overcome the deep fades resulting from multipath propagations in rich scattering environments. Knowing this, we will in this thesis always assume that mutual coupling impedes MIMO performance and we will work to reduce it.

### 2.3.9 Technical Definition of Mutual Coupling

We can quantify mutual coupling between any number of antennas via network theory. If we assume an MEA with N-elements, then we can develop an equivalent N-port network; that is a network with N-inputs or N-outputs [VAUG 03]. For simplicity we will consider two antennas, which form a two-port network, and can then formulate the impedance matrix  $[Z]$  so that  $[V]=[Z][I]$ , in other words

$$\begin{bmatrix} V_1 \\ V_2 \end{bmatrix} = \begin{bmatrix} Z_{11} & Z_{12} \\ Z_{21} & Z_{22} \end{bmatrix} \begin{bmatrix} I_1 \\ I_2 \end{bmatrix} \quad (2.3-55)$$

where  $Z_{mn} = R_{mn} + jX_{mn}$  and is known as the self-impedance for ( $m = n$ ), and the mutual impedance ( $m \neq n$ ) between ports  $m$  and  $n$ . If we write out the compact notation (2.3-55) in full, we of course have [STUT 13]

$$V_1 = Z_{11} I_1 + Z_{12} I_2 \quad (2.3-56)$$

$$V_2 = Z_{21} I_1 + Z_{22} I_2 \quad (2.3-57)$$

where the mutual impedances are

$$Z_{12} = \left. \frac{V_1}{I_2} \right|_{I_1=0}, Z_{21} = \left. \frac{V_2}{I_1} \right|_{I_2=0} \quad (2.3-58)$$

and the self-impedances of the antennas are

$$Z_{11} = \left. \frac{V_1}{I_1} \right|_{I_2=0}, Z_{22} = \left. \frac{V_2}{I_2} \right|_{I_1=0} \quad (2.3-59)$$

The input impedances seen by each of the antenna elements are (under operational conditions)

$$Z_{in}^1 = \frac{V_1}{I_1} = Z_{11} + Z_{12} \left( \frac{I_2}{I_1} \right) \quad (2.3-60)$$

$$Z_{in}^2 = \frac{V_2}{I_2} = Z_{21} \left( \frac{I_1}{I_2} \right) + Z_{22} \quad (2.3-61)$$

This implies that the impedance seen at one antenna is affected by the impedance of the “other” antenna due to mutual coupling. The amount of mutual coupling in a system is captured in the off-diagonal terms in the scattering matrix, or  $S_{12}$  in this two-antenna case. We are able to transform impedance matrix terms  $[Z]$  into scattering matrix parameters  $[S]$  by using the following equations [POZA 98].

$$S_{11} = \frac{(Z_{11} - Z_0)(Z_{22} + Z_0) - Z_{12}Z_{21}}{(Z_{11} + Z_0)(Z_{22} + Z_0) - Z_{12}Z_{21}} \quad (2.3-62)$$

$$S_{12} = \frac{2Z_{12}Z_0}{(Z_{11} + Z_0)(Z_{22} + Z_0) - Z_{12}Z_{21}} \quad (2.3-63)$$

$$S_{21} = \frac{2Z_{21}Z_0}{(Z_{11} + Z_0)(Z_{22} + Z_0) - Z_{12}Z_{21}} \quad (2.3-64)$$

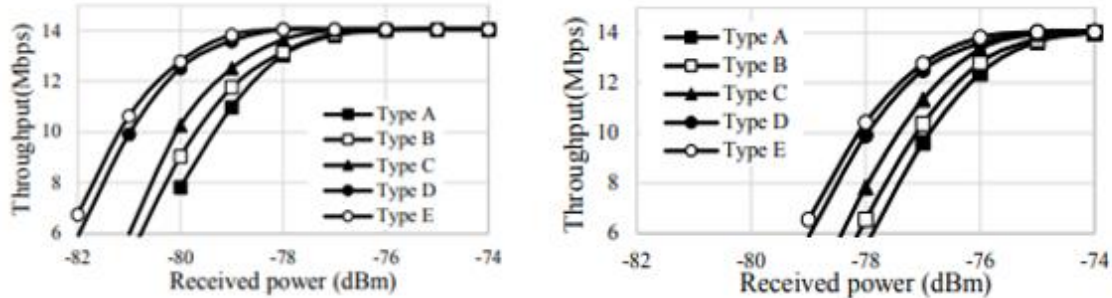
$$S_{22} = \frac{(Z_{11} + Z_0)(Z_{22} - Z_0) - Z_{12}Z_{21}}{(Z_{11} + Z_0)(Z_{22} + Z_0) - Z_{12}Z_{21}} \quad (2.3-65)$$

This enables us to connect the S-parameter representations (used previously in Sections 2.3.6 and 2.3.7) to the mutual impedance viewpoint. In this thesis ‘mutual coupling’ will mean  $S_{ij}$ .

### 2.3.10 Inter-Element Coupling Requirements in MIMO Systems

Having established that mutual coupling can degrade the performance of MIMO systems, an important question arises: *How low does the mutual coupling need to be?* An attempt to answer this question was made in [MEI 18], by conducting a measurement campaign for antennas with different mutual coupling values. The antennas studied were inverted-F antennas (IFA’s) which made use of a lumped capacitor for decoupling. The capacitance of the lumped element was varied to yield different decoupling (mutual coupling) values, while all other antenna parameters remained the same. The testing was performed in a chamber which made use of the Umi (micro-

cell) and Uma (macro-cell) channel models which are the standard channel models for urban LTE MIMO micro- and macro- cell testing, respectively.



**Figure 2.3-11: Data throughput versus received power for antennas with different levels of mutual coupling. Macro-cell on the left and micro-cell on the right. (After [MEI 18])**

The results are shown in Figure 2.3-11. They display the maximum data throughput achieved by the antennas for different received power levels. The mutual coupling levels of the labelled antennas are : A (-11 dB), B (-13 dB), C (-16 dB), D (-20 dB) and E (-30 dB). The return loss is between -14 and -15 dB for all antenna types, and the efficiency between 80% and 84%. The measured results, for both scenarios, show that the throughput seems to increase as the mutual coupling is decreased. From the observation that there is no significant throughput increase in going from type E (-30 dB) to type D (-20 dB) it was concluded that -20 dB of mutual coupling is sufficient for good MIMO performance [MEI 18].

Other works have reported that mutual coupling of -15 dB is enough for ultra-wide-band MIMO [LEI 13]. For massive-MIMO basestations, the industry rule of thumb is more restrictive, as it requires mutual coupling to be -30 dB or lower [CHEN 18]. In Chapters 3 and 4 of this thesis we will work to decouple antennas by setting an upper limit to the mutual coupling and not defining a lower limit. This will allow the shaping process to minimize mutual coupling as much as possible within the rules of physics.

### **2.3.11 Radiation Pattern Requirement for MIMO Antennas**

Different propagation environments require different antennas, yet we should design an antenna which would yield acceptable performance in a range of environments. For an antenna to be effective in a MIMO system it must enhance the channel capacity by illuminating as much of the propagation environment as possible to encourage multipath propagation. Antennas which allow signals to be transmitted and received over a wide range of angles increase the channel capacity in MIMO systems [BALA 16]. For this reason, an isotropic antenna would be ideal for MIMO applications, but of course such antennas are not physically possible. The conclusion of a number of MIMO measurement campaigns [WITT 09] [LI 14] was that antennas with broad beams are best.

## **2.4 Existing Coupling Reduction Methods in MEA Systems**

### **2.4.1 Preliminary Notes**

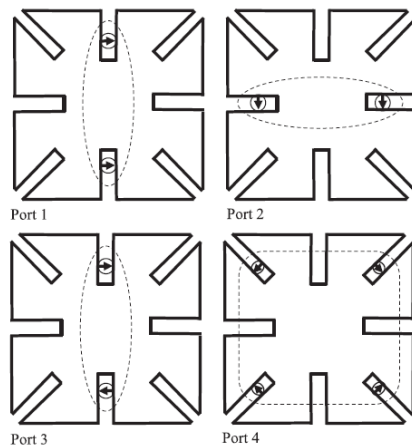
This section will deal with the existing methods for antenna decoupling in the literature. We will attempt to briefly explain the principles of each method and discuss selected examples, and list reported advantages and disadvantages offered by each approach. We will classify the methods that have so far been used for coupling reduction as eigenmode decomposition schemes, inserted component schemes, coupled resonator schemes, and decoupling surface schemes. We also indicate where some sort of automated optimization has been used in the design of MEA systems for MIMO applications. There are some existing methods which attempt to mitigate the effects of mutual coupling using various digital signal processing schemes. However as [CHEN 18] stated it is never a good idea to hope that the unwanted coupling can be completely removed using such an approach. It is better to design the MEA elements so that they have low coupling, and then (if necessary) use signal processing to make relatively small improvements.

### **2.4.2 Eigenmode Decomposition Schemes**

This method is based on the excitation of  $N$  orthogonal modes denoted as  $(J_n)$  for an  $N$ -element MIMO system. The modes correspond to surface currents which exist on the conductor geometry under study. Each mode has an associated radiation pattern. The orthogonal electromagnetic modes are obtained for the eigenvalues of the method of moment impedance matrix. Essentially,

the (orthogonal) modes are (orthogonal) solutions to Maxwell's equations. Since each characteristic mode yields a radiation pattern, in a multipath environment one radiation pattern of the antenna might undergo fading while the other radiation pattern does not. As a result, if the right modes are excited good antenna diversity can be achieved [SVAN 02]. This method has been shown to provide significant improvement in cases where the antennas are highly correlated over a narrow bandwidth. Characteristic modes research in MIMO has given rise to a multimode antenna which is essentially a single antenna structure fed through one or more ports. Each feed excites a set of orthogonal modes which allow the antenna to behave as a multi antenna system. One of the obvious advantages of such designs is their compactness which waive the need for multiple physical antenna or decoupling surface structures. The authors of [SVAN 02] propose the use of multi-mode antennas to increase MIMO channel capacity without the use of additional antenna elements. The additional modes which are excited essentially represent different solutions to Maxwell's equations that fulfill the boundary conditions for the geometry at hand. The author shows how a single circular microstrip antenna can be excited with two modes to achieve higher diversity.

In [MANT 16], the theory of characteristic modes is utilized to design massive MIMO MEA's. The design is comprised of an 11 by 11 planar array arrangement of physical antennas which are excited by four ports each. Figure 2.4-1 shows the four ports carefully placed in order to excite four orthogonal characteristic modes effectively yielding  $121 \times 4 = 484$  antennas with good isolation and reflection coefficients.



**Figure 2.4-1: Four ports feeding a single multimode antenna [MANT 16]**

Among the many interesting works which were conducted on MEA design using theory of characteristic modes (TCM), is that of [LI 12], where the characteristic modes excited on the mobile station's chassis (ground) were studied in order to determine the different antenna placements. In this study, it was revealed that if one considers the magnitude of the electric field resulting from the first characteristic mode shown in Figure 2.4-2, the designer could obtain better isolation by placing the antennas closer to each other. This is of course done by studying the results in Figure 2.4-2 which shows the normalized magnitude of the E-field of the dominant characteristic mode (at  $f = 1.35$  GHz) over the entire area (100 mm x 40 mm) of the mobile device's chassis which is shown at the bottom of the figure. The characteristic mode (CM) has a sinusoidal behaviour. Therefore, placing one antenna at the location of the electric field's maxima (edge of vertical component of chassis) and the other at the electric field's minima (center of chassis) will yield better isolation. The insight offered by TCM analysis produced an unintuitive result in which the center-edge placement (less distance) yields better isolation compared to the edge-edge placements (more distance).

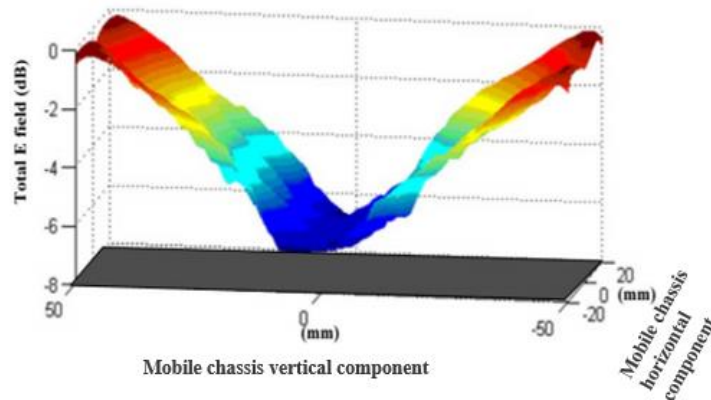
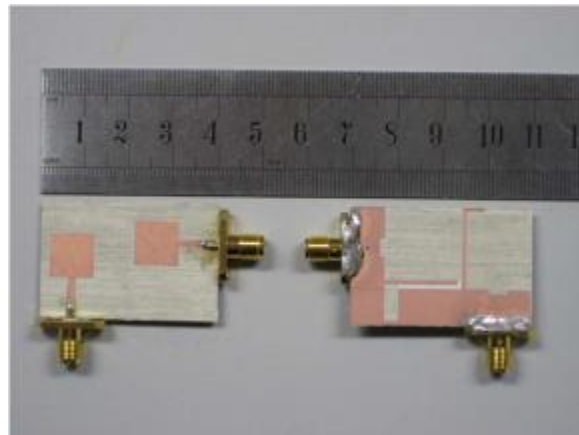


Figure 2.4-2: Normalized magnitude of the total electric field of first CM vs. chassis geometry [LI 12]

### 2.4.3 Inserted Component Schemes

The most popular inserted component schemes entail inserting a parasitic element which acts as a decoupling structure. Such structures are usually made of perfect electric conductor (PEC) and work to deflect surface and space waves radiated by an antenna away from the other antennas in

the MEA system. There are several examples in the literature which make use of this technique, for example [LIU 13]. Although the authors of [LIU 13] combine three different decoupling techniques, their use of the parasitic structures is especially interesting. In the vast majority of papers, parasitic structures are placed next to the antennas on the same side of the substrate. The authors of [LIU 13] used a defected ground plane<sup>20</sup> and etched the parasitic elements on the substrate side which contains the defected ground as shown in Figure 2.4-3.



**Figure 2.4-3: Top view of substrate (left), and (right) bottom view (After [LIU 13]).**

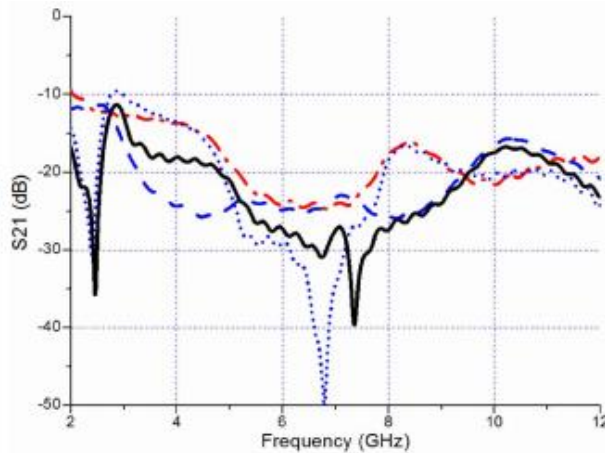
In Figure 2.4-3, the design consists of two planar monopole antennas which are placed orthogonal to each other. The antennas were designed for UWB operation (3.1 to 10.2 GHz). The antennas in the very compact design were printed on one side while the parasitic stubs were printed on the other side, the design also featured a defected ground plane as shown in the bottom view in Figure 2.4-3. Each monopole has a parasitic element which is a long stub, the parasitic stubs were oriented perpendicular to each other similar to the antenna orientations shown in Figure 2.4-3.

The simulated results obtained are presented in Figure 2.4-4. The isolation for the bandwidth of operation (3.1 to 10.2 GHz) is shown and is -15 dB or better. Similar results are achieved when only the horizontal parasitic stub is implemented. However, as reported by the authors, implementing only the horizontal stub, as opposed to vertical and horizontal stubs, results in 20

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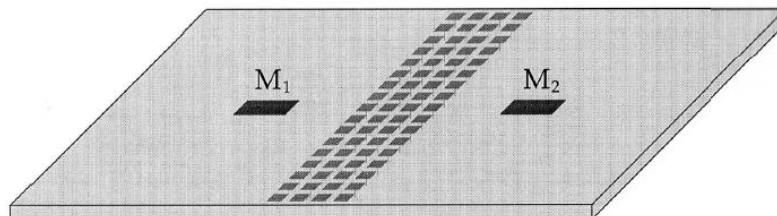
<sup>20</sup> Defected ground plane is a ground plane where certain areas of the grounding PEC material is removed. The parts which are removed (intentionally) are referred to as “defects”, hence the name defective ground.

dB increase for  $S_{11}$  in the 8 - 10 GHz bandwidth. As such two stubs are implemented to isolate the antennas and minimize the return loss.



**Figure 2.4-4:  $S_{21}$  Plots: (-.-) no parasitic stubs, (- - -) only vertical stub, (....) horizontal stub only, and (\_\_\_\_) both stubs (After[ LIU 14]).**

A thorough study presented in [YANG 03] focused on decoupling MEA antennas using electromagnetic bandgap (EBG) structures. An EBG structure is a periodic structure consisting of small parasitically-excited elements usually placed in between the antenna elements and is used to decouple the antenna elements by suppressing the surface waves present on the substrate. Effectively, the EBG can be thought of as a band-stop filter for surface waves. Two rectangular microstrip antennas operating at 5.8 GHz [YANG 03], with their E-planes coincident (“E-plane coupling” case is the term used) were considered as depicted in Figure 2.4-5, with the spacing between the patches being about 38.8 mm (or  $0.75 \lambda_{5.8 \text{ GHz}}$ ).



**Figure 2.4-5: EBG and two patch antennas coupled in E-Plane. (After [YANG 03])**

The EBG structure presented in [YANG 03] consists of 3 mm square mushroom-like patches and has an inter-element spacing of 0.5 mm. This mushroom-like EBG consists of 4 main components: a ground plane, a dielectric substrate, square patches, and vias which connect the patches to the ground plane. As seen in Figure 2.4-6, the introduction of the EBG resulted in 8 dB reduction in the coupling levels and enhanced the return loss of the antennas by about 3 dB. However, one general drawback of EBG's is the valuable space which is wasted in inter-element separation. Additionally, this particular mushroom-like EBG is inherently difficult to implement as it requires intricate manufacturing to connect the EBG elements' vias to ground.

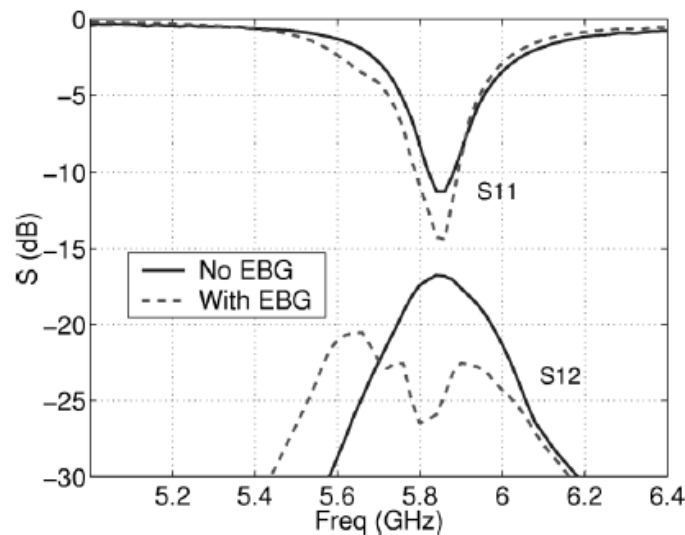


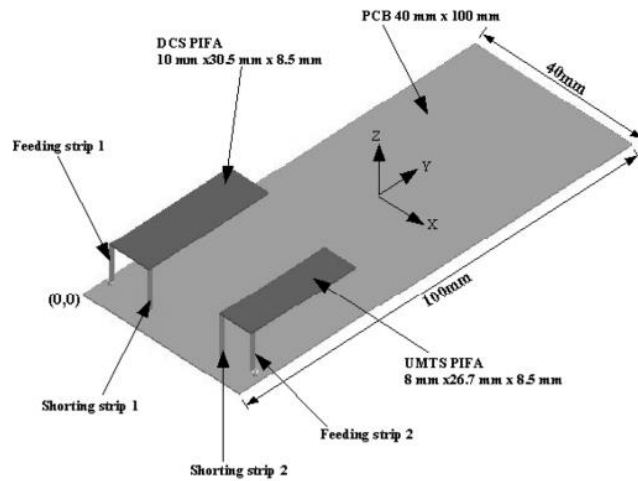
Figure 2.4-6: Measured results of microstrip antennas with and without EBG.

#### 2.4.4 Coupled Resonator Schemes

This method includes the neutralization line (NL) and the defected ground (DG) methods. The neutralization line (NL) is a microstrip link that connects two antennas and acts as a neutralization device by picking up some of the mutual coupling signal on antenna B and returning it out of phase to Antenna A, hence reducing or “neutralizing” the mutual coupling effect [DIAL 06]. Good results are achieved when the NL technique is implemented. However, it has been shown that this method is only effective for narrow band applications. Since the dimensions of microstrip lines are dependent on frequency, decoupling antennas at low frequencies such as the GSM band (800-

900 MHz) would require a NL with large physical area and is not feasible. Other coupled resonator methods include the defected ground plane method which entails removing parts of the PEC that makes up the ground plane of a substrate. The dimensions of the removed parts are selected carefully in order to reduce mutual coupling between the antenna elements. However, this method can lead to a decrease in the gain of the MEA. Specifically, in printed MEA's having a defected ground will always lead to a deteriorated Front-to-Back ratio in the far-field patterns. This is a result of radiation which “leaks” through the defected ground plane.

The first design intended to decouple antennas using the NL technique is [DIAL 06]. Although the term is not explicitly mentioned in this paper, the author makes use of a microstrip line suspended between two PIFA's to neutralize the coupling by producing an opposite coupling signal in order to enhance isolation. The PIFA's consisted of different sizes which were intended to operate at different frequencies. Both PIFAs were placed on top of a finite ground structure as shown in Figure 2.4-7.



**Figure 2.4-7: PIFA antennas on PCB (After [DIAL 06])**

Before arriving at the arrangement shown above, [DIAL 06] conducted a study to better understand how different positioning would affect the S-parameter response of the entire system. It was later found that placing the PIFAs in the above arrangement provides maximum isolation for the entire desired band of operation. Figure 2.4-9 shows the S-parameter response of the antenna arrangement shown above, where the shorting strips of the antennas are facing each other and the

antennas are separated by a distance of 18 mm ( $0.11\lambda_0$ ). These results were then taken as the reference case to quantify the effectiveness of the NL technique.

A parametric study was conducted to find the width and length parameters of the neutralization line which yield the best isolation. The authors of [DIAL 06] found that a width of 0.5 mm and a length of 18 mm yielded the best decoupling results. Figure 2.4-8 shows how the NL was implemented in [DIAL 06]. As shown in Figure 2.4-8, the NL was connected to each of the shorting strips of both PIFA's. Figure 2.4-9 shows the S-parameters of the PIFA's before the NL is implemented. As shown in Figure 2.4-9,  $S_{12}$  has a maximum value of -10 dB. After the implementation of the NL, the S-parameter graphs in figure B show a decrease in the coupling parameter, which now has flat response with a maximum value of about -20 dB. It should be noted that a frequency shift is also observed after the implementation of the NL; however, the antenna's operational bandwidths remain unchanged.

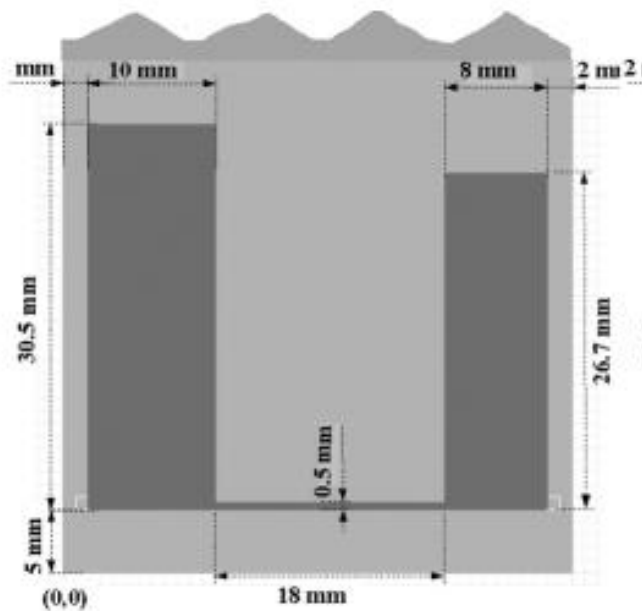


Figure 2.4-8: 2 PIFA's with NL's connected (After [DIAL06])

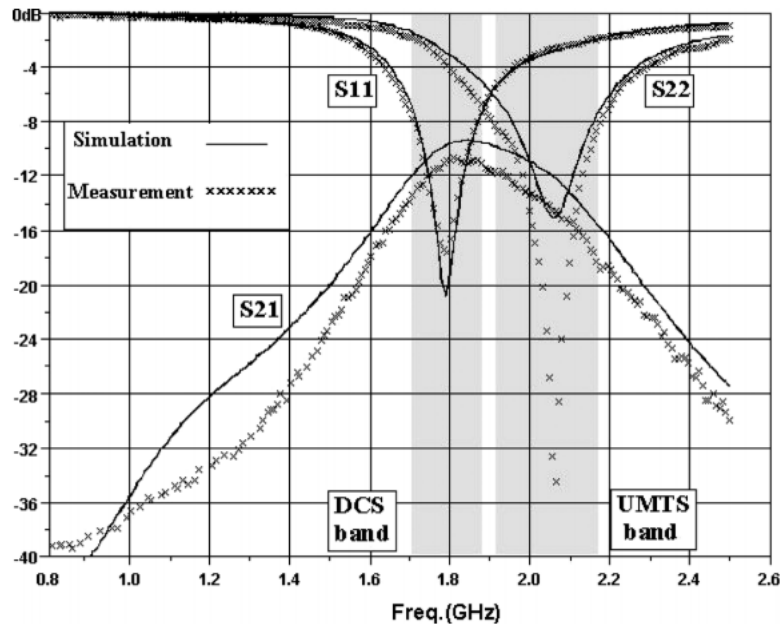


Figure 2.4-9: The S-parameter response before introducing the NL (After [DIAL 06])

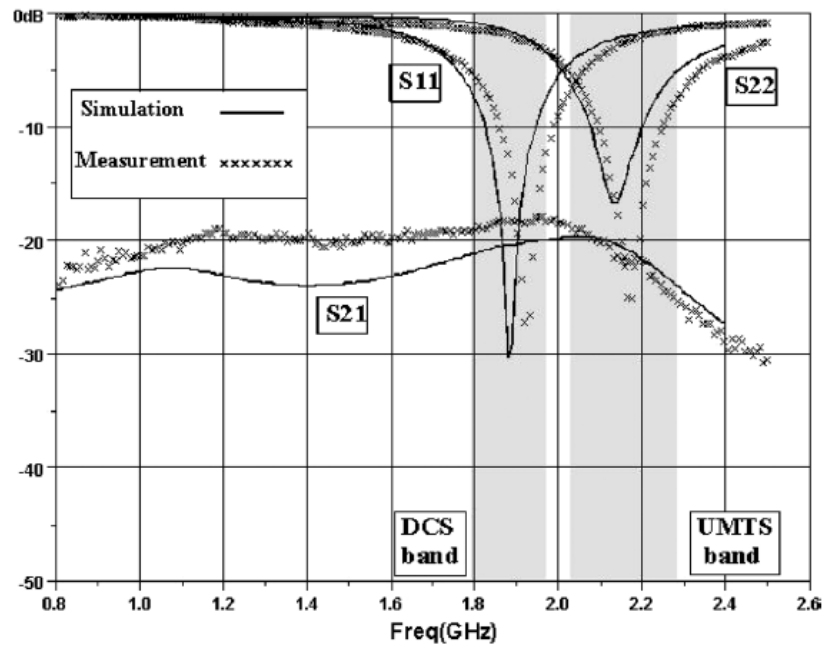


Figure 2.4-10: The S-parameter response after introducing the NL (After [DIAL 06])

During the parametric study which was conducted by [DIAL 06] for the NL, it was noted that increasing the width and/or the length of the NL, results in shift of the null in  $S_{12}$  graph. As such, if the antenna designer intends to decouple antennas at lower frequencies, a longer and/or thicker NL is needed. This relationship between the size of the NL and frequency is a limitation of this method. Although the NL provides good results for decoupling at high frequencies, implementing a NL on a mobile unit operating at lower frequencies (eg 900 MHz) is not feasible.

NL have been also reported to have narrow decoupling bandwidth for antennas other than PIFA's. There have been recent studies which were successful at introducing variations to the NL in an effort to increase the decoupling bandwidth. Among those studies is [ZHAN 16] where the two monopoles shown in Figure 2.4-11, were designed for an (3.1 – 5 GHz) Ultra Wide Band (UWB) MIMO application. As seen in Figure 2.4-11, a circular disk was inserted in the NL in order to widen the decoupling bandwidth of the NL. Moreover, the NL was placed on top of the ground plane to allow for a more compact design.

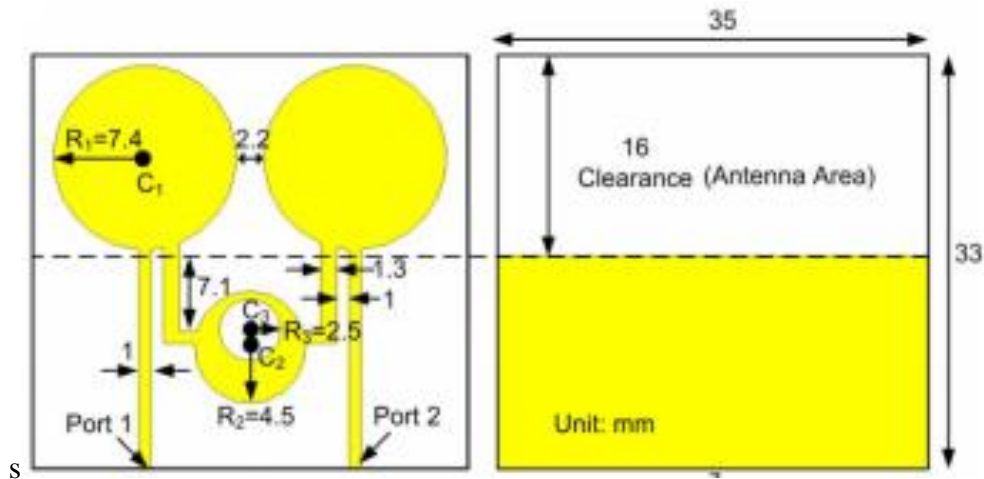


Figure 2.4-11: Design of two Antennas with NL (After [ZHAN 16])

Impressive decoupling results were achieved as shown in Figure 2.4-12 below, the maximum  $S_{21}$  value was reduced from -8 dB to -20 dB. However, the  $S_{11}$  graph value was degraded by a few dB. Nonetheless, the authors successfully decoupled the UWB monopole antennas using NL.

