



National Library
of Canada

Bibliothèque nationale
du Canada

Canadian Theses Service · Service des thèses canadiennes

Ottawa, Canada
K1A 0N4

NOTICE

The quality of this microform is heavily dependent upon the quality of the original thesis submitted for microfilming. Every effort has been made to ensure the highest quality of reproduction possible.

If pages are missing, contact the university which granted the degree.

Some pages may have indistinct print especially if the original pages were typed with a poor typewriter ribbon or if the university sent us an inferior photocopy.

Previously copyrighted materials (journal articles, published tests, etc.) are not filmed.

Reproduction in full or in part of this microform is governed by the Canadian Copyright Act, R.S.C. 1970, c. C-30.

AVIS

La qualité de cette microforme dépend grandement de la qualité de la thèse soumise au microfilmage. Nous avons tout fait pour assurer une qualité supérieure de reproduction.

S'il manque des pages, veuillez communiquer avec l'université qui a conféré le grade.

La qualité d'impression de certaines pages peut laisser à désirer, surtout si les pages originales ont été dactylographiées à l'aide d'un ruban usé ou si l'université nous a fait parvenir une photocopie de qualité inférieure.

Les documents qui font déjà l'objet d'un droit d'auteur (articles de revue, tests publiés, etc.) ne sont pas microfilmés.

La reproduction, même partielle, de cette microforme est soumise à la Loi canadienne sur le droit d'auteur, SRC 1970, c. C-30.

NUMERICAL ALGORITHM FOR A CLASS OF OPTIMAL CONTROL PROBLEMS
WITH APPLICATIONS TO ROBOTICS AND ECOLOGY

by

TIJANI SELLAMI

A thesis

presented to the University of Ottawa

in partial fulfillment of the

requirements for the degree of

Master of Applied Science

in

The Department of Electrical Engineering

Faculty of Engineering



Tijani Sellami, Ottawa, Canada, 1988.

Permission has been granted to the National Library of Canada to microfilm this thesis and to lend or sell copies of the film.

The author (copyright owner) has reserved other publication rights, and neither the thesis nor extensive extracts from it may be printed or otherwise reproduced without his/her written permission.

L'autorisation a été accordée à la Bibliothèque nationale du Canada de microfilmer cette thèse et de prêter ou de vendre des exemplaires du film.

L'auteur (titulaire du droit d'auteur) se réserve les autres droits de publication; ni la thèse ni de longs extraits de celle-ci ne doivent être imprimés ou autrement reproduits sans son autorisation écrite.

ISBN 0-315-46853-X



UNIVERSITÉ D'OTTAWA
UNIVERSITY OF OTTAWA

To

my parents Aifa S. and Abdellah O.

ABSTRACT

A methodology has been developed for computing the time optimal controls for a system governed by a set of linear or nonlinear ordinary differential equations. The method has been used to compute the optimal controls for two types of robots : Cylindrical robot and a robot with an articulated arm with two links and an ecological system. The method is based on successive approximation of the time optimal controls by controls which are optimal for a suitable sequence of terminal problems. These later problems are easily solved by conjugate gradient technique. The method avoids solving nonlinear two-point-boundary-value problems thereby gaining some computational simplicity. Further, if a system is not even controllable our method will still provide a control that takes the system to a point nearest to the target. In this case it is not even necessary to assume the existence of time optimal controls.

Techniques of optimal control theory have been utilized in identifying the parameters of a physical system. For illustration these results are then used to determine the unknown of certain robotics and ecological systems.

ACKNOWLEDGEMENTS

The author wishes to express his indebtedness and deep gratitude to his advisor prof. N.U. Ahmed for his generous encouragement, understanding and guidance throughout this work, without which this thesis would not have been possible.

Special thanks are also due to Dr. T. Dabbous for his helpful suggestions and discussions.

Special thanks to the members of the system and control group, especially to Mr. S.S. Lim and Mr. L. Peng, for their suggestions and discussions.

The financial assistance of the Tunisian Government during the period of this research is gratefully acknowledged.

CONTENTS

ABSTRACT	iv
ACKNOWLEDGEMENTS	v
<u>Chapter</u>	<u>Page</u>
I. INTRODUCTION	1
II. TIME OPTIMAL CONTROL AND ALGORITHM	8
2.1 Introduction	8
2.2 General Control Problem	9
2.3 Time Optimal Control and Theoretical Basis for the Algorithm	11
2.4 Shooting Method and Gradient Technique	18
2.4.1 Shooting Method	18
2.4.2 Gradient Technique	21
2.5 Algorithm	22
2.6 Summary	25
III. NUMERICAL SIMULATION OF TIME OPTIMAL CONTROL WITH APPLICATIONS TO ROBOTICS AND ECOLOGY	26
3.1 Introduction	26
3.2 Robot with Cylindrical Coordinates	27
3.3 Horizontal Articulated Arm Robot With Two Links	39
3.4 Ecological System	56
3.5 Summary	61
IV. NUMERICAL SIMULATION OF SYSTEM IDENTIFICATION WITH APPLICATIONS TO ROBOTICS AND ECOLOGY	
4.1 Introduction	62
4.2 Problem Formulation and Necessary Conditions of Optimality	63

4.3 Algorithm and Flowchart	65
4.3.1 Algorithm	65
4.3.2 Flowchart	67
4.4 Numerical Simulation	68
4.4.1 Automelec ACR robot	68
4.4.2 IBM Articulated Arm Robot	72
4.4.3 Ecological System	75
4.5 Summary	78
V. CONCLUSIONS	79
<u>Appendix</u>	<u>Page</u>
A. LAGRANGE EXTRAPOLATION	81
B. FORTRAN CODES	82
REFERENCES	94

LIST OF FIGURES

<u>Fig.</u>	<u>Page</u>
3.1 Sketch of the cylindrical robot	27
3.2.a Optimal control u_1^*	32
3.2.b Optimal control u_2^*	32
3.3.a Optimal State d	33
3.3.b Optimal state \dot{d}	33
3.3.c Optimal state θ	34
3.3.d Optimal state $\dot{\theta}$	34
3.4 Cost as a function of number of iterations	35
3.5.a Optimal costate ψ_2^*	36
3.5.b Optimal costate ψ_1^*	36
3.6 Optimal Controls u_1^* and u_2^* at different final times (not optimal time).	37-38
3.7 Sketch of the articulated arm robot	39
3.8.a Optimal control u_1^* (case a)	43
3.8.b Optimal control u_2^* (case a)	43
3.9.a Optimal states θ_1^* and $\dot{\theta}_1^*$	44
3.9.b Optimal states θ_2^* and $\dot{\theta}_2^*$	45
3.10 The cost function φ	46
3.11 Optimal controls at different final times (not the optimal time)	47-48
3.12 Optimal controls u_1^* and u_2^* (case b)	49
3.13.a Optimal states θ_1^* and $\dot{\theta}_1^*$	50

3.13.b	Optimal states θ_2^* and $\dot{\theta}_2^*$	51
3.14	The cost function φ	52
3.15	Optimal controls u_1^* and u_2^* (case c)	53
3.16.a	Optimal states θ_1^* and $\dot{\theta}_1^*$	54
3.16.b	Optimal states θ_2^* and $\dot{\theta}_2^*$	55
3.17.a	Optimal states x_1^* and x_2^*	58
3.17.b	Optimal states x_3^*	59
3.18	Optimal control u_1^*	59
3.19	The cost function φ	60
4.1	Init., comp. and true trajectories (cylindrical robot)	70-71
4.2	Init., comp. and true trajectories (articulated arm robot)	73-74
4.1	Init., comp. and true trajectories (ecological system)	76-77

LIST OF TABLES

<u>Table</u>	<u>Page</u>
3.1 Cost as a function of number of iterations	35
4.1 Convergence of the unknown parameter for different guesses (cylindrical robot)	68-69
4.2 Convergence of the unknown parameter for different guesses (articulated arm robot)	72
4.3 Cost as function of number of iterations (ecological system)	75

LIST OF SYMBOLS

\mathcal{U} =admissible set of controls

H =Hamiltonian

ψ =adjoint vector

$x^u(t)$ =response of the state equations to the control u at time t

x_d =desired state

\mathcal{A} =attainable set

ρ_H =Hausdorff metric

$F(t,x)$ =contingent function

$\varphi(t)$ =minimum distance between the attainable set and the target

$K(R^n)$ =class of all compact subsets of R^n .

F =force

T =torque

u =control vector

x =state vector

CHAPTER I

INTRODUCTION

In recent years much attention has been focused upon optimizing the behavior of systems described by a set of differential equations. The search for the control which attains the desired objective while minimizing (or maximizing) a defined system criterion constitutes the fundamental problem of optimization theory. A particular problem of optimization may concern with the minimization of the time required to drive a system from one desired state x_0 to some required target state x_d ; this is known as the time optimal control problem. Another class of optimization may deal with the identification of some unknown parameters introduced in the system equations. The main aim of this thesis is to study numerically the last two problems for a class of systems governed by a set of ordinary differential equations.

1.1 TIME OPTIMAL CONTROL

In the time optimal control problem, it is of interest to find the best control in the set of admissible controls that drives the system from one desired state to another in minimum time. It was LaSalle [29] who first gave a complete solution to this problem. In the last two decades the interest in time optimal control has increased significantly (see for example [1-13]). The time optimal control theory has been developed in the literature and has been used in many applications such as industrial robot.

The time optimal control of a system of rigid bodies connected in series by single-

degree of freedom joint is studied in [1]. A suboptimal feedback control, which provides a close approximation of the optimal control is developed. The suboptimal control is expressed in terms of switching curves for each of the controls. These curves are obtained from the linearized equations of motion for the system. Approximations are made for the effects of gravity loads and angular velocity terms in the nonlinear equations of motion. However, the possibility of singular time-optimal controls was not investigated.

About ten years later, the interest in time-optimality of robot motions increased significantly [2]-[6]. In [2], a theoretical procedure has been developed for comparing the performance of arbitrary selected admissible controls among themselves and with the optimal solution of a nonlinear optimal control problem. A recursive scheme converging to the time optimal solutions was presented. The approach has been applied to the approximately optimal control of a trainable manipulator with seven degrees of freedom, where the controller is used for motion coordination and optimal execution of objecthandling tasks. A similar, adaptive scheme for a two-link robot was given in [3]. In [4], the optimization problem was presented completely and the available mathematical design tools were discussed. In [5], an approximate time-optimal controller was combined with a linear regulator for the final phase of the trajectory. The dynamic programming method of synthesis of optimal trajectories was treated in [6]. Since the complexity of the necessary calculations for time-optimal controls is considerable, several schemes for simplification were proposed in [6]. The only method found to be suitable for such a complex optimization was based on dynamic programming. An algorithm for determining optimal velocity distribution for a given manipulator tip trajectory was elaborated in detail. The problem of time-optimal motions along arbitrary prespecified partial trajectories was studied in [7] and [8]. In

[7], a solution has been presented to the problem of minimizing the cost of moving a robotic manipulator along a specified geometric path subject to input torque/force constraints, taking the coupled, nonlinear dynamics of the manipulator into account. The proposed method uses dynamic programming to find the positions, velocities, accelerations, and torques that minimize the cost. As a numerical example, the path planning method was simulated for a two-jointed robotic manipulator. In [8], an optimal control policy which results in minimum-time motion for a robotic manipulator along any predetermined path in three dimensional space was presented. The technique permits the manipulator user to specify completely the path of the arm. The special case of rectilinear motion was studied in [9]. In this method there is only one degree of freedom, (the acceleration along the trajectory). The problem of time-optimal control motion of a load by a two-link manipulator is examined. The control functions are moments of forces applied to the axes of the manipulator joints and are limited in absolute values. Optimal and quasi-optimal controls were derived for rectilinear motion of the center of mass of a load from one point to another in such a way that the system becomes quiescent following the completion of the process. It was assumed that the mass of the manipulator is negligibly small in comparison to the mass of the load and that the manipulator links have equal lengths. The method of parameter optimization was used in [10] and [11]. It reduces the optimal control problem to the problem of finding a suitable parameterization for the control variables. In [12], a simple algebraic solution is obtained to an optimal control problem in joint space for a general robotic manipulator. Open-loop and closed-loop suboptimal solution were found by solving algebraic equations.

Recently, Geering et al [13] have presented a complete analysis for time optimal motions of three types of industrial robots: robot with cylindrical coordinates, robot

with spherical coordinates and a robot with an articulated arm. Techniques of optimal control theory have been utilized in the time optimal control problem. The maximum principle was used to formulate the problem as two-point boundary-value differential equations. In order to solve this problem the shooting method was used. It has been shown also that in the case of the IBM articulated arm robot, the shooting method alone did not work. Therefore, both the parameter optimization and the shooting techniques were utilized in order to find the switching times for the controls, starting with various control parametrizations, minimizing the problem duration and using the penalty approach for the prescribed final state. Convergence of the shooting algorithm was achieved with the Lagrange multiplier of the penalty term of the previous step as an initial guess for the final costate vector. In the other hand, to solve the optimal control problem, another popular technique so called gradient has been widely considered.

The gradient method consists of updating the controls using the gradient vector, in a way such that successive iterates produce maximum reduction of the cost. This technique has been applied to several optimal control problems by Kelley [15, 16], and Bryson and Denham [17]. Various modifications to include penalty functions and other methods of treating equality and inequality constraints have been presented in [18]-[24]. The gradient method has the ability to generate successively improved trajectories with very poor starting values. However, they tend to converge slowly as convergence is approached. To improve the convergence of the gradient method near the optimal trajectory, the conjugate gradient method can be used to update the controls. The convergence of the gradient and conjugate gradient methods for the singular, finite-dimensional quadratic minimization problems have been developed in [30]. A class of associated nonsingular quadratic problems is defined to show that the

gradient method has slower convergence on singular problems than on corresponding nonsingular approximations to the singular problems while the conjugate gradient method has more rapid convergence.

In this thesis, we have proposed a simpler technique for solving time optimal control problems without using the shooting method. Our algorithm is based on the conjugate gradient technique computing the optimal controls for a class of simpler terminal control problems over an arbitrary but variable time interval giving the distance of the attainable set to the desired target as a function of time and then locating the first minimum of this function over the given interval. If the first minimum is zero, then the corresponding time is the optimal time and the associated control is the time optimal control. In case there is no time optimal control our method still provides a control that takes the system to a point in the attainable set closest to the target. Thus our method does not require the assumption of existence of time optimal controls and it is also insensitive to the presence or absence of singular arcs. An added advantage of the algorithm is that it also provides a numerical technique for direct verification of controllability of the system. Our method also avoids the complexity of solving two-point boundary-value problems thereby gaining some computational simplicity.

In order to illustrate the usefulness of the proposed method, the algorithm has been used to compute the time optimal control of two robots : Cylindrical robot and a robot with an articulated arm with two links and an ecological system.

1.2. PARAMETER IDENTIFICATION

An important and essential aspect of modelling any physical system is the iden-

tification of parameters in the model equation.

Techniques of optimal control theory have been utilized in identification of parameters in a system described by a set of differential equations. Note that the identification problem could be considered as a special case of the general control problem [Ahmed 14].

The time optimal control algorithm developed in this thesis can be easily modified to compute the unknown parameters introduced in the differential equations which describe the evolution of a physical system. The modified algorithm can determine the unknown parameters so that the corresponding response of the model equation approximates as closely as possible the actual response of the physical system. For numerical illustration, the modified algorithm has been applied to two robots and an ecological system.

1.3 MAJOR CONTRIBUTION OF THE THESIS

Development of an efficient algorithm that can be used for computing time optimal controls as well as the unknown parameters for a class of systems governed by ordinary differential equations applied to robotics and ecology.

1.4 OUTLINE OF THE THESIS

In chapter II, a simple but efficient algorithm and its theoretical basis for computing the time optimal control are developed. Also, the necessary conditions of optimality for a general control problem and a brief discussion of the the gradient technique and the shooting methods are provided.

In chapter III, we consider the application of the proposed algorithm to two robots and an ecological system. In chapter IV, the algorithm developed in chapter II is modified to compute the unknown parameters introduced in the differential equations which describes the behavior of the system. In order to illustrate the effectiveness of the (modified) algorithm, we have considered the identification problem for practical systems (robotic and ecology). Finally, in chapter V conclusions of the thesis are presented.

CHAPTER II

TIME-OPTIMAL CONTROL AND ALGORITHM

2.1 INTRODUCTION

The behavior of a physical system over a certain time interval is generally described by a set of equations. These equations represent the mathematical relations between the states and the inputs to the system. Each system has one or more control variables (inputs) which can be altered in such a way as to create a desired response from the system. These controls may not be known a priori and in order to determine their magnitudes it is necessary to solve a control problem.

In many Engineering issues, some of the control problems seek the control variables that can transfer a system from an arbitrary fixed initial state to another desired in minimum time. This is known as the time optimal control problem. The solution of such problem requires the solution of a two-point boundary-value problem which is a difficult task.

In this chapter, we propose a simple but efficient algorithm for solving the time optimal control problem without using the shooting method. The method is based on the conjugate gradient technique. It avoids the solution of a two-point boundary-value problem and it consists of computing the optimal controls for a class of simpler terminal control problems over an arbitrary but variable time interval giving the dis-

tance of the attainable set to the desired target as a function of time and then locating the first minimum over the given interval. If the first minimum is zero, then the corresponding time is the optimal time and the associated control is the time optimal control. Further, if a system is not even controllable our method will still provide a control that takes the system to a point in the attainable set nearest to the target.

The algorithm and its theoretical basis are presented in section 5 and 3, respectively. The formulation of a general control problem and the corresponding necessary conditions of optimality are given in section 2. A brief discussion of the shooting method and the gradient technique is provided in section 4.

2.2 GENERAL CONTROL PROBLEM

Consider the system

$$\begin{aligned} \dot{x}(t) &= f(t, x(t), u(t)), & 0 < t \leq T \\ x(0) &= x_0 \end{aligned} \tag{2.1}$$

with the cost functional

$$J(x, u) = \int_0^T l(t, x(t), u(t)) dt + \phi(x(T)) \tag{2.1}'$$

Here $x(t) \in R^n$ denotes the state variable; $x_0 \in R^n$ is the initial condition and $u(t) \in R^m$ is the control variable. We assume that the control u takes values from a compact convex set $U \subset R^m$. Let

$$\mathcal{U} \equiv \{u : u \text{ is measurable; } u(t) \in U; t \geq 0\}$$

denote the set of admissible controls and

$$S \equiv \{(x, u) : x \in AC([0, T], R^n), u \in \mathcal{U}, \dot{x} = f(t, x, u) \text{ a.e. } t \in [0, T]\}.$$

denote the set of admissible pairs where $AC(I, R^n)$ denotes the class of all absolutely continuous functions on $I=[0, T]$ with values in R^n . The basic control problem is to find a pair $(x^*, u^*) \in S$ such that

$$J(x^*, u^*) \leq J(x, u) \text{ for all } (x, u) \in S.$$

A pair satisfying this property is called optimal.

The following theorem gives the necessary conditions that an optimal pair must satisfy.

Theorem 2.1 (necessary conditions of optimality)

Consider the control problem (2.1) – (2.1)' and suppose that f is measurable in t , C^1 in x and continuous in u and l is C^1 in all the variables. Define the Hamiltonian $H: [0, T] \times R^n \times R^n \times R^m \rightarrow R$ by

$$H(t, x(t), \psi(t), u(t)) = (f(t, x(t), u(t))\psi(t) + l(t, x(t), u(t))).$$

In order that the pair (x^*, u^*) be optimal, it is necessary that there exists a $\psi^* \in AC([0, T], R^n)$ such that the triple (x^*, u^*, ψ^*) simultaneously satisfies the following equations and inequalities

$$(i) \quad \dot{x}^*(t) = f(t, x^*(t), u^*(t)) \text{ a.e. on } [0, T]$$

$$x^*(0) = x_0$$

$$(ii) \quad \dot{\psi}^*(t) = -l_x(t, x^*(t), u^*(t)) - f_x'(t, x^*(t), u^*(t))\psi^*(t)$$

$$\psi^*(T) = (\partial\phi/\partial x)(x(T))$$

and

$$(iii) \quad H(t, x^*(t), \psi^*(t), u^*(t)) \leq H(t, x^*(t), \psi^*(t), u)$$

for all $u \in U$

Remark 2.2

The time optimal control problem is a special case of the optimal control problem for $l = 1$ and $\phi = 0$.

2.3 TIME OPTIMAL CONTROL AND THEORETICAL BASIS FOR THE ALGORITHM

In this section we present a general statement of the time optimal control problem and develop the theoretical basis for our algorithm.

Consider the system of nonlinear differential equations (2.1). We assume that for each $u \in U$ and $x_0 \in R^n$, the system (2.1) has a unique solution $x = x(t, x_0, u)$ for all finite $t \geq 0$. Let $x_d \in R^n$ be another given point. Our goal is to find a control $u^* \in U$ that drives the system (2.1) from the initial state x_0 to the desired target x_d in minimum time.

Let $\mathcal{U}(x_0, x_d)$ denote the set of all controls $\{u\} \in U$ that transfer the system (2.1) from the state x_0 to the state x_d in finite time. Clearly if $\mathcal{U}(x_0, x_d)$ is empty the question of time optimal control does not arise. However if $\mathcal{U}(x_0, x_d) \neq \emptyset$ then one may ask the question as to whether or not there exists a time optimal control ?

Suppose $\mathcal{U}(x_0, x_d) \neq \emptyset$ and let $u \in \mathcal{U}(x_0, x_d)$ and define

$$\tau(u) \equiv \inf \{t \geq 0 : x(t, x_0, u) \equiv x^u(t) = x_d\}$$

Clearly the question of existence of time optimal control is equivalent to the question of existence of a $u^* \in \mathcal{U}(x_0, x_d) \subset U$ such that

$$\tau(u^*) \leq \tau(u) \quad \forall u \in \mathcal{U}(x_0, x_d) \subset U.$$

In general the answer to the question of existence of time optimal control is negative. However if the contingent function $F(t, x) \equiv f(t, x, U)$ is convex for all $(t, x) \in [0, \infty) \times R^n$ and the system is controllable then one can prove the existence of time optimal controls. Given that a time optimal control exists, the *Pontryagin minimum principle* [24] gives a set of necessary conditions that an optimal control must satisfy.

For convenience of the reader we quote this result below.

Theorem 2.3 (Pontryagin's n.c of optimality)

In order that the pair $\{u^*, x^*\}$ be optimal with τ^* the optimal time it is necessary that there exists a $\psi^* \in AC([0, \tau^*], R^n)$ such that the triple $\{u^*, x^*, \psi^*\}$ simultaneously satisfies the following set of differential equations and inequalities

$$\dot{x}^* = f(t, x^*, u^*), \quad t \in [0, \tau^*], \quad x^*(0) = x_0, \quad x^*(\tau^*) = x_d$$

$$\dot{\psi}^* = -f_x(t, x^*, u^*)\psi^*$$

$$H(t, x^*(t), u^*(t), \psi^*(t)) \leq H(t, x^*(t), v, \psi^*(t))$$

for all $v \in U$ and for almost all $t \in I_{\tau^*} = (0, \tau^*]$, (2.2)

and

$$H(\tau^*, x^*(\tau^*), u^*(\tau^*), \psi^*(\tau^*)) + 1 = 0, \quad \text{where } H = (f, \psi). \quad \blacksquare$$

If a time optimal control exists then the above necessary conditions can be used to identify the controls that contain the optimal ones. In case the problem (2.2) has

a unique solution, any control computed by use of the necessary conditions is the optimal control.