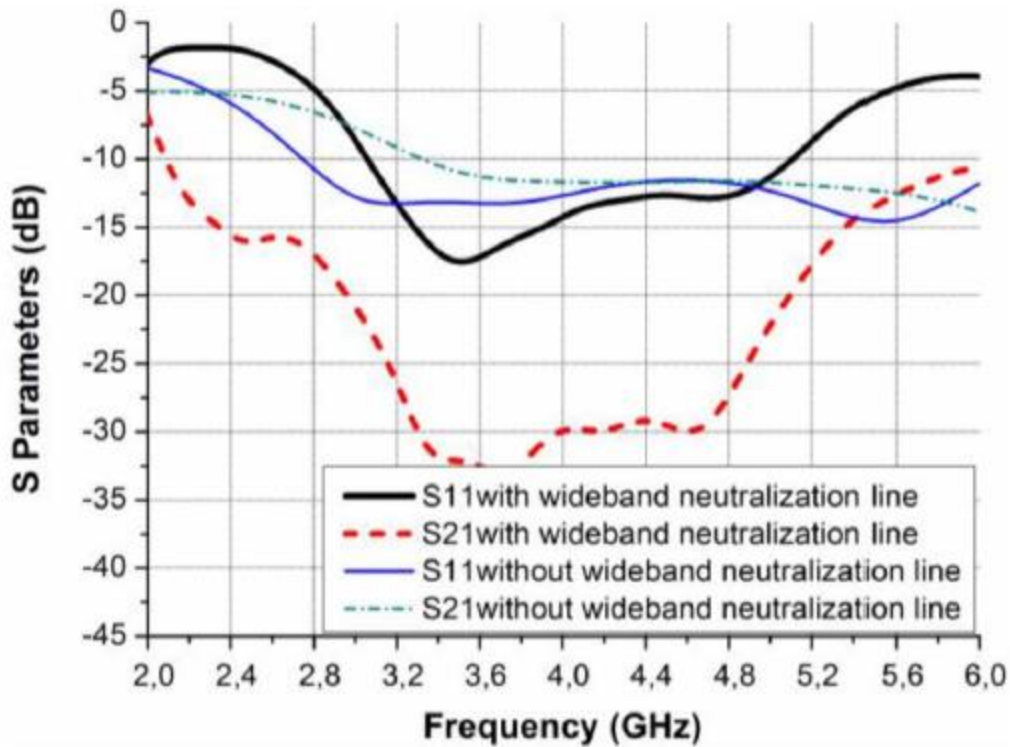


Figure 2.4-12: S-parameters with and without NL (After [ZHAN 16])

### 2.4.5 Decoupling Surface Schemes

The decoupling surface is similar to an EBG, however, one key difference is that the decoupling surface (DS) is not usually made up of regular shapes, such as rectangles or squares, and is not periodic. Additionally, EBG's are usually placed in the same plane of the antennas (coplanar) whereas DS can be implemented in a non-coplanar fashion. In such schemes a decoupling surface might or might not be coplanar with the radiating elements.

Before we begin the discussion of Decoupling Surfaces (DS), we must first take note of the differences between EBG's and DS. Designs which entail the use of EBG's have the EBG typically placed in between the antenna elements, whereas a DS can be placed anywhere near the antennas. As such, the DS offers more flexibility in design. Among the studies completed on DS schemes is that of [WEI 17] where a simple decoupling surface was used to decouple four printed quasi-yagi antennas, operating at 5.5 GHz. The antennas are co-linear as shown in Figure 2.4-13. The DS

(Decoupling Surface) is placed in the same plane as the antennas with a separation distance  $D_a = 0.38 \lambda_0$  and the inter-element spacing between the antennas is  $0.54 \lambda_0$ .

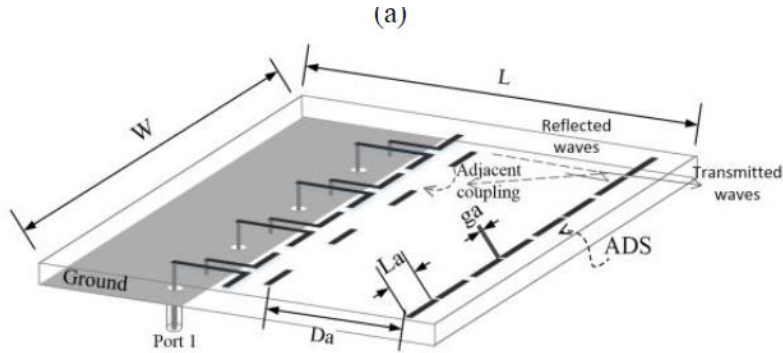


Figure 2.4-13: Quasi-Yagi antennas and antenna decoupling surface (After [WEI 16])

The results achieved by the decoupling surface are shown in Figure 2.4-14. An isolation improvement of -13 dB is noted for adjacent elements when comparing results before and after adding the DS. The coupling between the non-adjacent elements  $S_{14}$  deteriorated, however, the increase in coupling is only about 4 dB and the non-adjacent elements are still very well isolated compared to the adjacent elements. The return loss  $S_{11}$  is unchanged as shown in Figure 2.4-14. Moreover, the authors of this work also show that the radiation patterns remain unchanged both before and after adding the DS.

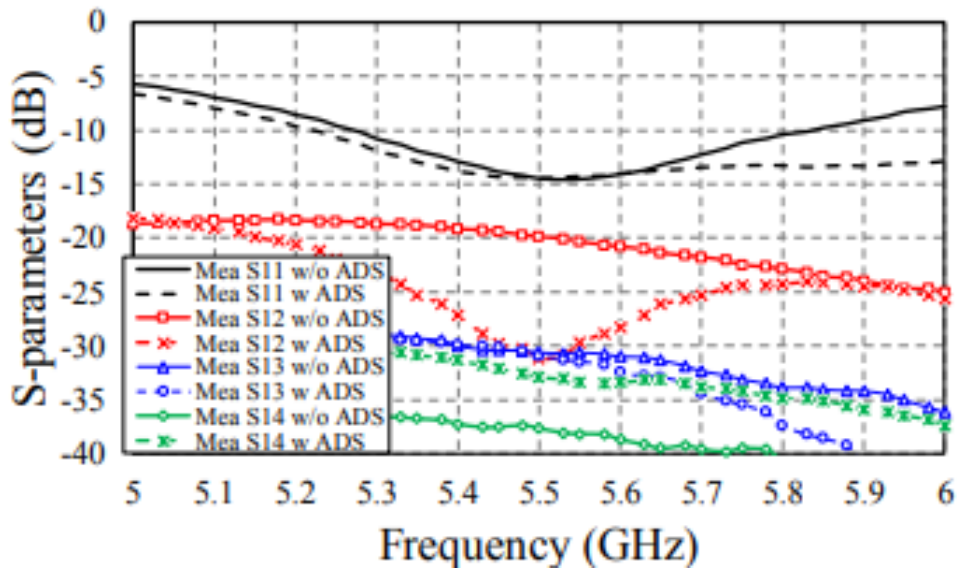


Figure 2.4-14: Effect of the DS on the S-parameters ( After [WEI 16])

A more sophisticated DS was designed in [WU 17]; a cross section of the design is shown in Figure 2.4-15. The setup comprised of 2-by-2 dual polarized dipole antennas. Each of the dual polarized antennas was made of two perpendicularly oriented dipole antennas, which the authors refer to as cross-dipole antennas. The individual antennas in the cross-dipole elements were oriented at  $45^\circ$  and at  $135^\circ$  with respect to the horizontal axis. This arrangement yielded a total of eight antenna elements. The antennas were designed to operate at 3 GHz – 3.8 GHz and were placed on top of a ground plane while the decoupling surface was etched on a separate substrate which was placed on top (non-coplanar) of the antenna elements.

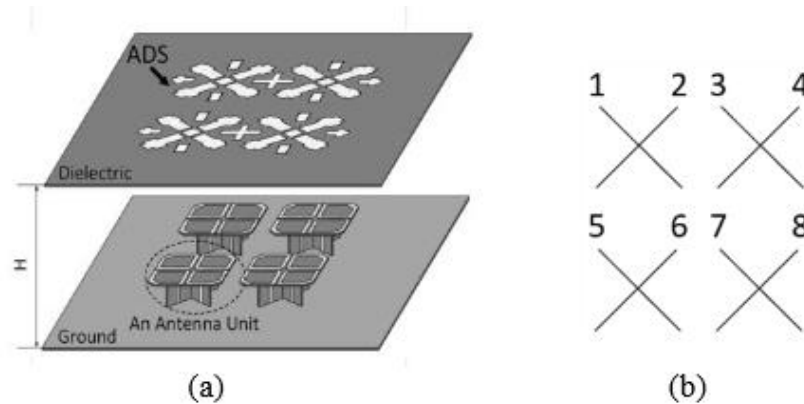


Figure 2.4-15: (a) Antennas and DS (b) Cross dipole arrangement (after [WU 17])

The substrate was selected to be a low-loss one with  $\tan(\delta) = 0.001$  and a thickness of 1 mm in order to maintain the radiation efficiency of the antennas. Horizontally adjacent antennas were separated by 45 mm ( $0.51 \lambda_0$ ) and vertically adjacent antennas were separated by 60 mm ( $0.68 \lambda_0$ ). The results reported in [WU 17] are shown in Figure 2.4-16 where it can be seen that that the decoupling surface greatly reduces the mutual coupling between co-polarized and horizontally adjacent elements as in the case of  $S_{13}$  which is reduced from -14 dB to -25 dB. In the case of cross-polarized horizontally adjacent elements, such as  $S_{14}$  and  $S_{23}$ , the coupling is slightly improved by 2 dB and about 6 dB, respectively.

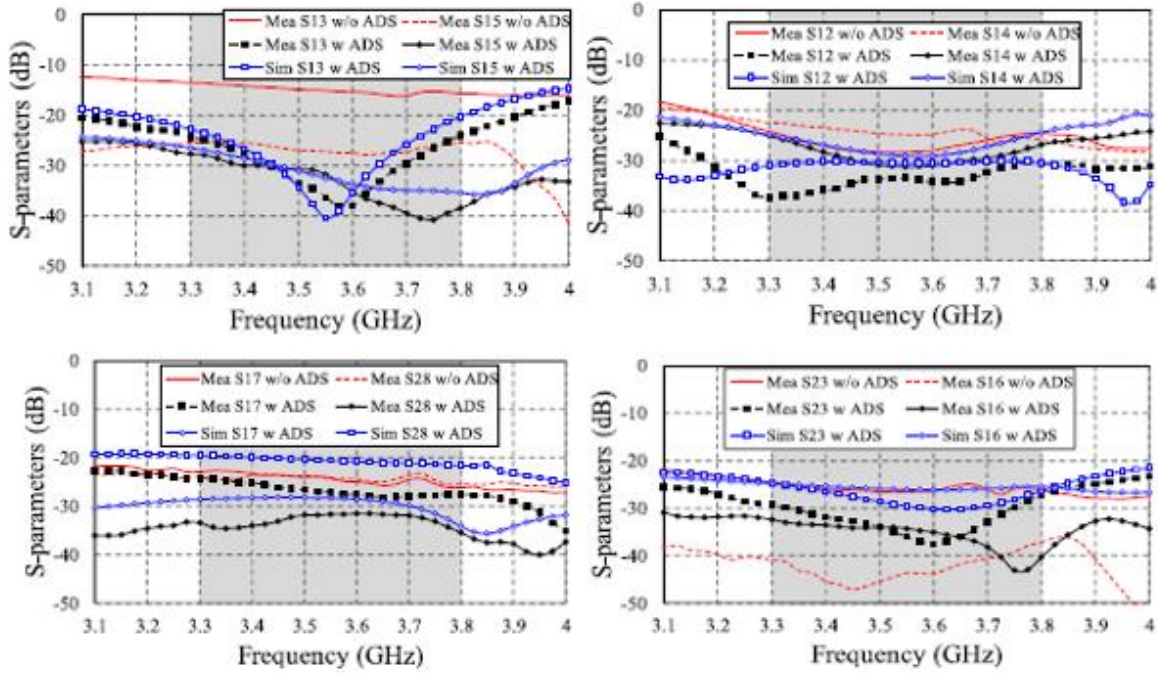


Figure 2.4-16: Simulated and Measured S-parameters (After [WU 17])

The coupling between cross-polarized vertically adjacent elements such as  $S_{16}$  is deteriorated from -40 dB to -32 dB, however, the mutual coupling level is still very low. As for non-adjacent cross-polarized dipoles such as  $S_{28}$ , the mutual coupling is inherently low as reported by the authors, however we can still note an improvement of about 5 dB for  $S_{28}$  and 2 dB for  $S_{16}$ . The authors also show measured results which depict no changes with respect to the return loss parameters  $S_{11}$  and  $S_{22}$ .

This work is presented for elements which are separated by distances greater than  $\lambda/2$ , which is known to be a distance at which the mutual coupling is reduced. However, the contribution of this work was that the DS is placed in a different plane with respect to the antenna, the authors of [Wu 17] were able to demonstrate how such surfaces can be used to reduce coupling for multiple antenna elements, maintain the cross-polarization and front-to-back ratio levels, and improve the antenna gain figures [WU 17].

#### 2.4.6 Existing Use of Optimization in MIMO Antenna Design

There is limited literature for MIMO MEA system designs involving optimization, and none that use actual shape synthesis<sup>21</sup>. In [KARL 09], three cases are presented as examples for utilizing particle swarm optimization in MEA design. In this study, a feeding network for the MEA's is optimized, the antenna's impedances as well as the feeding network are modeled in an equivalent circuit with lumped elements, the value of each of the lumped elements (capacitance, inductance or resistance) is then optimized in order to arrive at a desired design objective. In the first example included in [KARL 09], an RLC circuit is used as an equivalent model to a feed network for two dipole antennas of equal length, the circuit's lumped element values are then optimized to achieve a higher radiation efficiency. The optimization does result in ~ 8% increase in the radiation efficiency. The second example studies two inverted-F antennas (IFA's) which are placed in an orthogonal arrangement with respect to each other. The IFA's were designed to operate at 1.8 GHz, the portion of IFA's which extends parallel to the ground was then shortened (physically cut), and the optimization process was run for the feeding network of the antennas. Optimization was conducted for two objectives: first, to arrive at the highest possible diversity gain between both antennas and second, to recover the radiation efficiency lost after the antennas were shortened. Diversity gain is a measure of the effect of diversity on the communication system with respect to SNR, uncorrelated signals received on different channel paths will increase the SNR level at the receiver when these signals are combined [SHAR 13]. The measurement results of the second optimized design are shown in Figure 2.4-17 and Figure 2.4-18.

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<sup>21</sup> This is done in the present thesis for the first time.

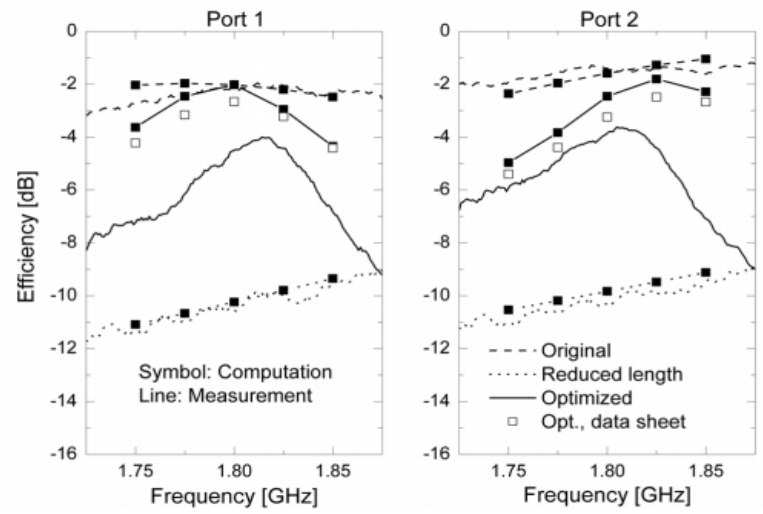


Figure 2.4-17: Efficiency of the antennas before and after optimization (After [KARL 09])

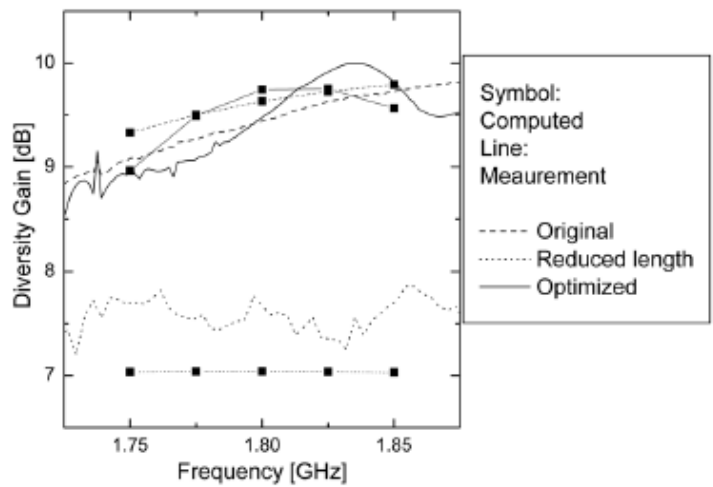
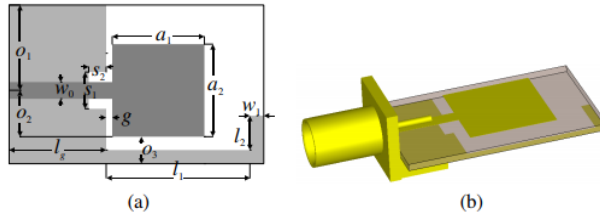


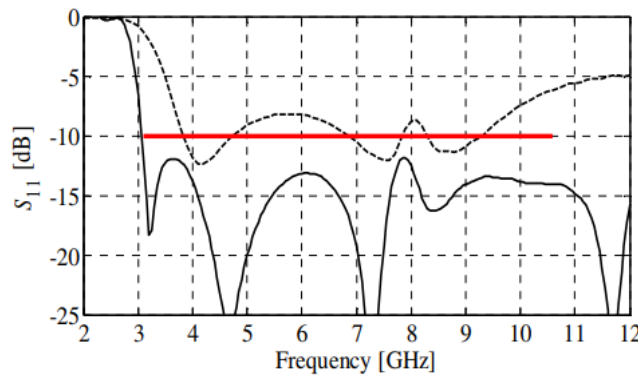
Figure 2.4-18: Diversity Gain before and after optimization (After [KARL 09])

In [BEKA 16], parameter optimization is performed for two monopole antennas in order to obtain a reduced size antenna with good electrical performance. The design process starts with a pre-defined geometry of the UWB printed monopole antenna in Figure 2.4-19. The dimensions of the geometrical features shown in Figure 2.4-19 are then optimised to reduce the value of some objective function defined in terms of the desired performance parameters, as needs to be done whenever optimization is used.



**Figure 2.4-19: Monopole parameters to be optimized, (b) 3D view of the monopole antenna (After [BEKA 16])**

The fitness function reported by the authors contains an objective function which selects the antenna with the smallest footprint ( $a_1 \times a_2$ ) as the most-fit antenna, as well as a constraint function which calculates the value of  $(\max\{|S_{11}|\} + S_{max})/S_{max}$ , where  $S_{max} = -12.5 \text{ dB}$ . Both the objective function and the penalty function are summed to yield the final value of the fitness function. Keeping in mind that the goal of the optimizer is to arrive at a design which minimizes the fitness function, if the value of  $S_{11}$  exceeds the threshold of  $-12.45 \text{ dB}$ , the fitness function value increases and the design is considered “less fit”. Similarly, if the design yields a large footprint, it is considered “less fit”. Figure 2.4-20 shows the simulated  $S_{11}$  results of the initial and the optimized antenna.



**Figure 2.4-20:  $S_{11}$  simulation results for initial (- -) and optimized (—) antennas (After [BEKA 16])**

After the optimization process is completed the “most-fit” antenna is replicated and combined by the designer to arrive at the final MIMO antenna design. Figure 2.4-21 shows the optimum design from [BEKA 16], while Figure 2.4-22 shows simulated results of the combined antennas in the final design. It should be noted that this study takes advantage of two other isolation methods in addition to optimization. First, there is a neutralization line connecting the two different antennas

as can be seen in Figure 2.4-21. Second, the antennas are placed in an orthogonal fashion with respect to each other, hence employing polarization diversity.

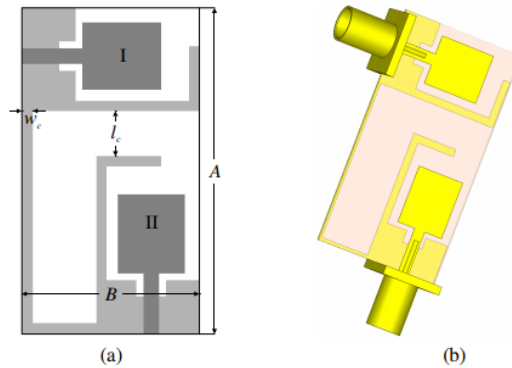


Figure 2.4-21: Arrangement of the optimized antennas (After [BEKA 16])

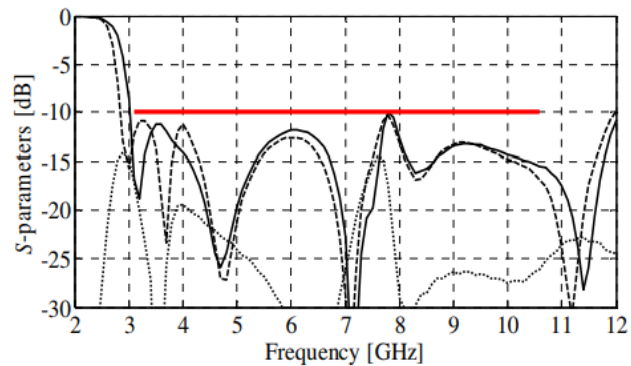
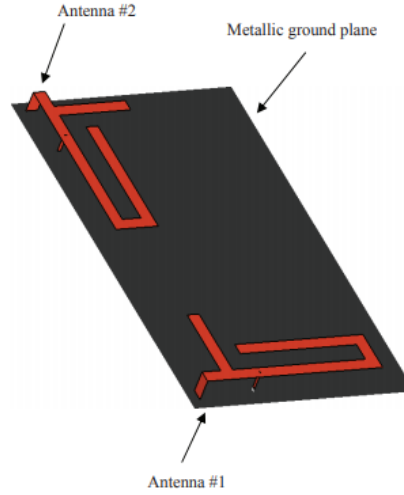


Figure 2.4-22: Simulated S-parameters of MIMO antennas:  $S_{11}$  (—),  $S_{22}$  (- -) and  $S_{12}$  ( ... ) (After [BEKA 16])

Another paper which discusses the use of optimization is [DERN 11], where antennas shown in Figure 2.4-23 are optimized. The way that this study tackles the requirements of a MEA system shows similarity to [BEKA 16]: essentially, the antenna's shapes and arrangement are manually selected by the designers based on their experience. The optimizer is then used to optimize the widths and lengths of the different antenna arms shown in Figure 2.4-23. The arrangements used here again makes use of polarization diversity through orthogonal placement of the antennas.



**Figure 2.4-23: Initial antenna arrangement "pre-optimization" (After [DERN 11])**

The objectives are selected to maximize the radiation efficiency, minimize the envelope correlation coefficient (ECC), and to minimize the total enclosed antenna area. The authors describe the scattering efficiency of an element “a” as  $\eta_x$ .

$$\eta_1 = 1 - |S_{11}|^2 - |S_{21}|^2 \quad (2.4-1)$$

$$\eta_2 = 1 - |S_{12}|^2 - |S_{22}|^2 \quad (2.4-2)$$

The “scattering efficiencies” for both elements can then be averaged,

$$\eta_{avg} = \frac{\eta_1 + \eta_2}{2} \quad (2.4-3)$$

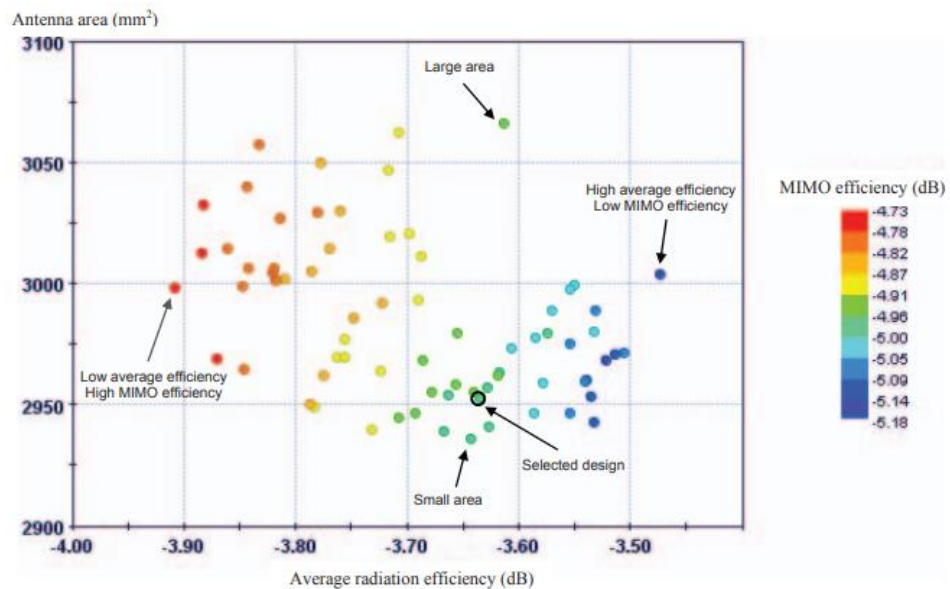
The ECC can be calculated below as shown below,

$$\rho_e = \frac{|S_{11} S_{12}^* + S_{12} S_{22}^*|}{\eta_1 \eta_2} \quad (2.4-4)$$

The authors of [DERN 11] then use what they refer to as MIMO efficiency which is calculated from the correlation coefficient

$$\eta_{MIMO} = \sqrt{\eta_1 \eta_2 (1 - \rho_e)} \quad (2.4-5)$$

Each of the above objectives (antenna area,  $\eta_{MIMO}$  and average radiation efficiency) was treated as a separate fitness function therefore the designer had to select the weights for each of the objectives. It should be noted that this is a multi-objective optimization which is different from the single objective optimization where the values are summed, as was done in [BEKA 16]. Using a multi-objective approach allows the designer to select the weights of the objectives after the simulation begins. In contrast, in a single objective design environment the weights of each term must be determined by the designer before the optimization process starts.



**Figure 2.4-24: Different results of the simulation.**

Figure 2.4-24 shows a selected sample of the results achieved by different design trials of the optimizer. The vertical axis represents the total area (footprint) of the antenna in millimeters, the horizontal axis represents the average radiation efficiency in decibels ( $\eta_{avg}$ ), and the color gradient represents the “MIMO efficiency” ( $\eta_{MIMO}$ ) which is used by the author to calculate the correlation coefficient. The optimum design selected is highlighted with a circle in Figure 2.4-24, this selection is based on a trade-off between all three objectives considered in this study (average radiation efficiency, antenna area, and MIMO efficiency). The weights of the optimum design were found to be 0.38, 0.36 and 0.26 for ( $\eta_{avg}$ ), ( $\eta_{MIMO}$ ) and antenna area, respectively. The

simulated results of the optimized design are shown in Figure 2.4-25 and Figure 2.4-26. Unfortunately, the design does not report any initial (pre-optimization) results for comparison.

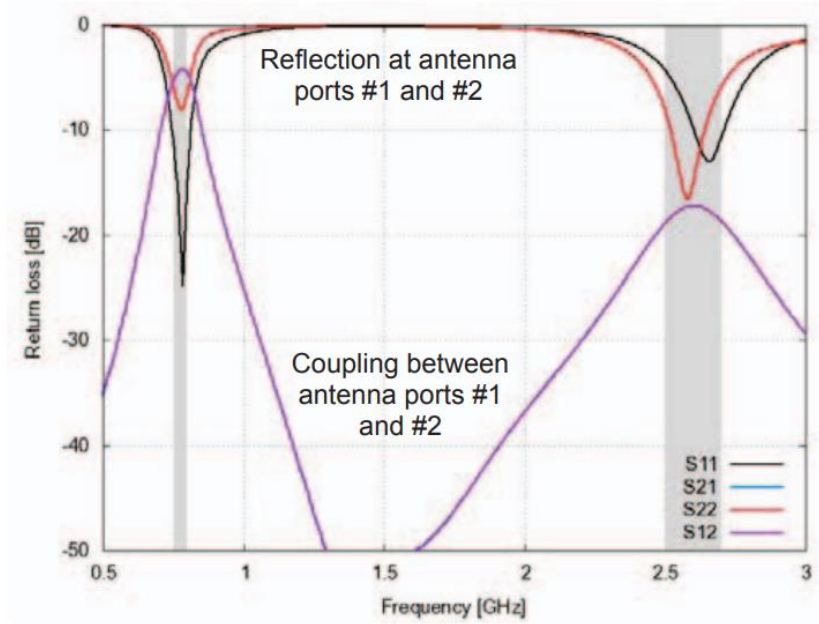


Figure 2.4-25: Simulated S-parameter of the optimized design (After [DERN 11]).

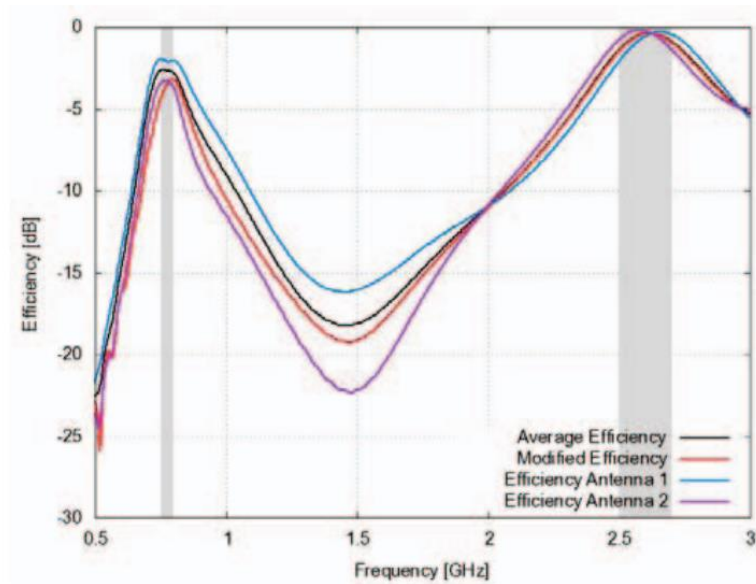


Figure 2.4-26: Simulated radiation efficiencies, average efficiency, and MIMO efficiency (After [DERN 11]).

## 2.5 Computational Approach Harnessed in the Thesis for Shape Synthesis

The research in this thesis makes heavy use of computational electromagnetics (CEM) to achieve its goals. It does not, however, undertake research or development on CEM methods themselves. Instead, it uses the commercially available computational engine known as the Hyperworks/FEKO suite [FEKO 18]. The background material is truly well-known, and adequately described in the textbook literature, amongst which are [PETE 97], [VOLA 12] and [KRIS 16]. The structure under study is composed of perfectly conducting shapes located on dielectric substrates of finite thickness, some of which may be backed by perfectly conducting “ground planes”. The electromagnetics problem to be modelled is driven by concentrated sources that allow one to define antenna feedpoint locations (ports). The electric current density on the conducting surfaces forms the unknown quantity in an integral equation that constitutes the rigorous full-wave electromagnetic model of the geometry. This model is solved numerically using the moment method (MM), the presence of the substrate material being accounted for through the use of a special Green’s function [JAKO 00] in the integral equation. Once the current density has been found in this way, the electromagnetic fields everywhere (eg. far-zone patterns; near-zone fields) may be found, and all the antenna figures of merit<sup>22</sup> defined in Section 2.3.

A software code associated with FEKO, called Hyperstudy [HYPE 20], allows the user to divide the conducting surfaces into rectangular cells, define objective functions in terms of the performance of the antenna structure being modelled, and perform shape optimization using a genetic algorithm inherent to [HYPE 20] by removing/retaining these cells. In Chapters 3 and 4 we will use this to shape antenna decoupling surfaces, and so more detail will be provided there.

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<sup>22</sup> Not all are computed automatically by FEKO. In such cases the author has developed scripts (in MATLAB) that use the FEKO outputs to do this.

## 2.6 Conclusions

This chapter assembled the definitions of, and useful expressions for, the performance metrics needed to evaluate MEA systems. Some numerical studies were performed to assess and compare (eg. for ECC and TARC) the suitability of some of these expressions for the stated purpose. These will be used in Chapters 3 and 4 in order to define objective functions whose minimization, accomplished through an optimization algorithm that controls the shape synthesis of the MEA structure (its overall geometry, namely the elements proper and any decoupling surfaces) to arrive at an MEA shape that has good performance. The different approaches used by others to achieve MEA systems with low inter-element coupling were also surveyed, in order to put the new approaches developed in this thesis, which follows this chapter, into perspective.

## Chapter 3 The Shape Synthesis of Coplanar Decoupling Surfaces

### 3.1 Goals

The series of computational experiments presented in this chapter will feature a parasitically excited rectangular structure placed between two printed rectangular microstrip patch antennas. The parasitic rectangular structure will be used as a decoupling surface (DS). After the shape synthesis process is completed, we expect to arrive at a completely different geometric shape than the one we started with. The ultimate goal driving the shaping process is the reduction of mutual coupling ( $S_{12}$ ) as well as improving other multi-element antenna (MEA) figures of merit for the design. It should be noted that for all the numerical experiments presented in this chapter, both radiating elements and the decoupling surface are placed on the same geometrical plane.

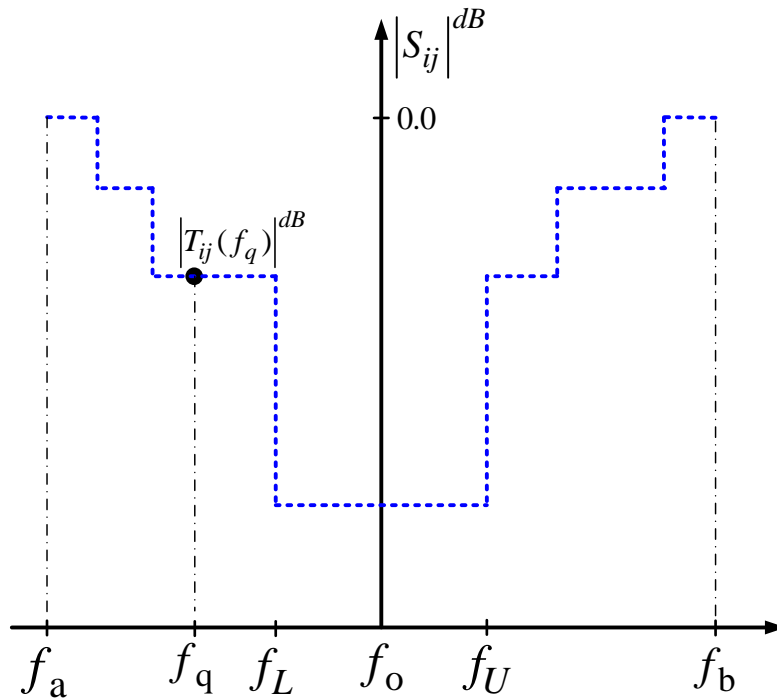
Section 3.2 describes the construction of the objective functions that will be used in the shape synthesis work of this chapter. The computational basis/tool for the implementation of the shape synthesis process has already been mentioned in Section 2.5; some further details are given here in Section 3.3. The shape synthesis example (the first to be shown in this thesis) in Section 3.4 concerns two adjacent microstrip patches, separated along their common H-plane, with a decoupling surface between them. The shape synthesis example in Section 3.5 is for two such patches that are separated along their common E-plane. Some additional problem-specific comments on the topic will be included in Section 3.4.5 and Section 3.5.5, the need for such additions being explained there as well. Section 3.6 discusses polarizations of the antennas and Section 3.7 concludes the chapter.

### 3.2 Objective Functions

The objective function is an important quantity as it must encapsulate the set of performance goals such that, once it has been minimised through execution of the shape synthesis process, the resulting MEA actually has the desired performance. The numerical value that the objective function yields is used to determine how “*fit*” an offspring geometry is. We have found that objective functions can be hard to realize without a few trials. Several mathematical expressions can be used when wanting to minimize the mutual coupling ( $S_{12}$ ) parameter for instance. One can for example use a very simple expression where  $F_{Obj} = \min(\text{avg}(S_{12}))$  or  $F_{Obj} =$

$\min (\max(S_{12}))$  and so on, where the average is taken over a set of frequencies (about some center frequency) where  $S_{12}$  is shaped to be low. From our numerical experiments, we found that use of the averaging objective function achieved good results, but it resulted in a shift of the lowest value away from the center. On the other hand, we were not able to achieve good isolation using the  $\min (\max(S_{12}))$  objective function.

Several other objective function variations, which will not be discussed, were tested in this numerical experimentation. It was found that the best approach is to employ a *mask* objective function. When we speak of a mask, we are referring to a line that is drawn by the antenna designer, and which resembles the desired S-parameter response the designer would like to achieve. An example of a mask objective function is shown in Figure 3.2-1.



**Figure 3.2-1: An example of S-parameter mask function**

The dashed line mask selected by the designer is denoted as  $T_{ij}$ . We see that in this example the mask is “relaxed” around frequencies  $f_a$  and  $f_b$ , but tougher at frequencies close to the centre frequency  $f_o$ . The mask is specified at several frequency points  $\{f_1, f_2, \dots, f_q, \dots, f_Q\}$  in the band

of interest. We then subtract the mask value (desired,  $T_{ij}$ ) from the simulated S-parameter (actual,  $S_{ij}$ ) of the device under design value for each frequency point of interest as

$$e_{ij}(f_q) = \begin{cases} |S_{ij}(f_q)|^{dB} - |T_{ij}(f_q)|^{dB} & \text{if } |S_{ij}(f_q)|^{dB} - |T_{ij}(f_q)|^{dB} < 0 \\ 0 & \text{Otherwise} \end{cases} \quad (3.2-1)$$

for  $i, j = (1,2)$  and  $q = 1,2, \dots, Q$ .

The difference value  $e_{ij}(f_q)$  is referred to as the error contribution; the term “error” is used as this value captures the discrepancy between the desired value and the actual value of  $S_{ij}$ . Notice that for values where  $S_{ij}(f_q) \leq T_{ij}(f_q)$  the error contribution is zero. This is because any value of  $S_{ij}(f_q)$  which is less than or equal to the (desired) mask  $T_{ij}(f_q)$  value is considered a “good” value that meets the objective and therefore should not contribute to the error sum. In other words, if  $S_{ij}(f_q)$  is less than or equal to the mask value, the model actual performance is considered to be “fit “ at this frequency point, and no further enhancement is needed by the algorithm. On the other hand, an error value greater than zero signifies the actual performance which has not met the fitness requirement, the shaping algorithm will then try a different shape to achieve an error value of zero. We can similarly set up a masked objective function value for the ECC as

$$e_{ECC}(f_q) = \begin{cases} |ECC(f_q)|^{dB} - |T_{ECC}(f_q)|^{dB} & \text{if } |ECC(f_q)|^{dB} - |T_{ECC}(f_q)|^{dB} < 0 \\ 0 & \text{Otherwise} \end{cases} \quad (3.2-2)$$

for,  $q = 1,2,\dots,Q$ .

After the error contributions are calculated for each performance-related quantity, the sum of the various error contribution terms is calculated in order to obtain the component objective functions. For example, the component objective function,  $F_{obj}^{11}$ , is the sum of error contribution terms of

$S_{11}$ ,  $F_{obj}^{12}$  is the sum of error contribution terms of  $S_{12}$ , and  $F_{obj}^{ECC}$  is the sum of error contribution terms of  $ECC$  <sup>23</sup>.

$$F_{obj}^{11} = \sum_{q=1}^Q |e_{11}(f_q)|^2 \quad (3.2-3)$$

$$F_{obj}^{12} = \sum_{q=1}^Q |e_{12}(f_q)|^2 \quad (3.2-4)$$

$$F_{obj}^{ECC} = \sum_{q=1}^Q |e_{ECC}(f_q)|^2 \quad (3.2-5)$$

The final step of formulating the objective function is to combine the component objective functions into a single expression, which allows the shaping algorithm to assign a single fitness value to a particular shape. This is done by squaring each error term then multiplying each component objective function by a weight value and finally taking the sum of the products (sum-of-squares) as seen in the expression below.

$$F_{obj} = w_1 F_{obj}^{11} + w_2 F_{obj}^{12} + w_3 F_{obj}^{ECC} \quad (3.2-6)$$

$$w_1 + w_2 + w_3 = 1 \quad (3.2-7)$$

Since the genetic algorithm will work to reduce the objective function's value, the error term contribution must always be a positive number, therefore the absolute value is taken when the terms are combined in the objective function. The weight values allow the shaping algorithm to set a preference for one objective component over the others. For example, if we were to set  $w_1 = 0$ ,  $w_2 = 1$ ,  $w_3 = 0$ , the algorithm would focus all of its efforts on minimizing  $S_{12}$  and simply ignore minimizing  $S_{11}$  and  $ECC$ . But doing so could yield an antenna that has a very high  $S_{11}$ ,  $ECC$ , and a very low  $S_{12}$ . Clearly, an "antenna" with high return loss would not be of practical use.

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<sup>23</sup> The objective function can include other design parameters as well such as directivity for example, although in this work far-zone pattern related quantities will be incorporated instead as constraints.

Thus, care must be taken when selecting the weighting values. In this thesis, the value for a given weighting term was selected by considering the importance of achieving the design goal corresponding with the given weighting term, as well as the level difficulty associated with achieving the desired design goal.

### 3.3 The Shape Synthesis Process

The shape synthesis process begins by setting up an initial antenna geometry. This initial shape is referred to as the mother structure<sup>24</sup>. Each cell is encoded with a binary design variable equal to one or zero. In genetic algorithm (GA) terminology, this is referred to as a gene. If the gene variable is equal to one then the metallic cell of the structure physically exists, whereas a gene equal to zero means that the metallic cell is removed, forcing the current on the corresponding cell of the model to be zero. The mother configuration is then exported into the GA optimizer in HyperStudy, along with the number of genes and their values. HyperStudy then triggers the CEM simulator FEKO in order to simulate the mother structure. The simulation results include the antenna figures of merit which are to be optimized. The simulation results are passed to HyperStudy allowing it to build a reference case for fitness evaluation of future shapes.

The desired objective function specified in HyperStudy is evaluated for the mother structure. At this point the GA algorithm can be run in HyperStudy to begin the shape optimization process. There are a number of parameters related to the GA, these parameters are configurable in HyperStudy at the designer's discretion, and these include:

- **Population:** This value controls the number of individuals (shapes) in a single generation (iteration). A generation is one iteration of the GA algorithm.
- **Chromosome:** The genetic material that represents an individual, for the purpose of this thesis a chromosome will be a binary string of 1's and 0's.
- **Minimum Number of Generations:** This value allows the user to set a minimum number of generations (iterations); the GA is not allowed to terminate before reaching the minimum number of generations.

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<sup>24</sup> We adopt the terminology used in [JOHN 99].

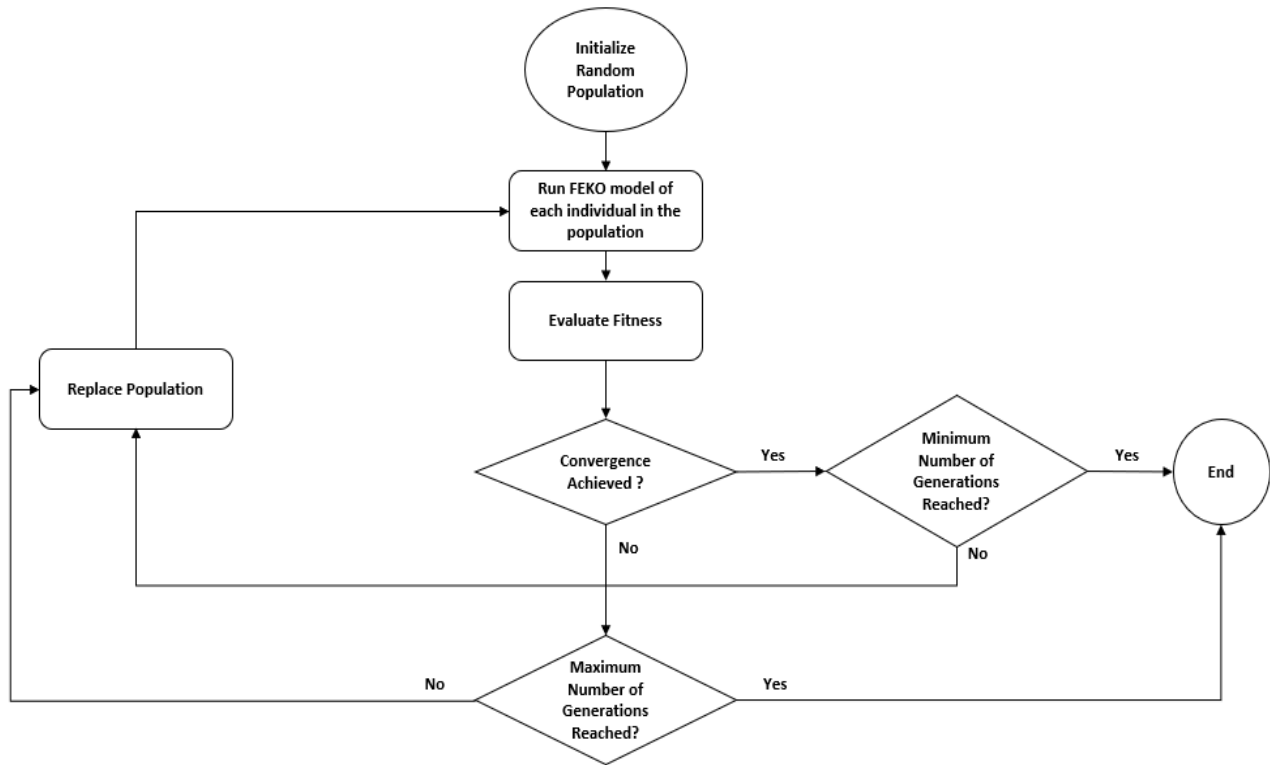
- **Maximum Number of Generations:** This number allows the user to set a maximum number of generations; once reached the optimizer will terminate whether or not convergence has been achieved.
- **Convergence:** This is the optimum fitness value, an individual (shape) achieving this value indicates that the desired objective function value has been met.
- **Mutation:** A value typically set between 0.01 – 0.1, it allows the GA to negate a random gene in a chromosome. Mutation is used by the GA to add new genetic material to the population pool, which allows it to better explore the solution space.

After the GA settings are selected, the iteration process of the GA begins by initializing a starting population<sup>25</sup>. The sequence that then takes place within HyperStudy is outlined in the flowchart in Figure 3.3-1. In the first step, the GA randomly selects the chromosomes of each individual (shape), these individuals make up the initial population. The fit individuals<sup>26</sup> of a population are carried forward to the next generation. Populations are replaced with new offspring, which are generated by crossing the chromosomes of fit individuals, mutation also occurs in this step. This iterative process carries on until an individual is found with an acceptably low objective function value (“convergence”) or the maximum number of generations is reached. In this thesis, the values for the different GA parameters were selected in a two-step process. We first selected values based on what was used in previous studies that used GA for antenna optimizations such as [JOHN 97] and [JOHN 99], some values were then slightly adjusted based on trial and error.

---

<sup>25</sup> This will be a population of different structure shapes.

<sup>26</sup> These shapes with the lowest objective function values.



**Figure 3.3-1: Flowchart describing the HyperStudy GA operation**

Since GA is a global search algorithm, its final results are independent of the initial population [JOHN 97]. As such, in most GA codes, the starting population is randomly selected from the solution space. During the population replacement step, fit individuals mate to produce individuals of the subsequent population called offspring. Mating occurs by copying the first half of the chromosome of parent one and the second half of the chromosome of parent 2 and combining them to make up a complete chromosome for the offspring. It should be noted that unfit individuals are also selected for mating to include more diversity in the population. Population diversity is an important factor in GA, as it is often the case that the fittest individuals within the population at some iteration are far from the optimal solution. Population diversity is needed to expand the search within the solution space. Mutation is also used in the GA to increase the population diversity; it is a simple mechanism whereby a single gene in an individual's chromosome is changed from a binary one to binary zero, or vice versa.

## 3.4 Linearly Polarized Microstrip Patches: H-Plane Coupling

### 3.4.1 Goals

In this section, a simple antenna configuration is shaped in order to achieve low mutual coupling while maintaining the return loss levels. Figure 3.4-1 shows the initial antenna arrangement before the shaping. This configuration is comprised of two patch antennas intended to operate within the ISM band with a center frequency  $f_0 = 5.7$  GHz and a bandwidth of 300 MHz. The antennas are placed along the x-axis and a decoupling surface is inserted between them. An infinite substrate and ground plane are used in the FEKO model with  $\epsilon_r = 4.5$ ,  $\tan(\delta) = 0.0037$ , and a thickness of 1.57 mm. Since the microstrip antennas are positioned collinearly along the H-plane we will refer to this as the H-plane co-planar coupling case, this is to follow the terminology used in similar works in the literature [BALA 16], [YANG 03]. The antenna arrangement is geometrically symmetrical about the yz-plane, and therefore  $S_{11} = S_{22}$ . Since the materials used are isotropic  $S_{12} = S_{21}$ . Hence, we will only consider  $S_{11}$  and  $S_{12}$  to capture the return loss and mutual coupling of the antennas.

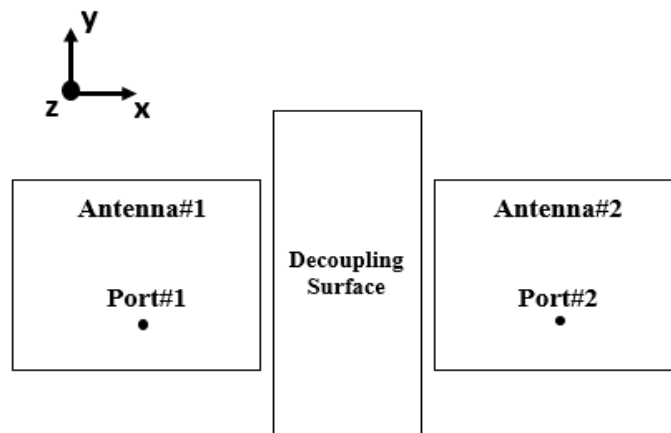


Figure 3.4-1: Unshaped MEA (mother structure)

### 3.4.2 Single Probe-fed Microstrip Patch

The first step in this study is to analyze the properties of a single microstrip patch antenna, which is placed on an infinite substrate. This patch was designed using the cavity model approximation [BALA 16] to obtain the initial length, width and probe feed location. The same parameters were then “tuned” using FEKO modeling to arrive at the microstrip antenna shown in Figure 3.4-2. The diameter of the feed probe is 0.8 mm.

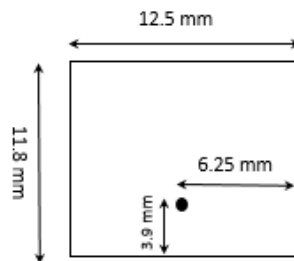


Figure 3.4-2: Dimensions of a single probe-fed microstrip patch antenna

We begin the numerical experiment by examining the properties of the single antenna’s reflection coefficient in addition to the far-field radiation pattern.

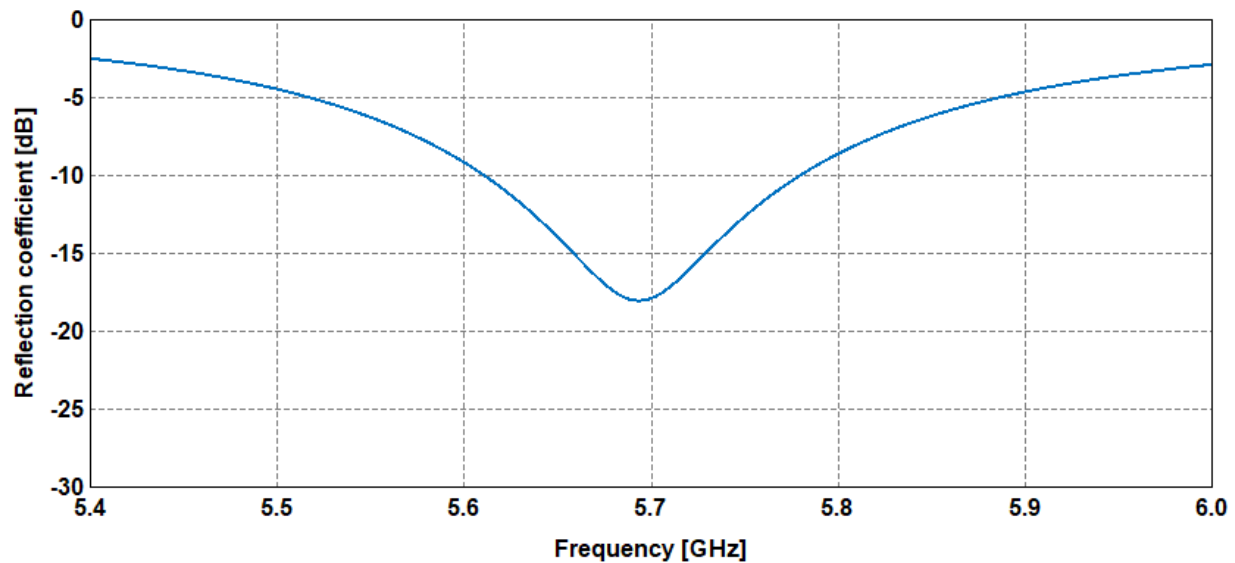
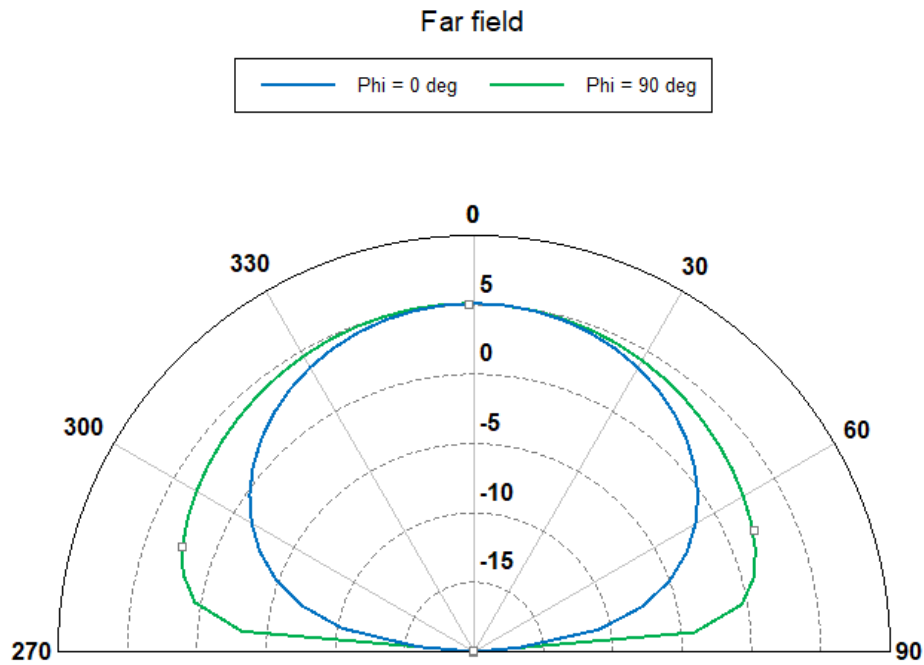


Figure 3.4-3: Computed reflection coefficient for the single patch antenna in Figure 3.4-2

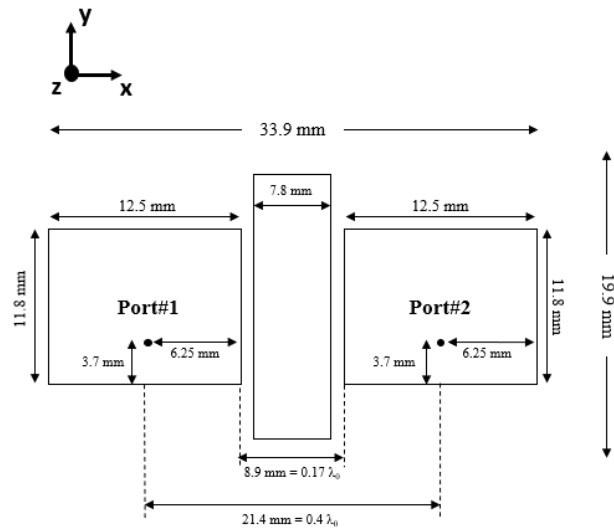


**Figure 3.4-4: Computed total directivity patterns for a single antenna at 5.7 GHz. The pattern coordinate system is shown in Fig 3.4-1, and defines angle  $\phi$ .**

Figure 3.4-3 and Figure 3.4-4 show that the single microstrip patch antenna has good properties considering that pin-fed microstrip antennas offer a bandwidth of 2%-5% [POZA 92]. Given that a reflection coefficient value of even -6 dB is acceptable for mobile communication applications [REHM 18] the bandwidth of this patch antenna is 5.3%.

### 3.4.3 MEA consisting of two microstrip patch antennas

Having examined an isolated single antenna, we add a second one shifted by 8.9 mm ( $0.17\lambda_0$ ) as shown in Figure 3.4-5. This will establish the microstrip antenna's performance as an element of MEA, when there is no decoupling method applied between the MEA elements.



**Figure 3.4-5:Dimensions of MEA design**

Figure 3.4-6 shows the simulated results for the geometry in Figure 3.4-5. The values of  $S_{11}$  are below  $-5$  dB across the entire bandwidth while the mutual coupling is lower than  $-12$  dB. In Figure 3.4-7, we see a diagram of the surface current on the structure when port 1 is excited and port 2 terminated with a matched load. The magnitude of the induced surface current density at the location of port#2 is about 12 dB below that at the driven port#1. Having an induced current on the load-terminated antenna is an undesired result of mutual coupling which we hope to reduce using shape synthesis. We aim to reduce the mutual coupling across the 300 MHz bandwidth from 5.55 GHz to 5.85 GHz.

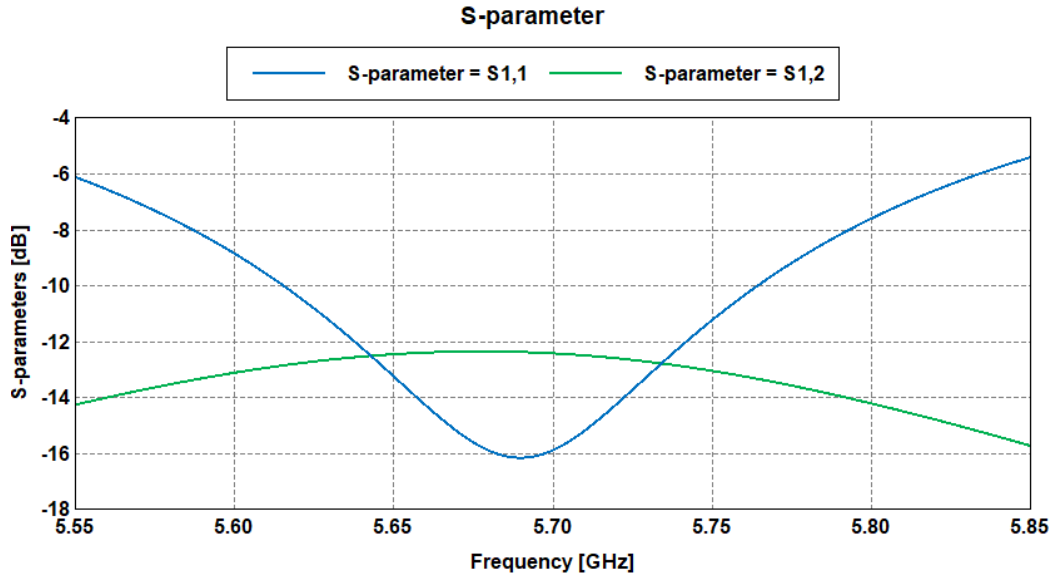


Figure 3.4-6: Computed S-parameter plot for MEA

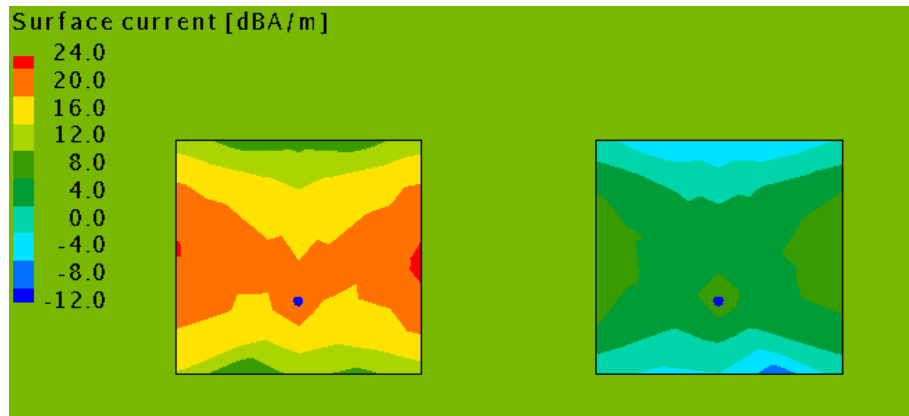
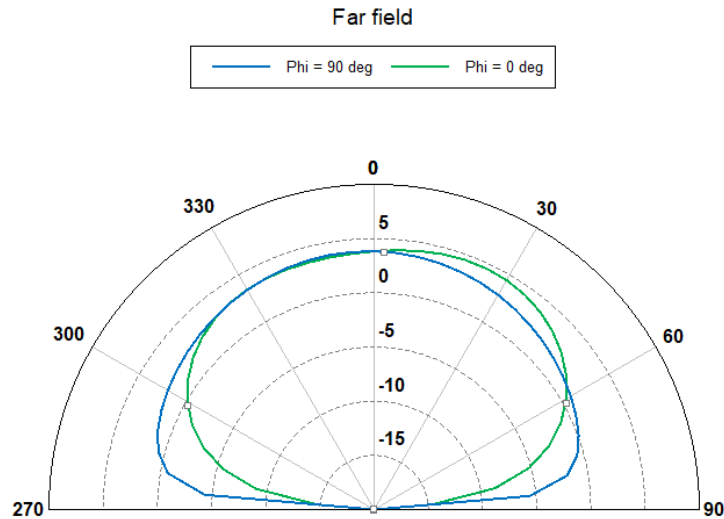


Figure 3.4-7: Surface current density for MEA at 5.7 GHz

To ensure that the microstrip patch remains functional as antenna we will also consider the radiation pattern when the second patch is added to the geometry. Figure 3.4-8 shows the far-field pattern when antenna 1 is excited and antenna 2 terminated in a matched load. Comparing this far-field pattern to the pattern of a single antenna shown in Figure 3.4-4, we note that in the  $\phi = 0$  cut

(xz-plane) the pattern is slightly skewed towards antenna 2, with the maximum directivity  $D(\theta = 30^\circ, \phi = 0^\circ) = 5$  dBi. We will aim to correct this tilt in the peak directivity by using the shape synthesis process to have the peak point back in the broadside direction.