If the set $\mathcal{U}(x_0, x_d)$ is empty, or, in particular, if there is no time optimal control, then the necessary conditions have no meaning. In that case one may like to find a control that drives the system from the point x_0 to one that is closest to the target x_d . Indeed for each $t \geq 0$ let

$$\mathcal{A}(t) \equiv \{\xi \in R^n : \xi = x^u(t) \text{ for some } u \in \mathcal{U}\}$$

denote the attainable set. Under fairly general assumptions we can show that the set $\mathcal{A}(t)$ is compact for each finite $t \geq 0$, and that $t \rightarrow \mathcal{A}(t)$ is continuous in a suitable metric. Before we can prove this result we need the following definition.

Let $K(R^n)$ denote the class of all compact subsets of R^n . For $A \in K(R^n)$ define

$$\rho(z, A) = \inf\{\|z - a\|, a \in A\}.$$

Clearly for $z \in A$, $\rho(z, A) = 0$. For $A, B \in K(R^n)$ define

$$\rho_H(A, B) = \max\{\sup_{z \in B} \rho(A, z), \sup_{z \in A} \rho(z, B)\}$$

and note that $(K(R^n), \rho_H)$ is a metric space. The metric ρ_H is called the *Housdorff metric*.

Lemma 2.4

Suppose f is measurable in t for $t \geq 0$, and continuous in the rest of the variables and that U is a compact subset of R^r with the contingent function $F(t, x) = f(t, x, U)$

being a closed convex subset of R^n . Suppose there exists an $h \in L_1^{\text{loc}}$, possibly depending on the set U such that

$$\|f(t, x, u)\| \leq |h(t)|[1 + \|x\|] \quad \text{a.e}$$

for all $x \in R^n$ and $u \in U$. Then for each finite $t \geq 0$, the attainable set $\mathcal{A}(t)$ is compact and further the set valued map $t \rightarrow \mathcal{A}(t)$, $t \geq 0$, is continuous in the Hausdorff metric ρ_H .

Proof

Let x^u denote any solution of (2.1) corresponding to $u \in \mathcal{U}$. Then under the given assumptions it follows from Gronwall's lemma [14] that

$$(1 + \|x^u(t)\|) \leq (1 + \|x_0\|)\exp\left(\int_0^t |h(\theta)| d\theta\right) \text{ for } u \in \mathcal{U} \text{ and } 0 \leq t < \infty.$$

Hence, for $t < \infty$, the attainable set $\mathcal{A}(t)$ is a bounded subset of R^n . Further, one can show that for any compact interval $I = [0, T]$, $T < \infty$, the set of trajectories $\mathcal{X} \equiv \{\{x^u\} \in C(I, R^n) \text{ satisfying (2.1) for some } u \in \mathcal{U}\}$ is a bounded subset of $C(I, R^n)$ and it is also equicontinuous. Hence by *Ascoli-Arzelà* theorem [31] the set \mathcal{X} is a relatively compact subset of $C(I, R^n)$. That the set \mathcal{X} is closed follows from the fact that the contingent function $F(t, x) \equiv f(t, x, U)$ is closed and convex for $(t, x) \in I \times R^n$. This, combined with relative compactness, implies that the set \mathcal{X} is a compact subset of $C(I, R^n)$. Hence for each $t \in I$, the set $\mathcal{A}(t) \equiv \{\xi \in R^n : \xi = x(t) \text{ for } x \in \mathcal{X}\}$ is closed. Thus $\mathcal{A}(t)$ is both closed and bounded and consequently compact. The continuity of the set valued map $t \rightarrow \mathcal{A}(t)$ follows from the inequality

$$\rho_H(\mathcal{A}(t_1), \mathcal{A}(t_2)) \leq \beta \int_{t_1}^{t_2} |h(\theta)| d\theta, \quad 0 \leq t_1 \leq t_2 \leq T, \quad (2.3)$$

where β is a finite positive number depending only on the bound of the set \mathcal{X} restricted to the interval $[0, t_2]$. Since $0 \leq T < \infty$ but otherwise arbitrary, the lemma is proved. \blacksquare

The following result is an immediate consequence of the above lemma.

Theorem 2.5

Consider the system (2.1) with x^u denoting its solution corresponding to $u \in \mathcal{U}$. Let g be any real valued continuous function on R^n and suppose the hypotheses of lemma (2.4) hold. Then for any finite $\tau > 0$, there exists an $u^* \in \mathcal{U}$ such that $g(x^{u^*}(\tau)) \leq g(x^u(\tau))$ for all $u \in \mathcal{U}$. That is

$$g(x^{u^*}(\tau)) = \inf\{g(\xi), \xi \in \mathcal{A}(\tau)\}.$$

Proof

Since g is continuous and $\mathcal{A}(\tau)$ is compact for $\tau < \infty$, g restricted to $\mathcal{A}(\tau)$, attains both its minimum and maximum on $\mathcal{A}(\tau)$. Hence there exists an $u^* \in \mathcal{U}$ satisfying the assertion of the theorem. ■

For each finite $\tau \geq 0$, define

$$\varphi(\tau) = \rho(x_d, \mathcal{A}(\tau)) = \inf\{\rho(x_d, \xi), \xi \in \mathcal{A}(\tau)\}. \quad (2.4)$$

Since for each $0 \leq t < \infty$, $\mathcal{A}(t)$ is compact by lemma (2.4), and the function $\xi \rightarrow \rho(x_d, \xi)$ is continuous on R^n , it follows from theorem (2.5), that $\rho(x_d, \xi), \xi \in \mathcal{A}(t)$, attains both its maximum and minimum on $\mathcal{A}(t)$. Hence $\varphi(\tau), 0 \leq \tau < \infty$, is well defined and it gives the distance of the points in $\mathcal{A}(\tau)$ nearest to the target x_d . Since $\mathcal{A}(\tau)$ is compact, the set

$$\mathcal{N}_0(\tau) \equiv \{\xi \in \mathcal{A}(\tau) : \|\xi - x_d\| = \varphi(\tau)\} \quad (2.5)$$

is nonempty and there is at least one control $u^* \in \mathcal{U}$ such that $x^{u^*}(\tau) \in \mathcal{N}_0(\tau)$. Hence the control u^* must satisfy the necessary conditions of optimality for the following

terminal control problem:

$$\dot{x}(t) = f(t, x, u), 0 \leq t \leq \tau < \infty, x(0) = x_0, \quad (2.6)$$

$$J(\tau, u) = (1/2)\|x^u(\tau) - x_d\|^2 \equiv \min. \quad (2.7)$$

The corresponding necessary conditions of optimality is given by the Pontryagin's minimum principle which is quoted below for convenience of reference.

Theorem 2.6

In order that $u^* \in \mathcal{U}$ or equivalently the pair $\{u^*, x^*\}$ be optimal for the terminal control problem (2.6), (2.7) it is necessary that there exists $\psi \in AC(I_\tau, R^n)$, $I_\tau = [0, \tau]$, (absolutely continuous functions from I_τ to R^n) such that the triple $\{u^*, x^*, \psi^*\}$ satisfies the following differential equations and inequalities:

$$(i) \dot{x}^* = f(t, x^*, u^*) \text{ for a.e } t \in (0, \tau], x^*(0) = x_0, \quad (2.8i)$$

$$(ii) \dot{\psi}^* = -f_x'(t, x^*, u^*)\psi^* \text{ for a.e } t \in [0, \tau), \psi^*(\tau) = x^*(\tau) - x_d, \quad (2.8ii)$$

and

$$(iii) H(t, x^*(t), u^*(t), \psi^*(t)) \leq H(t, x^*(t), v, \psi^*(t)) \text{ a.e in } [0, \tau], \quad (2.8iii)$$

and for all $v \in U$. ■

In general the system (2.1) may not be controllable from x_0 to x_d , or the time optimal control may not have a solution but, by virtue of theorem (2.5) and the subsequent discussion, the problem (2.6) -(2.7) has always a solution. Therefore the necessary conditions of optimality (2.8i)-(2.8iii) are nonvacuous and can be used

to determine $\varphi(\tau)$ (see 2.4) or equivalently the controls satisfying (2.7). This is particularly true whenever the necessary conditions (2.8) are also sufficient or they have a unique solution. In case the Hamiltonian is convex in the control variable, the above necessary conditions are also sufficient. For this the reader is referred to [Clarke, Theorem 5.4.2, pp.219-220]. If the original time optimal control problem has a solution the set

$$I_0 \equiv \{t \geq 0 : \varphi(t) = 0\} \quad (2.9)$$

is nonempty and hence $\tau^* \equiv \inf(I_0)$ exists and there is a control $u^* \in \mathcal{U}$ such that $x^{u^*}(\tau^*) = x_d$. If on the otherhand $\varphi(t) > 0$ for all $t \geq 0$ then the system is not controllable with respect to the pair $\{x_0, x_d\}$. In this situation one may be content with a control that steers the system from x_0 to an attainable point, that is a point $\xi^* \in \bigcup \mathcal{A}(\tau)$, $\tau \geq 0$ closest to the target x_d . Indeed since $t \rightarrow \mathcal{A}(t)$ is continuous from $[0, \infty)$ to $K(R^n)$ in the Housdorff metric ρ_H , and

$$|\varphi(t) - \varphi(s)| \leq \rho_H(\mathcal{A}(t), \mathcal{A}(s)) \quad (2.10)$$

for all $t, s \in [0, \infty)$, the function φ is continuous and non-negative and therefore has a minimum over any given compact interval $I_T = [0, T], T < \infty$.

Letting $m^0(T)$ denote the minimum of φ over the interval $I_T \equiv [0, T]$, and $\tau^0(T)$ the first time in I_T at which φ equals $m^0(T)$ it is clear that $T \rightarrow m^0(T)$ is a monotone nonincreasing function of T and that $\tau^0(T) \leq T$. The algorithm developed in the paper and presented in the following section is able to compute $m^0(T)$ and $\tau^0(T)$ for each finite T .

Note that if T^* is the first time at which $m^0(T^*) = 0$ (in the event time optimal control exists) then $\varphi(T^*) = 0$, and $m^0(T) = 0$ for all $T \geq T^*$, though $\varphi(T), T > T^*$,

need not be zero unless $\mathcal{A}(T^*) \subset \mathcal{A}(T)$ for all $T \geq T^*$.

As indicated earlier, the time optimal control problem has been (numerically) solved in the literature by the use of the shooting method which involves the solution of two-point boundary value problem. It should be noted, however, that the application of gradient technique to the (numerical) solution of time optimal control has not been considered yet in the literature. In the next section we present, for convenience, a brief description for both shooting and gradient techniques.

2.4 SHOOTING METHOD AND GRADIENT TECHNIQUE

In this section, we present a brief discussion of the shooting method and the gradient technique.

2.4.1 Shooting Method

Two-point boundary-value problems associated with systems of linear and non-linear ordinary differential equations occur in many branches of mathematics, engineering, and the various sciences. In these problems, conditions are specified at the endpoints of an interval, and a solution of the differential equations over the interval is sought which satisfies the given endpoint conditions.

Consider the control problem

$$\text{minimize}\{J(u) = \int_{t_1}^{t_2} l(t, x(t), u(t))dt\} \quad (2.11)$$

subject to the dynamic constraint

$$\begin{aligned}\dot{x}(t) &= f(t, x(t), u(t)), & u \in \mathcal{U} \text{ and } t_1 < t_2 \\ x(t_1) &= x_0 \\ x(t_2) &= x_1\end{aligned}\tag{2.12}$$

where $x(t) \in R^n$ and $u(t) \in R^m$ denote the state and control variables, respectively. The initial and terminal state vectors are given by $x_0 \in R^n$ and $x_1 \in R^n$, respectively.

Define the Hamiltonian $H(t, x, \psi, u)$ by

$$H(t, x, \psi, u) = l(t, x(t), u(t)) + (f(t, x(t), u(t)), \psi(t))\tag{2.13}$$

where the costate vector $\psi(t) \in R^n$ is the solution of the adjoint system

$$\begin{aligned}\dot{\psi}(t) &= -f_x'(t, x(t), u(t))\psi(t) - l_x(t, x(t), u(t)), & t_1 \leq t < t_2 \\ \psi(t_2) &= 0\end{aligned}\tag{2.14}$$

Suppose there exists a function $\eta : R^n \times R^n \rightarrow U$ such that

$$M(x, \psi) \equiv \max_{u \in U} H(t, x, \psi, u) = H(t, x, \psi, \eta(x, \psi))$$

for all $(x, \psi) \in R^n \times R^n$.

Using the function η and the canonical equations $\dot{x} = H_\psi$, $\dot{\psi} = -H_x$, we can transform the original control problem (2.11)-(2.12) to obtain

$$\begin{aligned}\dot{x} &= f_1(x, \psi), & x(t_1) = x_0, & x(t_2) = x_1 \\ \dot{\psi} &= f_2(x, \psi)\end{aligned}\tag{2.15}$$

where

$$f_1(x, \psi) = f(x, \eta(x, \psi))$$

and

$$f_2(x, \psi) = -f_x'(x, \eta(x, \psi))\psi - l_x(x, \eta(x, \psi))$$

The system(2.15) consists of $2n$ nonlinear sets of differential equations with n initial and n terminal conditions. These problems are known as two-point boundary-value problems. Generally, these problems cannot be solved analytically and a numerical approach must be provided.

In the shooting method, a set of values of the unspecified conditions is assumed ($\psi(t_1) = \xi$), and the differential equations (2.15) are numerically solved to obtain $x = x(t, x_0, \xi)$ and $\psi = \psi(t, x_0, \xi)$. If the computed terminal values satisfy the specified terminal conditions, the problem is completely solved. If they do not, one has to introduce an error function

$$E(t_2, \xi) = \|x(t_2, x_0, \xi) - x_1\|^2 \quad (2.16)$$

to adjust the missing initial conditions.

The original control problem is then reduced to that of minimizing the error function. In order to minimize the error function one uses an iterative process based on the gradient technique. The computation of the gradient to update the missing initial conditions is extremely laborious and time consuming. In this case, the gradient vector $g_\xi \in R^n$ is given by

$$g_\xi = \lim_{\epsilon \rightarrow 0} (E(t_2, \xi + \epsilon) - E(t_2, \xi)) / \epsilon$$

The next iterative set of the missing conditions is taken as

$$\psi(t_2) = \xi + \alpha g_\xi$$

where $\alpha \in R$ is sufficiently small so that

$$E(t_2, \xi + \alpha g_\xi) \leq E(t_2, \xi)$$

The procedure of updating the missing initial conditions will be repeated until the error function cannot decrease anymore.

2.4.2 Gradient Method

Consider the control problem (2.11)-(2.12).

The gradient method consists of updating the control using the gradient vector, in a way successive iterates produce maximum reduction of the cost. The iterative scheme attempts to generate a sequence of controls u_i which eventually converge to a minimizing control u^* . Further, it involves the iterative rule

$$u_{i+1} = u_i + \alpha_i s_i \tag{2.17}$$

An initial control $u_0 \in R^m$ is chosen arbitrary. At each iteration a direction $s_i \in R^m$ is determined and a step size $\alpha_i \in R$ is chosen sufficiently small such that

$$J(u_i + \alpha_i s_i) \leq J(u_i + \lambda s_i) \text{ for all } \lambda$$

In the gradient method $s_i = -g_i$, where $g_i \in R^m$ is given by

$$g_i(t) = (\partial H / \partial u)(t, x_i(t), \psi_i(t); u_i(t)) \tag{2.18}$$

with H is the Hamiltonian given by (2.13).

The gradient method has the property that the computed solution tends to converge to a local minimum. Thus once a minimum is attained, it is necessary to change the initial guess and search for other minima in the control space.

The control sequence generated by the gradient method does converge to a local minimum even if the initial guess is poor. However, convergence becomes slower as the minima is approached. In computer application, convergence of the gradient method can be substantially improved by using conjugate gradient technique for updating the controls. In the conjugate gradient technique, s_i is taken as

$$s_i = -g_i + (\langle g_i, g_i \rangle / \langle g_{i-1}, g_{i-1} \rangle) s_{i-1} \quad \text{with } s_0 = -g_0$$

where $\langle a, b \rangle = a'b$ denotes the inner product in R^m , for $a, b \in R^m$.

2.5. ALGORITHM

Using the necessary conditions of optimality presented in section 3, we develop an iterative algorithm to compute the time optimal control. The algorithm is based on the conjugate gradient technique and it is presented as follows

Step 0 : Let τ be an arbitrary positive number (equations (2.6)-(2.7))

Step 1 : Guess $u_1 \in \mathcal{U}$ and set $n=1$.

Step 2 : Solve the initial value problem (2.6) with $u(t) = u_n(t)$ giving $x_n(t) = x(t, x_0, u_n)$.

Step 3 : Using the data x_n and u_n , solve the adjoint system

$$\dot{\psi}(t) = -f_x'(x_n(t), u_n(t))\psi(t)$$

$$\psi(\tau) = x_n(\tau) - x_d$$

backward in time to determine ψ_n .

Step 4 : Compute the gradient vector (equation 2.18)

Step 5 :

- (i) If $n = 1$ set $s_n(t) = -g_n(t)$
- (ii) If $n \neq 1$ set $s_n(t) = -g_n(t) + (\|g_n(t)\|^2 / \|g_{n-1}(t)\|^2) s_{n-1}(t)$

Step 6 :

Step 6.1: If $g_n(t) \neq 0$, then modify $u_n(t)$ to $u_{n+1}(t) = u_n(t) + \epsilon s_n(t)$ by choosing $\epsilon > 0$ sufficiently small so that for all t , $u_{n+1}(t) \in U$ and $J(\tau, u_{n+1}) \leq J(\tau, u_n)$.

- (i) If $u_{n+1}(t) \geq u_{\max}$, for some $t \in [0, \tau]$, then $u_{n+1} = u_{\max}$.
- (ii) If $u_{n+1} \leq u_{\min}$, for some $t \in [0, \tau]$, then $u_{n+1} = u_{\min}$.

(Where the control constraints are $u_{\min} \leq u(t) \leq u_{\max}$, for all t .)

- (iii) If $|J(\tau, u_{n+1}) - J(\tau, u_n)| \leq \delta$, for small $\delta > 0$, then set $\varphi(\tau) = J(\tau, u_n)$ and go to step 7, otherwise set $u_n = u_{n+1}$, $n = n + 1$, and go to step 2.

Step 6.2: If at the n^{th} stage $g_n(t) = 0$ for $t \in [0, \tau]$, then u_n is a local minimizing element of $J(\tau, u)$ and set $\varphi(\tau) = J(\tau, u_n)$.

Step 7 : Take $\Delta\tau > 0$, go to step 1 and compute $\varphi(\tau + \Delta\tau)$ and $\varphi(\tau - \Delta\tau)$ by solving (2.6) and (2.7) replacing τ by $\tau \pm \Delta\tau$.

- (i) If $\varphi(\tau - \Delta\tau) < \varphi(\tau)$, set $\tau = \tau - 2\Delta\tau$ and go to step 1.
- (ii) If $\varphi(\tau + \Delta\tau) < \varphi(\tau)$, set $\tau = \tau + 2\Delta\tau$ and go to step 1.

(iii) Stopping criterion : If $\varphi(\tau + \Delta\tau) > \varphi(\tau)$ and $\varphi(\tau - \Delta\tau) > \varphi(\tau)$ for $\Delta\tau$ sufficiently small, then $\varphi(\tau)$ is a local minimum and stop.

Remark 2.7

In step 6, the step size ϵ to update the control is determined using the Lagrange Extrapolation. For more details, the reader is referred to appendix A.

2.6 SUMMARY

In this chapter, we have proposed a simple algorithm and its mathematical justification for the computation of the time optimal control. The method is based on approximation of time optimal controls by a sequence of controls which are optimal for a suitable class of terminal control problems. These later problems are easily solved by conjugate gradient technique. The algorithm avoids the solution of a two-point boundary-value problem which is a difficult task. Our method is also able to determine if a system is controllable or not. Further, Pontryagin's minimum principle of optimality for a general control problem and a brief discussion of the gradient technique and shooting method have been provided. Three numerical examples illustrating the efficiency of the algorithm will be presented in the next chapter.

CHAPTER III

NUMERICAL SIMULATION OF TIME OPTIMAL CONTROL

3.1 INTRODUCTION

In chapter II, the time optimal control problem (in a general set up) has been considered and the corresponding necessary conditions of optimality has been presented. Using the conjugate gradient technique, we have developed an iterative scheme with the help of which time optimal controls can be computed. In this chapter, we consider the application of this algorithm to two kinds of robots and an ecological system. This chapter is organized as follows:

The applications of the algorithm to the time optimal control problem of Autom-elec ACR robot with cylindrical coordinates, and IBM articulated arm robot with two links are given in section 2 and 3, respectively. Dynamic models of the two robots were taken from [13]. The complete solution of the time optimal control of an ecological system is given in section 4. The ecological model has been developed in [25].

3.2 ROBOT WITH CYLINDRICAL COORDINATES

In order to maximize the productivity of the robot, the total assembly time must be minimized. In other words, both the layout of the component's stacks around the workpiece must be optimized and each of the steps of assembly must be executed in minimum time. The problem is equivalent to find a control in the set of admissible controls that minimizes both the motion of the robot to the component location and the transportation of that component to the location of assembly. However, the problem of time optimal motions of articulated robotic manipulators is computationally cumbersome, mainly because of their highly coupled and nonlinear dynamics.

The cylindrical robot has three degrees of freedom and its geometry is shown in Fig.3.1.

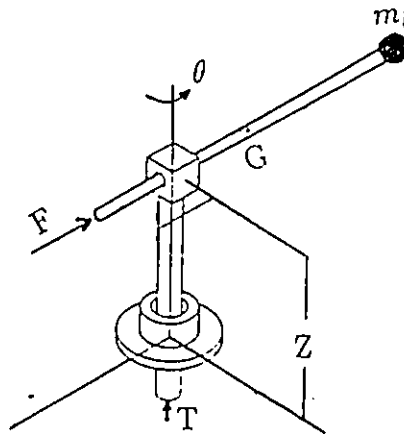


Fig.3.1. Sketch of the Cylindrical Robot.

The first degree of freedom is the azimuth rotation θ which is driven by a limited torque T . The second degree of freedom is the radial translation d which is driven by a limited force F . The third degree of freedom is the vertical translation Z . The remaining degrees of freedom are in the robot's hand. Their effects are neglected and the hand and the load are lumped into a point mass m_l .