**Figure 3.4-8: Computed total directivity patterns of antenna 1, when antenna 2 is terminated in a matched load, at 5.7 GHz.**

### 3.4.4 Insertion of an unshaped decoupling surface

An unshaped decoupling surface (DS) was placed in between the antennas as shown in Figure 3.4-9. Thus maintaining the original separation distance between the antennas. As expected, we notice some change in the S-parameter plots of the antennas now shown in Figure 3.4-10 compared to that in Figure 3.4-6. The  $S_{11}$  values where the lowest value deteriorated slightly, but the mutual coupling  $S_{12}$  was reduced by 4 dB to less than  $-16$  dB. This reduction is an expected result of adding the DS surface, which is that it acts as a “shield” between the antennas’ radiating edges.

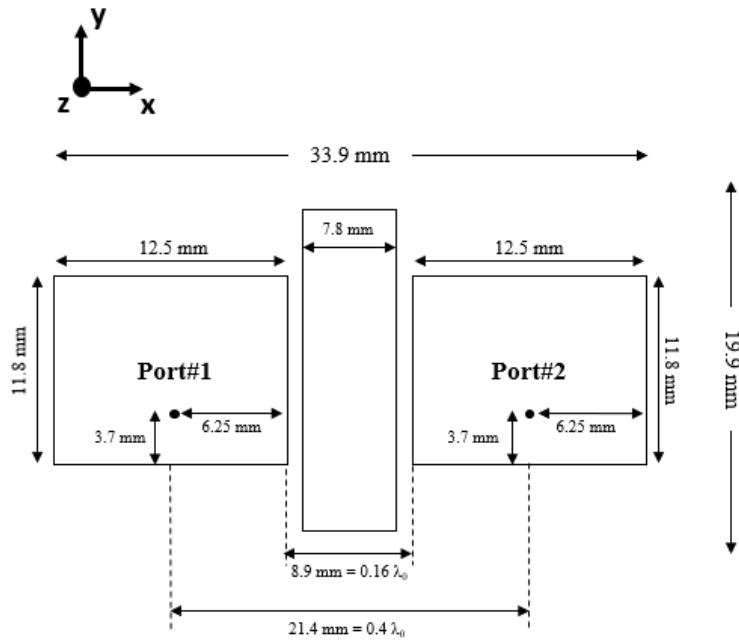


Figure 3.4-9: Dimensions of MEA after introducing the DS.

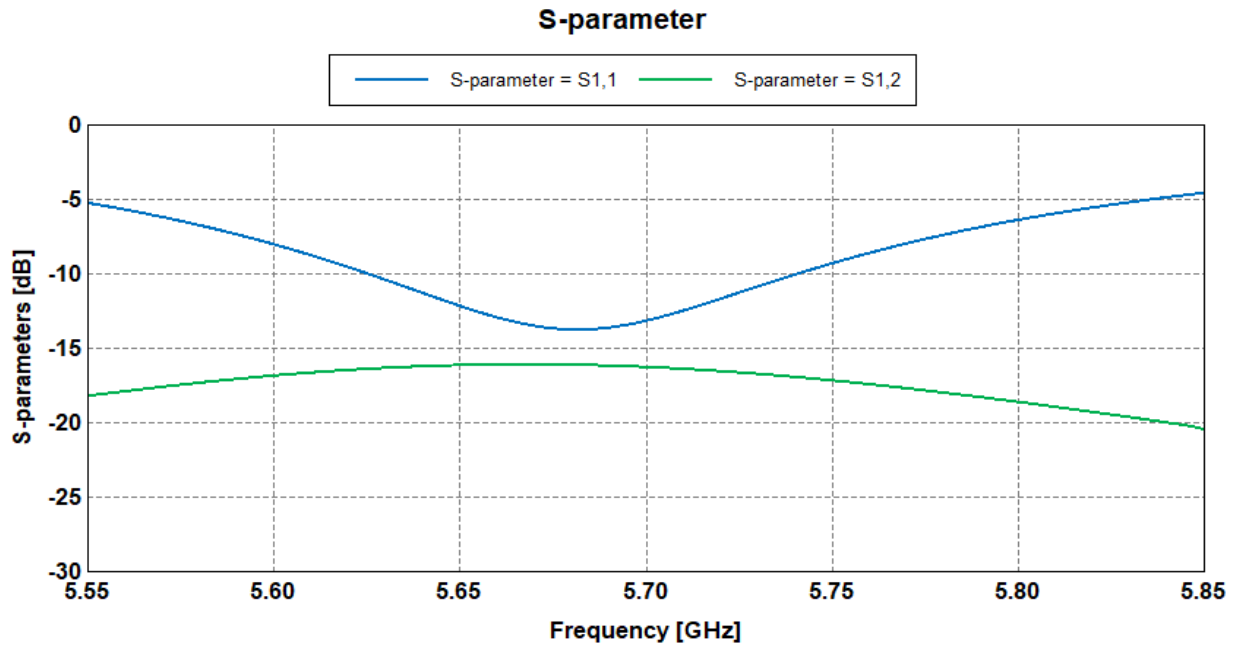
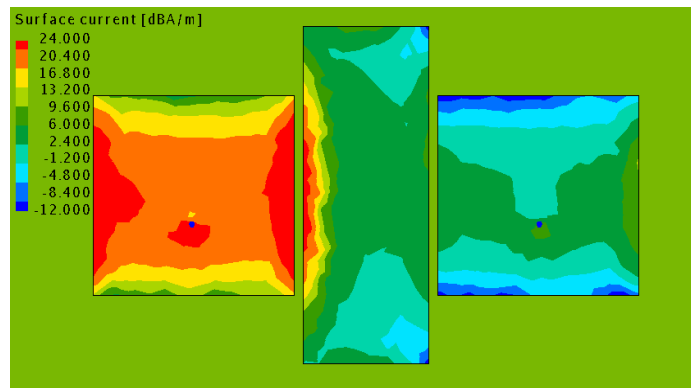


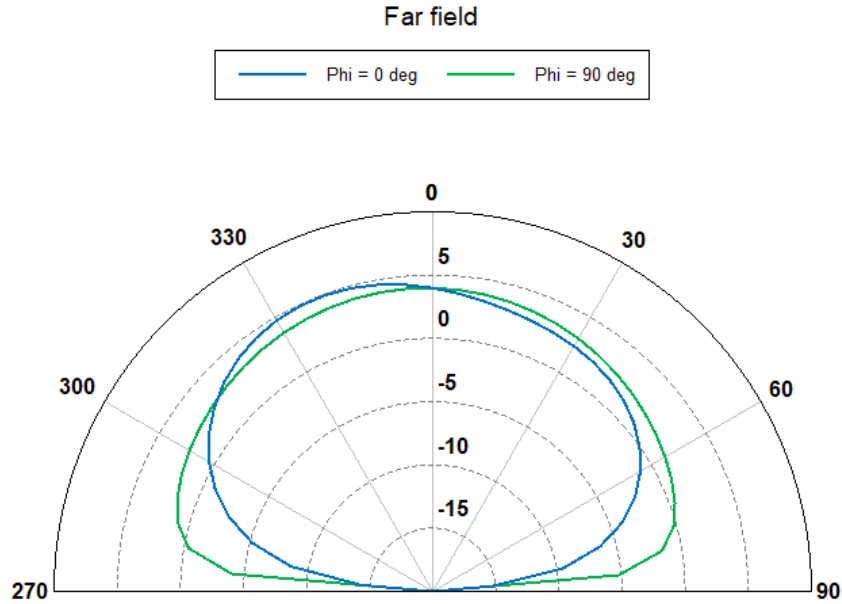
Figure 3.4-10: Computed S-parameter plots after the DS is introduced

The reduction in coupling is also reflected in Figure 3.4-11 which shows a reduction in the induced surface currents on the terminated antenna. In the absence of the DS, we saw that the maximum induced current on antenna 2 was about 8 dBA/m, this was reduced to about 4 dBA/m after the introduction of the DS. Additionally, we note that the portions of the DS closest to the excited antenna, are more strongly excited than the rest of the DS. This is due to the geometric proximity of the DS to the excited antenna. The magnitude of the induced surface current density at the location of port#2 is about 18 dB below that at location of port#1, after the introduction of the DS (compared to 12 dB without the DS).



**Figure 3.4-11: Surface current density of MEA with DS, with antenna 1 excited and antenna 2 terminated with a matched load, at 5.7 GHz.**

From Figure 3.4-12 we see that the introduction of the DS tilts the far-field pattern (in the  $xz$ -plane) more severely than, and in a direction opposite to that in, Figure 3.4-8, although the maximum value of total directivity remains 5 dBi. This pattern “skewing” is due to the presence of induced currents on both the terminated radiating element and the unshaped DS.



**Figure 3.4-12: Computed total directivity patterns at 5.7 GHz, when antenna 1 is excited, with the unshaped DS present.**

### 3.4.5 Application of Shape Synthesis

We aim to minimize the mutual coupling and return loss levels by shape synthesizing not only the decoupling surface but also the two antennas. The main goal of the shape synthesis is to reduce  $S_{12}$ , and adjust the magnitude and direction of the peak directivity in the far-field. The far-field pattern skewing will be corrected in the shaping process through introducing constraints. The constraints do not alter the objective function expression. Instead, individuals in a population (at each iteration of the GA) that do not satisfy the constraint are effectively assigned very-high objective function values. For example, in this case, the constraint was set such that the peak directivity must be at least 5 dBi and occur within +/- 10 degrees of broadside. The GA will thus work to minimize the objective function defined via the S-parameter masks while ensuring that the above directivity pattern constraint is not violated. If an individual in a population has “good” S-parameter properties but violates the constraint, it will be deemed unfeasible by the GA. This is not to say that such individuals will be disqualified from being passed on to the next generations; although such shapes (individuals) will not be considered optimum solutions in the overall process, some of their chromosomes might still be passed on to next generations. Implementing constraints in this manner offers flexibility in the shape synthesis process as it allows the designer to eliminate undesired characteristics and thereby have greater control over the final solution. However, as with

any optimization process, one should be careful to not impose constraints that are not physically possible. Imposing too many constraints can also shrink the solution space of feasible shapes. A reduced solution space could ultimately prevent the shaping process from arriving at a shape that meets the fitness criteria. The far-field directivity pattern of the two antennas and DS at the center frequency ( $f_0 = 5.7$  GHz) was simulated in FEKO and was passed to HyperStudy as a string of 36 values ( $a_1, a_2, \dots, a_{36}$ ). Each magnitude value corresponds to a value in theta ranging from  $-90^\circ$  to  $90^\circ$  in increments of  $5^\circ$ . If the maximum directivity value  $\max(a_1, a_2 \dots a_{36})$  is not  $a_{16}, a_{17}, a_{18}, a_{19},$  or  $a_{20}$  which corresponds to the values measured at  $-10^\circ, -5^\circ, 0^\circ, 5^\circ,$  and  $10^\circ$  then the peak directivity is not pointing in broadside and the shape (individual) is considered “not feasible”. Similarly, if  $\max(a_1, a_2 \dots a_{36}) < 5$  dBi the shape will be considered not feasible. The S-parameter masks for the present “H-plane coupling” case is shown in Figure 3.4-13. Although the associated objective function is written in general form in section 3.2, for clarity we repeat it here for this specific case:

We begin by defining the error contribution terms as shown below<sup>27</sup>:

$$e_{11}(f_q) = \begin{cases} |S_{11}(f_q)|^{dB} - |-5|^{dB} & \text{if } |S_{11}(f_q)|^{dB} - |-5|^{dB} < 0 \\ 0 & \text{Otherwise} \end{cases} \quad (3.4-1)$$

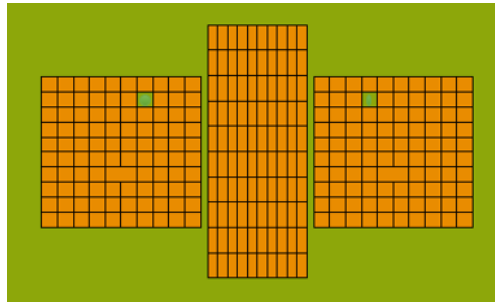
and

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<sup>27</sup> Note that we are merely “asking” the shaping process to maintain the  $S_{11}$  performance and not improve it. The “mask” value for  $S_{11}$  was selected to be flat with a relatively high maximum value in order to simplify this introductory example and focus all of the optimization efforts on reducing  $S_{12}$ .



Figure 3.4-14 shows the starting geometry (which will be shaped) divided into pixels<sup>28</sup>. To ensure left/right symmetry is retained for all shapes that are synthesized, we made sure that each single optimization bit (gene) only controls two pixels which are mirrored reflections of each other about the yz-plane. For example, in Figure 3.4-14, only a single bit is set to zero with all the remaining variables in the chromosome are set to 1. Yet corresponding cells are removed on either side. Note that two of the pixels (are on either side) are larger than others. The reason is that these are points on the geometry where the probe-feeds are attached to the patches. FEKO plants special attachment expansion functions there. It may not fall at the intersection of two pixels, and so in this case the two pixels in question were simply enlarged to encompass the feed points.

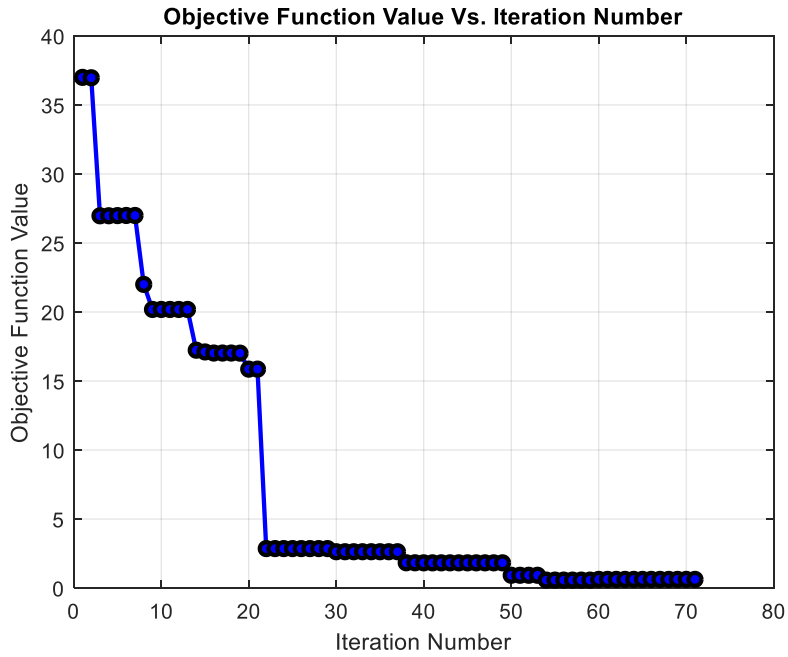


**Figure 3.4-14: Symmetry enforced on the pixelated shape of the MEA to be shaped.**

Figure 3.4-15 shows the objective function value of the fittest shape in each shaping. Significant reductions are achieved in the first few iterations. As the iterations progress the room for improvement decreases due to the governing physics and progress is slower.

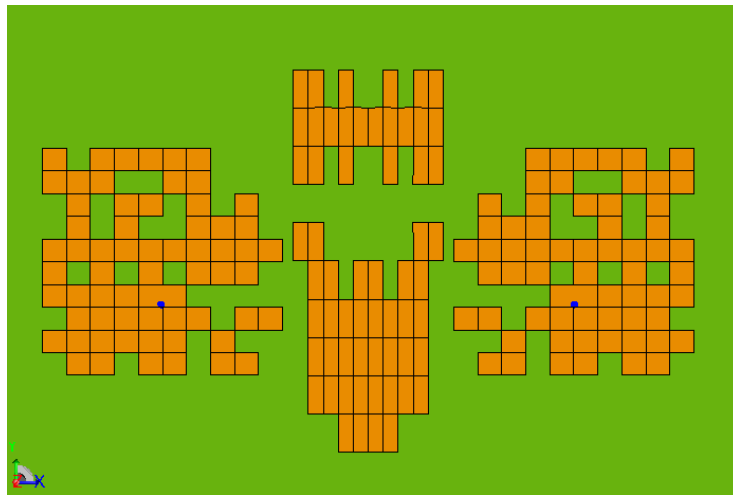
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<sup>28</sup> This is not identical to the mesh used or the computational electromagnetic analysis done by FEKO.



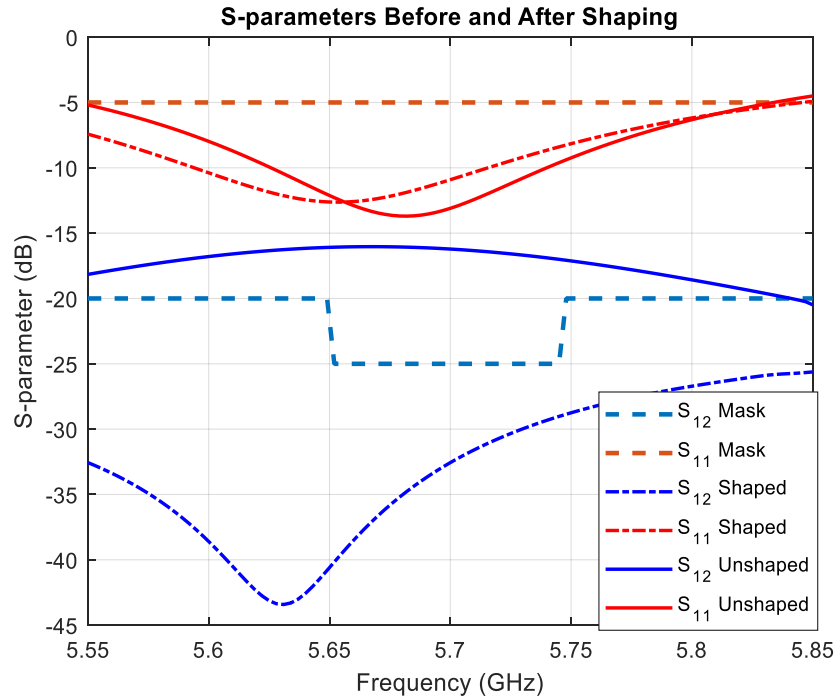
**Figure 3.4-15: Objective function values during the shape synthesis process**

The outcome of the optimization process is the structure shown in Figure 3.4-16. We see that there has been significant removal of cells from the starting geometry in Figure 3.4-14.



**Figure 3.4-16: Shape Synthesized MEA**

The computed S-parameter performance of the shaped MEA in Figure 3.4-16 is provided in Figure 3.4-17.



**Figure 3.4-17: Simulated S-parameters of shaped MEA**

In order to evaluate what the shaping has achieved, we will use the result in Figure 3.4-10 as a reference; recall that it applies to the geometry in Figure 3.4-9. We notice from Figure 3.4-17 that  $S_{11}$  remains below -5 dB (which is all we asked for via the objective function), but that the largest  $S_{12}$  value has decreased by roughly 10 dB compared to Figure 3.4-10. We can also examine the surface current diagram shown in Figure 3.4-18 to gain an insight on how the shaping process affected the current flow from the excited port 1 to the terminated port 2. We see that there is a significant reduction in the parasitic excitation of antenna 2. Compared to the reference model (no DS), where the maximum surface current density on antenna 2. Recall that the magnitude of the induced surface current density at the location of port#2 was only 12 dB below that at the driven port#1 when there was no DS. After adding and shaping the DS, Figure 3.4-18 shows that this difference is around 36 dB. The results presented in the surface current diagram are clearly aligned with reduction in mutual coupling ( $S_{12}$ ) which was noted in Figure 3.4-17. As such, we can conclude that the shaping synthesis process was successful in improving the decoupling between the antennas. Note that the cells which are only attached through corners in the CEM model have

no current flowing in them, and therefore, during the fabrication process these cells must be physically disconnected from each other.

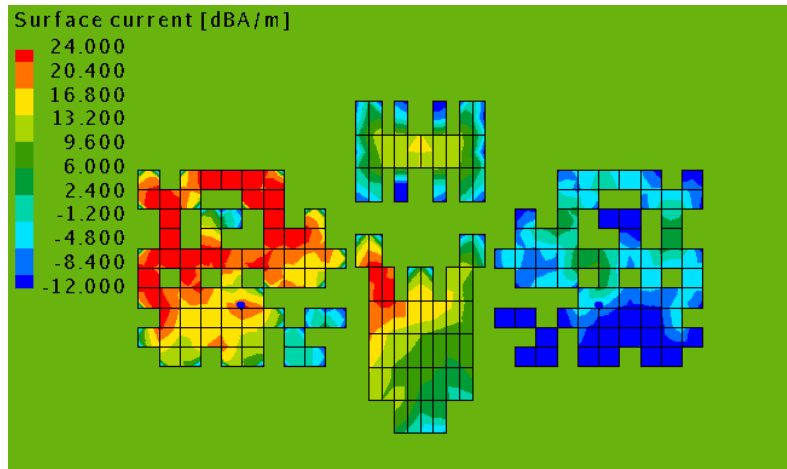


Figure 3.4-18: Surface current density for shaped H-plane model

Figure 3.4-19 shows the pattern of the shaped MEA. We can see that the peak directivity is slightly larger than 5dBi and lies within an angular sector centered on the broadside direction. The constraints set on the pattern have been satisfied; the  $xz$ -plane pattern skewing recorded in Figure 3.4-8 and Figure 3.4-12 has been largely removed.

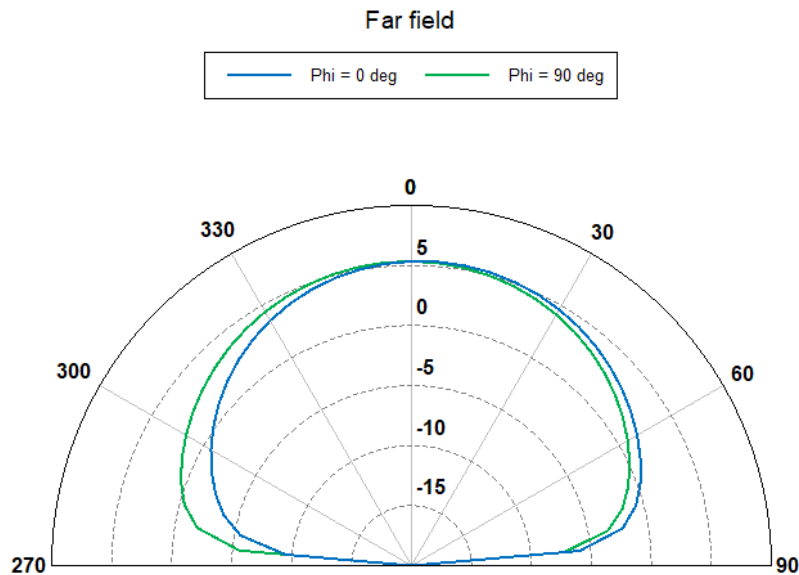


Figure 3.4-19: Computed total directivity pattern of the shaped MEA.

The simulated radiation efficiency plots in Figure 3.4-20 show that there is no significant change of this metric after the shaping. We note a 10% improvement in the lower half of the bandwidth after the shaping. In the upper half, the efficiency plots do not show any significant changes.

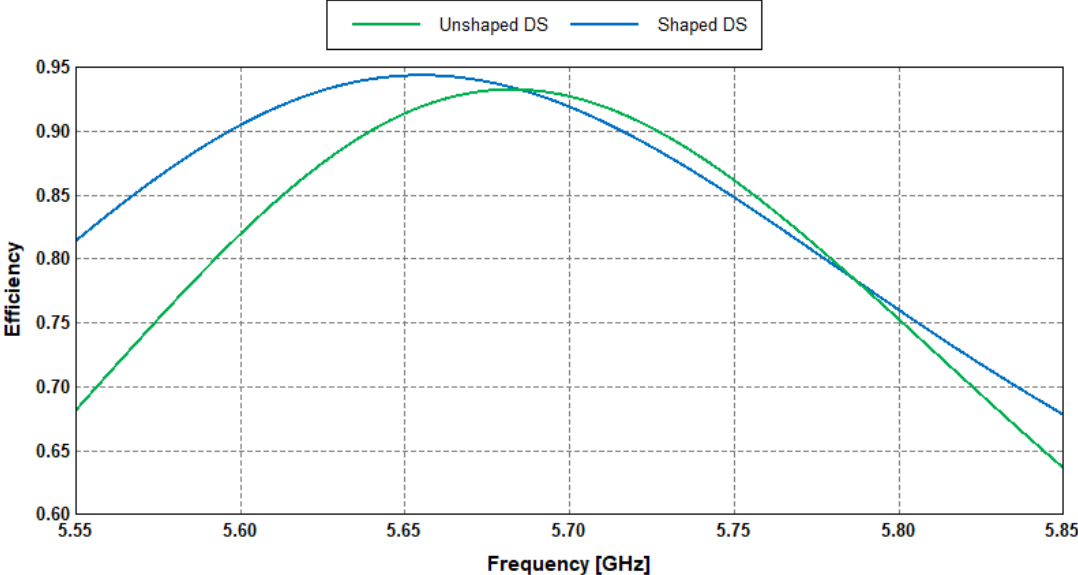


Figure 3.4-20: Simulated radiation efficiency before and after shaping

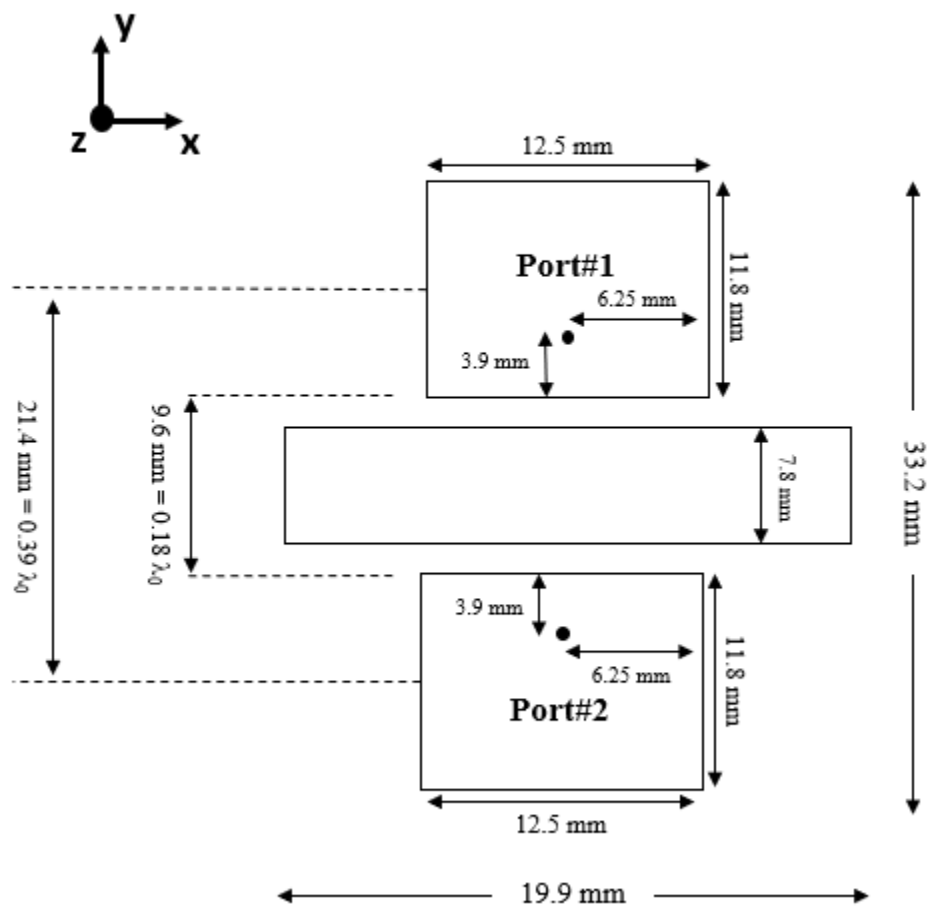
## 3.5 Linearly Polarized Microstrip Patches: E-Plane Coupling

### 3.5.1 Preamble

We next apply the shaping process to a case similar to that in Section 3.4, except that the two patches are aligned along their E-plane, as shown in Figure 3.5-1. The feedpoint locations of the antennas are on opposite sides (out-of-phase) in order to maintain symmetry. It should be noted that the dimensions of all the elements making up the MEA arrangement in Figure 3.5-1 were slightly altered from those used for the H-plane case. The substrate used is the same in every way to that specified in Section 3.4.1. In Section 3.4 the shaping process was required to maintain  $S_{11}$  at its unshaped value, and reduce  $S_{12}$  to values below a specified mask, while constraining the total directivity pattern to have its peak within a prescribed angular sector about the broadside direction. In the present section we add additional requirements : In addition to lowering  $S_{12}$  below a prescribed mask and constraining the total directivity pattern as before, we will require the shaping process to decrease  $S_{11}$  and the ECC below their unshaped values. This will be done by adjusting the mask level for  $S_{11}$  and adding an ECC-term to the composite objective function. As was done in Section 3.4, we will study this “E-plane” case in steps.

### 3.5.2 Single Microstrip Patch Reference

The computed performance of single probe-fed patch with dimensions shown in Figure 3.5-1 is shown in Figure 3.5-2 and Figure 3.5-3, for reference. The reflection coefficient  $S_{11}$  of the single patch shows acceptable properties within the frequency range 5.4 GHz to 5.7 GHz, and this will be the frequency range of interest. Figure 3.5-3 shows the total directivity patterns with expected maxima (5 dBi in this case, similar to Figure 3.4-4) in the broadside direction.



**Figure 3.5-1: Dimensions of the unshaped MEA arrangement (E-plane coupling) that will form the starting shape for the shape synthesis application.**

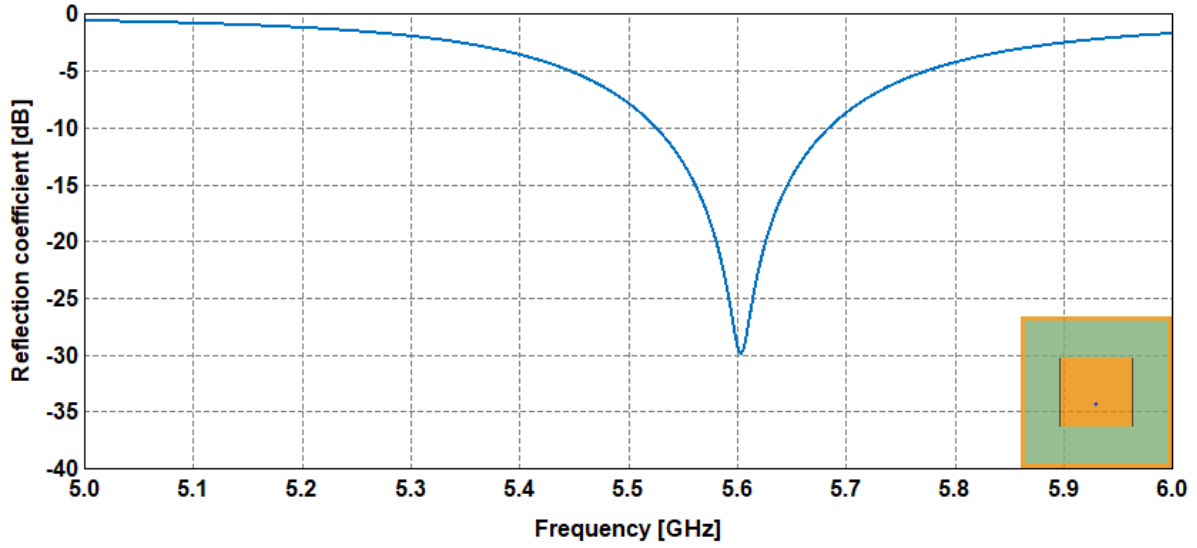


Figure 3.5-2: Computed reflection coefficient of the single isolated patch antenna.