Neglecting the friction, the cylindrical robot is governed by the following set of differential equations [13],

$$\begin{aligned}\dot{x}_1(t) &= x_2(t) \\ \dot{x}_2(t) &= \{u_1(t) + m_k x_1(t)x_4^2(t) + m_l[d_0 + x_1(t)]x_4^2(t)\}/(m_k + m_l) \\ \dot{x}_3(t) &= x_4(t) \\ \dot{x}_4(t) &= \{u_2(t) - 2m_k x_1(t)x_2(t)x_4(t) - 2m_l[d_0 + x_1(t)]x_2(t)x_4(t)\}/M[x_1(t)]\end{aligned}\tag{3.1}$$

with $x(t) \equiv [x_1(t), x_2(t), x_3(t), x_4(t)]' \in R^4$ and the control vector $u(t) \equiv [u_1(t), u_2(t)]' \in R^2$.

Where

x_1 = translation measured between the azimuth axis Z and the center of gravity G of the naked arm = d

x_2 = rate of the radial translation.

x_3 = azimuth rotation = θ

x_4 = angular velocity of the azimuth rotation.

u_1 = F

u_2 = T

m_k = mass of the naked arm.

m_l = mass of the hand and the load.

M_z = mass moment of inertia of the robot without the arm, hand and load, measured with respect to the azimuth axis Z.

M_c = mass moment of inertia of the naked arm without the load, measured with respect to its center of gravity G.

d_0 = distance between the mass m_l and G.

$$M[x_1] = M_z + M_c + m_k x_1^2 + m_l [d_0 + x_1]^2.$$

The control constraints are

$$|u_1(t)| \leq F_{\max}$$

$$|\dot{u}_2(t)| \leq T_{\max}$$

Let us find admissible controls u_1^* and u_2^* satisfying the control constraints which transfer the robot from the fixed initial state $[.15, 0, 0, 0]$ to the fixed final state $[.15, 0, \pi/2, 0]$ in minimum time.

For numerical simulation we use the following parameters as given in the above reference,

$$m_k = 3.7Kg, m_l = 4.6Kg, d_0 = .37m, M_z = .28m^2Kg,$$

$$M_c = .09m^2Kg, F_{\max} = 15N, T_{\max} = 5Nm.$$

Using the algorithm in chapter II, the optimal cost $\varphi(\tau)$, the optimal trajectories and controls $u_1^*(t)$ and $u_2^*(t)$ are computed as shown in Fig.3.2-3.4. The optimal time is $\tau^* = 1.16$ (sec) which coincides with the result of Geering et al [13]. In Fig.3.6, controls $u_1^*(t)$ and $u_2^*(t)$ are shown for different final times. The required time to compute the optimal time was about 44 minutes (in CPU time in AMBDAHL 5860).

The function $\varphi(\tau)$ is monotone decreasing and $\varphi(t)=0$ for all $t \geq \tau^*$. In this situation the attainable set $\mathcal{A}(t)$ is monotone noncontractive, that is,

$$\mathcal{A}(t_1) \subset \mathcal{A}(t_2) \text{ for } t_1 \leq t_2$$

The time optimal control is purely bang-bang for both of the control variables. Further, from Fig.3.2 and 3.5, it is clear that

$$\begin{aligned} u_1^*(t) &= F_{\max} \text{ for } \psi_2(t) < 0 \\ u_1^*(t) &= -F_{\max} \text{ for } \psi_2(t) > 0 \end{aligned} \tag{3.2}$$

and

$$\begin{aligned} u_2^*(t) &= T_{\max} \text{ for } \psi_4(t) < 0 \\ u_2^*(t) &= -T_{\max} \text{ for } \psi_4(t) > 0 \end{aligned} \tag{3.3}$$

The above results coincide with the theoretical ones, this is shown below.

The Hamiltonian function is given by

$$H = 1 + (f, \psi) = 1 + \psi_1(t)\dot{x}_1(t) + \psi_2(t)\dot{x}_2(t) + \psi_3(t)\dot{x}_3(t) + \psi_4(t)\dot{x}_4(t)$$

After $\dot{x}_i(t)$, $i=1, \dots, 4$ take the values given in the system equation (3.1), it follows that the Hamiltonian is affine in the controls and it has the form

$$H = g(x, \psi) + g_1(x, \psi)u_1 + g_2(x, \psi)u_2$$

where

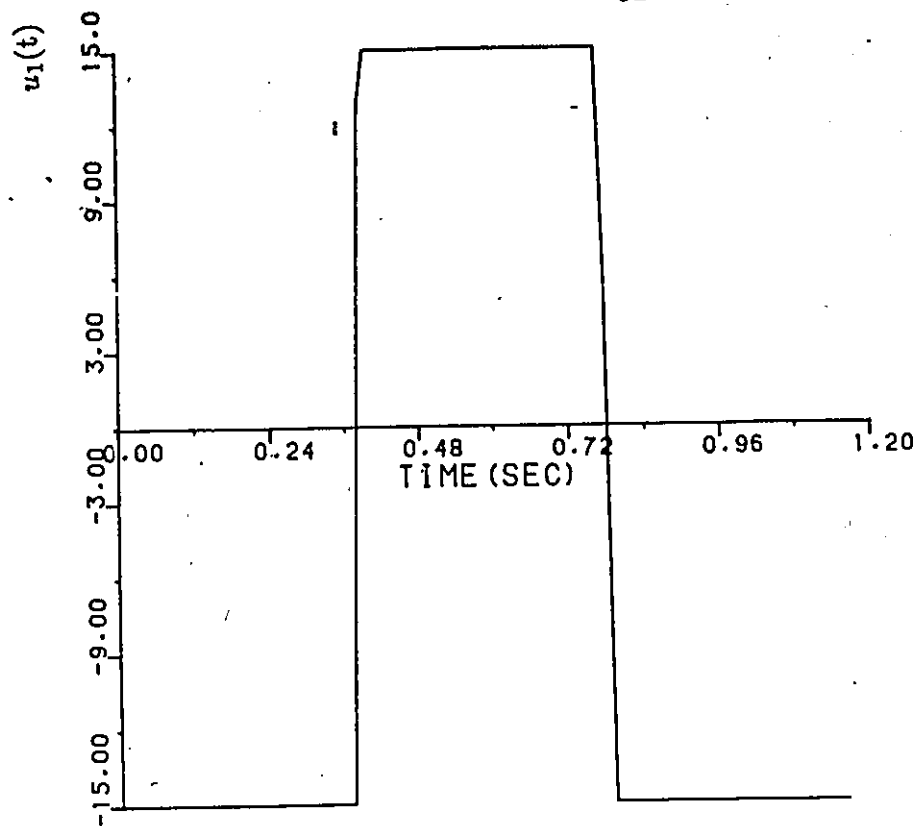
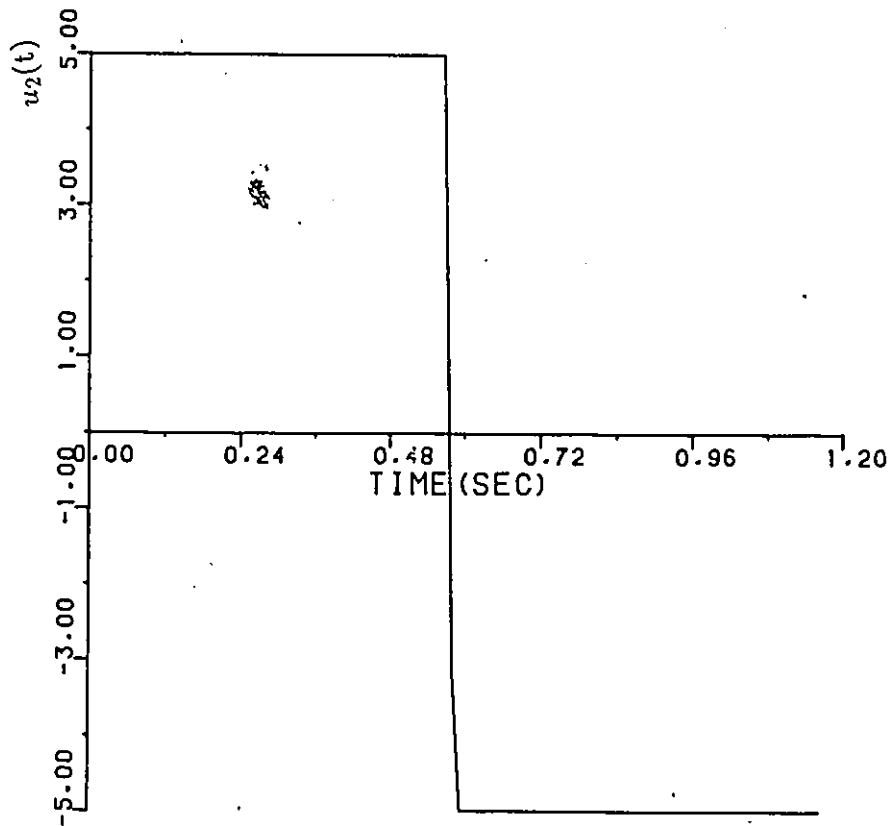
$$\begin{aligned} g_1(x, \psi) &= \psi_2(t)/(m_k + m_l) \\ g_2(x, \psi) &= \psi_4(t)/M[x_1] \end{aligned} \tag{3.4}$$

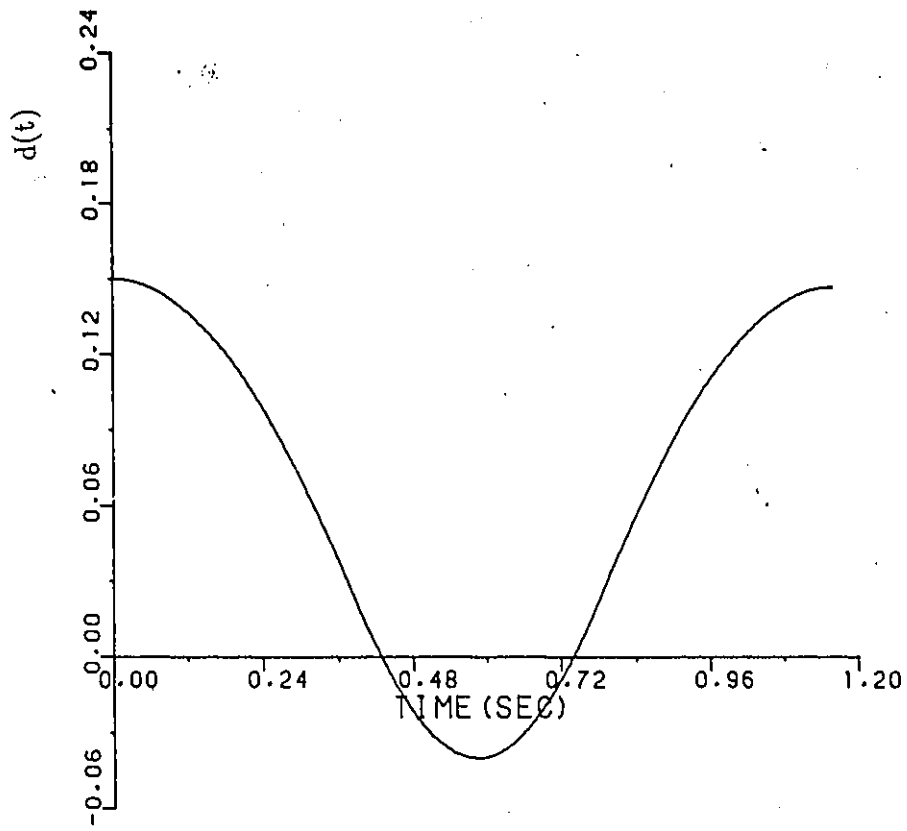
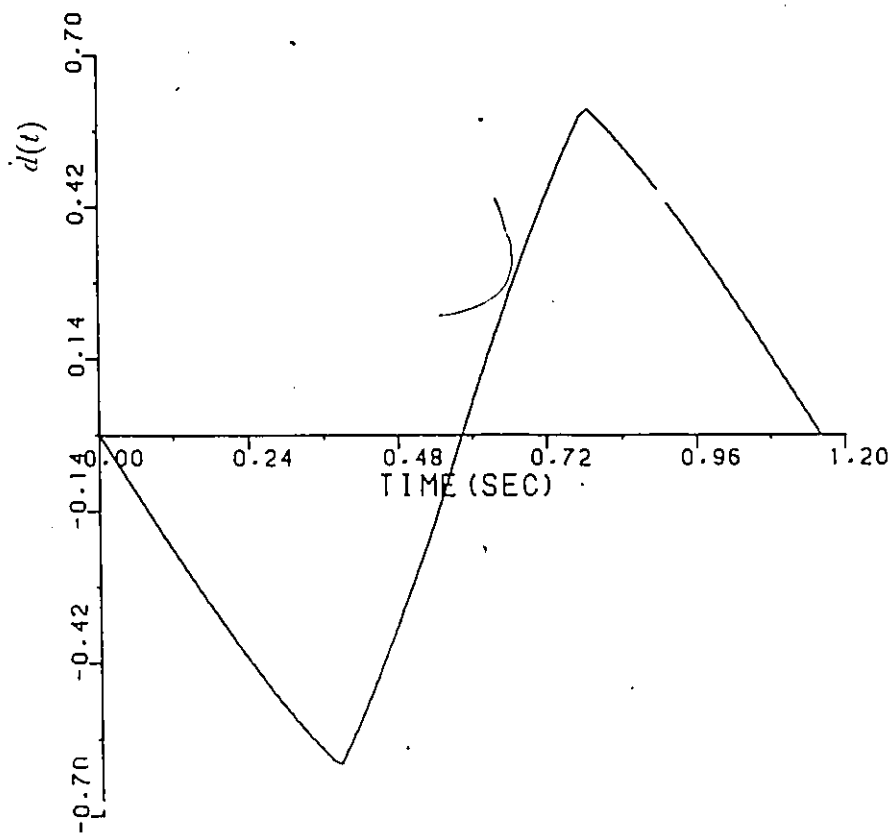
and $g(x, \psi)$ collects the remaining terms.

The optimal controls are obtained by globally minimizing the Hamiltonian with respect to the controls u_1 and u_2 . Hence, the optimal controls are essentially determined by the sign of the switching functions g_1 and g_2 . Since both of the denominators in (3.4) are positive, the sign of $\psi_2(t)$ and $\psi_4(t)$ are relevant, respectively.

Remark 3.1

For a given set of masses m_i , one has to compute the control policies for each mass separately and store them. Each time the robot load changes one has to recall the corresponding policy.

Fig.3.2.a. Optimal Control u_1^* .Fig.3.2.b. Optimal Control u_2^* .

Fig.3.3.a. Optimal State d .Fig.3.3.b. Optimal State \dot{d} .

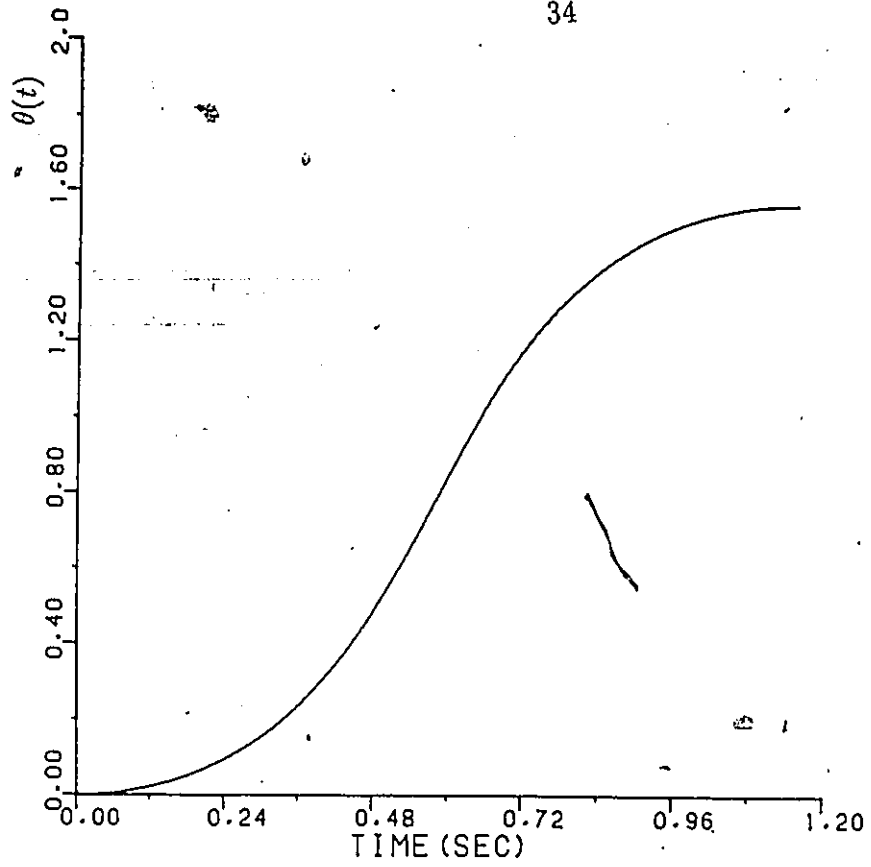


Fig.3.3.c. Optimal State θ .

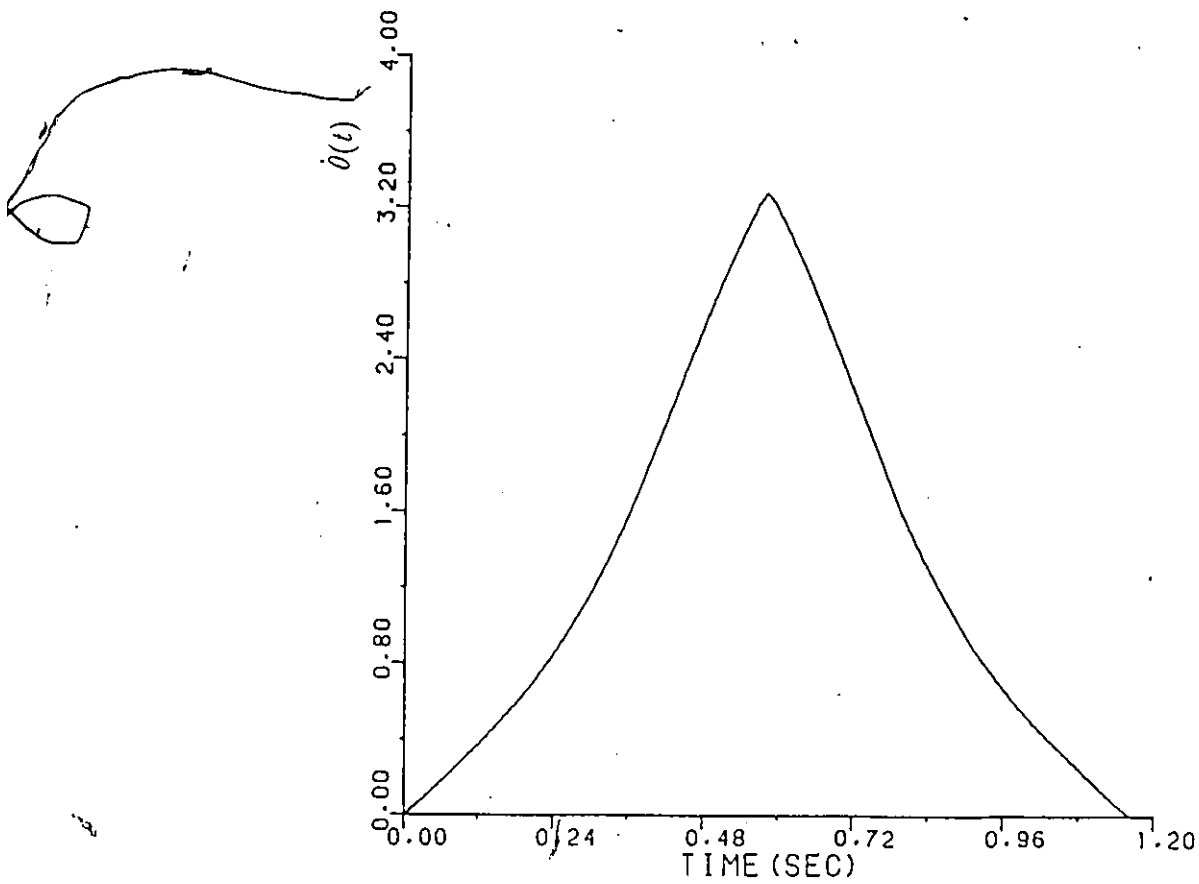


Fig.3.3.d. Optimal State $\dot{\theta}$

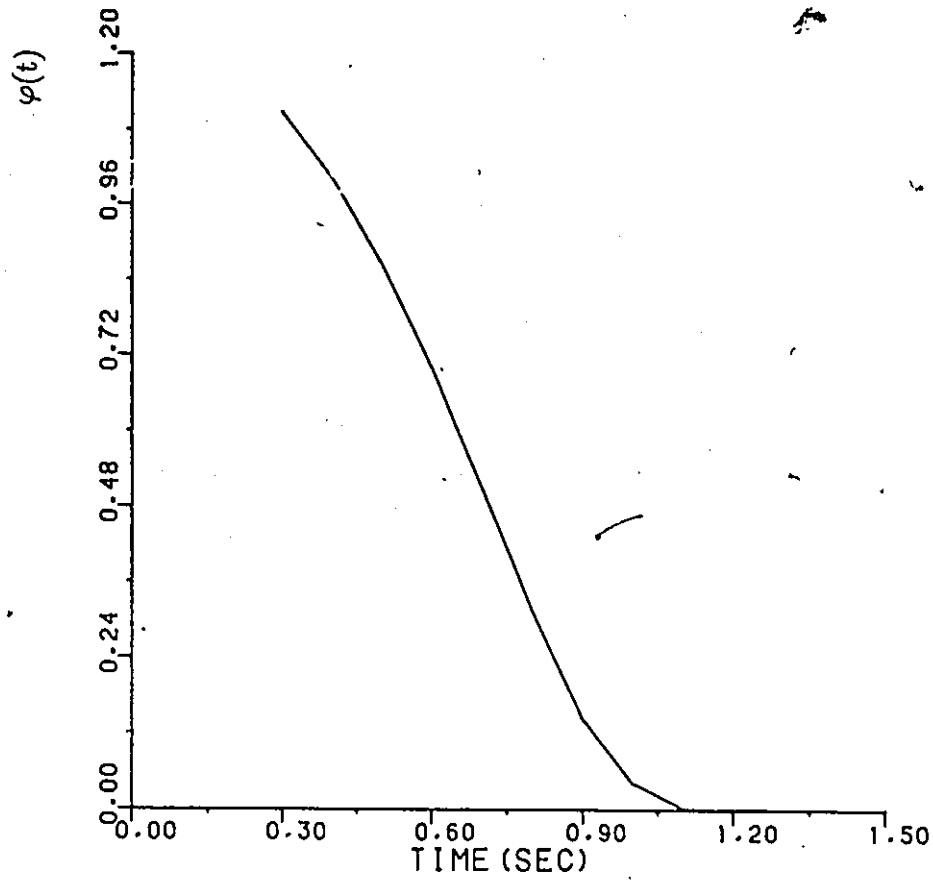
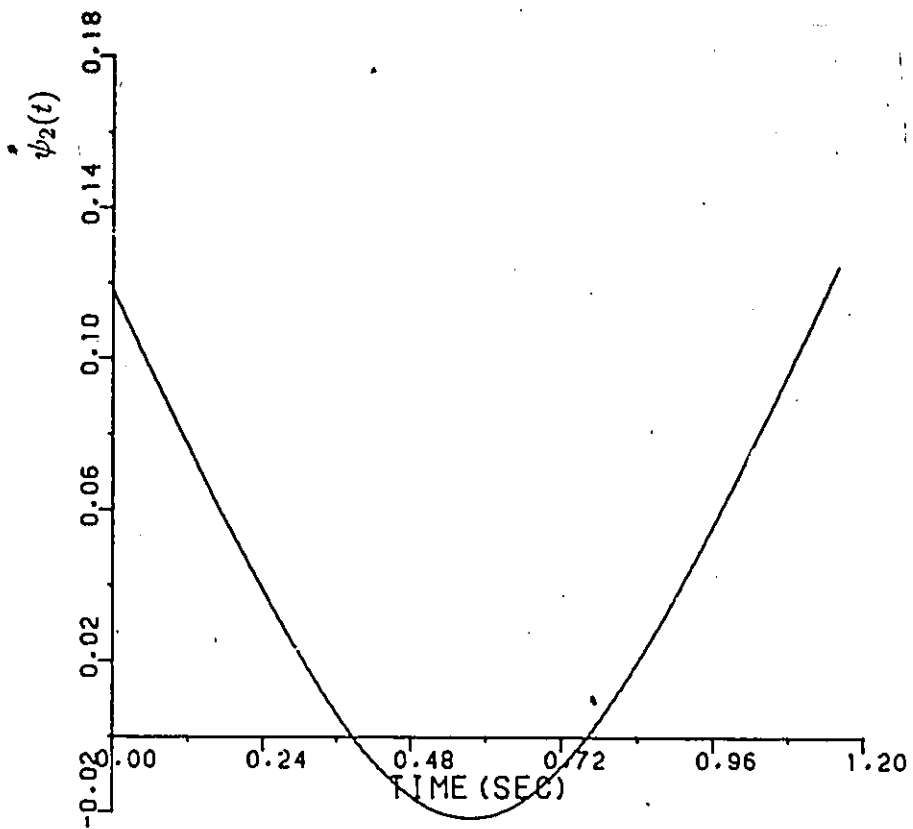
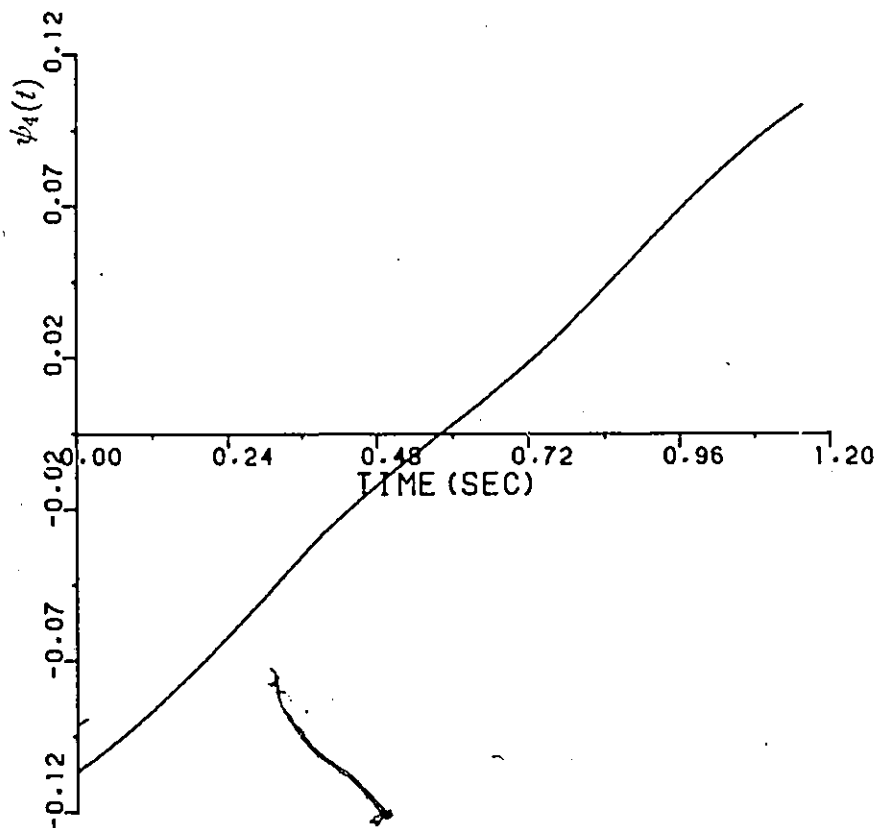


Fig.3.4. The cost Function $\varphi(t)$.

Iter.	cost
0	$.903 \times 10^1$
10	$.113 \times 10^{-1}$
20	$.901 \times 10^{-2}$
30	$.873 \times 10^{-2}$
60	$.511 \times 10^{-3}$
100	$.299 \times 10^{-5}$
150	$.372 \times 10^{-8}$

Table-3.1. Cost as a function of Number of Iterations.

Fig.3.5.a. Optimal Costate ψ_2^* .Fig.3.5.b. Optimal Costate ψ_4^* .

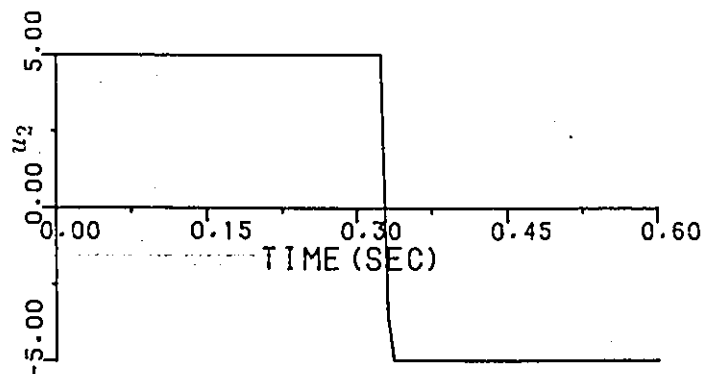
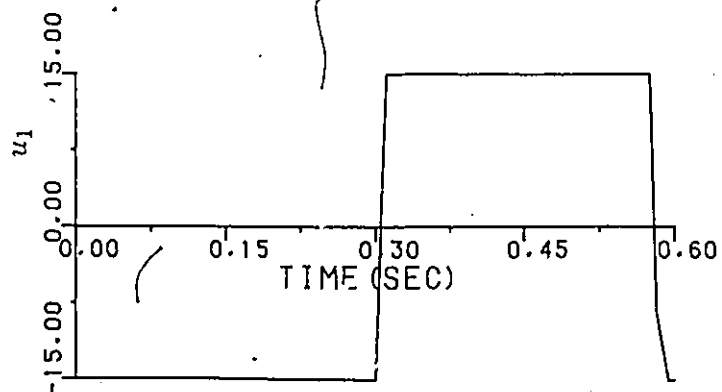
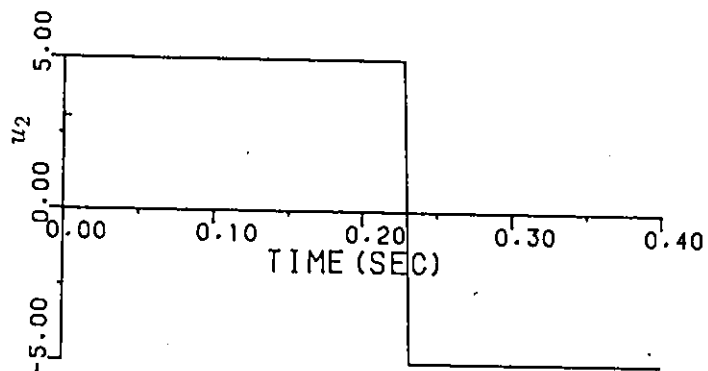
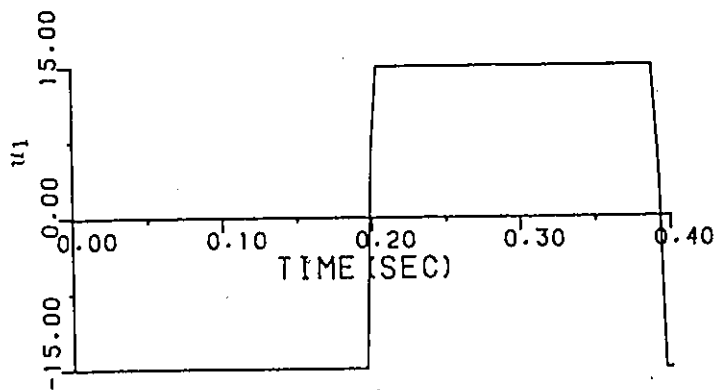
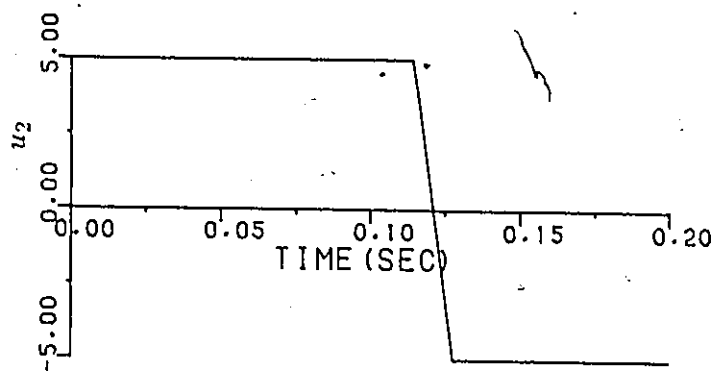
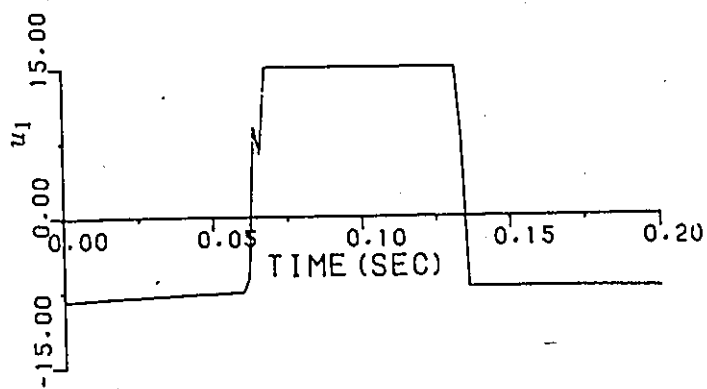
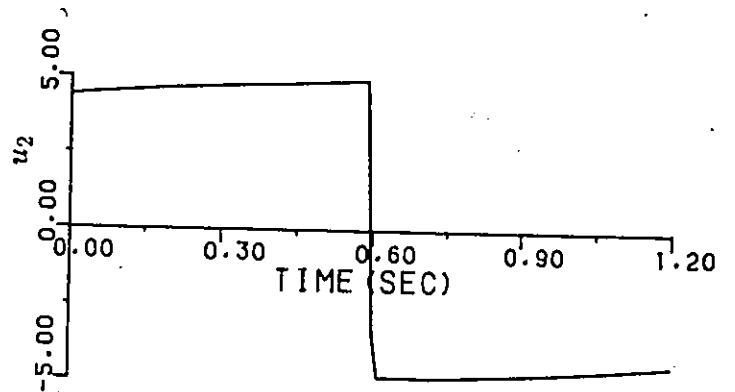
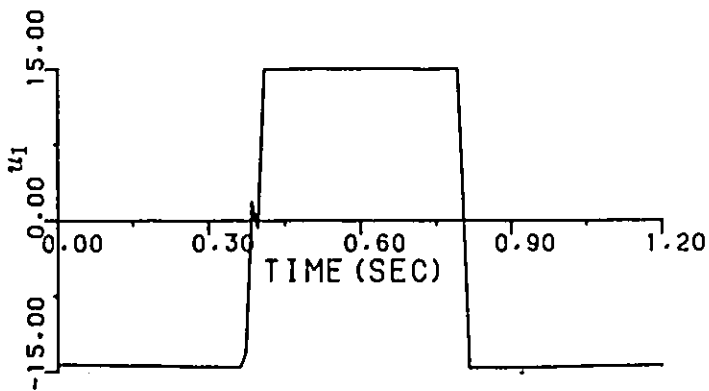
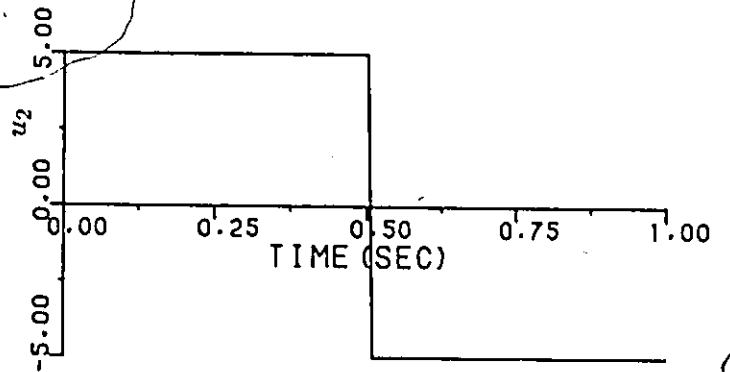
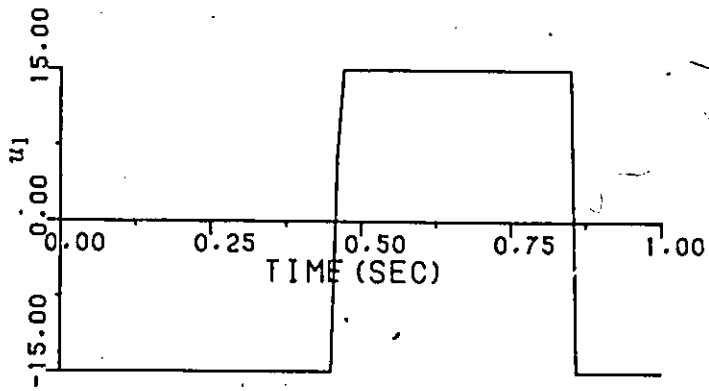
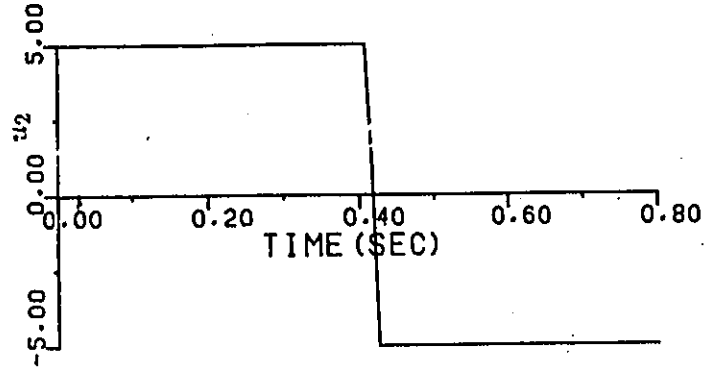
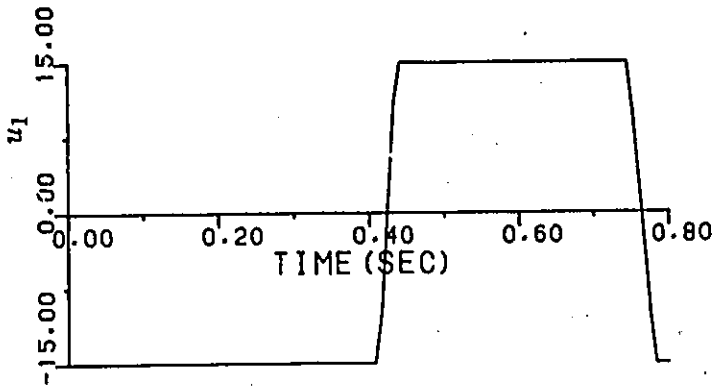


Fig.3.6. Optimal Controls u_1^* and u_2^* at Different Final Times not optimal time



3.3 HORIZONTAL ARTICULATED ARM ROBOT WITH TWO LINKS

This robot can be regarded as the simplest nontrivial manipulator. It has two degree of freedom and its geometry is shown in Fig.3.7. The first degree of freedom is the azimuth rotation θ_1 of the inner link which is driven by a limited torque T_1 . The second degree of freedom is the angular rotation θ_2 of the outer link which is driven by a limited torque T_2 . The remaining degrees of freedom used for positioning the endeffector are neglected.

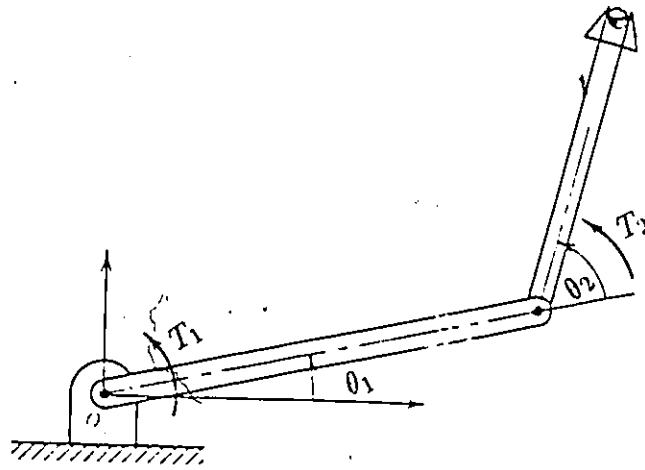


Fig.3.7. Sketch of the Articulated Arm Robot with Two Links.

Neglecting the friction, the articulated arm robot is governed the following set of equations [13],

$$\begin{aligned}
 \dot{x}_1 &= x_2 \\
 \dot{x}_2 &= [M_4\{u_1 - u_2 + M_3(x_2 + x_4)^2 \sin(x_3)\} - M_3 \\
 &\quad \cdot \{u_2 - M_3 x_2^2 \sin(x_3)\} \cos(x_3)] / [M_4 M_2 - M_3^2 \cos^2(x_3)] \\
 \dot{x}_3 &= x_4 \\
 \dot{x}_4 &= [(M_2 + M_3 \cos(x_3))\{u_2 - M_3 x_2^2 \sin(x_3)\} - (M_4 + M_3 \cos(x_3)) \\
 &\quad \cdot \{u_1 - u_2 + M_3(x_2 + x_4)^2 \sin(x_3)\}] / [M_4 M_2 - M_3^2 \cos^2(x_3)]
 \end{aligned} \tag{3.5}$$

with the state vector $x(t) = [x_1(t), x_2(t), x_3(t), x_4(t)]' \in R^4$ and the control vector $u(t) = [u_1(t), u_2(t)]' \in R^2$.

where

x_1 = angular rotation of the inner link.

x_2 = rate of change of x_1 .

x_3 = angular rotation of the outer link.

x_4 = rate of change of x_3 .

$u_1 = T_1$

$u_2 = T_2$

M_f = mass moment of inertia of the first link w.r.t. the first axis.

M_s = mass moment of inertia of the second link w.r.t. the second axis.

M_h = mass moment of inertia of the hand and load w.r.t. the position of the hand.

$M_1 = M_s + L_2^2 m_h$

$M_2 = M_f + L_1^2 (m_s + m_h)$

$M_3 = L_1 (d_s m_s + L_2 m_h)$



$$M_4 = M_1 + M_h$$

L_1 = length of the first link.

L_2 = length of the second link.

m_s = mass of the second link.

m_h = mass of the hand and the load.

d_s = distance between the center of gravity of the second link and its driving axis.

The control constraints are

$$|u_1(t)| \leq T_{1,\max}$$

$$|u_2(t)| \leq T_{2,\max}$$

The structure of the time-optimal solution will be discussed for the special case where the robot's arm is stretched both in the initial and in the final position, i.e., for the boundary conditions

$$x_1(0) = x_2(0) = x_3(0) = x_4(0) = 0$$

$$x_1(\tau) = \theta_1 > 0$$

$$x_2(\tau) = 0$$

$$x_3(\tau) = \theta_2 = 0, \text{ or } \pm 2\pi, \dots$$

and

$$x_4(\tau) = 0$$

Obviously, for $\theta_1 \neq 0$ the solution is symmetric.