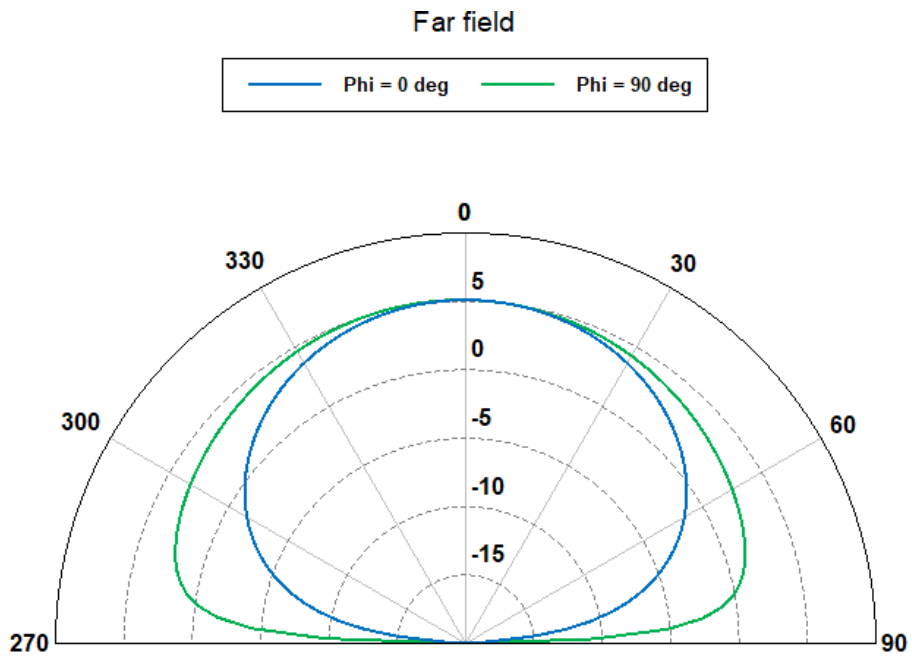


Figure 3.5-3: Computed total directivity radiation pattern of the single patch antenna in the two principal planes. The frequency is 5.55 GHz.

### 3.5.3 MEA Consisting of Two Patch Antennas: E-Plane Alignment

Before proceeding to shaping, we examine the properties of the two patches shown in Figure 3.5-1 but with the unshaped DS absent. The patches are separated by 9.6. The Antenna's combined computed characteristics will be used as a reference and compared to the results that will be produced through shape optimization. Figure 3.5-4 reveals that there is not a significant change in  $S_{11}$  compared to what it was (Figure 3.5-2) when the 2<sup>nd</sup> patch was absent. The  $S_{12}$  rises as high as -12 dB; in MIMO applications [MEI 18] the coupling between any two elements should preferably be below -20 dB.

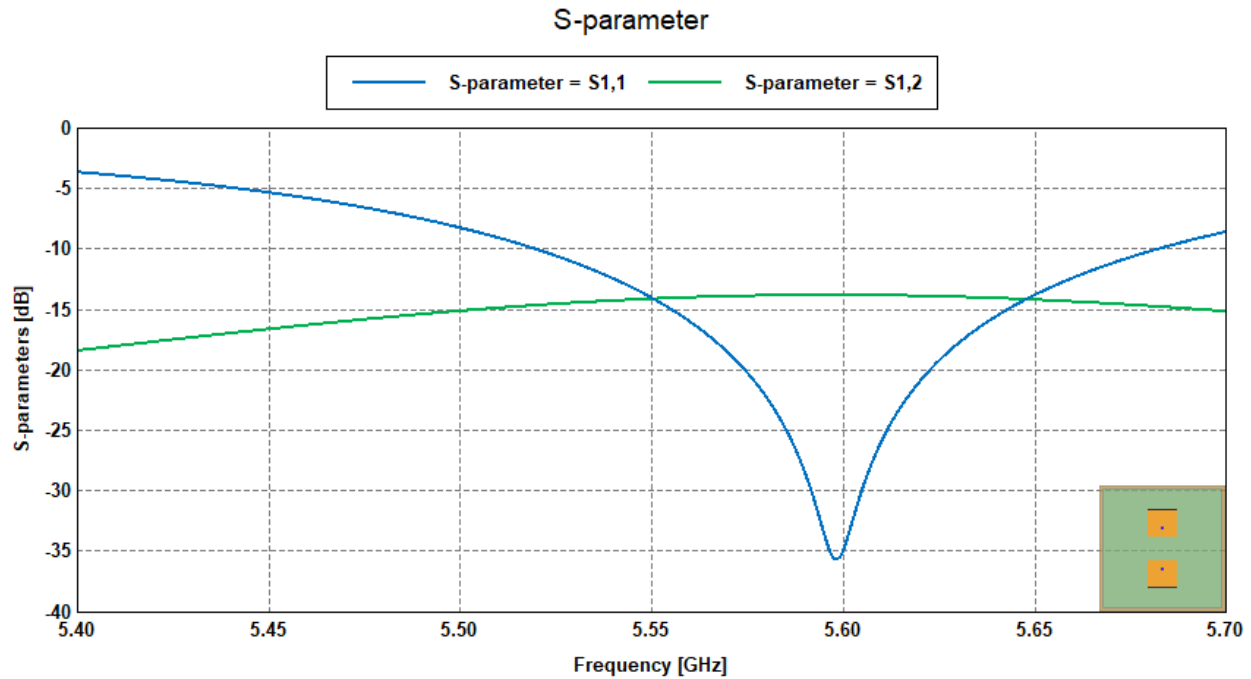
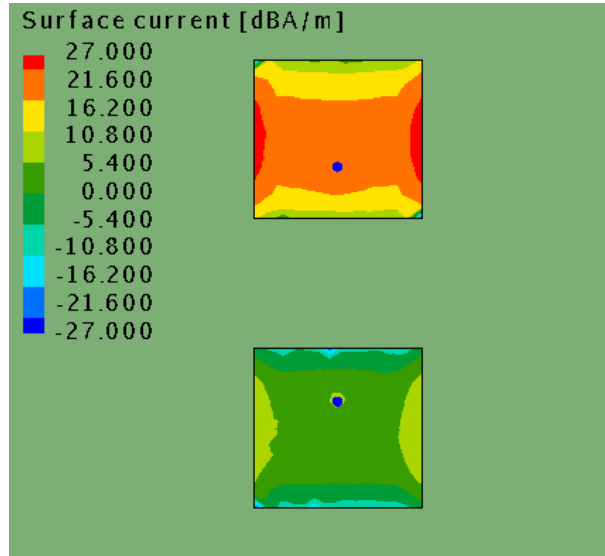


Figure 3.5-4: Computed S-parameters of the MEA (two patches without a DS).

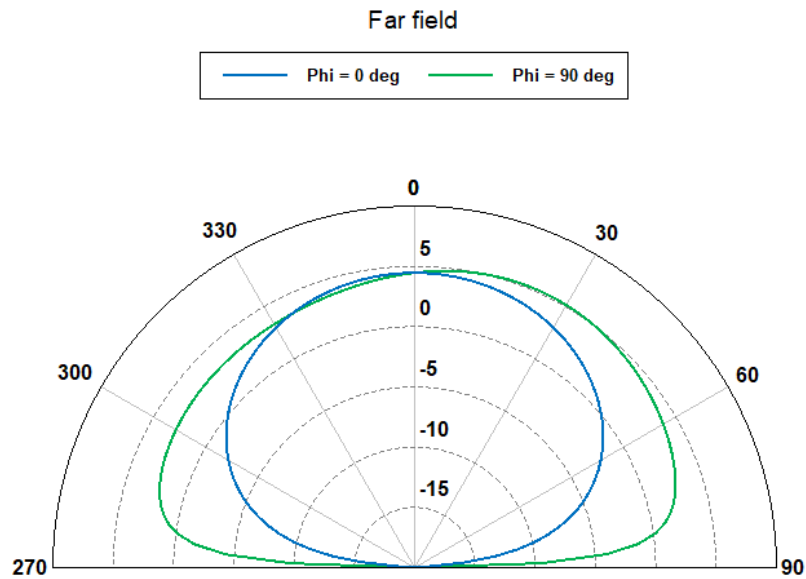
The surface current density when antenna 1 is excited and antenna 2 is terminated with a  $50\Omega$  load is shown in Figure 3.5-5. The magnitude of the induced surface current density at the location of port#2 is about 11dB below<sup>29</sup> that at the driven port#1. The radiation pattern of antenna#1 under

<sup>29</sup> We aim to reduce this through the shape synthesis process with a DS present.

the same conditions is now, as expected, skewed in the yz-plane ( $\phi = 90^\circ$  plane), as is clear from Figure 3.5-6.

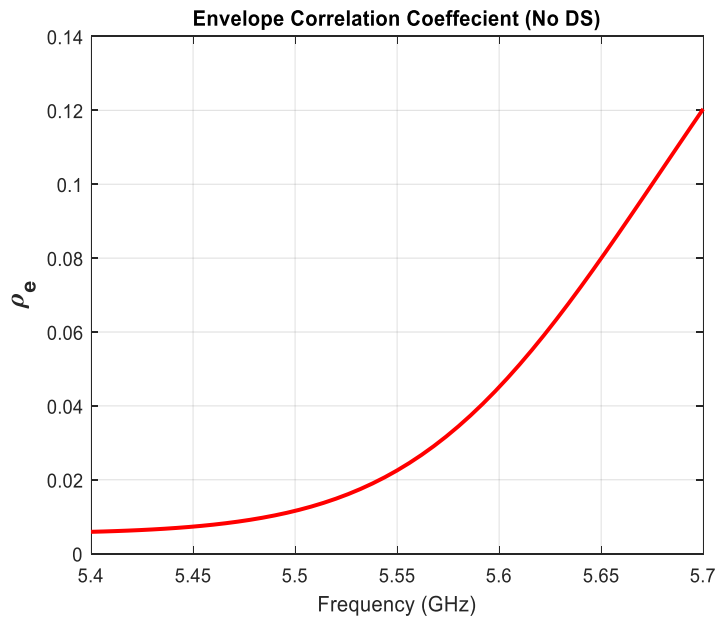


**Figure 3.5-5: Surface current density on the MEA (two patches and no DS) when port#1 is driven and port#2 terminated. The frequency is 5.55 GHz.**



**Figure 3.5-6: Computed principal plane total directivity patterns of the MEA (two patches and no DS) when port#1 is driven and port#2 terminated. The frequency is 5.55 GHz.**

If the antennas are intended for MIMO operation, the ECC figure of merit would be of interest. Figure 3.5-7 shows the computed<sup>30</sup> ECC plotted over the frequency band of interest. It is interesting to observe, by comparing Figure 3.5-4 and Figure 3.5-7 that low mutual coupling ( $S_{12}$ ) values alone are not sufficient to achieve low ECC values. In Figure 3.5-4 the highest  $S_{12}$  values occur around 5.6GHz, yet the ECC value is not peaked there, but has its highest value at 5.7 GHz. This is because the  $S_{12}$  only measures the coupling between the antenna ports and not the radiated far-field coupling [SHAR 13]. This provides motivation to include explicit ECC terms in the objective function, as we will do here<sup>31</sup>.



**Figure 3.5-7: ECC for the MEA (two patches and no DS).**

Although it will not be included in the objective function, we out of interest will monitor the total active reflection coefficient<sup>32</sup> (TARC) figure of merit in order to simply record any enhancements in the TARC brought about by the shape synthesis process. Since the TARC is

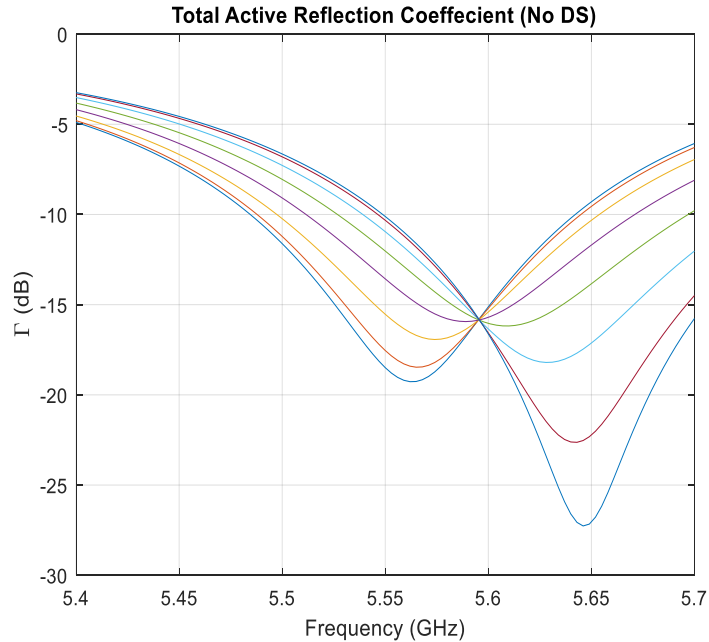
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<sup>30</sup> This is done using the radiation patterns, as explained in Section 2.3.7.

<sup>31</sup> The EEC is actually already sufficiently low, as the maximum value of 0.12 is still well below the 0.5 requirement mentioned for mobile terminals. Nevertheless, we wish to explore what the shape optimization process is able to achieve as regards further reducing the ECC.

<sup>32</sup> Defined in Section 2.3.6.

calculated strictly from the S-parameters, enhancing the port isolation between the antennas should as a consequence reduce the TARC values. For the current MEA configuration (that is, two patches and no DS) the TARC plotted in Figure 3.5-8 shows a worst-case value of -3 dB, which is considered high for a MIMO implementation [BROW 06]. We will see in Section 3.5.5 that adding a DS and shaping the entire resulting MEA is able to decrease this.



**Figure 3.5-8: Computed TARC for the MEA (two patches and no DS).**

### 3.5.4 MEA Consisting of Two Patch Antennas and an (Unshaped) Decoupling Surface

The next step in this study, once again for later reference, is to examine the behavior of an MEA consisting of the two patches and the unshaped DS, precisely as shown in Figure 3.5-1. Figure 3.5-9 shows that (compared to Figure 3.5-4) the location of the minimum  $S_{11}$  value (the “resonance”) and the maximum  $S_{12}$  value has shifted from 5.6 GHz to 5.53 GHz. The presence of the unshaped DS has detuned the antennas. From Figure 3.5-10 we see that the magnitude of the induced surface current density at the location of port#2 is about still about 11dB below that at the driven port#1, as it was in Figure 3.5-5. The radiation pattern of antenna#1 under the same conditions is now that in Figure 3.5-11. Surprisingly, introduction of the unshaped DS corrected

the skewing of the radiation pattern in the  $\phi = 90^\circ$  plane that was observed (without the unshaped DS present) in Figure 3.5-6, across the whole frequency band. The maximum total directivity has, however, decreased to 4.3 dBi, which is less than the 5 dBi we wish to achieve in the broadside direction. The ECC plot in Figure 3.5-12 is similar to the DS-less case in Figure 3.5-7; the maximum ECC value has fallen to 0.10 from 0.12 with the DS present (due to a small amount of additional isolation), but we do not attach too much significance to this.

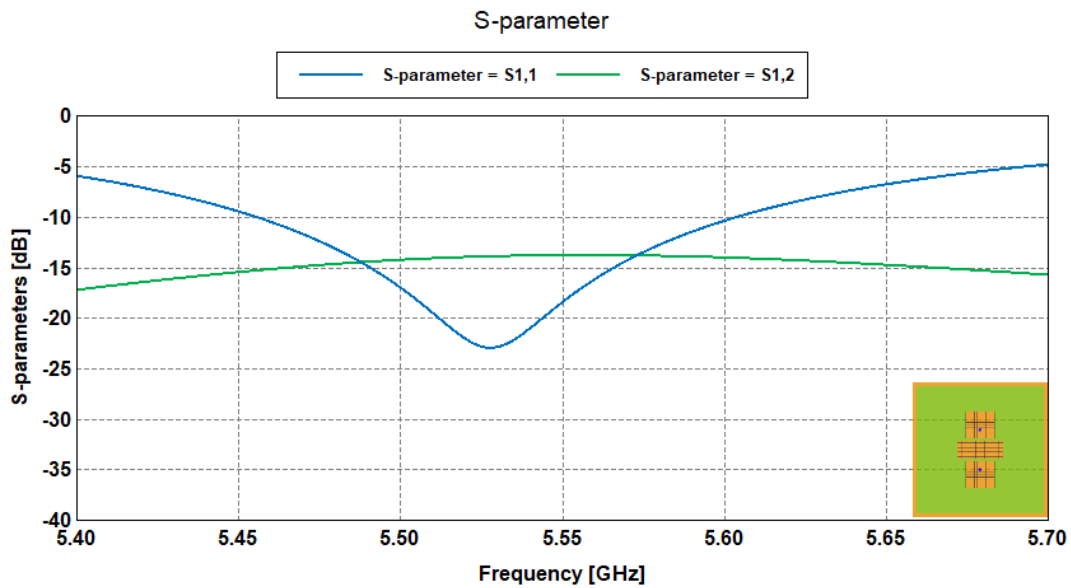


Figure 3.5-9: Computed S-parameters of the MEA consisting of two patches and the unshaped DS.

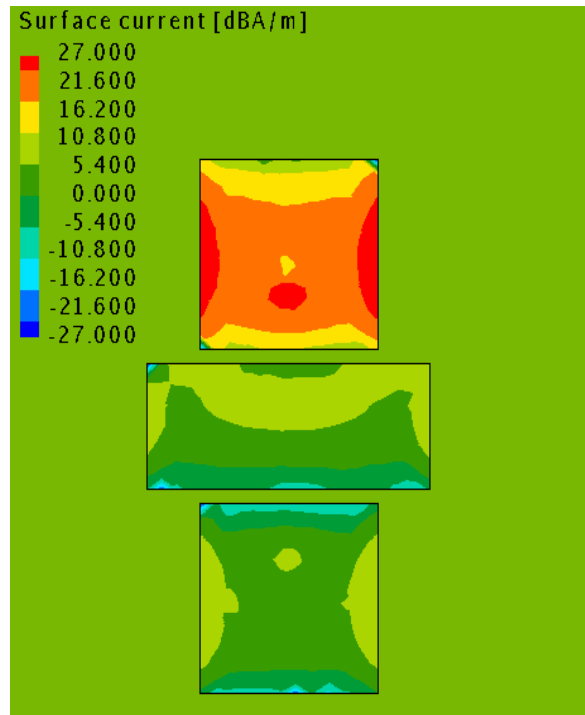


Figure 3.5-10: Surface current density on the MEA consisting of two patches and the unshaped DS when port#1 is driven and port#2 terminated. The frequency is 5.55 GHz.

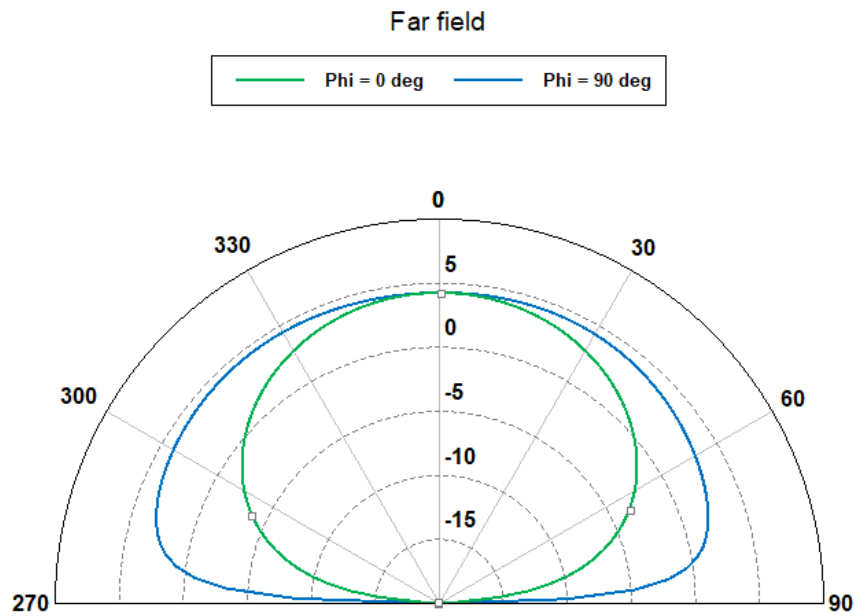
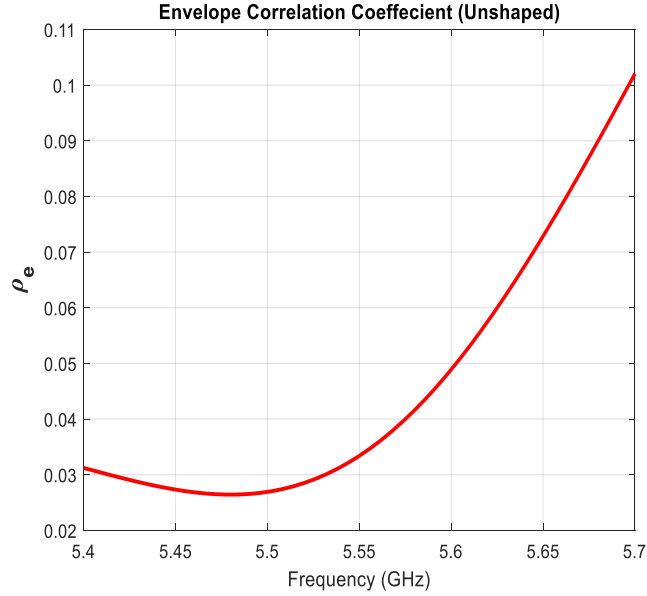


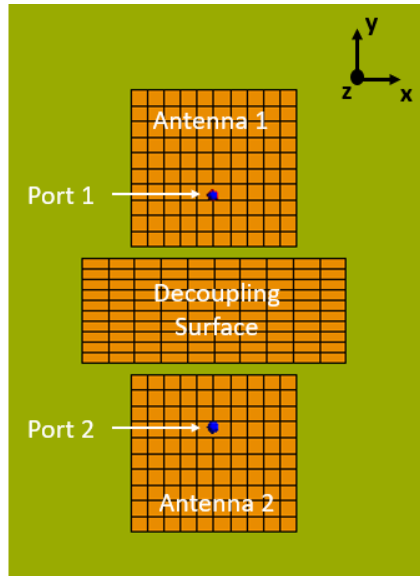
Figure 3.5-11: Computed total directivity principal plane patterns of the MEA consisting of two patches and the unshaped DS. The frequency is 5.55 GHz



**Figure 3.5-12: Computed ECC for the MEA consisting of two patches and the unshaped DS.**

### 3.5.5 MEA Consisting of Two Patch Antennas and a Shaped Decoupling Surface

We now apply the shaping procedure to the starting shape shown in Figure 3.5-1. The pixelated form is that in Figure 3.5-13. Top/down symmetry will be enforced during shaping.



**Figure 3.5-13: Starting Structure of E-Plane coupled MEA**

The objective function was formulated in a similar manner to that used for the ‘‘H-plane coupling’’ case in Section 3.4. The S-parameters masks were simply upper limit values of -10 dB and -20 dB for  $S_{11}$  and  $S_{12}$ , respectively. As promised in Section 3.5.1, a new term is added to the objective function to allow the GA to include the ECC between the pair of antennas when taking decisions on the shapes that arise in each iteration. The  $e_{ECC}$  error term can be formulated in the same way as  $e_{11}$  and  $e_{12}$ , and will be assigned an upper mask value<sup>33</sup> of 0.05 or -13 dB over the frequency band of interest. Thus, the objective function component error terms are

$$e_{11}(f_q) = \begin{cases} |S_{11}(f_q)|^{dB} - |-10|^{dB} & \text{if } |S_{11}(f_q)|^{dB} - |-10|^{dB} < 0 \\ 0 & \text{Otherwise} \end{cases} \quad (3.5-1)$$

$$e_{12}(f_q) = \begin{cases} |S_{12}(f_q)|^{dB} - |-20|^{dB} & \text{if } |S_{12}(f_q)|^{dB} - |-20|^{dB} < 0 \\ 0 & \text{Otherwise} \end{cases} \quad (3.5-2)$$

and

$$e_{ECC}(f_q) = \begin{cases} |\rho_e(f_q)|^{dB} - |-13|^{dB} & \text{if } |\rho_e(f_q)|^{dB} - |-13|^{dB} < 0 \\ 0 & \text{Otherwise} \end{cases} \quad (3.5-3)$$

with

$$|\rho_e|^{dB} = 10 \log \rho_e \quad (3.5-4)$$

---

<sup>33</sup> It should be noted that in the literature ECC is not often quoted in decibels; however, to formulate the objective function we had to make the decision to convert the values to dB in order to ensure consistency since  $S_{11}$  and  $S_{12}$  are expressed in decibels. This is because the ECC values expressed in decimal are less than 1, and when squared in the error term, ECC will have no or negligible impact on the objective function.

The complete objective function is

$$F_{Obj}(f) = 0.55 \sum_{q=1}^{13} |e_{11}(f_q)|^2 + 0.35 \sum_{q=1}^{13} |e_{12}(f_q)|^2 + 0.1 \sum_{q=1}^{13} |e_{ECC}(f_q)|^2 \quad (3.5-5)$$

In addition to minimizing  $S_{11}$ ,  $S_{12}$ , and ECC, we constrain the total directivity pattern when port#1 is driven and port#2 is terminated to be at least 5 dBi with the peak<sup>34</sup> in a 10° cone centered on the broadside direction at the centre frequency  $f_o = 5.55$  GHz, as was also done (and explained) in Section 3.4.

During pilot runs for this shape optimization problem we found that minimizing  $S_{11}$  was significantly harder to achieve by the GA compared to minimizing  $S_{12}$ . We initially formulated the objective function with equal weights on the  $e_{11}$  and  $e_{12}$  parts, but found that such weights always caused the process to terminate without achieving the  $S_{12}$  and  $S_{11}$  goals. A larger weighting multiplier was therefore assigned to  $e_{11}$ , as seen in expression (3.5-5). A low weight was assigned to the  $e_{ECC}$  terms because the ECC of the starting shape was already close to the target value of -13 dB (0.05).

The progress of the shaping process is represented by the plot of the objective function value versus the iteration number in Figure 3.5-14. Because of the similarity of the problem to that in Section 3.4, it is not surprising that the behaviour is much like that in Figure 3.4-15. The outcome is the shape synthesized MEA in Figure 3.4-15.

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<sup>34</sup> This is an important issue. In a session of the *18th International Symposium on Antenna Technology and Applied Electromagnetics (ANTEM)*, Waterloo, Canada, August 2018, comments were made to the effect that one cannot reap the MIMO benefits of low correlation between MEA elements if the patterns of these elements are not pointed in their intended directions. By using the shaping process we could demand slightly different pattern peak directions for the elements in an MEA if the system concepts require it. This would allow us to remove the geometrical symmetry that has been enforced during the shaping process here and in Section 3.4, but  $S_{22}$  would need to be included in the objective function as well since it will no longer be identical to  $S_{11}$ .

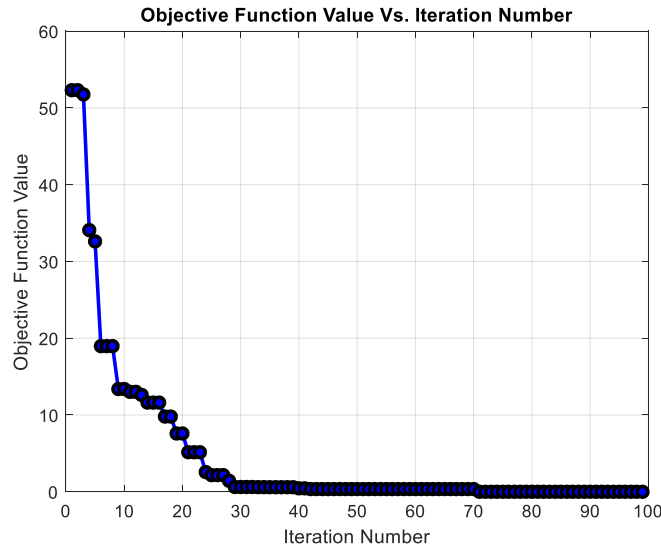


Figure 3.5-14: Objective function value during the shape optimization of the MEA consisting of two patches and DS

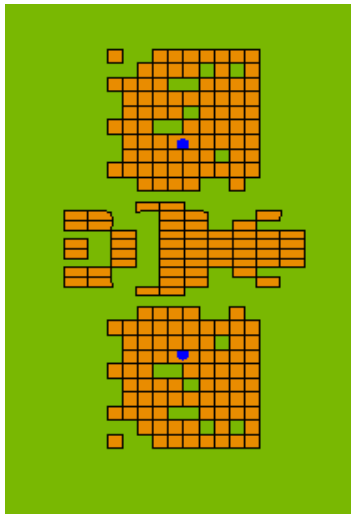


Figure 3.5-15: The MEA consisting of two patches and a DS after the shaping is completed

We see in Figure 3.4-16 that the S-parameter performance meets the mask limits specified. As “requested”, the shaping procedure has lowered the maximum value of  $S_{11}$  from (the reference case value of) -6dB to -10dB, and decreased the maximum value of  $S_{12}$  by 6dB so that  $S_{12}$  is below -20dB over the whole band. The latter is reflected in the magnitude of the induced surface current density at the location of port#2 (in Figure 3.4-17) which is now roughly than 21dB below

that at the driven port#1, some 10dB decrease from what it was in the reference case in Figure 3.5-5.

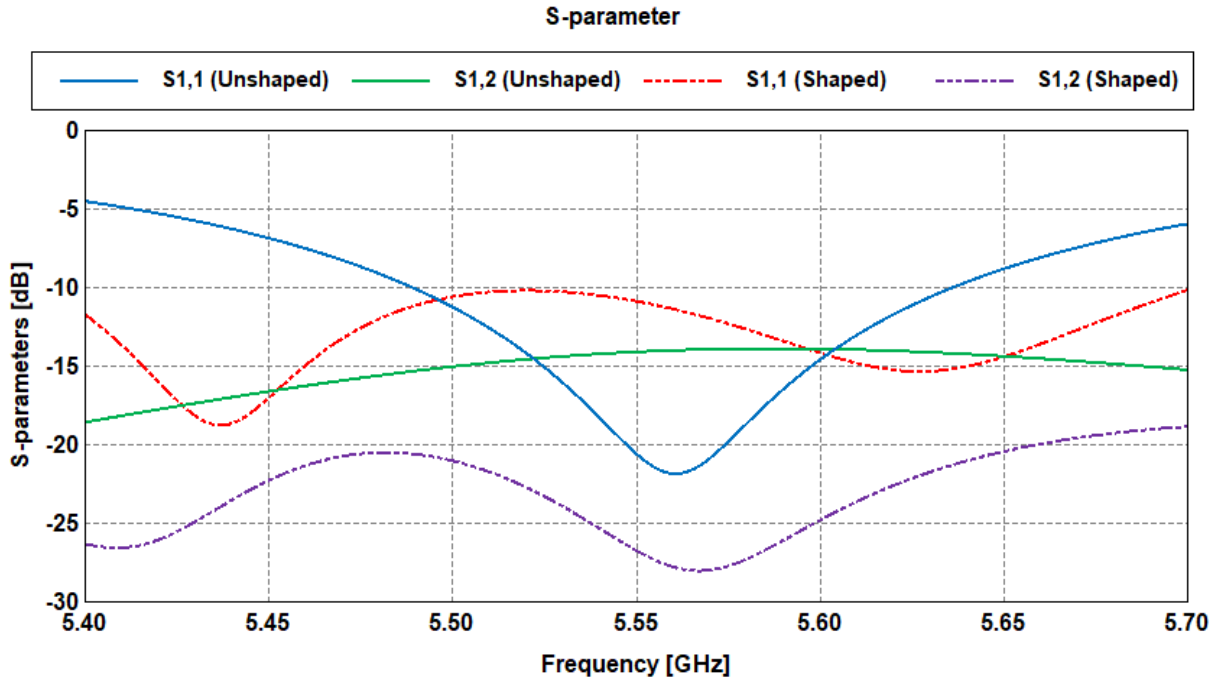
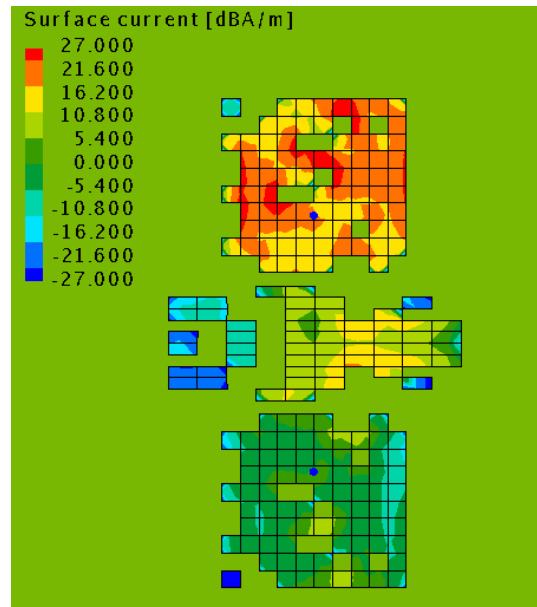


Figure 3.5-16: Computed S-parameter plots for the shaped MEA consisting of two patches and a DS.