For numerical simulation we use the following parameters as given in the above reference,

$$L_1 = .4\text{m}, L_2 = .25\text{m}, d_s = .125\text{m}, m_s = 15\text{Kg}, m_h = 6\text{Kg}, M_f = 1.6\text{m}^2\text{Kg},$$

$$M_s = .43\text{m}^2\text{Kg}, M_h = .01\text{m}^2\text{Kg}, T_{1,\max} = 25\text{Nm}, T_{2,\max} = 9\text{Nm}.$$

a. $\theta_1 = 2.0$ and $\theta_2 = -2\pi$

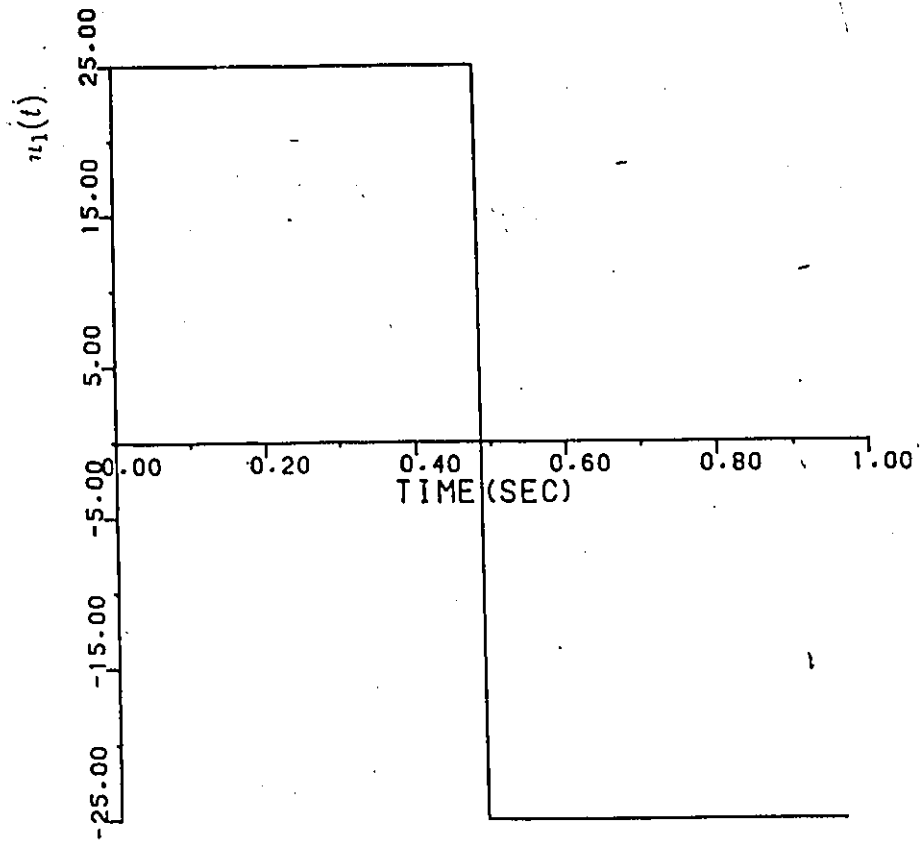
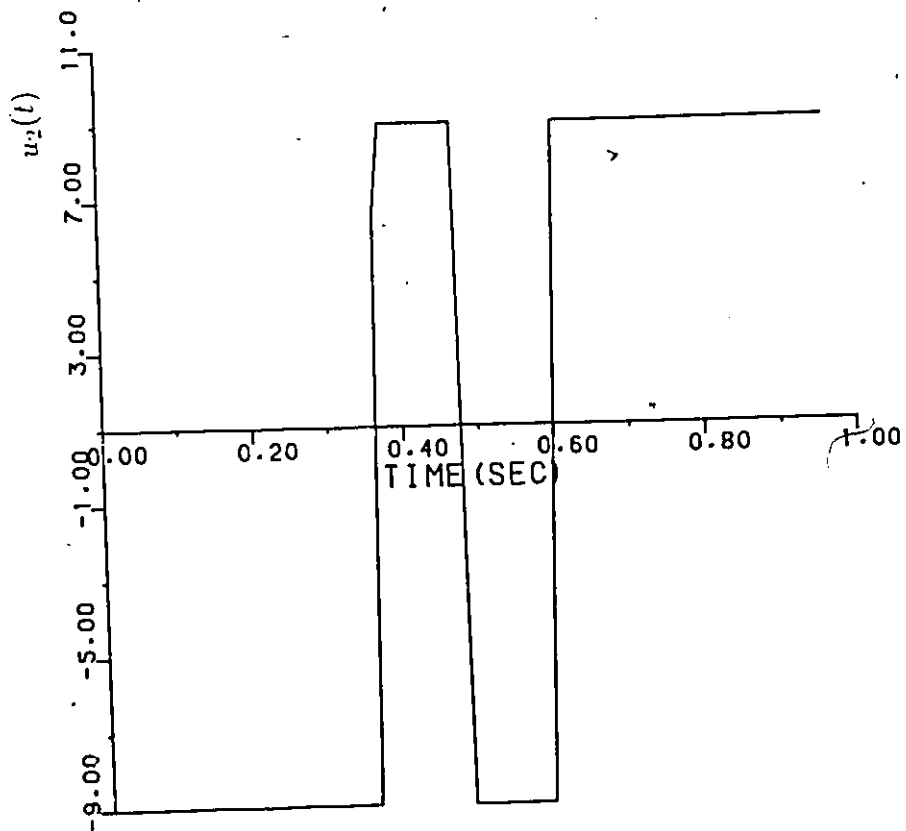
The function $\varphi(\tau)$, the optimal trajectories and controls $u_1^*(t)$ and $u_2^*(t)$ are computed using the algorithm in chapter II as shown in Fig.3.8-3.10. The time optimal control is bang-bang for both of the control variables. Mathematically, the analysis proceeds along the lines of example 1. The Hamiltonian is again affine in the controls u_1 and u_2 are nonzero over some interval. The optimal time is $\tau^* = .964$ (sec) which is 25 % better than that computed in [13]. In Fig.3.11, controls $u_1^*(t)$ and $u_2^*(t)$ are shown for different final times. The function $\varphi(\tau)$ is monotone decreasing and $\varphi(t) = 0$ for all $t \geq \tau^*$.

b. $\theta_1 = .4$ and $\theta_2 = 0$

The function $\varphi(\tau)$ is monotone decreasing as shown in Fig.3.14. The optimal controls and trajectories are computed as shown in Fig.3.12-3.13. The time optimal control is bang-bang for both of the control variables. The optimal time is $\tau^* = .759$ (sec) which is 30% better than that computed in [13]. In table (3.2), a summary of the cost at the optimal time is given as a function of the number of iterations.

c. $\theta_1 = 1.2$ and $\theta_2 = -2\pi$

Using the algorithm in chapter II, the optimal controls and trajectories are computed as shown in Fig.3.15-3.16. The time optimal control is purely bang-bang for both of the control variables. The optimal time is $\tau^* = .925$ (sec) which is better than that computed in [13]. The function $\varphi(\tau)$ is monotone decreasing and $\varphi(t) = 0$ for all $t \geq \tau^*$.

Fig.3.8.a. Optimal Control u_1^* .Fig.3.8.b. Optimal Control u_2^* .

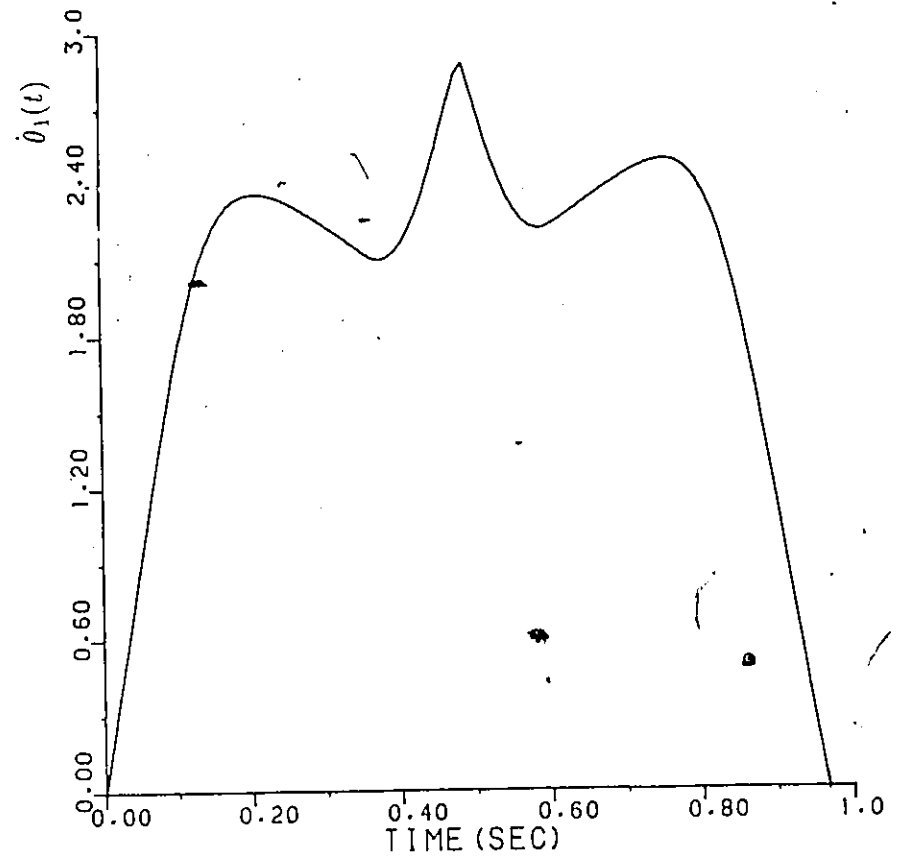
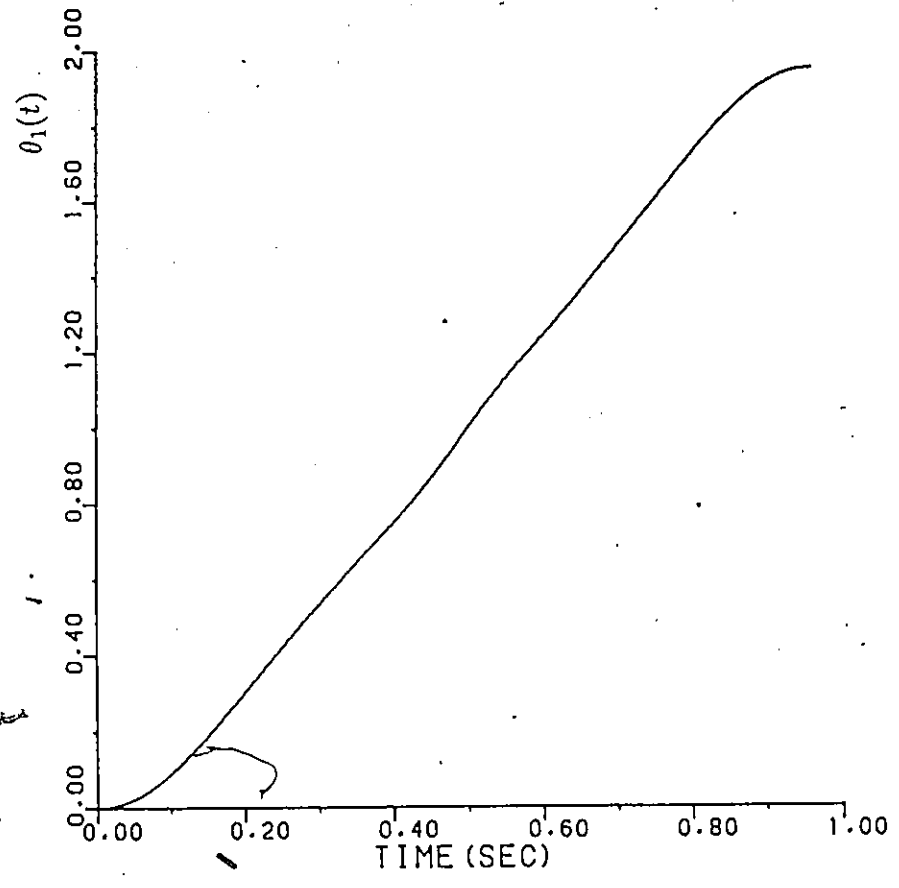
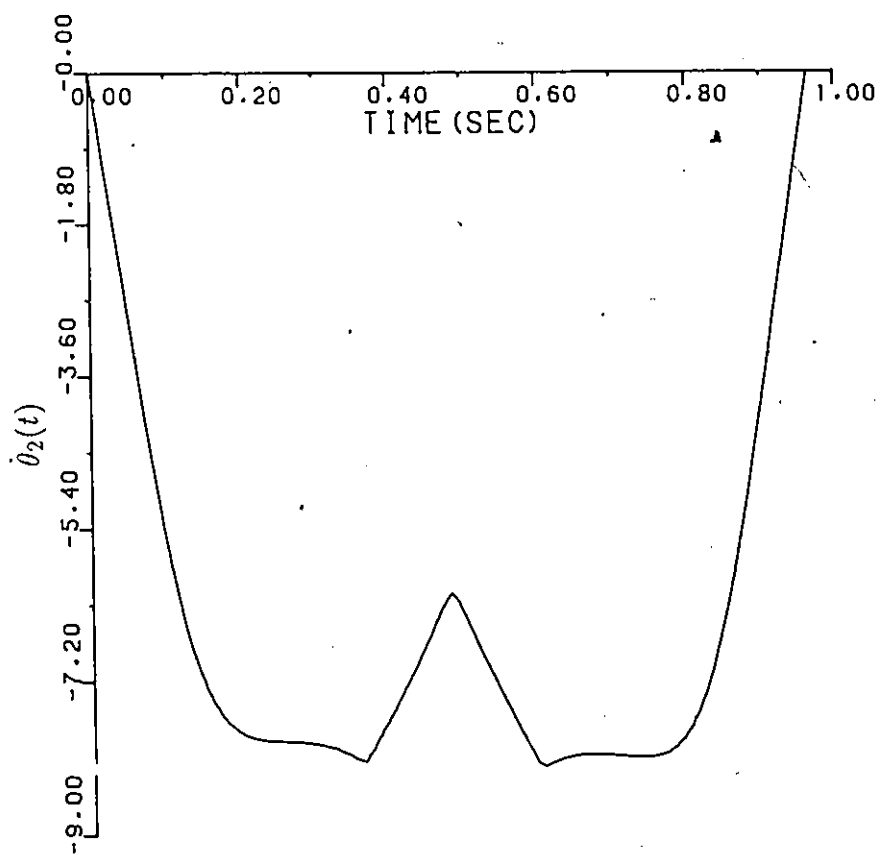
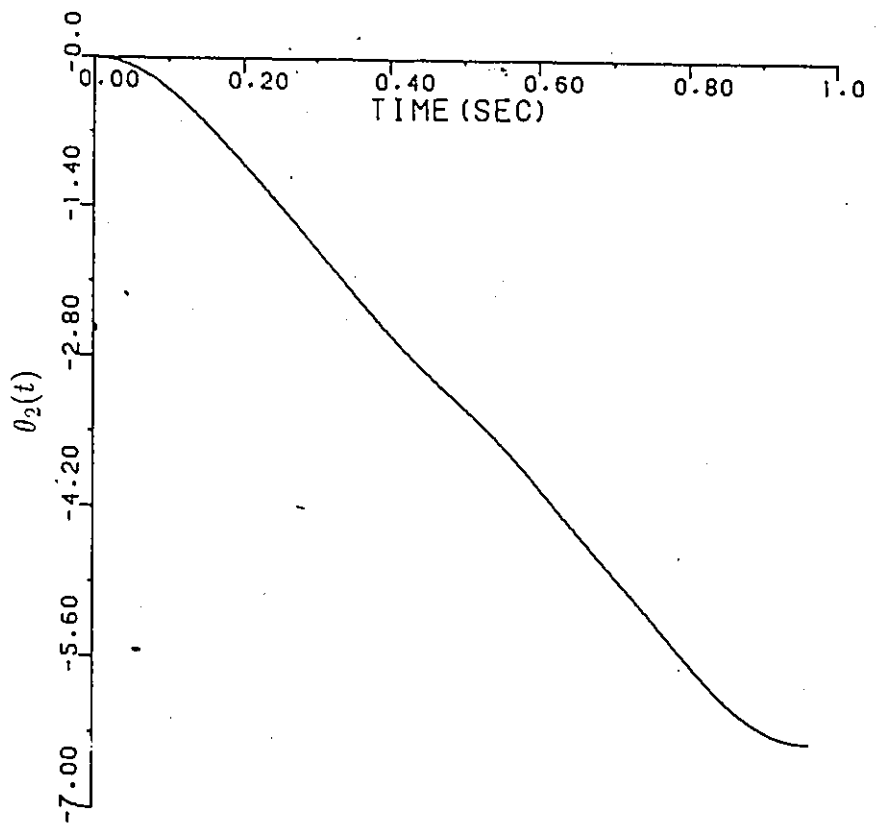
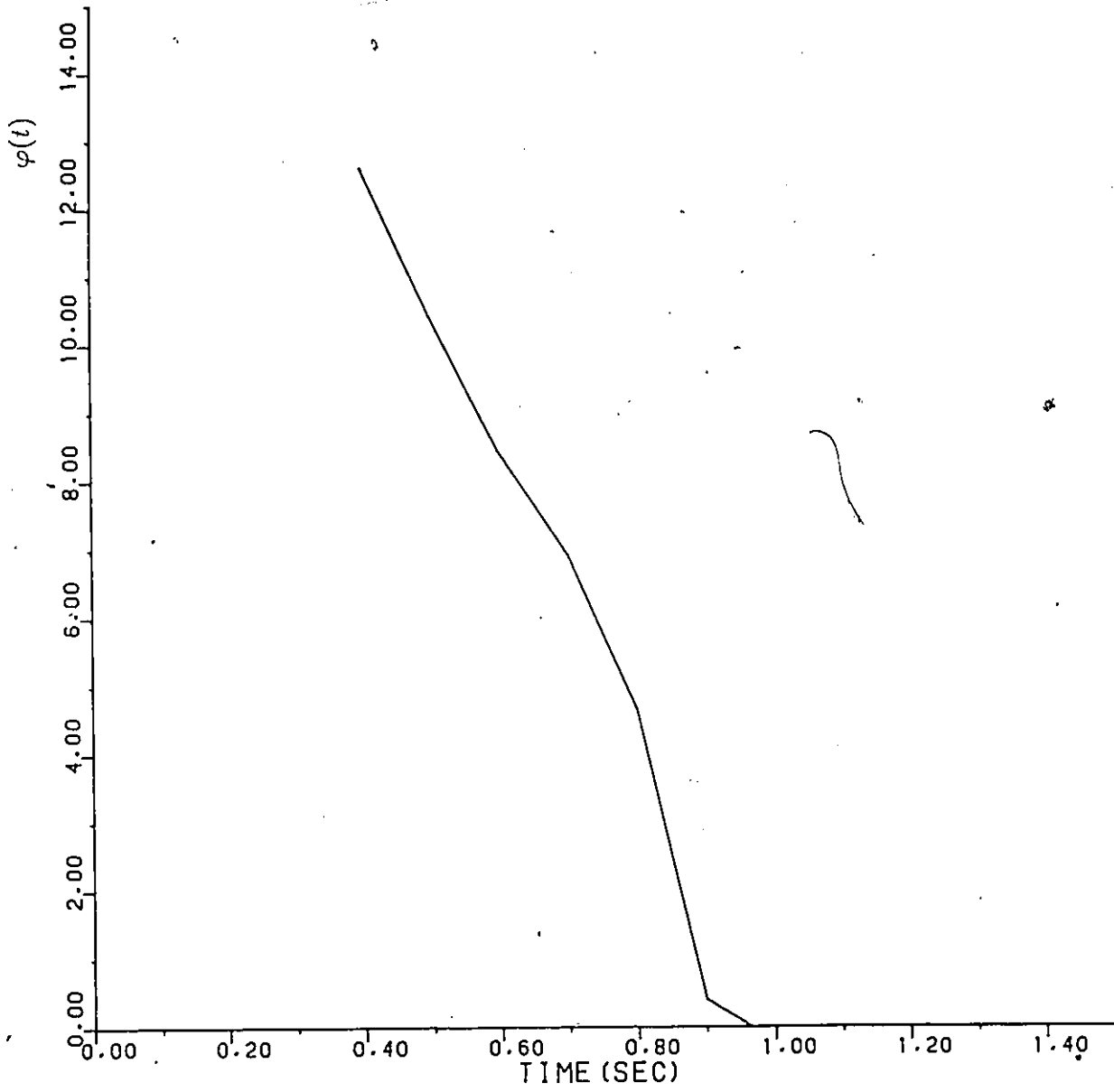


Fig.3.9.a. Optimal States θ_1 and $\dot{\theta}_1$

Fig.3.9.b. Optimal States θ_2 and $\dot{\theta}_2$

Fig.3.10. The Cost Function φ

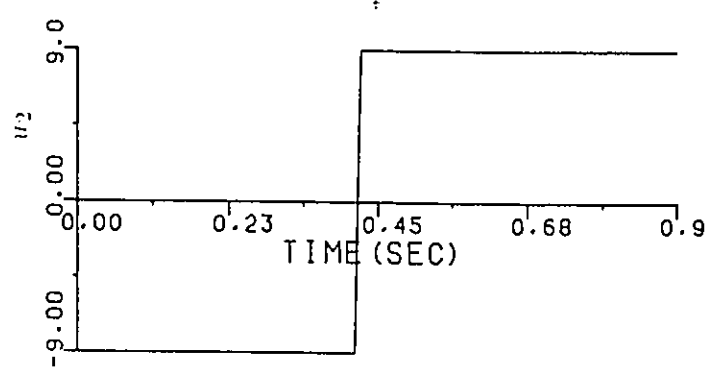
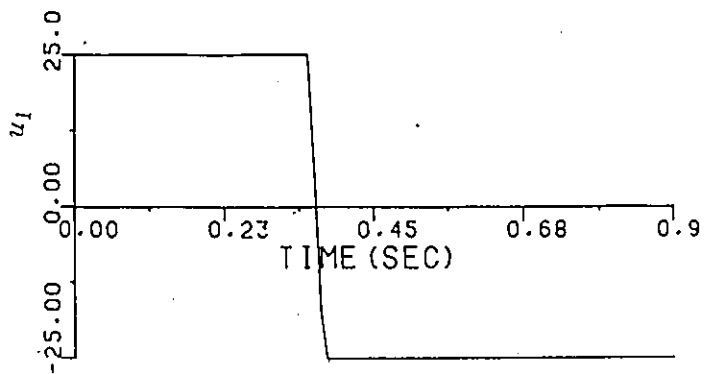
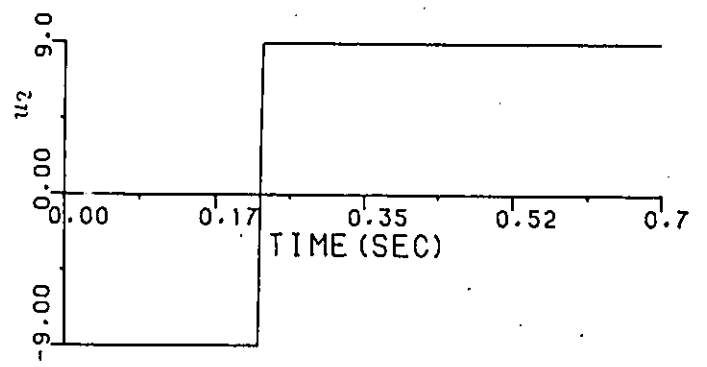
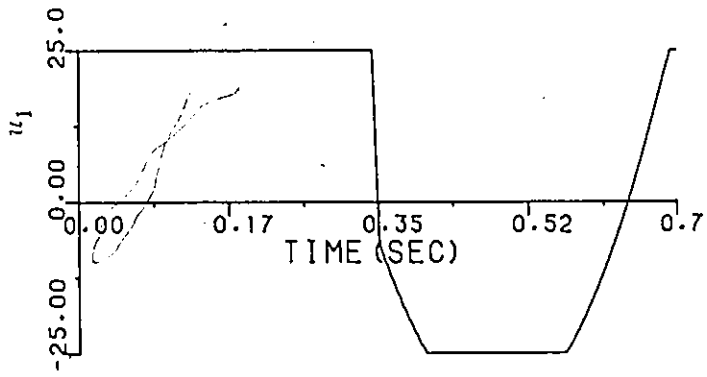
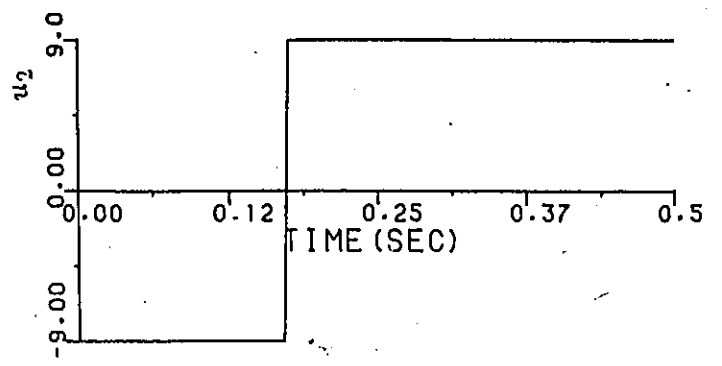
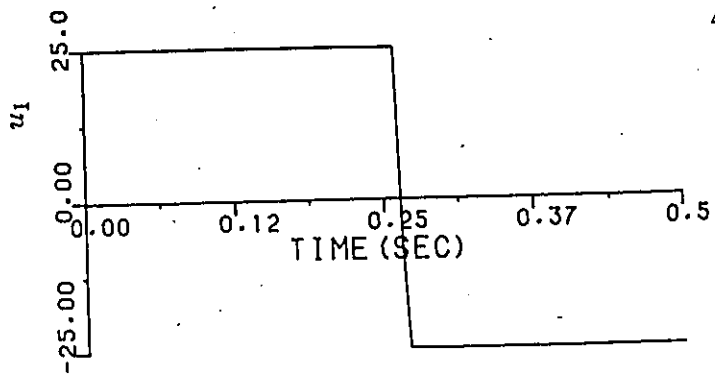
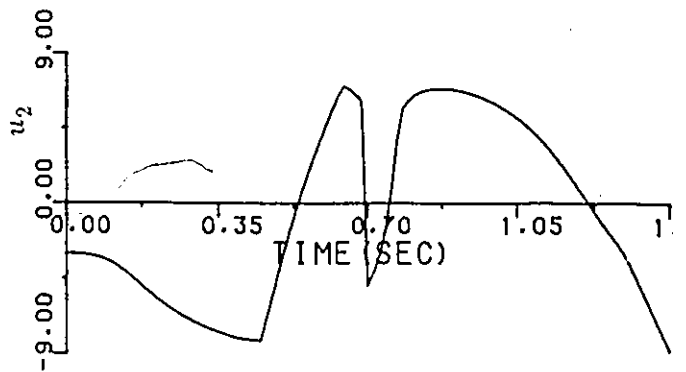
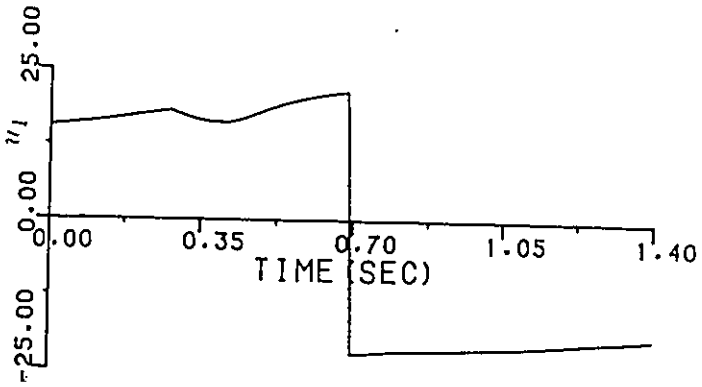
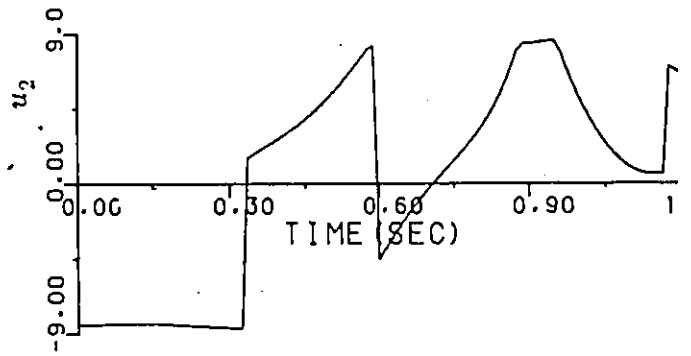
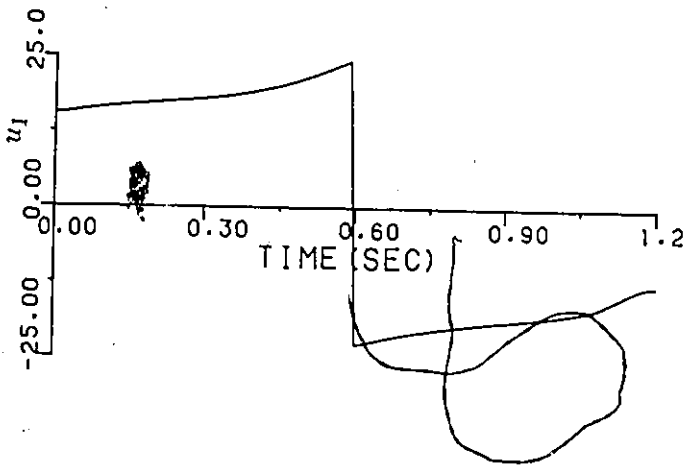
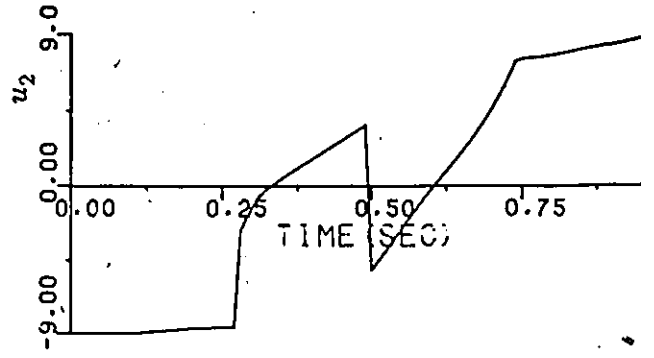
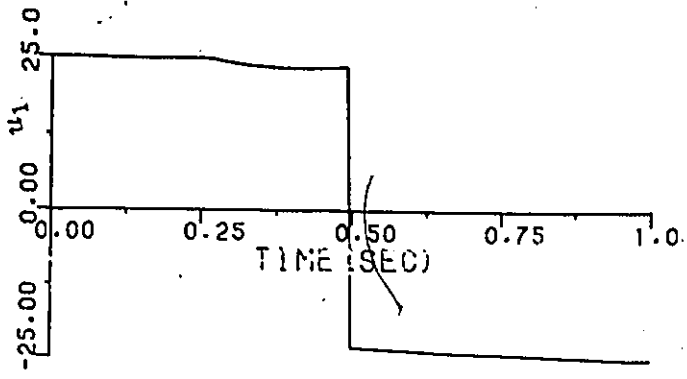


Fig.3.11. Optimal Control u_1^* and u_2^* at Different Final Times not optimal time



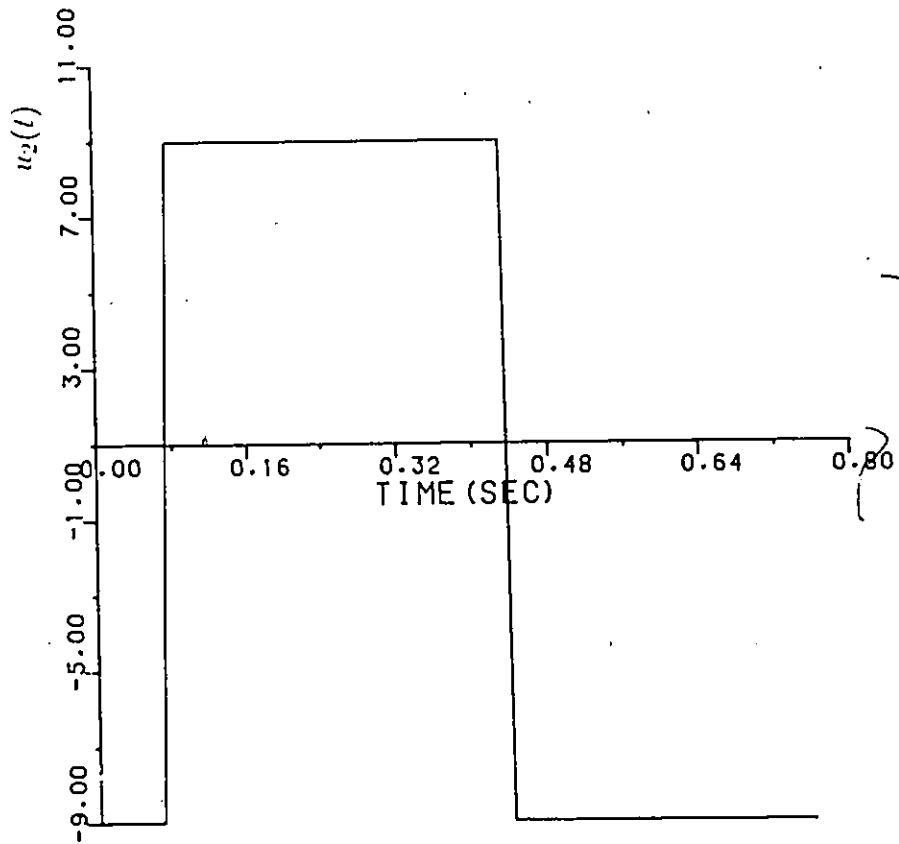
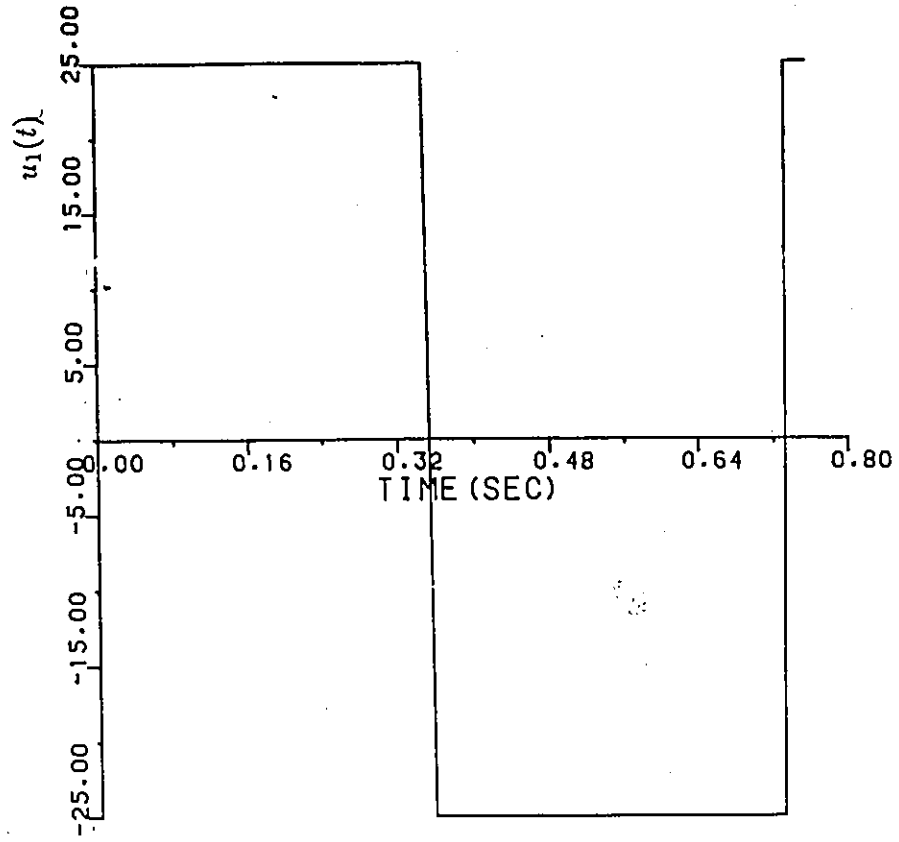
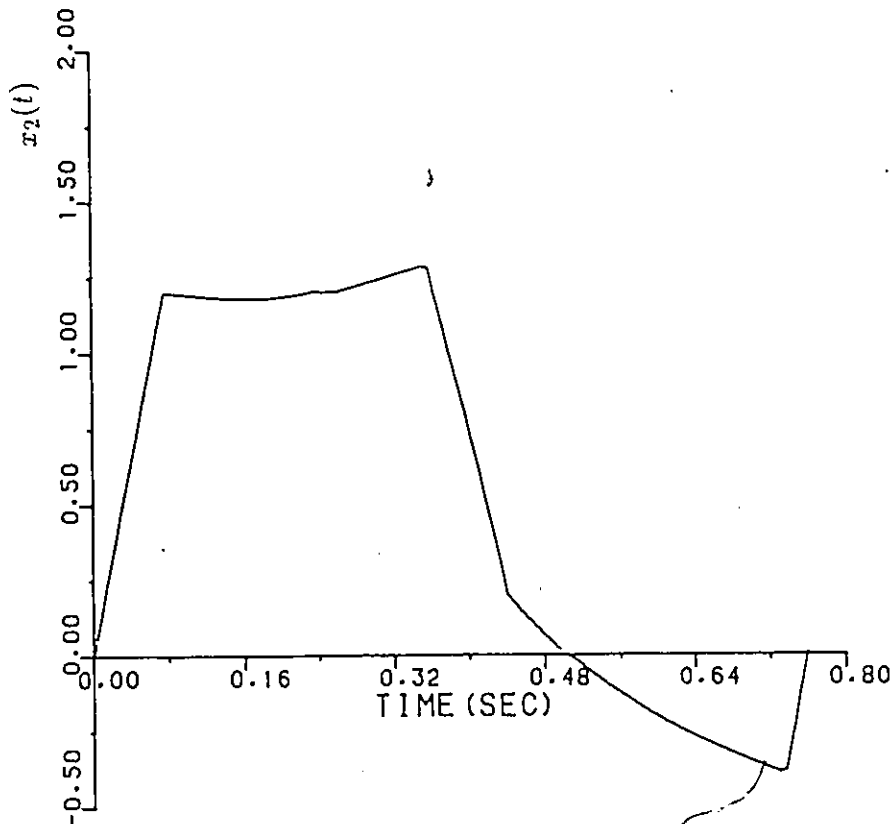
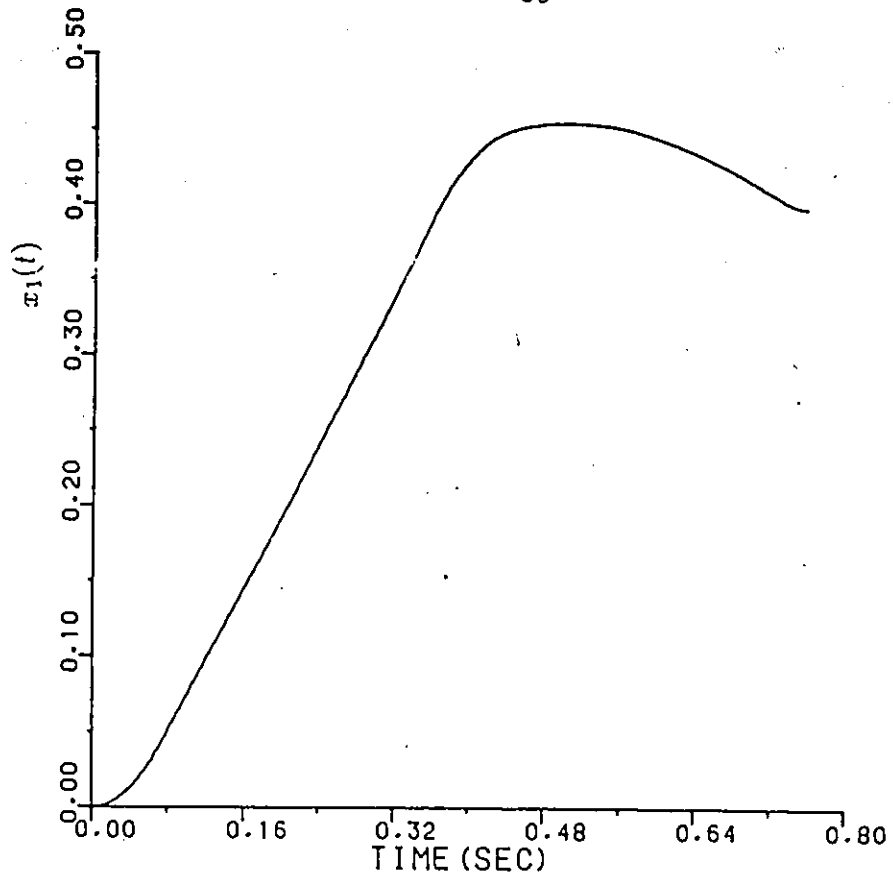


Fig.3.12. Optimal Controls u_1^* and u_2^* .

Fig.3.13.a. Optimal States x_1^* and x_2^* .

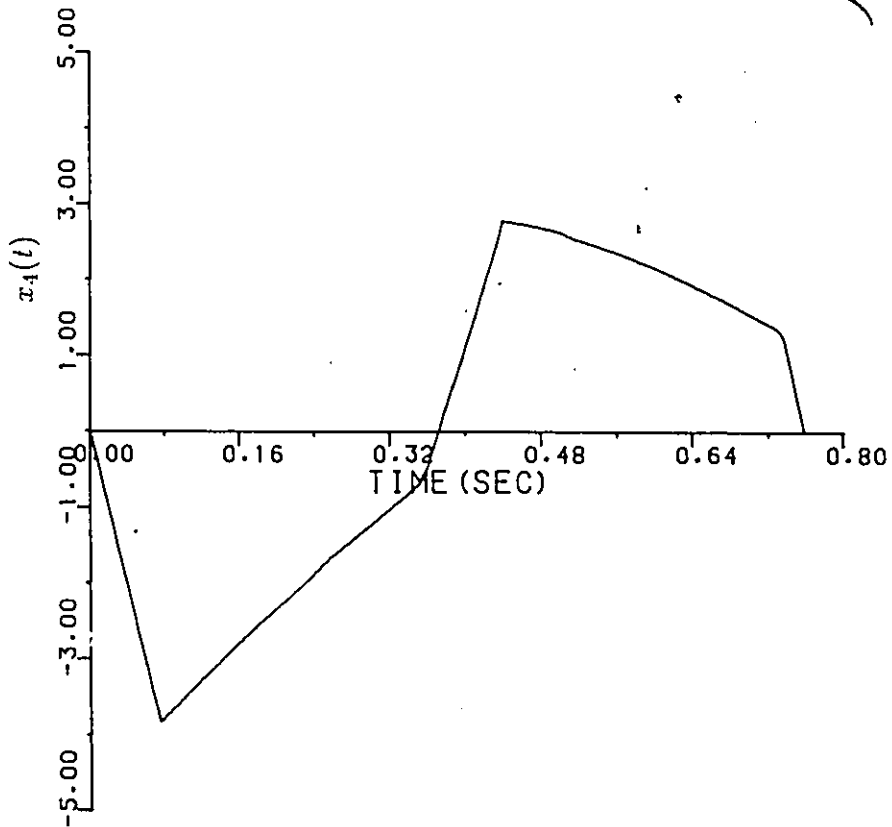
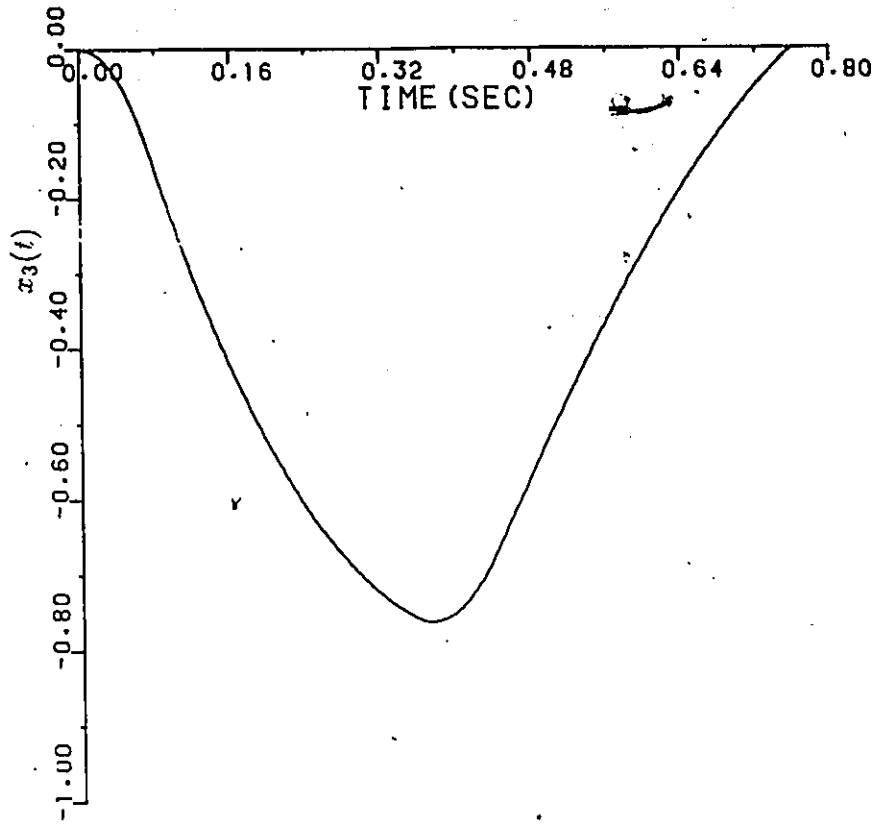
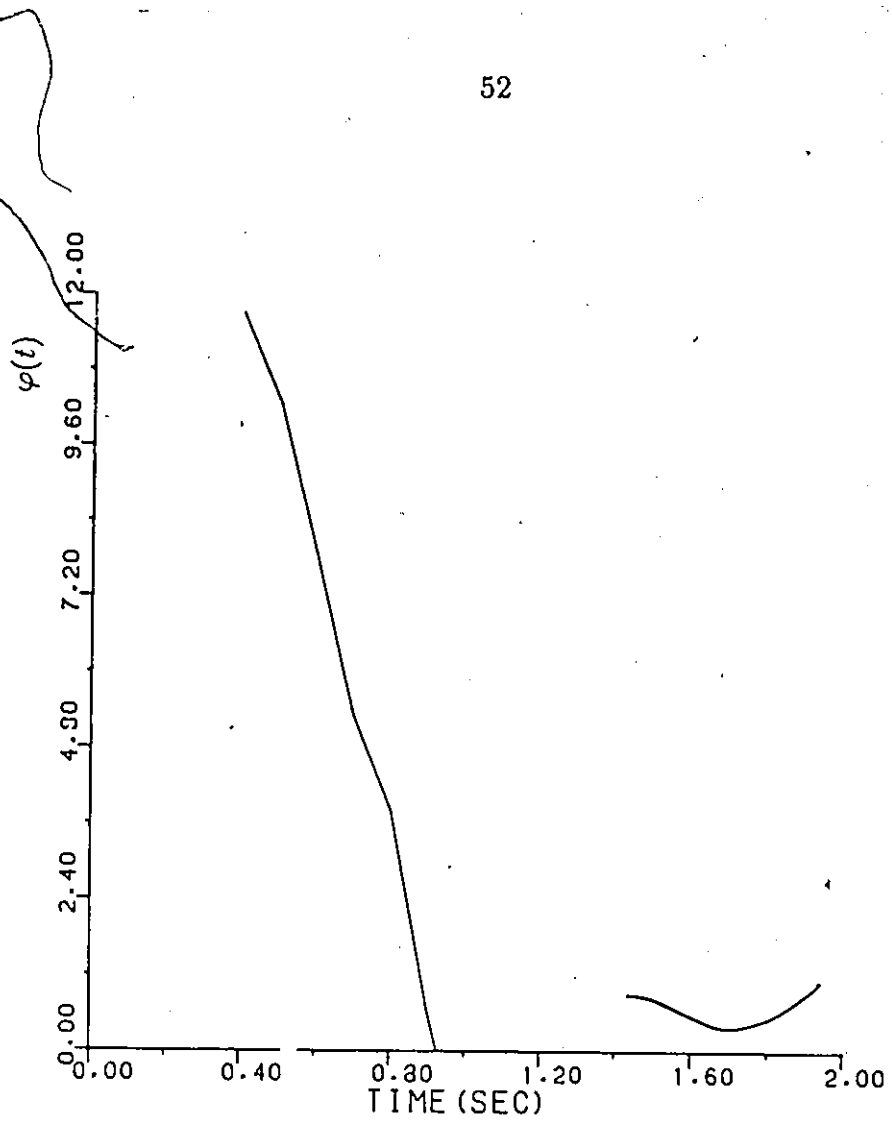
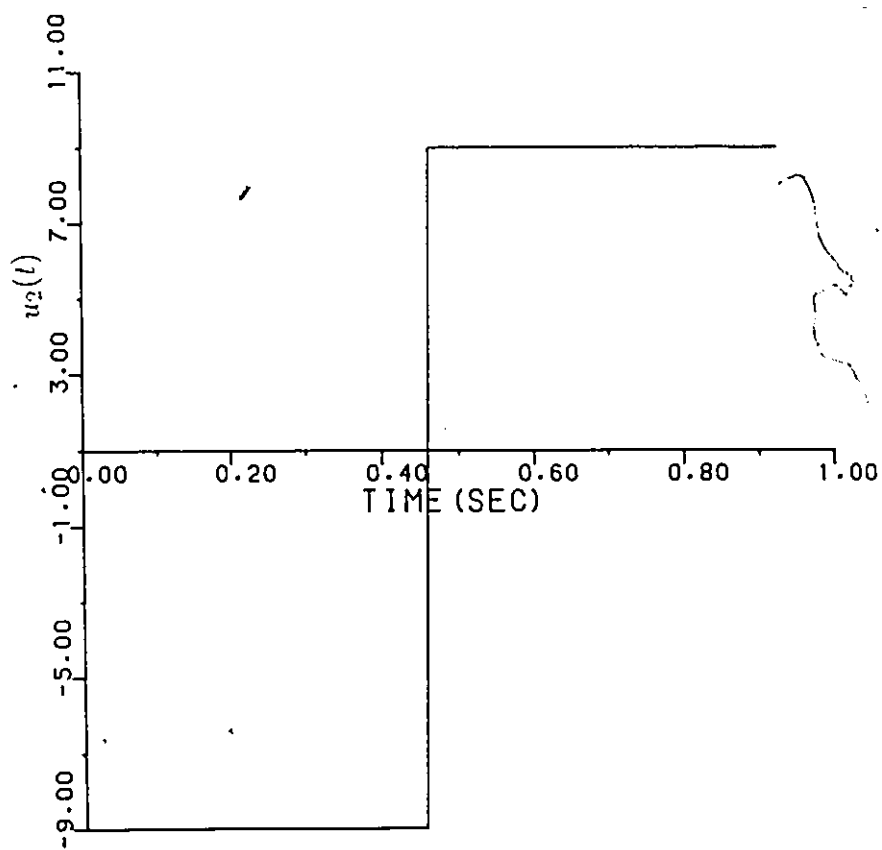
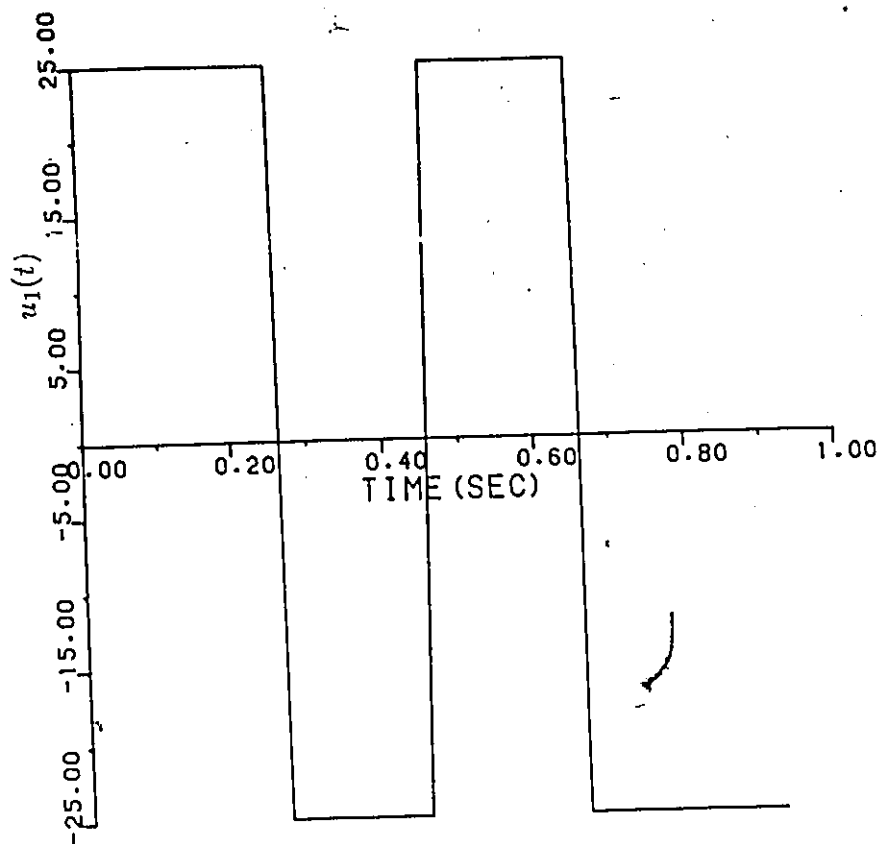