**Figure 3.5-17: Surface current density on the shaped MEA consisting of two patches and a DS, when port#1 is driven and port#2 terminated. The frequency is 5.55 GHz.**

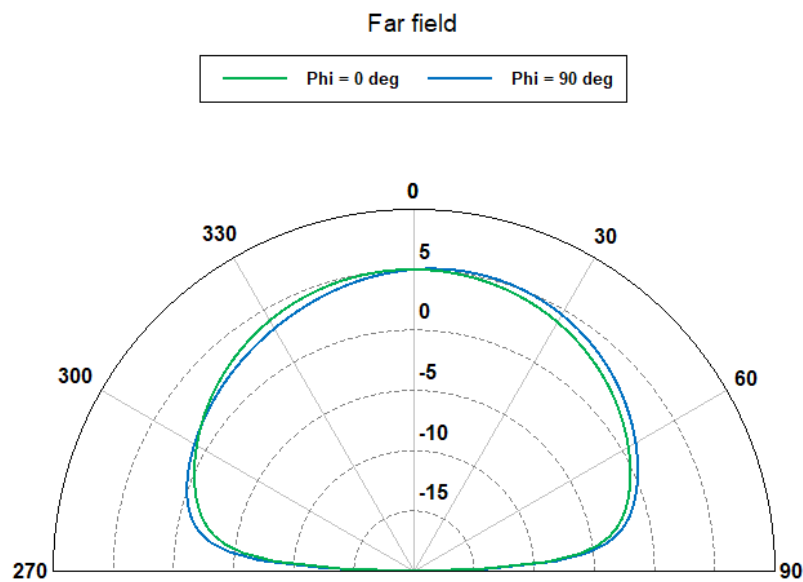
The individual MEA element pattern<sup>35</sup> is shown in Figure 3.5-18. The peak total directivity is indeed 5 dBi, and in the  $\phi = 90^\circ$  this peak occurs about  $10^\circ$  off broadside instead of  $30^\circ$  as for the skewed MEA with the two patches and no DS, as depicted in Figure 3.5-6. It is possible to “draw” the peak closer to the broadside direction, but only by sacrificing some isolation. The shape process can be used to establish such trade-offs, with the wider “design space” offered by performing shaping instead of accepting the geometry restrictions of pre-selected geometries.

Although the ECC behaviour (for the shaped MEA) in Figure 3.5-19 is much improved compared to the reference cases in Figure 3.5-7 and Figure 3.5-12, the ECC mask value of 0.05 has not been satisfied<sup>36</sup> over the complete frequency band, even though the shaping process terminated. This could be prevented by using a larger weight in front of the ECC term in expression (3.5-5), or lowering expectations on the other performance quantities, and so on. It became clear from many other shaping runs that one is always trying to “play the physics”, given the starting shape (in this case Figure 3.5-13). Another possibility would be to alter the geometry (eg. it need not be

<sup>35</sup> Recall that the geometrical symmetry enforced during shaping

<sup>36</sup> The ECC rises to about 0.067 at the upper end of the frequency band.

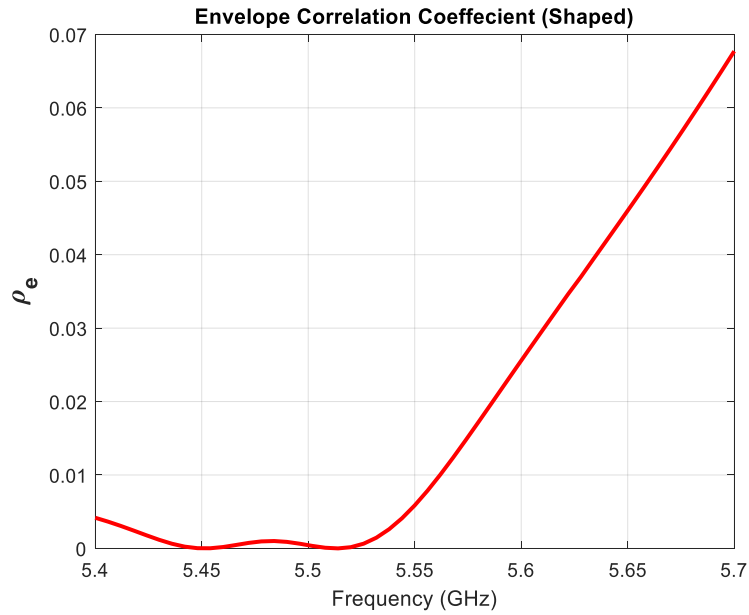
rectangular) of the unshaped DS to give the synthesis algorithm more conductor “to play with”. The possibilities are endless, and could be driven by specific size constraints in some particular application, for instance. The TARC curves in Figure 3.5-20 show a worst-case value of about -6 dB. Although TARC was not included in the objective function, we still see an improvement of about 3 dB compared to the worst case in Figure 3.5-8. If TARC were deemed to be a more important figure of merit than  $S_{11}$  it could in principle be included in the objective function<sup>37</sup> instead of  $S_{11}$ .



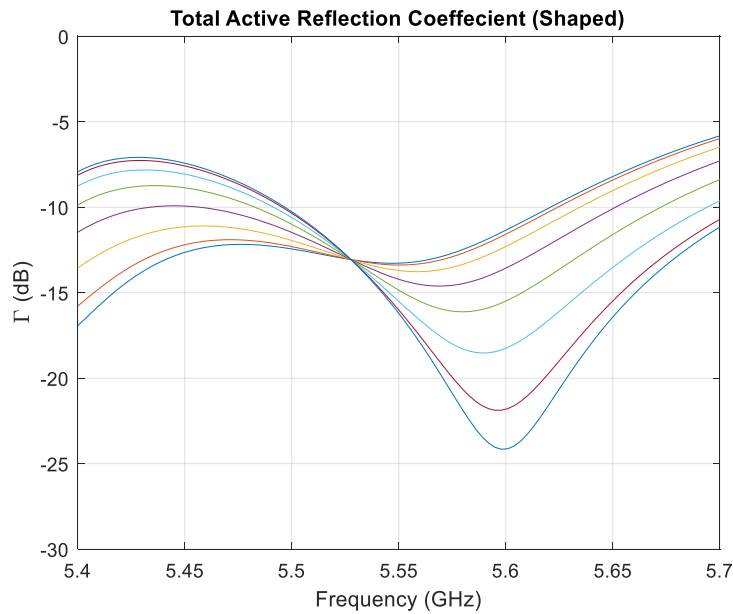
**Figure 3.5-18: Computed total directivity principal plane patterns on the shaped MEA consisting of two patches and a DS when port#1 is driven and port#2 terminated. The frequency is 5.55 GHz.**

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<sup>37</sup> The ability to use the S-parameter information to determine the TARC, as described in Section 2.3.6, and then incorporate this in the objective function, is at present not possible in the Hyperstudy suite [HYPE 20].



**Figure 3.5-19: Computed ECC for the shaped MEA consisting of two patches and a shaped DS.**



**Figure 3.5-20: TARC curves for the shaped MEA consisting of two patches and a DS.**

The radiation efficiency did improve slightly as seen in Figure 3.5-21, the improvement is less than 25% and is mostly present near the upper and lower edges of the bandwidth. The improvement

in efficiency was not intended in the initial design, however as with any other antenna metric the radiation efficiency can be included in the objective function if need be.

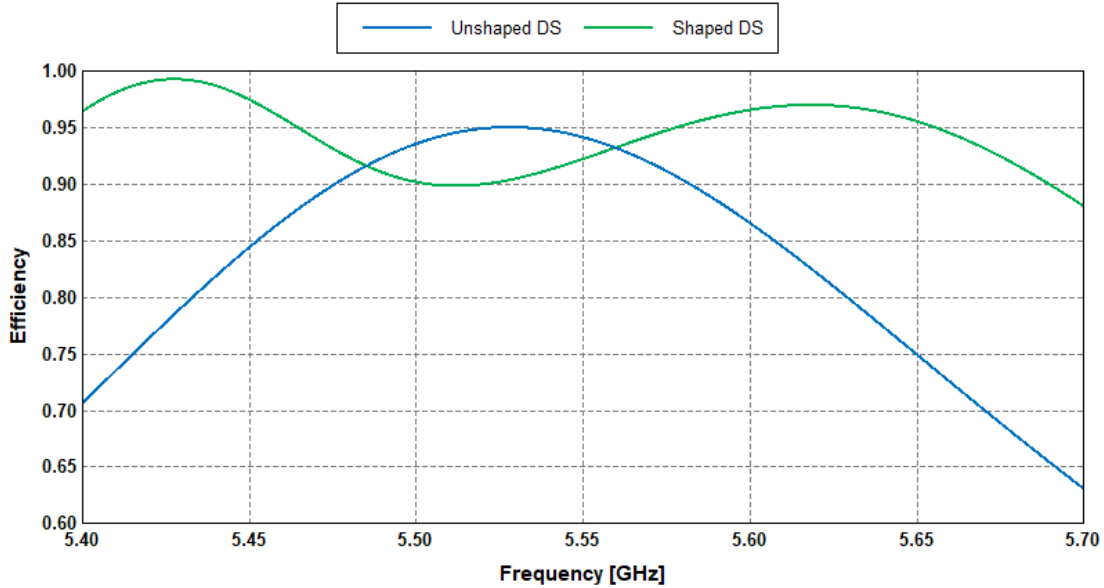
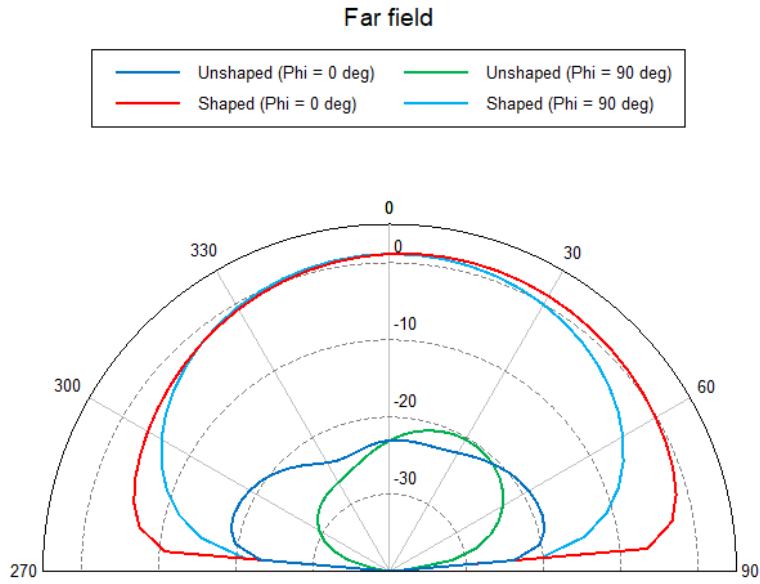


Figure 3.5-21: Simulated radiation efficiency plots before and after shaping two patches and DS

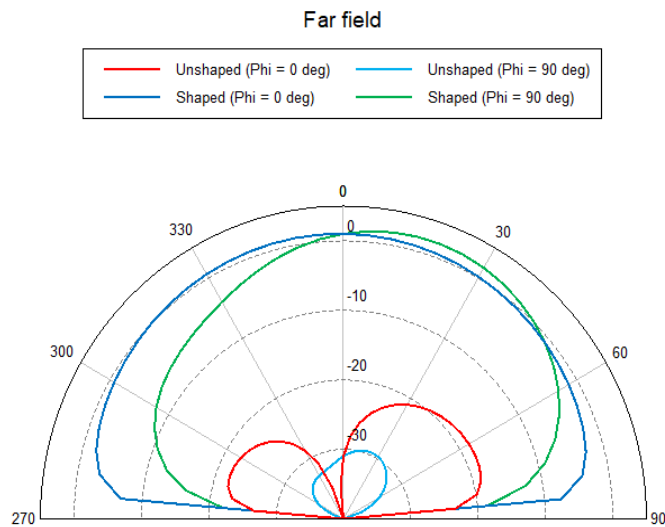
### 3.6 Patch Antenna Polarization

As discussed in sections 3.4 and 3.5, we have opted to optimize the farfield radiation patterns based on the total directivity pattern rather than the co-polarization directivity component. The reason for this is that MEA antennas are typically used for MIMO systems that rely on multipath to achieve the increased channel capacity. As a result, the polarization of the signals arriving at the antennas after traversing the multipath environment is not known and will likely be randomly varying. Therefore, optimizing for a high co-polarization (low cross-polarization) component would not have provided us with the complete radiation performance of the antenna for MIMO application. Nevertheless, the cross-polarization components (Ludwig-III) were examined for both H-plane and E-Plane cases. The cross-polarization components for the H-plane and E-plane cases are shown in Figure 3.6-1 and Figure 3.6-2, respectively. In both cases we can see that the cross-pol components have increased from less than -20dBi before the shaping, to about 2.5 dBi after the shaping. These results show that the shaping process has changed the polarization of the antennas from having vertical polarization to having 45° polarization. Since both E-plane and H-

plane cases are symmetrical, the polarizations of antenna #1 and antenna #2 will point in opposite directions after the shaping is completed. The results in this section demonstrate that global search algorithm for the shaping process forced the antennas to have orthogonal polarizations to meet the requirements set in the objective function.



**Figure 3.6-1: Computed cross-polarization directivity components in the farfield for two antennas and DS before and after shaping (H-plane case)**



**Figure 3.6-2: Computed cross-polarization directivity components in the farfield for two antennas and DS before and after shaping (E-plane case)**

### 3.7 Concluding Remarks

In this chapter, we described the shape synthesis process. This included the decision flowchart of the genetic algorithm that controls the shape synthesis process. Importantly, the construction of the objective functions, which is key to the success of the shape optimisation, was discussed at length in Sections 3.2, 3.4.5 and 3.5.5. Two examples of the application of shape synthesis in the design of decoupling surfaces were presented. Both examples involved probe-fed microstrip patch antennas separated by a decoupling surface located on the same plane as the patches. To demonstrate the effectiveness of the shape synthesis algorithm, the examples were purposefully selected so that the patches were closely spaced along their common H-plane and common E-plane, respectively.

In the first example (“H-plane” case in Section 3.4) we initially concentrated on reducing mutual coupling ( $S_{12} = S_{21}$ ) while preserving the input reflection coefficients ( $S_{11} = S_{22}$ ) through incorporating these in the objective function, and shaping both the decoupling surface and the patches. It was observed (even without shaping or without the decoupling surface) that there is a tilt in the far-field directivity pattern in the  $\phi = 0^\circ$  cut. To rectify this tilt, we introduced an additional term in the optimizer’s software tool to “pin” the peak directivity on broadside. The final result produced by the shape synthesis process showed that we were able to reduce the mutual coupling from -16 dB to less than -25 dB and correct the tilt in the far-field directivity pattern. In the second example (“E-plane” case in Section 3.5) we were successful in showing the flexibility of the shaping process by including yet additional goals in the objective function. This enabled us to not only reduce  $S_{ii}$  and  $S_{ij}$ , but also directly the ECC, all the while preserving the value and direction of the maximum directivity of the individual patches that would have otherwise (without shaping) been tilted in the  $\phi = 90^\circ$  cut. Through the shape synthesis process, we were able to reduce the  $S_{ii}$ ,  $S_{ij}$ , and ECC by 6 dB, 6 dB, and 50% respectively. As far as the author is aware, this is the first time that shape synthesis has been used for the automated shaping of an MEA system with a decoupling surface.

# Chapter 4 The Shape Synthesis of Non-Coplanar Decoupling Surfaces

## 4.1 Preliminary Remarks

In Chapter 3, we explored how shape synthesis can be used as a method to decouple closely packed antennas, with the antennas and the decoupling structures lying on the same plane ('coplanar'). We demonstrated the flexibility of this method by incorporating several performance goals in the objective function. In the present chapter, our aim is to further demonstrate, also for the first time, the use of shape synthesis for an antenna arrangement where the decoupling surface (DS) is not located in the same plane as the antenna elements proper ('non-coplanar'). The MEA system is extended to one consisting of four elements (albeit still probe-fed patches) that lie in a single plane.

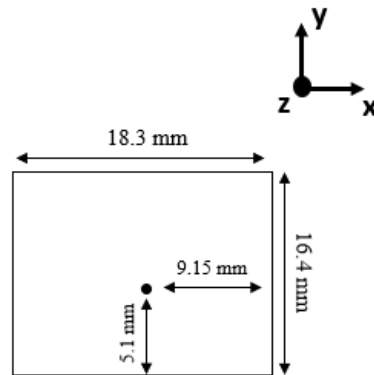
The MEA system, without a DS, is introduced in Section 4.3 and its performance parameters ( $S_{ij}$ , TARC, ECC and radiation patterns) computed. These serve as a reference for the performance achieved when a DS is later added. In Section 4.4 a DS is introduced into the MEA system (still consisting of four elements), consisting of an unshaped rectangular conducting plate located above (not in the same plane as) the four rectangular patch elements. The performance parameters are again computed and, as expected when one simply "invades" a radiating structure by inserting an additional piece of conductor, all are degraded. In Section 4.6 the shape synthesis process is applied to this structure by the shaping of the DS, utilising an objective function defined in Section 4.5. The contrast between the performance with the shaped DS and those in Section 4.3 (no DS) and Section 4.4 (unshaped DS) are very satisfactory indeed. Section 4.7 concludes the chapter.

## 4.2 Single Probe-Fed Microstrip Patch

The shape synthesis process used in this chapter is similar to that developed in Chapter 3. However, since we are dealing with a four-element MEA system, there are some adjustments which have to be made on the fitness function side to account for the additional S-parameters

resulting from the additional antenna elements. The other aspects of the shaping process, such as the process flow from HyperStudy to FEKO, and the decision-making flowchart in Figure 3.3-1, remain unchanged.

We begin the study by analyzing a single patch antenna which will be replicated to construct the 4-element MEA system. Figure 4.2-1 shows the dimensions of the pin-fed single patch on an infinite Rogers RT-Duroid 5880 substrate ( $\epsilon_r = 2.2$ ,  $\tan \delta = 0.0009$ ) with a thickness of 1.57 mm.



**Figure 4.2-1: Dimensions of a single patch antenna which will be used in MEA**

The pin-fed microstrip antenna was designed using the cavity approximation method [BALA 16]. The preliminary dimensions obtained from the cavity approximation were then “tuned” by verifying results obtained using a method of moments analysis in FEKO. The operational bandwidth selected for these antennas is 300 MHz (5.55 GHz – 5.85 GHz), which is a fractional bandwidth of 5.3%. The computed S-parameter responses for this single patch is shown in Figure 4.2-2; the highest reflection coefficient magnitude of - 4.6 dB occurs at 5.55 GHz, with a deep null at about<sup>38</sup> 5.78 GHz. Figure 4.2-3, shows the far-field radiation pattern of the antenna at the center

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<sup>38</sup> The lowest input reflection coefficient is not precisely at the centre frequency, but will be when the four patch elements are arranged into a compact MEA system in Section 4.3.

frequency. The far-field pattern is symmetrical about the broadside direction, as expected from the patch antenna. The peak directivity of 6.5 dBi is in the broadside direction.

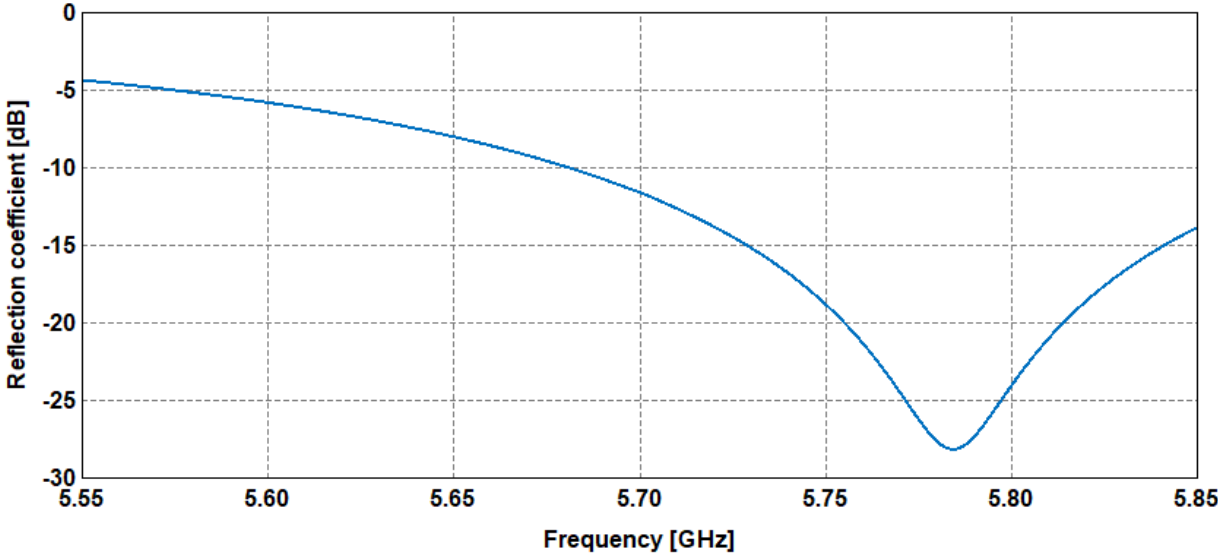


Figure 4.2-2: Computed reflection coefficient of the single isolated patch antenna

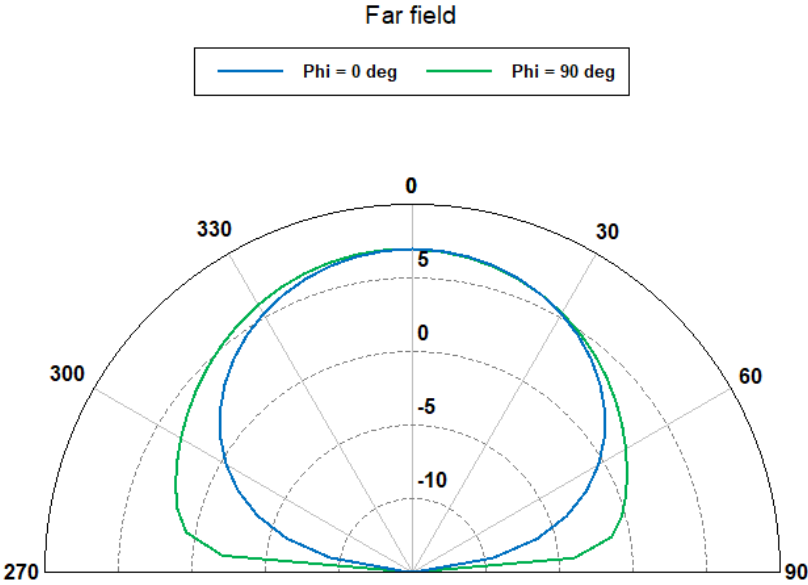


Figure 4.2-3: Computed total directivity of the isolated patch antenna at f=5.7 GHz.

### 4.3 MEA System Consisting of Four Patch Antennas

After validating the results for the single patch antenna, we introduce three additional yet identical antennas that are placed an equal distance from each other. Figure 4.3-1 shows the layout and dimensions of this 4-element MEA system. The antennas are equidistant from each other to maintain symmetry about the vertical and horizontal axes. The antennas are mirrored about the axes and separated by 7.6 mm or  $0.14\lambda_0$ . The design was forced to be symmetric to simplify the shaping process as will be discussed in the following sections.

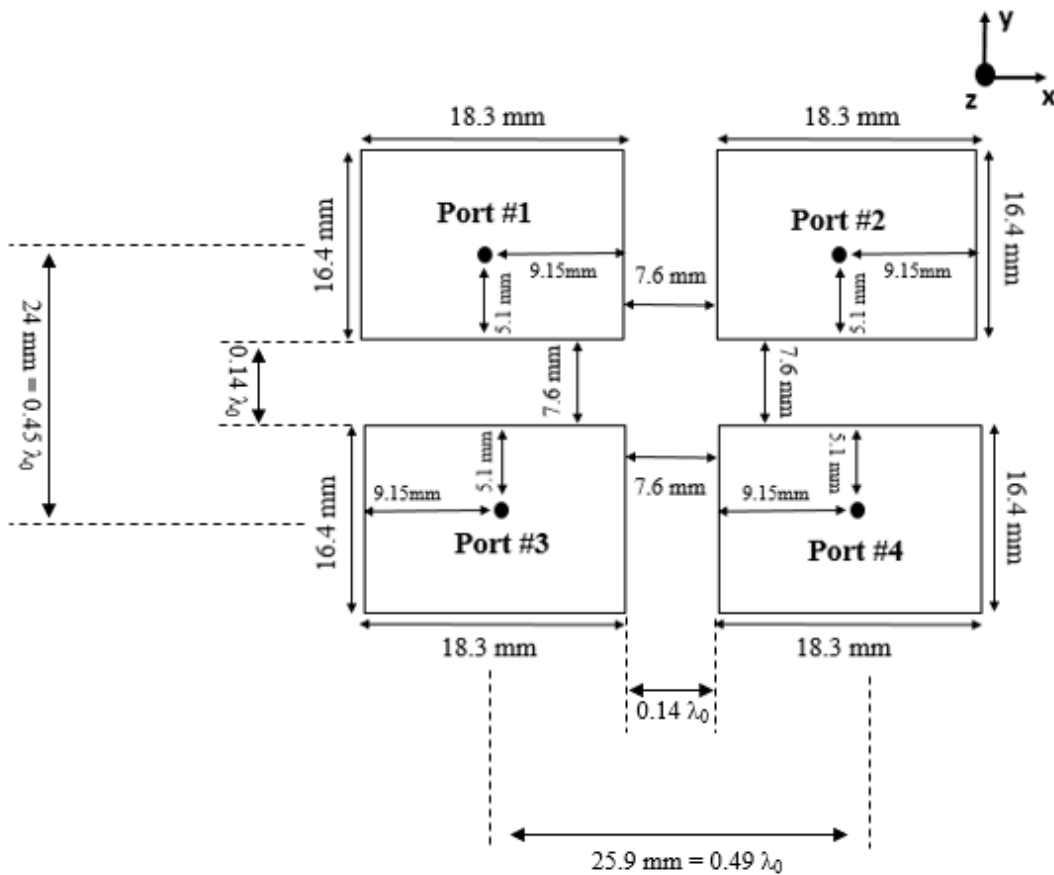


Figure 4.3-1: Dimension of MEA consisting of four patch antennas

Similar to the situation in Chapter 3, due to the symmetric nature of this design and the isotropic nature of the materials, many of the S-parameters are equal to each other. As such, we can focus

on examining a subset of the S-parameters when studying the S-parameter response of this MEA system. The equivalences are as follows:

- $S_{11} = S_{33} = S_{22} = S_{44}$
- $S_{12} = S_{21} = S_{34} = S_{43}$
- $S_{13} = S_{42} = S_{24} = S_{31}$
- $S_{14} = S_{23} = S_{32} = S_{41}$

For simplicity, we will only examine four S-parameters ( $S_{11}$ ,  $S_{12}$ ,  $S_{13}$ , and  $S_{14}$ ) since they capture all the information needed to evaluate the S-parameter properties of this model. Figure 4.3-2 shows the S-parameter response for the four antennas. We note that there is a slight improvement in the reflection coefficient ( $S_{11}$ ) where the maximum value is now 0.5 dB lower compared to the single antenna case in Figure 4.2-2. The highest coupling magnitude is  $S_{12}$  which is between the horizontally adjacent antennas #1 and #2, where the maximum value is -15 dB. The coupling coefficients for vertically adjacent antennas,  $S_{13}$ , also shows a plot with values which is similar to  $S_{12}$ . These are expected results since the horizontally and vertically adjacent elements are geometrically closer to each other than the diagonal elements pairs. Nonetheless, these coupling values are still significantly higher than the -20 dB mutual coupling requirement for MIMO applications [MEI 18].

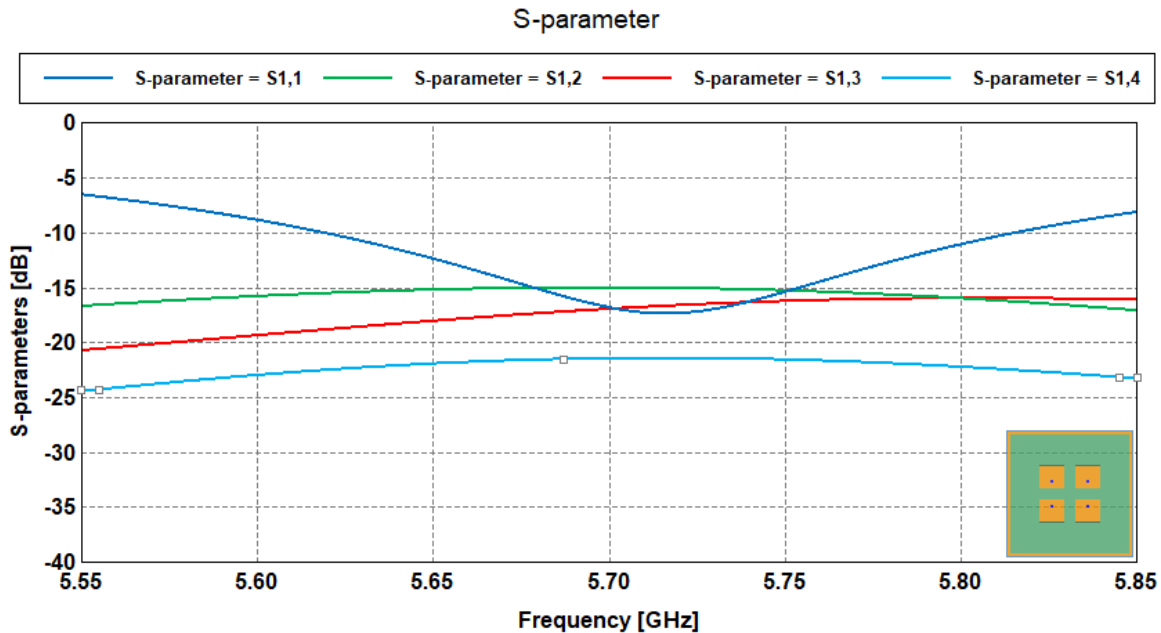
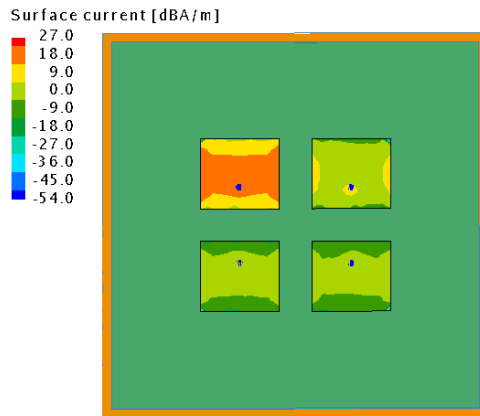


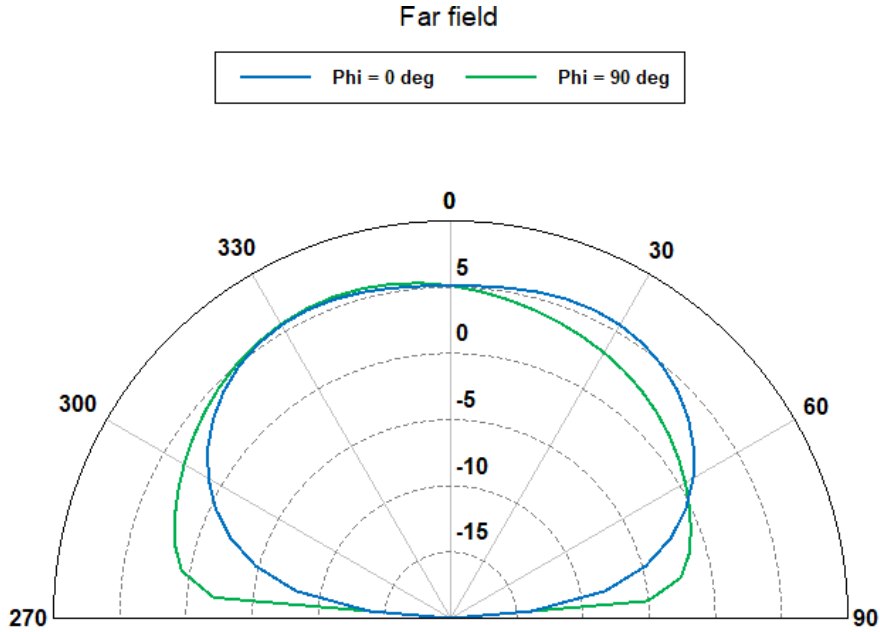
Figure 4.3-2: Computed S-parameters of the MEA (four patches without DS).

Figure 4.3-3 shows the surface currents on the model when antenna#1 is excited while the remaining three antennas are terminated with a 50 Ohm load. We can see that there are induced surface currents on the non-excited antennas due to coupling. The induced currents are highest on antenna#2; at the location of port#2 has an induced current which is 9 dB lower than that at port#1. At the location of ports #3 and #4 the induced surface currents are 18 dB lower than that at port#1.



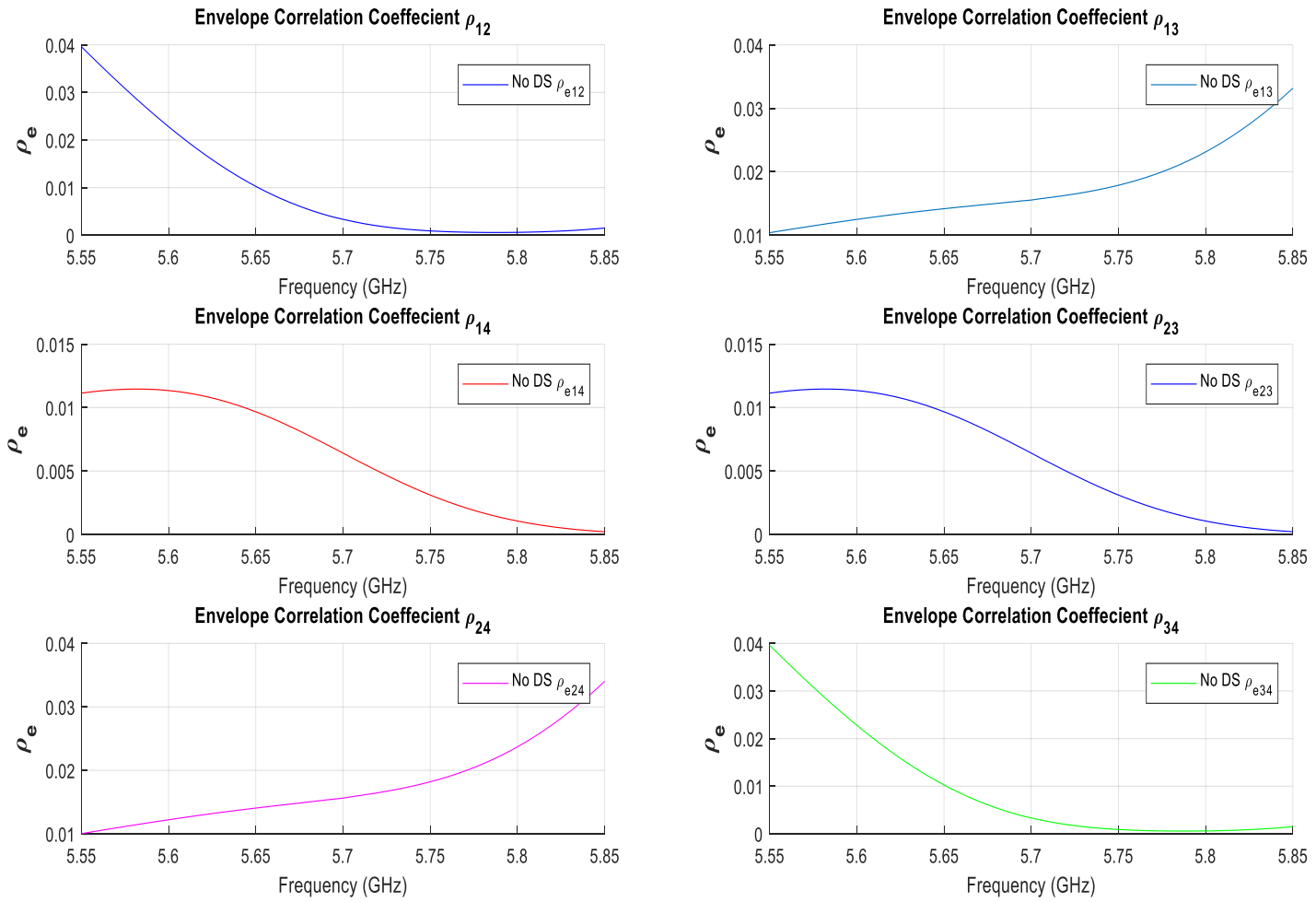
**Figure 4.3-3: Surface current density on the MEA (four patches and no DS) when port#1 is driven, and the remaining ports are terminated. The frequency is 5.7 GHz.**

Figure 4.3-3 shows the far-field radiation pattern of antenna#1 while the remaining antennas are terminated. The mutual coupling between antenna#1 and the remaining antennas has slightly deformed the radiation pattern. Compared to the isolated patch case in Figure 4.2-3, we see that Figure 4.3-4 shows the peak of the pattern is no longer pointing to broadside. We also note a slight decrease of 1.5 dB in the peak directivity; this is especially clear in the  $\phi = 90^\circ$  cut. However, there is no significant change noted with respect to the beam-width of the patterns.



**Figure 4.3-4: Computed total directivity pattern of the MEA when port#1 is driven and the remaining ports are terminated. The frequency is 5.7 GHz.**

The calculation of ECC between two antennas using the 3D far-fields method, which was presented in section 2.3.7, is still valid in the presence of additional antenna elements (i.e for N-element MEA where  $N > 2$ ). However, the S-parameter based method in equation 2.3-54 is not valid as it does not account for additional coupling from elements other than antenna#2 on antenna#1. The plots captured in Figure 4.3-5 are calculated from the radiation patterns of each antenna. We can see that due to the symmetry of the design, we have three unique plots. Namely, for horizontally adjacent antennas  $\rho_{e_{1,2}} = \rho_{e_{3,4}}$ , vertically adjacent antennas  $\rho_{e_{1,3}} = \rho_{e_{2,4}}$ , and diagonal antenna pairs  $\rho_{e_{1,4}} = \rho_{e_{2,3}}$ . Furthermore, the ECC results seem to be in agreement with the mutual coupling levels seen between the antennas. In Figure 4.3-2 we noted that the highest mutual coupling levels of -15 dB are seen between horizontally adjacent antennas. Figure 4.3-5 confirms these results, as it shows that horizontally adjacent antennas (1,2 and 3,4) have the highest envelope correlation of 0.04. Vertically adjacent antennas (1,3 and 2,4) saw the second highest mutual coupling levels in Figure 4.3-2 and this is once again confirmed by the ECC plots in Figure 4.3-5 where the highest ECC for  $\rho_{e_{1,3}} = \rho_{e_{2,4}}$  is approximately 0.038. Finally, the lowest ECC is between diagonal antennas (1,4 and 2,3), which have an ECC of 0.011 or better.



**Figure 4.3-5: ECC for MEA (four patches no DS)**

The specific expression for the TARC of a four-element MEA is slightly different compared to that of the 2-element MEA written down in Section 2.3.6. Starting from the general definition for TARC provided in expression (2.3-33), namely

$$\Gamma_a^t = \sqrt{\frac{\sum_{i=1}^N |b_i|^2}{\sum_{i=1}^N |a_i|^2}} \quad (4.3-1)$$

the relation

$$\mathbf{b} = \mathbf{S}\mathbf{a} \quad (4.3-2)$$

for the 4-element case becomes

$$\begin{bmatrix} b_1 \\ b_2 \\ b_3 \\ b_4 \end{bmatrix} = \begin{bmatrix} S_{11} & S_{12} & S_{13} & S_{14} \\ S_{21} & S_{22} & S_{23} & S_{24} \\ S_{31} & S_{32} & S_{33} & S_{34} \\ S_{41} & S_{42} & S_{43} & S_{44} \end{bmatrix} \begin{bmatrix} a_1 \\ a_2 \\ a_3 \\ a_4 \end{bmatrix} \quad (4.3-3)$$

Defining  $a_1 = a_0 e^{j\psi_0}$  as the reference excitation phase and amplitude; we can define the excitations of the remaining three antennas as having the same amplitude as well as a phase shift relative to  $a_1$ .

$$\begin{bmatrix} a_1 \\ a_2 \\ a_3 \\ a_4 \end{bmatrix} = \begin{bmatrix} a_0 \\ a_0 e^{j\psi_1} \\ a_0 e^{j\psi_2} \\ a_0 e^{j\psi_3} \end{bmatrix} \quad (4.3-4)$$

The system of equations can then be expanded to:

$$b_1 = a_0(S_{11} + S_{12}e^{j\psi_1} + S_{13}e^{j\psi_2} + S_{14}e^{j\psi_3}) \quad (4.3-5)$$

$$b_2 = a_0(S_{21} + S_{22}e^{j\psi_1} + S_{23}e^{j\psi_2} + S_{24}e^{j\psi_3}) \quad (4.3-6)$$

$$b_3 = a_0(S_{31} + S_{32}e^{j\psi_1} + S_{33}e^{j\psi_2} + S_{34}e^{j\psi_3}) \quad (4.3-7)$$

$$b_4 = a_0(S_{41} + S_{42}e^{j\psi_1} + S_{43}e^{j\psi_2} + S_{44}e^{j\psi_3}) \quad (4.3-8)$$

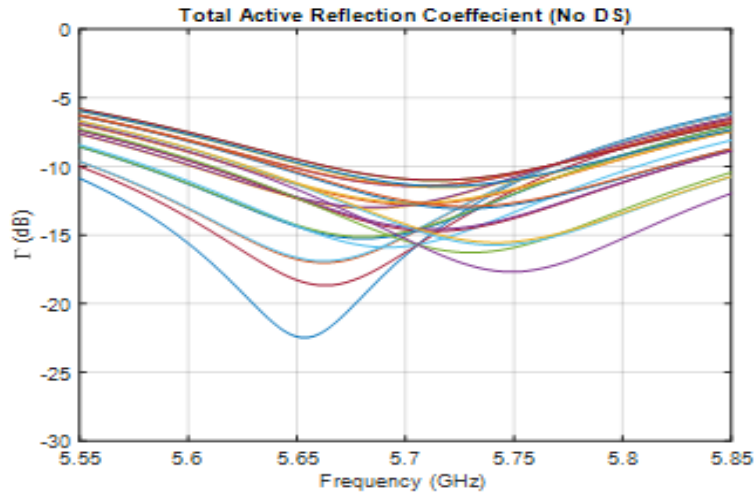
finally,

$$\Gamma_a^t = \sqrt{\frac{a_0^2 (b_1^2 + b_2^2 + b_3^2 + b_4^2)}{4 a_0^2}} \quad (4.3-9)$$

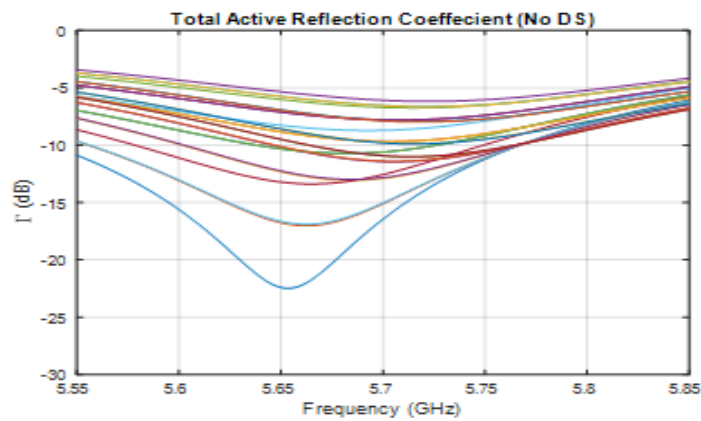
Capturing all possible variations between the variable phases will be difficult. So we will only consider a subset of plots by keeping one psi variable constant and changing the other two psi's in steps of 45°. For example, Figure 4.3-6 shows the TARC curves calculated when  $\psi_1$  is kept constant<sup>39</sup> while  $\psi_2, \psi_3$  are varied from 0° to 180° in steps of 45°. Figure 4.3-6 captures the TARC when excitation signals have constant phase between antenna#1 and antenna#2 and when antennas #3 and #4 have excitation signals with changing phase. The worst-case TARC in this scenario is less than -6 dB, which is acceptable.

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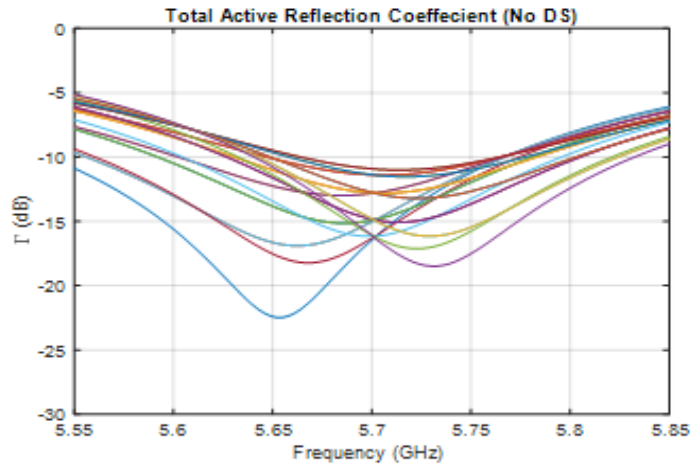
<sup>39</sup> In this context, the constant phase is fixed to 0°. Note that in all cases  $\psi_0$  which corresponds to antenna#1 is always fixed to 0°.



**Figure 4.3-6:** TARC of MEA consisting of 4 patch antennas and no DS. These plots are taken when Antenna #1 and #2 have constant phase, while antenna #3 and #4 have a varying phase.



**Figure 4.3-7:** TARC of MEA consisting of 4 patch antennas and no DS. These plots are taken when Antenna #1 and #3 have constant phase, while antenna #2 and #4 have a varying phase.



**Figure 4.3-8: TARC of MEA consisting of 4 patch antennas and no DS. These plots are taken when Antenna #1 and #4 have constant phase, while antenna #2 and #3 have a varying phase.**

In all three TARC scenarios presented above, Figure 4.3-7 shows that the worst TARC case is when antennas #1 and #3 are kept in phase while varying the phases for antennas #2 and #4. Particularly, this happens when antennas #2 and #4 are both  $180^\circ$  or completely out of phase with antennas #1 and #3. The TARC values in all scenarios are acceptable as they are close to the -6 dB requirement for MIMO, with the exception of scenario 2 in Figure 4.3-7 where the worst case plot is consistently greater than -6 dB [SHAR 13].

#### **4.4 The 4-Element MEA & an Unshaped Decoupling Surface**

Having established the benchmark performance of the 4-element antenna as MIMO antenna in section 4.3, we can introduce the non-coplanar decoupling surface. Figure 4.4-1 below shows the added decoupling surface which is printed on the same substrate material as the microstrip antennas RT-Duroid 5880 (thickness 1.57 mm). The substrate containing the DS is raised 20 mm above the antennas (along the normal axis), the DS is pixelated in 3.5 mm square cells. The 20 mm distance of separation between the antennas and DS was selected as such in order for the DS to be sufficiently close to the antennas so that any changes on the DS shape would have an impact

on the antennas S-parameters. The copper-backing on the board removed in order to allow for proper far-field transmission and decoupling<sup>40</sup> of the antennas. The DS is divided into small cells for shaping, as was done in Chapter 3, the dimensions are shown in Figure 4.4-2. To limit the number of variables controlling the shaping process, we have chosen to only shape the DS.

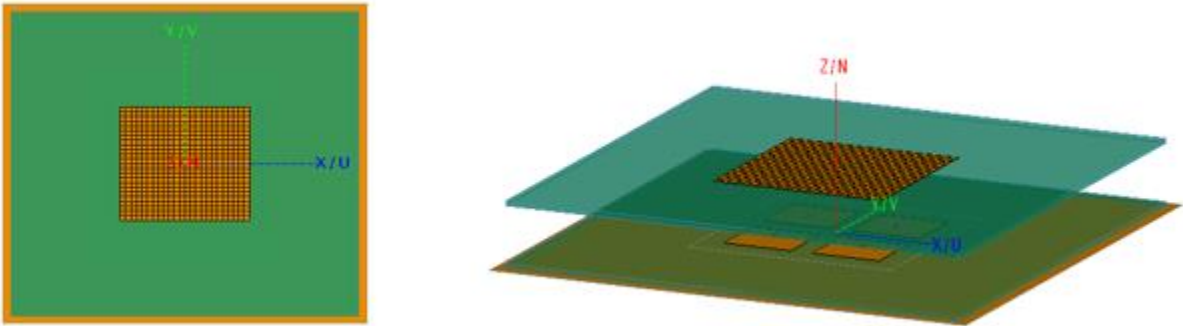


Figure 4.4-1: Top and Iso view of the MEA after introducing the DS

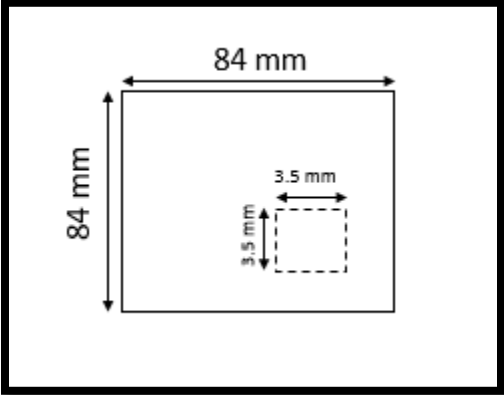


Figure 4.4-2: Dimensions of the DS

The S-parameter plots after introducing the DS are shown in Figure 4.4-3. We can see that all the reflection and mutual coupling coefficients ( $S_{12}$ ) deteriorate significantly. The maximum value for

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<sup>40</sup> Decoupling will not occur until after the DS is shaped.

$S_{11}$  is now over -5 dB in the upper frequencies.  $S_{14}$  shows significant increase with a maximum value of -6 dB after introducing the DS. This is due to the effect of the unshaped DS acting as reflector in close proximity to the antennas.

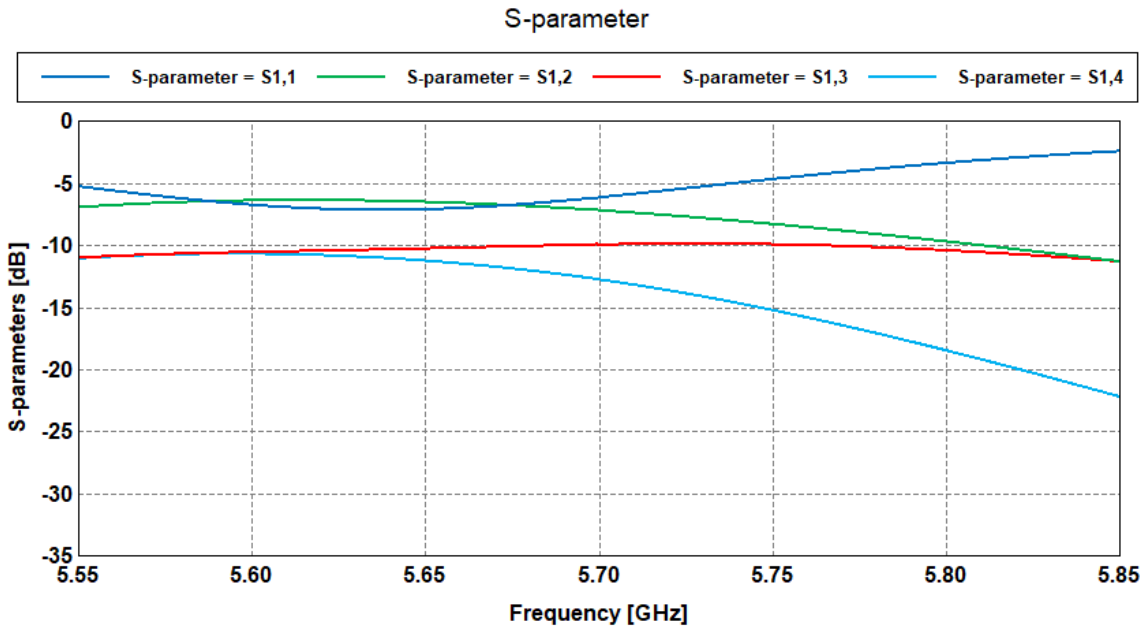
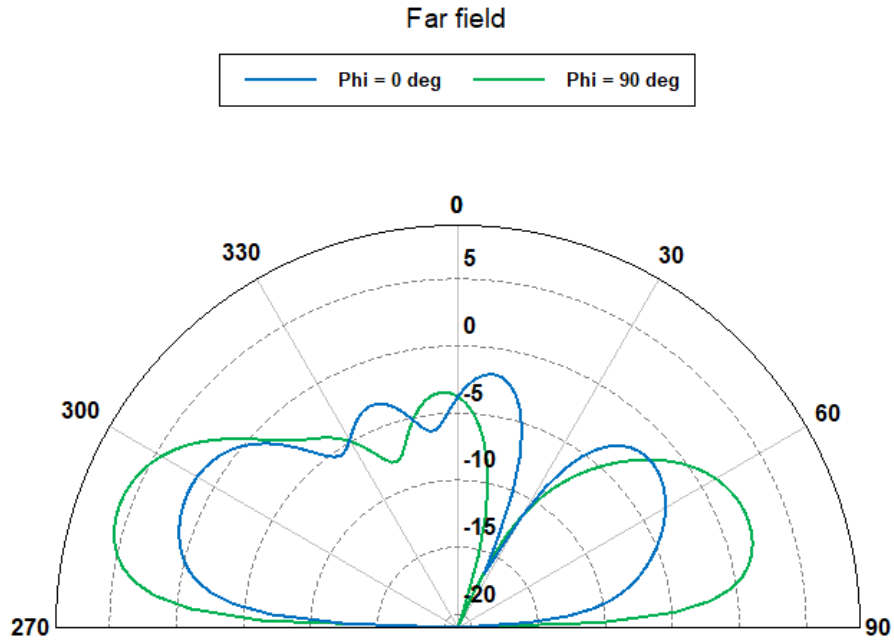
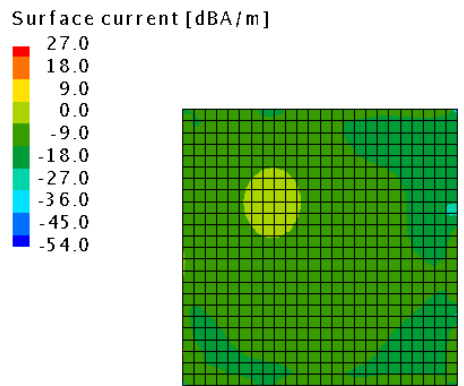


Figure 4.4-3: Computed S-parameter plots of the MEA consisting of four patches and the unshaped DS.

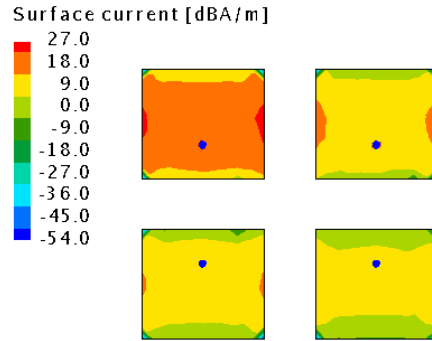
The introduction of the DS also deforms the far-field radiation pattern of antenna#1 as seen in Figure 4.4-4, since the unshaped decoupling surface is a sheet of copper the radiation pattern is reflected which causes deformation in the pattern. The setup and results here show that the far-fields are transmitted in the direction parallel to the antennas and the DS, thus resembling the far-field behaviour of waveguide antenna without the sidewalls.



**Figure 4.4-4: Computed total directivity pattern of the MEA consisting of four patches and the unshaped DS, when port#1 is driven and remaining ports are terminated. The frequency is 5.7 GHz**



**Figure 4.4-5: Surface current on DS when port#1 is driven and the remaining ports are terminated. The frequency is 5.7 GHz**



**Figure 4.4-6: Surface current on the MEA when port#1 is driven and the remaining ports are terminated.  
The frequency is 5.7 GHz.**

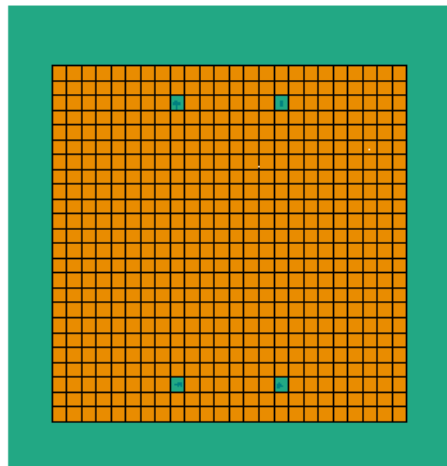
Figure 4.4-5 and Figure 4.4-6 show the surface currents on the structure when antenna#1 is excited while the remaining antennas are terminated. There is a significant increase of 9 dB for the induced surface currents on antennas #2, #3 and #4. Prior to introducing the DS, Figure 4.3-3 was showing that the current at port#2 was 9 dB lower than that at port#1; after introducing the unshaped DS, the induced current at port#2 is roughly equal to that at port#1. This validates the results seen in the S-parameter where there was a significant increase in the coupling after introducing the DS into this model. However, such a result is expected since the DS is constructed from copper, which reflects any incoming waves from antenna #1 (excited) back to the terminated antennas causing an increase in the induced currents on the surfaces of neighboring antennas, thus increasing the magnitude of induced currents.

## 4.5 Objective Function Used in the Shape Synthesis Process

The objective function for this numerical experiment was formulated with the following goals:

- To reduce the coupling between all antenna elements ( $S_{12}, S_{13}, S_{14}$ ) to a value of -20 dB or lower.
- To maintain the reflection coefficient at less than -5 dB.
- To ensure the total directivity peak in the far-field is no less than 5 dBi and points within +/- 5° of broadside.

One of the difficulties faced when completing this numerical experiment was the computation duration needed to complete a full-wave analysis of the model and evaluate the fitness function. The reason for the extended computation time is the large design footprint combined with a fine moment method meshing and a more sophisticated fitness function. In order to reduce the computational cost of this experiment, we opted not to include the ECC in the objective function, as this value was sufficiently low in the starting mother structure (without DS). Additionally, we chose to enforce left/right and up/down symmetry on the DS<sup>41</sup>. This was achieved by configuring a single binary bit (gene) to control retention/removal of four cells that are mirrored across the axes. Figure 4.5-1 shows an example where a single binary bit was set to zero whilst the remaining variables are set to one.



**Figure 4.5-1: Symmetry is retained after removal of cells in the DS**

To achieve the optimization goals for this section, we can formulate the objective function for the reflection coefficient value as follows<sup>42</sup>:

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<sup>41</sup> No pixelization was performed on the actual antenna elements since we do not plan to shape them.

<sup>42</sup> Note that we are once again merely asking the shape synthesizer to maintain the reflection coefficient at -5 dB and not improve it.

$$e_{11}(f_q) = \begin{cases} |S_{11}(f_q)|^{dB} - |-5|^{dB} & \text{if } |S_{11}(f_q)|^{dB} - |-5|^{dB} < 0 \\ 0 & \text{Otherwise} \end{cases} \quad (4.5-1)$$

The mutual coupling requirements can be encoded into the objective function:

$$e_{12}(f_q) = \begin{cases} |S_{12}(f_q)|^{dB} - |-20|^{dB} & \text{if } |S_{12}(f_q)|^{dB} - |-20|^{dB} < 0 \\ 0 & \text{Otherwise} \end{cases} \quad (4.5-2)$$

$$e_{13}(f_q) = \begin{cases} |S_{13}(f_q)|^{dB} - |-20|^{dB} & \text{if } |S_{13}(f_q)|^{dB} - |-20|^{dB} < 0 \\ 0 & \text{Otherwise} \end{cases} \quad (4.5-3)$$

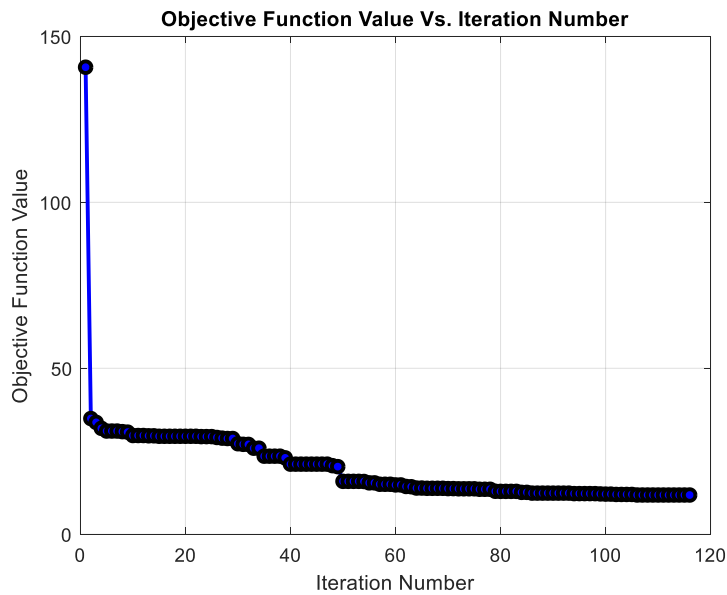
$$e_{14}(f_q) = \begin{cases} |S_{14}(f_q)|^{dB} - |-20|^{dB} & \text{if } |S_{14}(f_q)|^{dB} - |-20|^{dB} < 0 \\ 0 & \text{Otherwise} \end{cases} \quad (4.5-4)$$

After selecting the appropriate weights for each error term, we can combine these terms into a single function.

$$F_{Obj}(f) = 0.35 \sum_{q=1}^{13} |e_{11}(f_q)|^2 + 0.3 \sum_{q=1}^{13} |e_{12}(f_q)|^2 + 0.2 \sum_{q=1}^{13} |e_{13}(f_q)|^2 + 0.15 \sum_{q=1}^{13} |e_{14}(f_q)|^2 \quad (4.5-5)$$

To apply the far-field pattern requirement, we enforced a constraint on the far-field results for the function similar to what was done in Section 3.5. Figure 4.5-2 shows the objective function value during each iteration. We can see a big drop after the first iteration; this is because, as discussed earlier in this chapter, after adding the unshaped DS the S-parameter and radiation pattern plots were very far from meeting the objectives defined, and so there was a lot of room for rapid improvement. In the subsequent iterations, as the objective values approach the set values, the

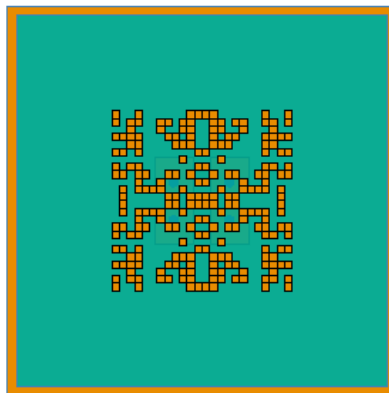
improvements get less and less significant. The objective function values finally plateau at the 110<sup>th</sup> iteration.