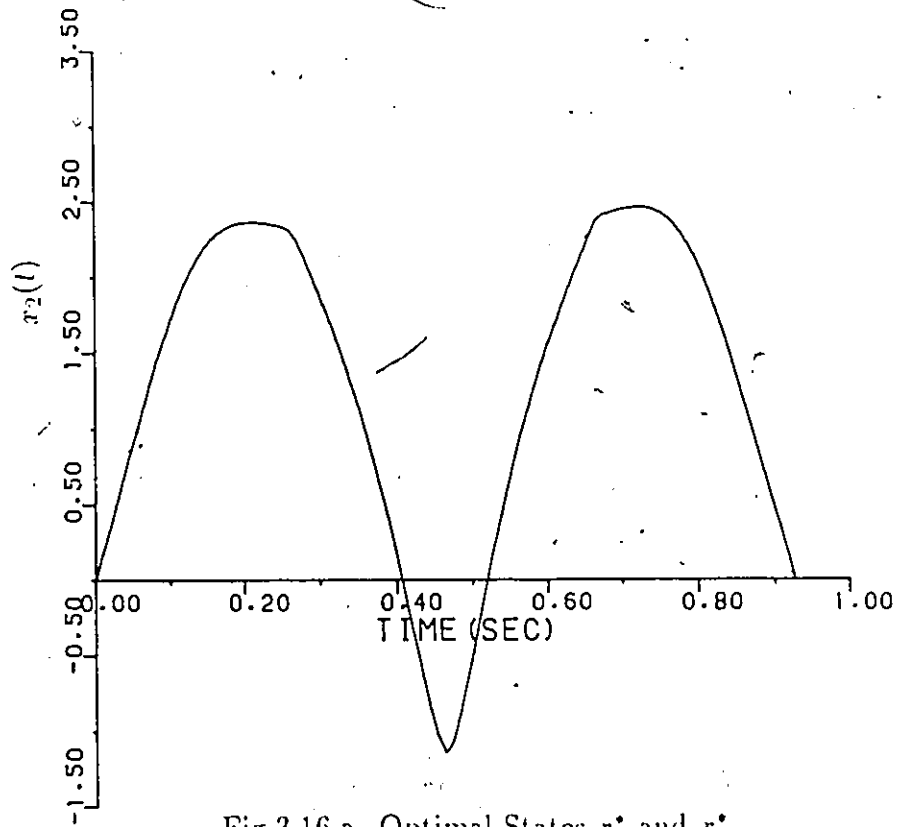
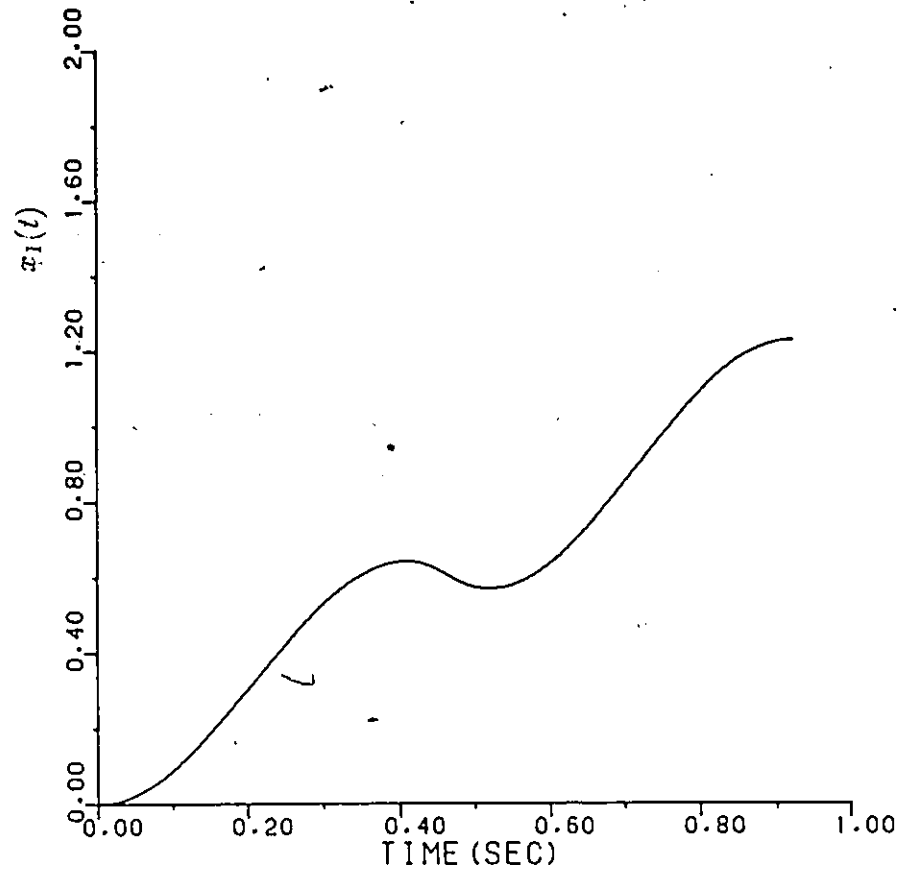
Fig.3.13.b. Optimal States x_3^* and x_4^* .

Fig.3.14. The Cost Function φ

Iter.	cost
0	$.228 \times 10^1$
10	$.701 \times 10^{-2}$
20	$.522 \times 10^{-2}$
30	$.133 \times 10^{-2}$
60	$.123 \times 10^{-3}$
80	$.268 \times 10^{-4}$
100	$.517 \times 10^{-5}$
150	$.273 \times 10^{-7}$
190	$.654 \times 10^{-10}$

Table 3.2.

Fig.3.15. Optimal controls u_1^* and u_2^* .

Fig.3.16.a. Optimal States x_1^* and x_2^* .

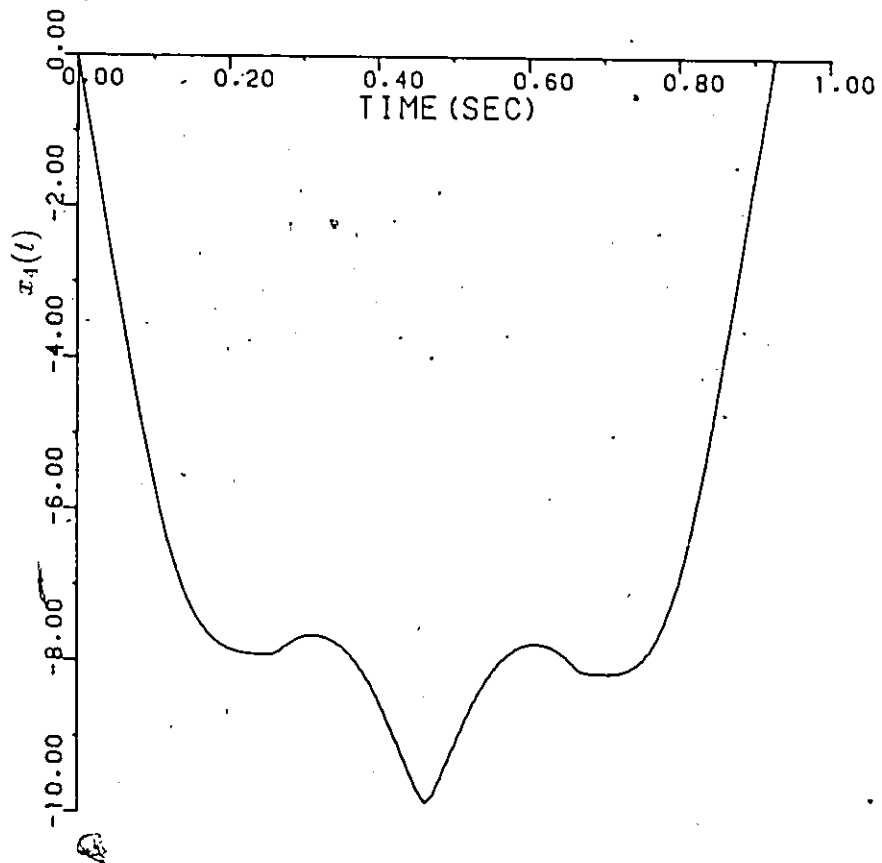
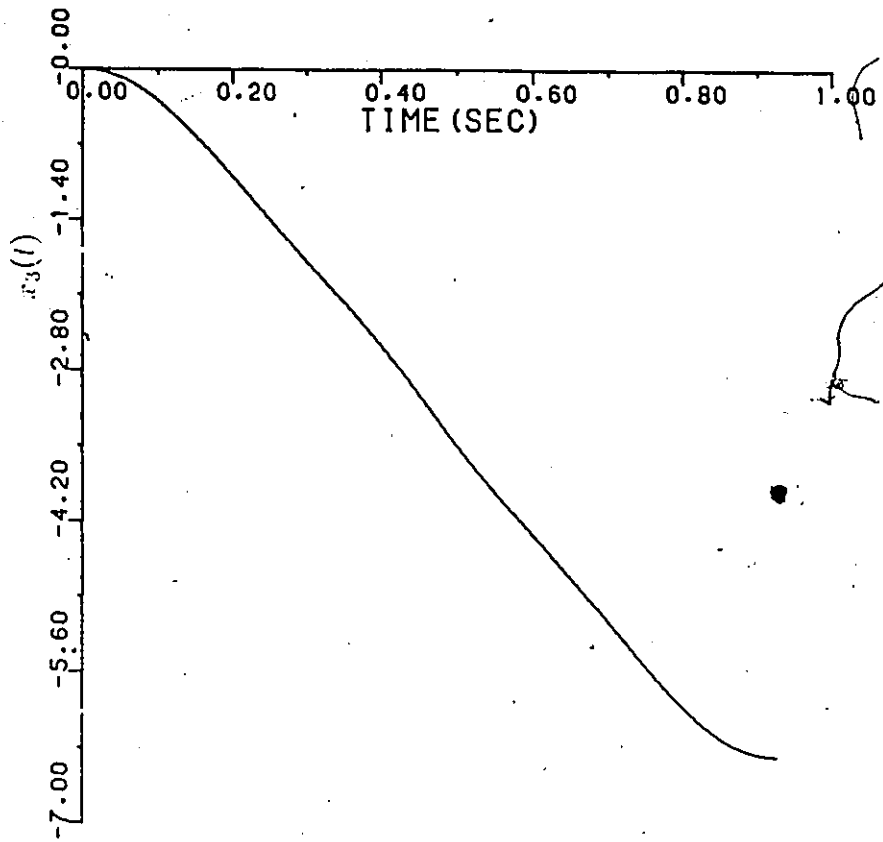


Fig. 2.10. Optimal States x_3 and x_4

3.3 ECOLOGICAL SYSTEM

In the eastern Canada and the northeastern United States, the spruce and fir forests have been periodically subject to ravages by a caterpillar called the spruce budworm. For a number of years, a given path of forest is seen to grow with hardly any budworm in evidence. When the trees have reached a certain level of maturity there is, however, an explosive increase in the number of these insects and they begin to defoliate the trees. When a stand of mature trees have been sufficiently denuded over several consecutive years, they wither and die.

Budworm growth depends on the amount of the habitat available, which is measured in average branch surface area per acre of land and on the amount of spruce and fir needles on these branches. Average budworm density is assumed to have logistic growth in which the carrying capacity is proportional to the habitat size.

The ecological system is governed by the following set of equations [25].

$$\begin{aligned}\dot{x}_1 &= ax_1(1 - x_1/bx_3) - [cx_1^2/(dx_3^2 + x_1^2)] - u_1x_1 \\ \dot{x}_2 &= \alpha x_2(1 - x_2) - \beta x_1x_2^2/x_3 \\ \dot{x}_3 &= \gamma x_3(1 - x_3/\delta x_2)\end{aligned}\tag{3.6}$$

with the state vector $x(t) = [x_1(t), x_2(t), x_3(t), x_4(t)]' \in R^4$ and the control $u(t) = u_1(t) \in R$.

Where

x_1 = average budworm density.

x_2 =percentage of foliage on the trees with x_2 close to unity indicating a healthy forest.

x_3 =average branch surface area per acre of land.

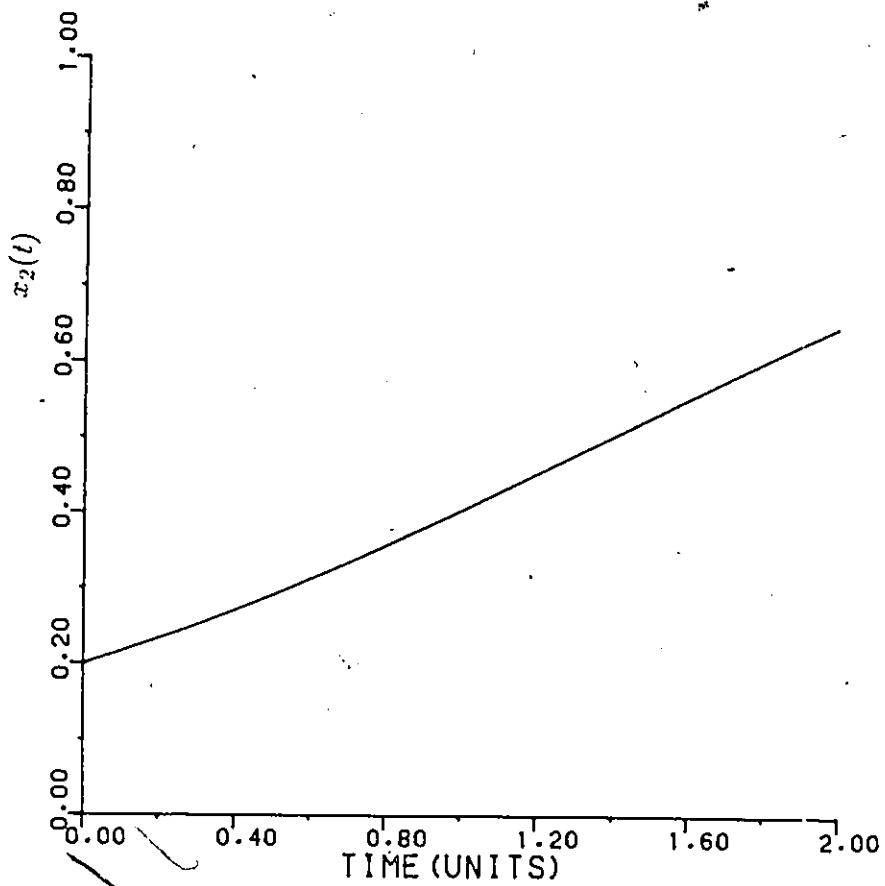
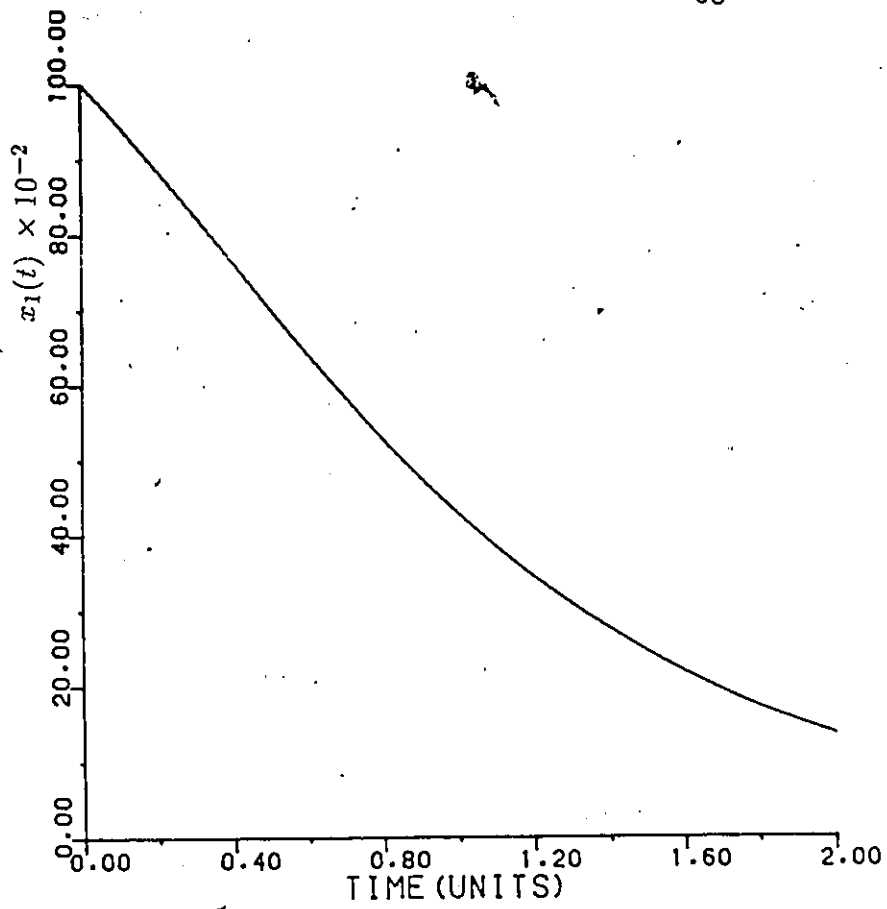
u_1 =the control giving fractional removal rate.