**Figure 4.5-2: Objective function value during the shape synthesis of the MEA consisting of four patches and a DS.**

## 4.6 Shape Synthesis Results

The optimum result from the shape synthesis process is shown in Figure 4.6-1. Symmetry is retained about the vertical and horizontal axes. No changes or shaping was done on the actual antennas that reside in the bottom board of this model.



**Figure 4.6-1: DS after the shaping process**

In Figure 4.6-2 the S-parameter plots are shown after the shaping process is completed. The reflection coefficient  $S_{11}$  has been maintained and is still better than the -5 dB requirement. The mutual coupling S-parameters  $S_{12}, S_{13}, S_{14}$  show improved isolation between the antenna elements. Prior to introducing the DS, in the reference model,  $S_{12}$  and  $S_{13}$  showed a maximum value of -15 dB. These values have been reduced to -20 dB or better. Similarly,  $S_{14}$  in the reference model had a maximum value of -20 dB; after shaping the DS this reduced to -22.5 dB.

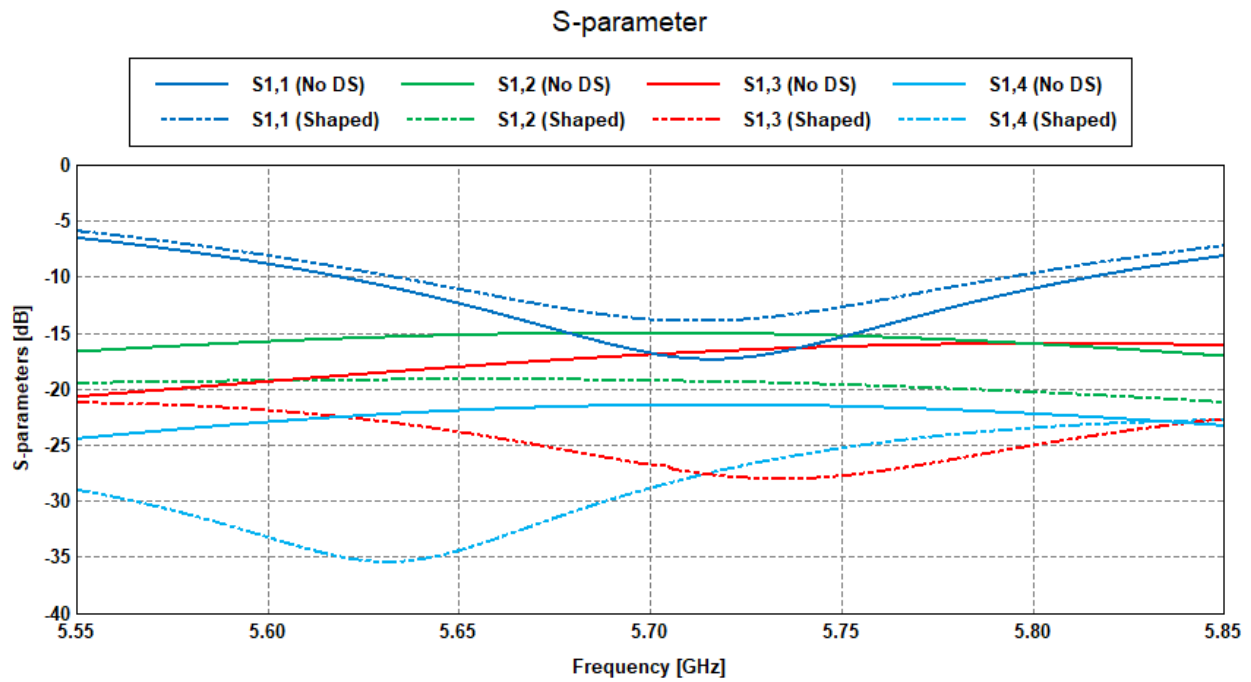
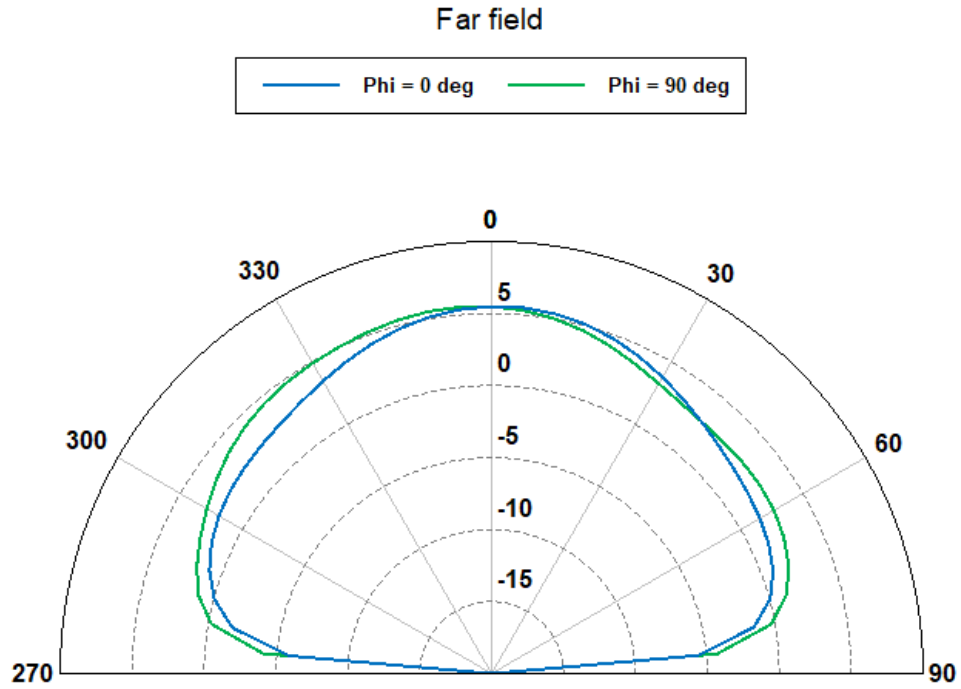


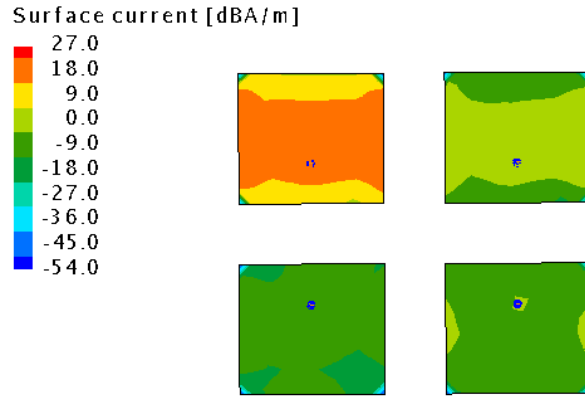
Figure 4.6-2: Computed S-parameter plots of the 4 patch MEA and shaped DS.

The radiation patterns meet the required constraints imposed in the objective function. Figure 4.6-3 shows the far-field total directivity patterns in the  $\phi = 0^\circ$  and  $\phi = 90^\circ$  cuts now have peaks of 5dBi in the broadside direction. Recall that Figure 4.3-4 revealed the distortion of the far-field pattern (before introduction of the DS), where the peak directivity was in the  $\theta = 30^\circ$  direction in the  $\phi = 0$  cut, and in the  $\theta = -25$  in the  $\phi = 90^\circ$  cut.

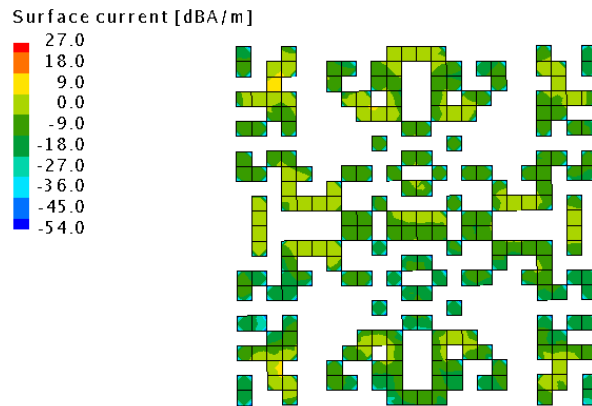


**Figure 4.6-3: Computed total directivity principal plane patterns of the MEA and shaped DS when port#1 is driven and the remaining ports are terminated. The frequency is 5.7 GHz**

The induced surface current density at the location of port#2 has been reduced to 18 dB lower than that at the driven port#1, as shown in Figure 4.6-4. Prior to introducing the DS, the induced current at port#2 was only 9 dB lower than at port#1, as was previously shown in Figure 4.3-3.

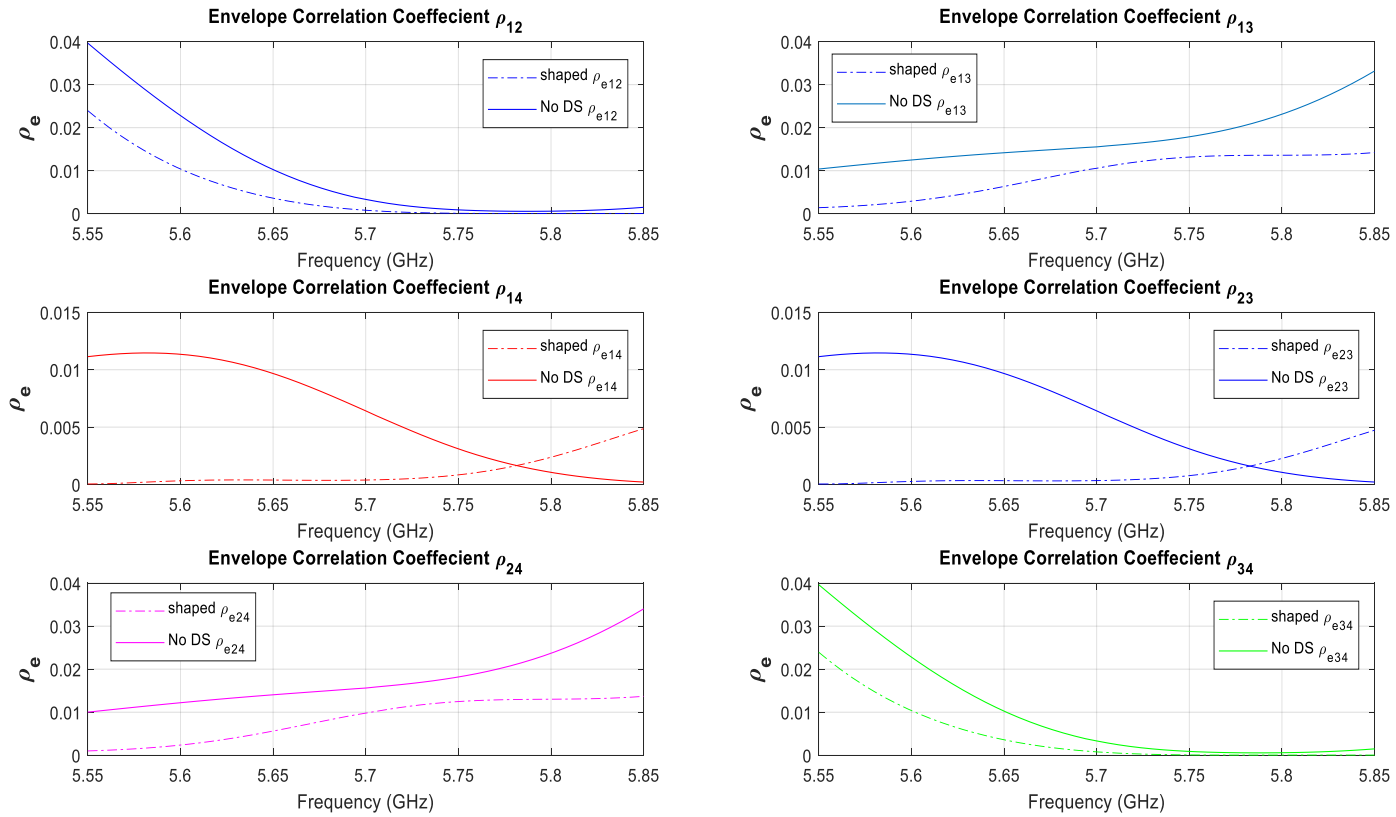


**Figure 4.6-4: The surface current density on the MEA after the DS has been shaped when port#1 is driven and the remaining ports are terminated. The frequency is 5.7 GHz.**



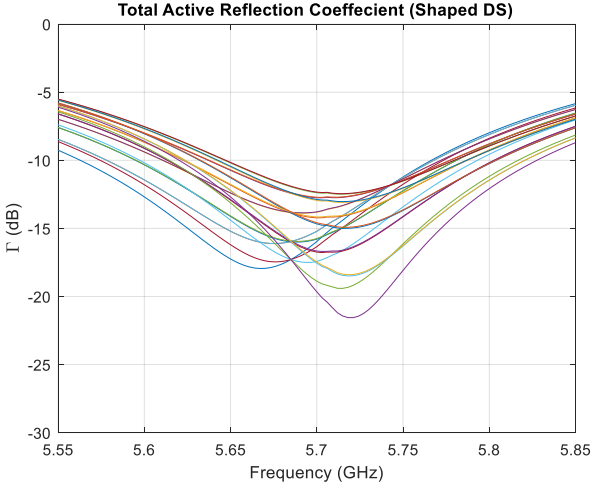
**Figure 4.6-5: The surface current density on the shaped DS when port#1 is driven and the remaining ports are terminated. The frequency is 5.7 GHz.**

The ECC plots are provided in Figure 4.6-6 for the structure without a DS, as well as the structure with a shaped DS. Although the ECC was not explicitly included in the objective function, the shaping process nevertheless still improve the value in most cases. The only exception was the ECC for diagonal pairs  $\rho_{e_{1,4}} = \rho_{e_{2,3}}$  where the ECC values increased compared to the reference case, but only in the upper portion of the band of interest. Nevertheless, in all scenarios the ECC is still sufficiently low for MIMO applications.

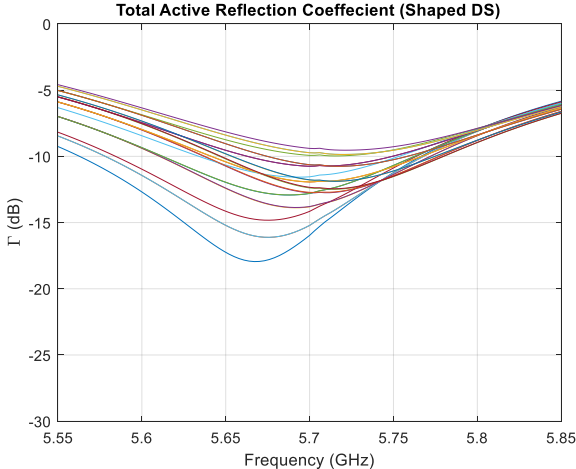


**Figure 4.6-6: ECC plots for MEA consisting of four patch antennas without a DS, compared to adding a shaped DS.**

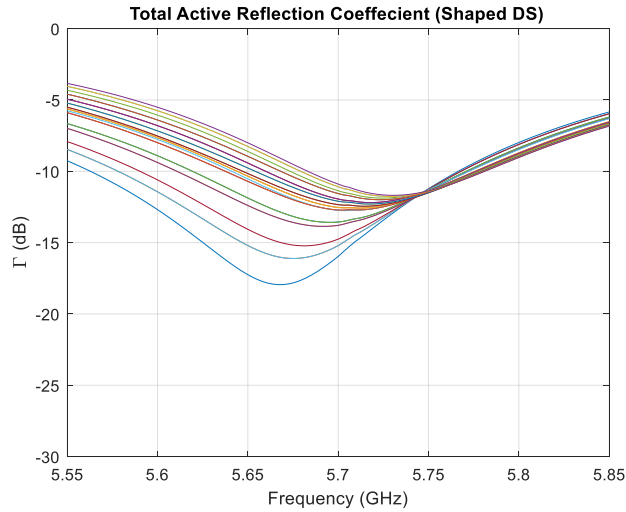
The TARC plots in Figures 4.6-7, 4.6-8 and 4.6-9 show a worst-case value of about -5 dB, with significant improvement in the lower frequencies of the band. Prior to introducing and shaping the DS, the highest TARC values were those shown in Figure 4.3-7, close to -5 dB across the entire bandwidth. After introducing and shaping the DS, we see that the values drop to -10 dB or lower in all cases over a significant portion of the frequency band.



**Figure 4.6-7: TARC of MEA consisting of 4 patch antennas and shaped DS. These plots are taken when Antenna #1 and #2 are constant while antenna #3 and #4 have a varying phase.**

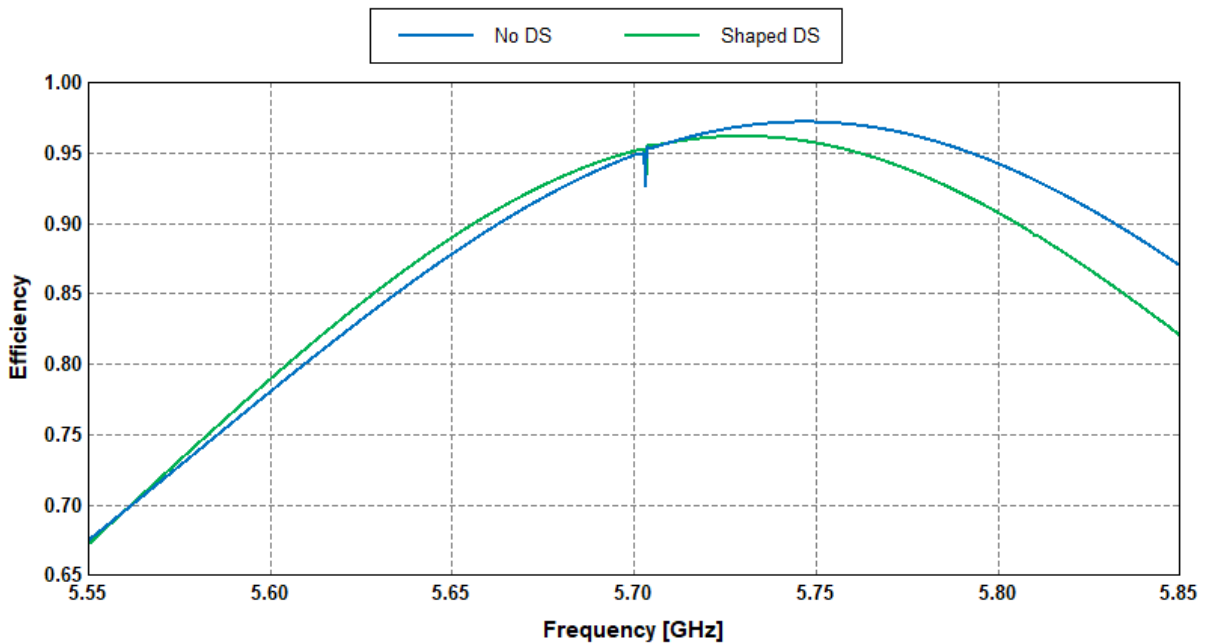


**Figure 4.6-8: TARC of MEA consisting of 4 patch antennas and shaped DS, taken when antenna 1 and 3 are constant and antenna 2 and 4 are varying phase.**



**Figure 4.6-9: TARC of MEA consisting of 4 patch antennas and shaped DS Taken when Antenna #1 and #4 are constant and antennas #2 and #3 are varying phase.**

The radiation efficiency plots provided in Figure 4.6-10 show that there are no significant changes between the reference case (4 patches without DS) and the 4 patches with a shaped DS.



**Figure 4.6-10: Simulated radiation efficiency plots before and after shaping DS**

## 4.7 Concluding Remarks

In this chapter we built on the numerical experiments in Chapter 3. We extended the application of the shape synthesis process to a 4-element MEA system, and one in which the decoupling surface (DS) is located in a different plane to that of the four rectangular patch elements of the MEA system. The success of the shape synthesis process was again demonstrated in a number of steps (viz. four elements without a DS; four elements with an unshaped DS, four elements with a shape synthesized DS). The shaping process was able to improve the inter-element isolation while maintaining acceptable input reflection coefficients and correcting the skewing of the radiation patterns of the individual elements<sup>43</sup>.

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<sup>43</sup> That had been present before introduction of the DS.

## Chapter 5 General Conclusions

### 5.1 Contributions

The main contribution of this thesis is the introduction, for the first time, of a shape synthesis process for the design of decoupling surfaces as a novel technique for decreasing the mutual coupling (and related performance parameters) between the elements of an MEA system. The procedure presented combines the use of a decoupling surface (DS) in an MEA system to decrease such coupling, and the use of shape synthesis to provide the physical structure of the antenna elements and the DS. The current state of knowledge in MEA decoupling contains several research works that start with having two or more antenna elements with relatively good characteristics which are then decoupled further using one or more schemes<sup>44</sup>. It appears to be rarely the case that the decoupling is considered in the initial antenna design stage. Although most of the existing works on this topic makes use of a DS, the procedure followed to arrive at a particular DS geometry is seldom explained in a quantitative manner. Much time-consuming (and no doubt ingenious) cut-and-try is most likely needed.

This inspired the author to introduce and apply the systematic method described in this thesis, which provides a detailed insight on how one can come up with a DS structure via an optimization algorithm. The optimization process utilizes the GA which makes use of inputs collected from the electromagnetic simulations in order to reduce the mutual coupling between different antenna elements without sacrificing other antenna characteristics.

Strong evidence supporting the soundness of the proposed shaping method is provided in this thesis by demonstrating its application to three different classes of radiating-element/DS arrangements. These are the 2-element “H-plane” case (Section 3.4) and the “E-plane” case (Section 3.5), in which the DS resides on the same plane as the radiating elements, and the 4-element case of Chapter 4 where the radiating elements and DS are located on different planes.

In the first numerical experiment, the 2-element co-planar “H-plane” case, we demonstrated how shape synthesizing both the antenna elements and the DS reduced the mutual coupling levels by at least 9 dB while maintaining the required input reflection coefficient levels. This was similarly

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<sup>44</sup> Reviewed in Section 2.4.

achieved for the 2-element coplanar “E-plane” case, but for the latter we showed how the skewing of the radiation pattern from the broadside direction<sup>45</sup> could be successfully included in the objective function used in the shaping process. Such flexibility, offered by the shape synthesis process for the class of problem at hand, has not been shown to be possible using any other approach. In the said coplanar “E-plane” case (the second numerical experiment), the ECC was incorporated into the objective function in addition to the pattern control, coupling levels and input reflection coefficients. We demonstrated how the shape synthesis process was able to reduce the input reflection coefficients and mutual coupling by up to 5 dB and 6 dB, respectively. The ECC was reduced by more than 30%.

The third numerical experiment consisted of a 4-element element MEA system with the decoupling surface placed on a plane some distance above that on which the four radiating elements reside. This demonstrated that the shape synthesis method can be applied to all classes of decoupling surface configuration thus far published in the open literature. The experiment also showed that the shaping approach can be applied to MEA systems with more than just two elements. By shaping the DS, the levels of the mutual coupling between elements were reduced by at least 5dB at their highest points over the frequency band and more so over the remainder of the band, and the “skewing” of the total directivity pattern removed.

A lesser yet noteworthy contribution of this thesis is the analysis and examples provided in Section 2.3.7. This showed the significant discrepancy in calculating ECC using the precise approach (in terms of radiation patterns) and that using the approximate (but convenient) route utilizing the multiport S-parameters, when lossless versus low-loss radiating elements are assumed. Although this discrepancy is often mentioned in the literature, it does not appear to have been presented as conspicuously as was done in Section 2.3.7, with actual numerical examples that emphasize how the discrepancy increases as the radiating elements become more lossy.

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<sup>45</sup> The skewing of the peak total directivity direction in the  $\phi = 0^\circ$  principal plane was corrected from  $30^\circ$  to be the broadside direction. We note that shaping could be done with pattern peaks pointing in directions other than broadside.

## 5.2 Possible Future Work

The work presented in this thesis provides adequate information to form a good starting point for incorporating shape synthesis tools in the development of MEA systems. Important future work in this topic could include the use of additional “non-shape related” variables in the optimization algorithm. For example, one could include the substrate material properties and thicknesses in the shaping algorithm as additional “optimizable” parameters (eg. a set of feasible values could be defined based on what is available commercially) to meet bandwidth and radiation efficiency design optimization goals. When performing DS shape synthesis for non-coplanar arrangements, one could also include the separation between the two substrates<sup>46</sup> as an optimization variable.

The shape synthesis procedure described is enormously computationally intensive. The numerical experiments completed in section 3.4 and 3.5 took the GA roughly one week to reach convergence. However, with constant increases computational power coupled with advancements in artificial intelligence and machine learning, the shape synthesis process could become increasingly attractive. The work in this thesis has brought together tools, to implement the novel shape synthesis application, that were not specifically developed for antenna problems. It is possible that customization of these tools to be more antenna-specific<sup>47</sup> would speed up the shaping procedure considerably. Given both the facts just mentioned, it might be that antenna designers could even explore shape synthesizing decoupling structures for the many-element MEA systems needed for massive MIMO applications that can contain tens or hundreds of antenna elements.

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<sup>46</sup> One for the radiating elements, and one for the decoupling surface. The properties of these two substrates also need not be the same.

<sup>47</sup> This would take considerable effort.

# Appendix A: Multiple-Input Multiple-output (MIMO) Systems

## A.1 MIMO Diversity Techniques

The principle of operation of MIMO systems is to divide the communication channel into several sub-channels. This is done in order to provide the receiver with multiple copies of the same signal, as shown in Figure A-1. That way, the likelihood of all the channels fading simultaneously is reduced.

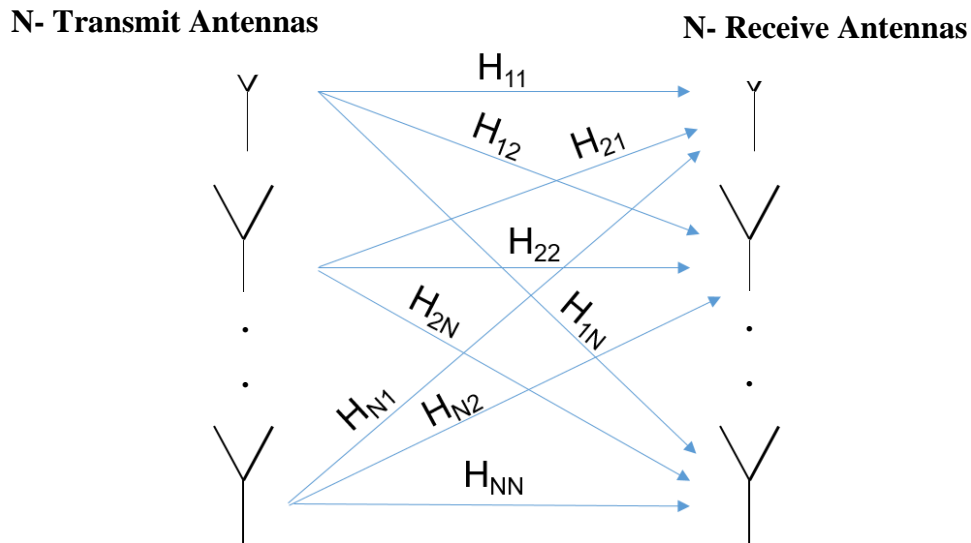


Figure A-1: Multipath Propagation in MIMO

The concept of MIMO has been enabled by the enhancements in modern signal processing systems. As mentioned in chapter 1, diversity is key in order to excite uncorrelated sub-channels and achieve good MIMO performance. Antenna diversity can be done in three different ways [FRAN 08]. The first method is spatial diversity where antennas are physical separated by a distance. The second method is polarization diversity, where the signals transmitted have orthogonal polarizations. It should be noted that the diversity techniques mentioned are not mutually exclusive [FRAN 08], as it is possible to have a system which combines different antenna diversity schemes. One example of this is two antennas which are physically separated, and one is rotated  $90^\circ$  with respect to other. Finally, the third method is pattern diversity; in this method the radiation patterns are orthogonal to each other. Pattern diversity can be achieved using a single or multiple physical structure. In other words, some pattern diversity designs are comprised of a

single physical antenna which excites different modes in order to achieve orthogonal radiation patterns, such designs are known as multi-mode antennas. In this thesis we are not interested in multi-mode antennas, we will only focus on decoupling antennas as a part of spatial diversity setup.

There are two main frequency band at which wireless communication MIMO is expected to operate. 1) 2.45 GHz - 6 GHz, conventional SISO systems have been operating at this frequency. The other frequency band is expected to be mm-wave band, namely 30-300 GHz. In this study we are interested in reducing the mutual coupling (decoupling) of antennas at microwave frequencies as it has been established that decoupling antennas is harder at lower frequencies [CHEN 18]. MEA's operating at the mm-wave bands undergo scattering differently, ultimately mm-wave MIMO faces a greater challenge which is ensuring that the environment has rich enough scattering and tackling the issue of increased material losses which ultimately leads to a higher pathloss [JENS 04].

## **A.2 Massive MIMO**

Massive MIMO is an extension of conventional MIMO where the number of antennas on a base station is in the order of tens or hundreds. The motivation for M-MIMO is the possibility to exploit the highly directive farfield pattern of MEA's in order to combat pathlosses at mm-wave frequencies. Implementation of M-MIMO has opened the door for non-linear MEA designs, typical designs are 2D and 3D (ex. cylindrical arrangement) MEA configurations. While such designs offer more compactness, they amplify the issues resulting from mutual coupling. In 2D and 3D designs an antenna element can have up to 6 adjacent elements [RUS13]. Although there is not a set number of antennas to define M-MIMO system, it has been reported that M-MIMO will employ a few hundred antennas [LARS 14]. As such, M-MIMO is strictly for base station applications as the geometric requirements for base station antennas is less stringent.

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