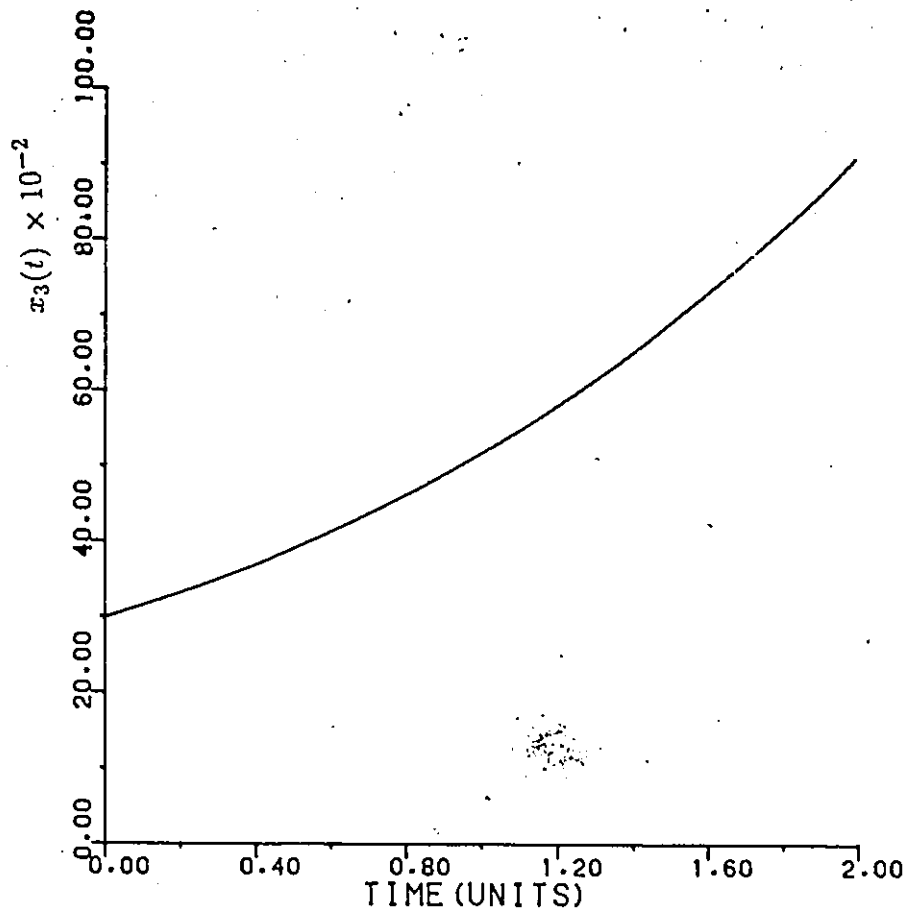
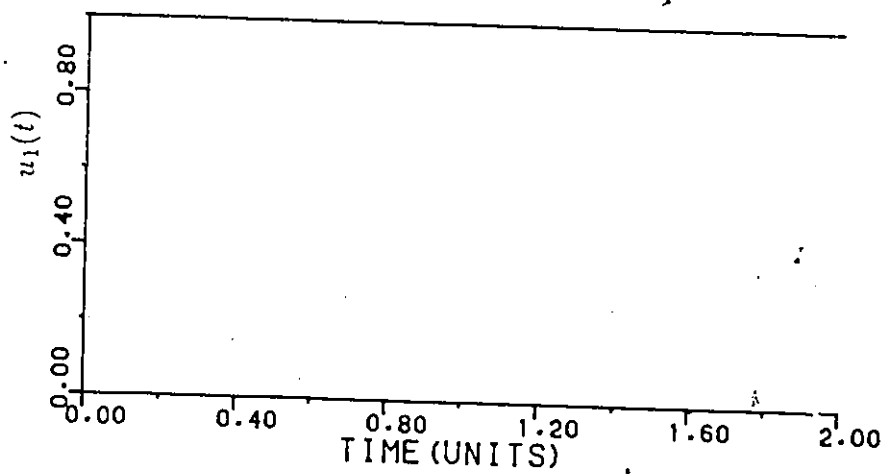
For numerical simulation we choose the following parameters:

$a = 0.5$, $b = 300$, $c = 10.5$, $d = 100$,

$\alpha = 1.0$, $\beta = 0.001$, $\gamma = 1.0$, $\delta = 3 \times 10^4$.

For initial state [10000, .20, 3000] and for terminal state [1360, .65, 9087] optimal trajectories, control, and $\varphi(\tau)$ are shown in Fig.3.17-3.19. In this case, the optimal cost $\varphi(\tau)$ increases after the optimal time.

Fig.3.17.a. Optimal States x_1^* and x_2^* .

Fig.3.17.b. Optimal State x_3^* .Fig.3.18. Optimal Control u_1^* .

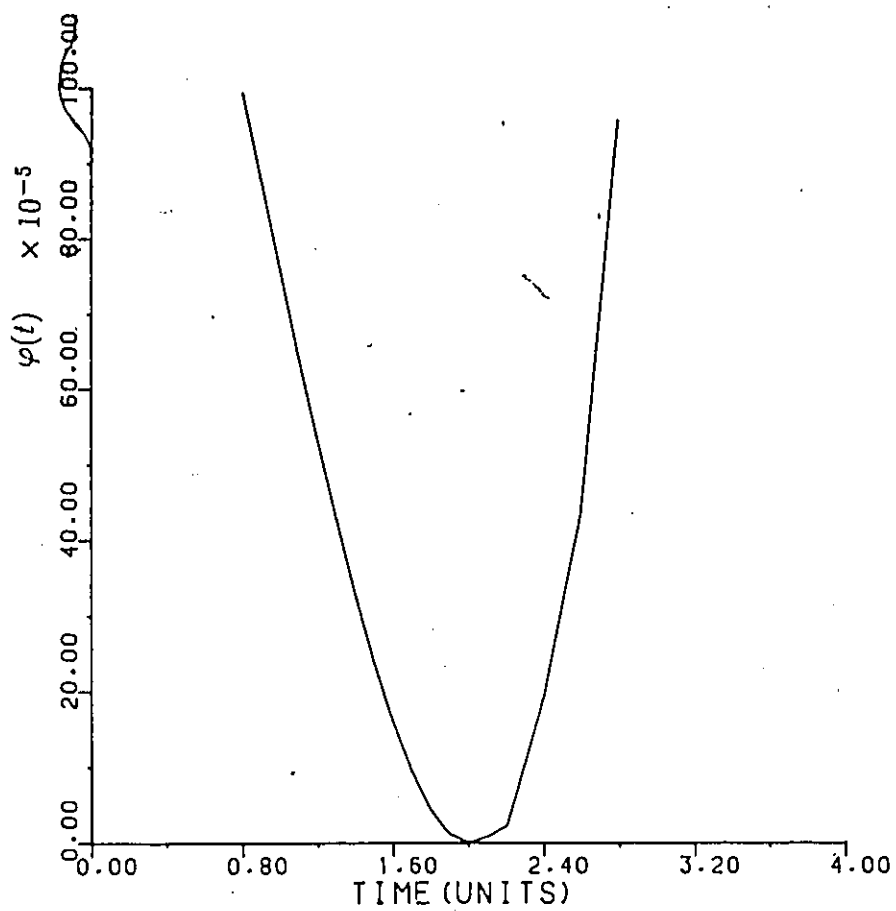


Fig.3.19. The Cost Function φ



3.4 SUMMARY

In this chapter, three numerical examples were considered in order to illustrate the effectiveness of the proposed algorithm for time optimal control problem (see chapter II). In the case of the IBM articulated arm robot, an improvement of the optimal time has been achieved. It is clear, from the three examples, that the proposed method provides the optimal control policy for any system, while using poor initial guesses.

In the next chapter we show how the time optimal control algorithm can be easily modified to be used for the identification problem of systems governed by ordinary differential equations.

CHAPTER IV

NUMERICAL SIMULATION FOR SYSTEM IDENTIFICATION WITH APPLICATIONS TO ROBOTICS AND ECOLOGY

4.1 INTRODUCTION

An important and essential aspect of modelling any physical system is the identification parameters in the model equation. Usually a physical system is modelled on the basis of intuition, some idealizing assumptions and knowledge of functional relationship between the state variables and their time rate of change. The model equations contains certain parameters which are completely unknown because of lack of precise understanding of the system, or only partly known because of poor measurement data. A fundamental problem in system modelling is, then, determination of these unknown parameters so that the corresponding response of the model equation approximates as closely as possible the actual response of the physical system. A very natural approach for solving this problem is to consider it as a control problem.

For numerical computation of the optimal parameters, we use the gradient method as discussed in chapter II. In fact the algorithm given in chapter II can be easily modified to be used for the identification problem. The algorithm is presented in section 3 and its theoretical basis is given in section 2. Three numerical examples illustrating the usefulness of the algorithm are provided in section 4.

4.2 PROBLEM FORMULATION

In this section, we present a general statement of the parameter identification problem and develop the necessary conditions of optimality.

Consider the system

$$\begin{aligned}\dot{x}(t) &= f(t, x, \alpha), \quad t \in [0, T] \\ x(0) &= x_0\end{aligned}\tag{4.1}$$

where $x(t) \in R^n$ is the state vector and α , taking values in a closed bounded subset P of R^m , is the unknown parameter. The function $f: I \times R^n \times R^m \rightarrow R^n$ is known, except for the vector α . It is assumed that the response of the physical system, which is modelled as (4.1), is given in the form of data $y(t)$, $t \in I$.

We define an identification error as the mean square difference between the system output and the observed data

$$J(\alpha) = (1/2) \int_0^T \|x(t, \alpha) - y(t)\|^2 dt\tag{4.2}$$

where $x(\cdot, \alpha)$ is the response of system equation (4.1) corresponding to the parameter α and the initial condition x_0 .

The identification problem may be stated as follows:

Find a parameter $\alpha^* \in P$ such that $J(\alpha^*) \leq J(\alpha)$ for all $\alpha \in P$, where $J(\alpha)$ is given by (4.2) for all $\alpha \in R^m$.

In view of the above discussion, it is clear that a very natural approach for solving the identification problem is to consider it as a control problem. In fact this could be

considered as a special case of the general control problem (see chapter II) where the parameter α can be regarded as the control vector. Using the minimum principle as given in theorem 2.1 (see chapter II), we can derive a set of necessary conditions of optimality for the identification problem. This is given below.

Theorem 4.1 (Necessary Conditions of Optimality for Identification)

In order that the $\alpha^* \in P$ or equivalently the pair $\{\alpha^*, x^*\}$ to be optimal it is necessary that there exists $\psi \in AC(I, R^n)$ $I=[0, T]$, (absolutely continuous function on I with values in R^n) such that the triple $\{\alpha^*, x^*, \psi^*\}$ satisfies the following differential equations and inequalities :

(i) $\dot{x}(t) = f(t, x(t), \alpha)$ for a.c. $t \in (0, T]$

$x(0) = x_0$

(ii) $\dot{\psi}(t) = -f_x'(t, x(t), \alpha)\psi(t) + (y(t) - x(t))$

$\psi(T) = 0$

(4.3)

and the inequality

(iii) $\int_0^T H(t, x^*(t), \psi^*(t), \alpha^*)dt \leq \int_0^T H(t, x^*(t), \psi^*(t), \alpha)dt$

for all $\alpha \in P$

where

$H(t, x, \psi, \alpha) = (f(t, x, \alpha), \psi) + (1/2)\|x(t, \alpha) - y(t)\|^2$ (4.4)

4.3 ALGORITHM and FLOWCHART

4.3.1 Algorithm

Based on the necessary conditions of optimality presented in the preceding section, an iterative algorithm can be developed for the determination of the optimal parameter α^* using the gradient technique as discussed in the chapter II. The time optimal control algorithm can be easily modified to be used in this case. Indeed, for the gradient method applied to identification problems, we have the following algorithm :

Step 1 : Guess $\alpha_1 \in P$ and set $n=1$.

Step 2 : Solve the initial value problem (4.1) with $\alpha = \alpha_1$ giving $x_n(\vec{t}) = x(t, \alpha_n)$.

Step 3 : Using the data x_n and α_n , solve the adjoint system

$$\begin{aligned} \dot{\psi}(t) &= -f_x'(x_n(t), \alpha_n)\psi(t) + (y(t) - x_n(t)) \\ \psi(T) &= 0 \end{aligned} \tag{4.5}$$

backward in time to determine ψ_n .

Step 4 : Compute the gradient vector

$$g_n = \int_0^T (\partial H / \partial \alpha)(t, x_n, \psi_n, \alpha_n) dt \tag{4.6}$$

where H is the Hamiltonian (see equation 4.4)

Step 5 :

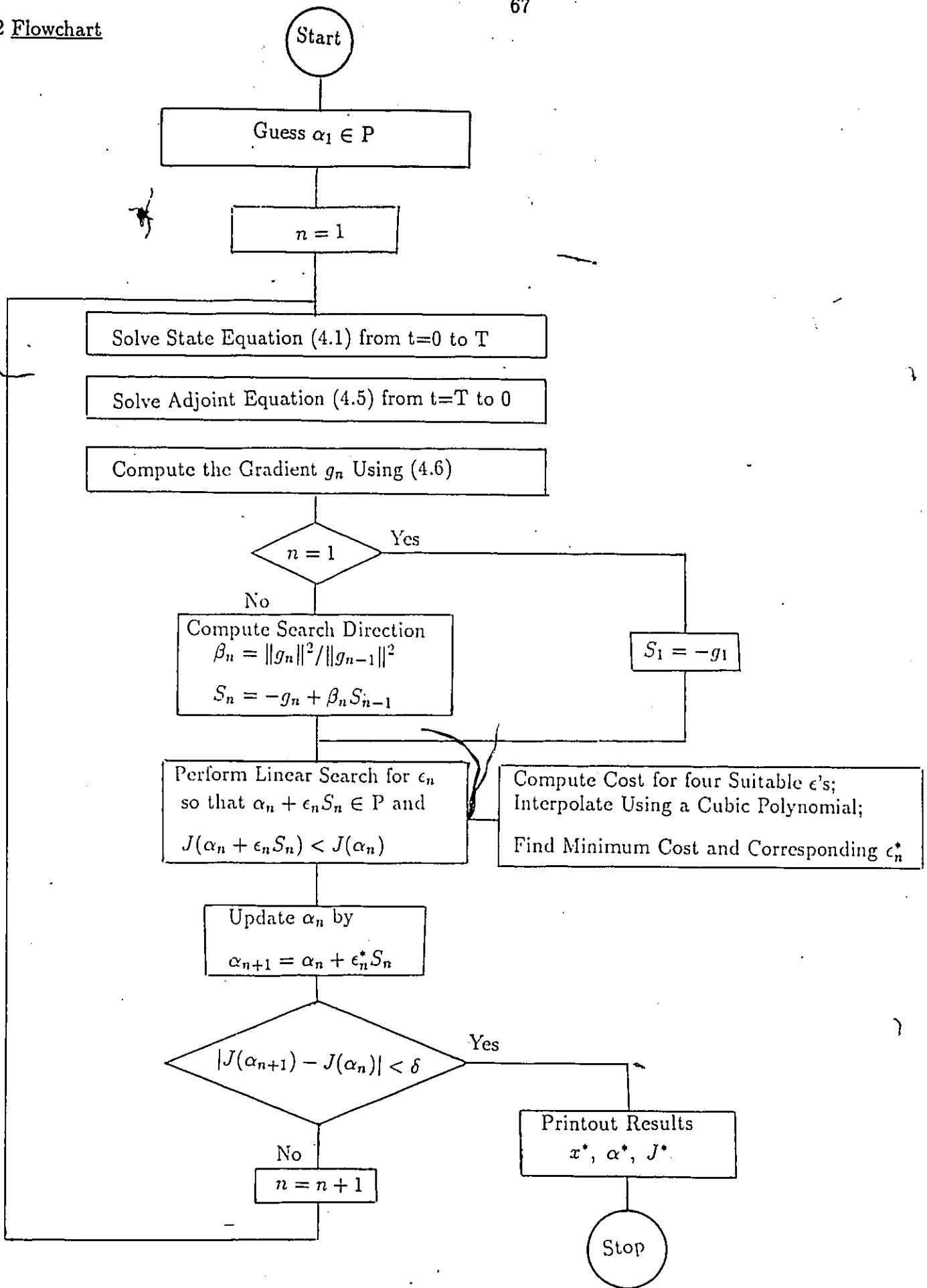
- (i) If $n = 1$ set $s_n = -g_n$
- (ii) If $n \neq 1$ set $s_n = -g_n + (\|g_n\|^2 / \|g_{n-1}\|^2) s_{n-1}$

Step 6 :

- a. If $g_n \neq 0$, then modify α_n to $\alpha_{n+1} = \alpha_n + \epsilon s_n$ by choosing $\epsilon > 0$ sufficiently small so that $\alpha_{n+1} \in P$ and $J(\alpha_{n+1}) \leq J(\alpha_n)$. A stopping criterion is used at this stage. If $|J(\alpha_{n+1}) - J(\alpha_n)| \leq \delta$, for small $\delta > 0$, then stop, otherwise set $\alpha_n = \alpha_{n+1}, n = n + 1$, and go to step 2.
- b. If at the n^{th} stage $g_n = 0$ then α_n is a local minimizing element of $J(\alpha)$.

Remark 4.2

As in any gradient method the iterated parameter would tend to seek a local minimum. This could be considered by repeating the procedure for different initial guesses for the unknown parameter.



4.4 NUMERICAL SIMULATION

The observation data for the actual system i.e. $y(t)$, were generated with known values of the parameter in the equations which, in the sequel, will be referred to as the "true parameter". In order to illustrate the effectiveness of our proposed scheme, we consider the identification problem for the three models that have been presented in chapter III.

Example 1: (Automelec ACR robot)

The identification problem arises naturally in the modelling of a robot manipulator. The system model given in chapter III contains certain unknown parameters such as

- a. moment of inertia
- b. masses of the arm and the load
- and
- c. lengths of the manipulator links

The true parameter is taken as

$$\alpha^* = [m_k, m_l, d_0, M_z, M_c]' \in R^5$$

$$= [3.7, 4.6, 0.37, 0.28, 0.09]'$$

In table (4.1.a)-(4.1.d), we summarize the convergence of the parameter α corresponding to different initial guesses of the unknown parameter.

(i)

Iter.	α_1	α_2	α_3	α_4	α_5	cost
0	1.5	3.1	1.6	1.3	1.0	$.283 \times 10^1$
30	2.978	3.818	0.1811	0.9675×10^{-1}	0.6255×10^{-1}	$.134 \times 10^{-2}$
80	3.729	4.322	0.3984	0.1781	0.1439	$.268 \times 10^{-2}$
120	3.849	4.393	0.3884	0.1864	0.1522×10^{-1}	$.138 \times 10^{-5}$
200	3.885	4.414	0.3855	0.1888	0.1517	$.037 \times 10^{-10}$

Table 4.1.a.

The initial, computed, and true system trajectories are shown in figure (4.1.a)-(4.1.d).

(ii)

Iter.	α_1	α_2	α_3	α_4	α_5	cost
0	6.0	7.0	0.2	1.5	2.0	$.256 \times 10^1$
10	4.006	6.490	0.1122	0.4301	0.1302	$.274 \times 10^{-1}$
20	2.667	5.732	0.3261	0.3587	0.5875×10^{-1}	$.102 \times 10^{-2}$
30	2.569	5.662	0.3075	0.3919	0.9190×10^{-1}	$.431 \times 10^{-4}$
90	2.614	5.686	0.2993	0.3951	0.9513×10^{-1}	$.359 \times 10^{-13}$

Table 4.1.b.

(iii)

Iter.	α_1	α_2	α_3	α_4	α_5	cost
0	7.0	1.0	3.0	1.0	1.0	$.279 \times 10^1$
20	4.359	3.496	0.4932	0.2270	0.0	$.243 \times 10^{-3}$
50	4.489	3.605	0.4927	0.1158	0.274×10^{-2}	$.150 \times 10^{-4}$
120	4.591	3.707	0.4591	0.1656	0.526×10^{-1}	$.113 \times 10^{-9}$
140	4.592	3.708	0.4590	0.1657	0.527×10^{-1}	$.193 \times 10^{-14}$

Table 4.1.c.

It is clear from the above tables, that there exists more than one optimal parameter that provide the desired trajectory.

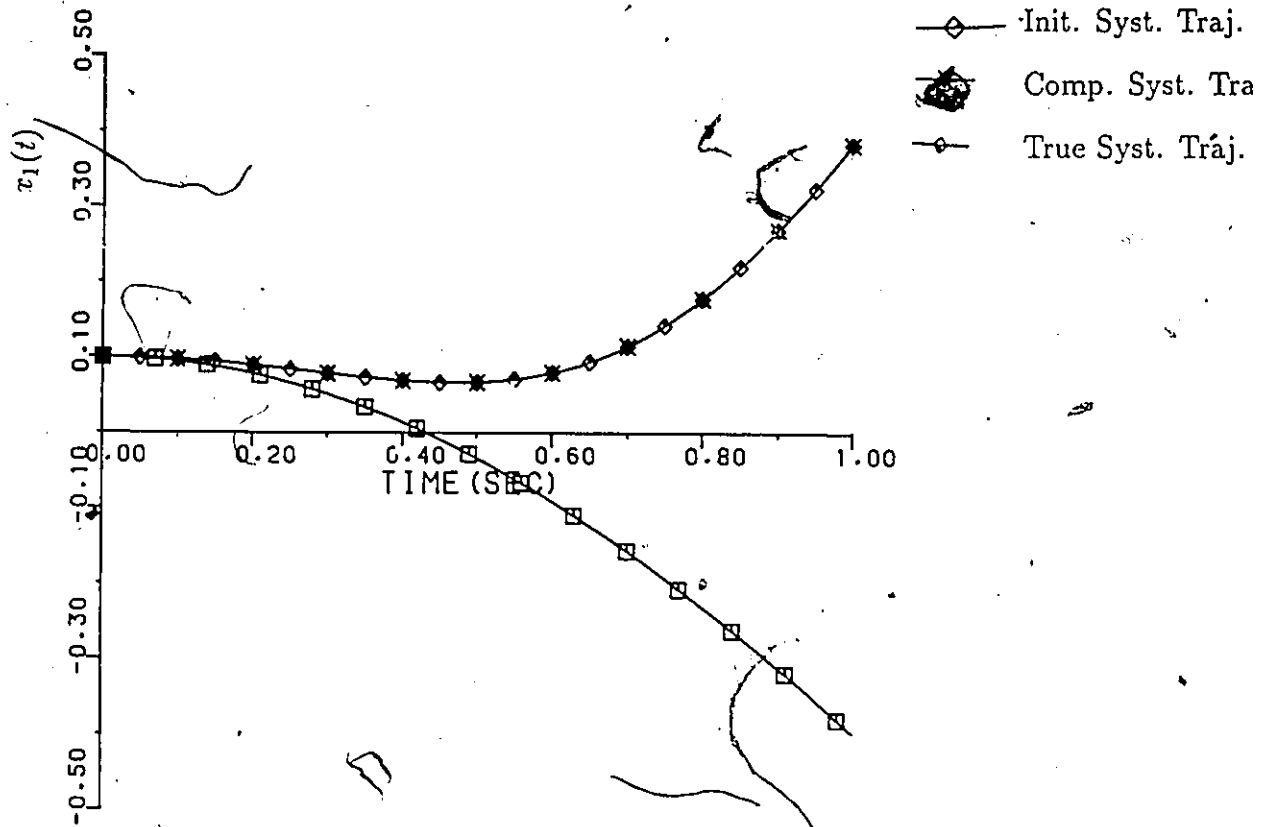


Fig. 4.1.a.

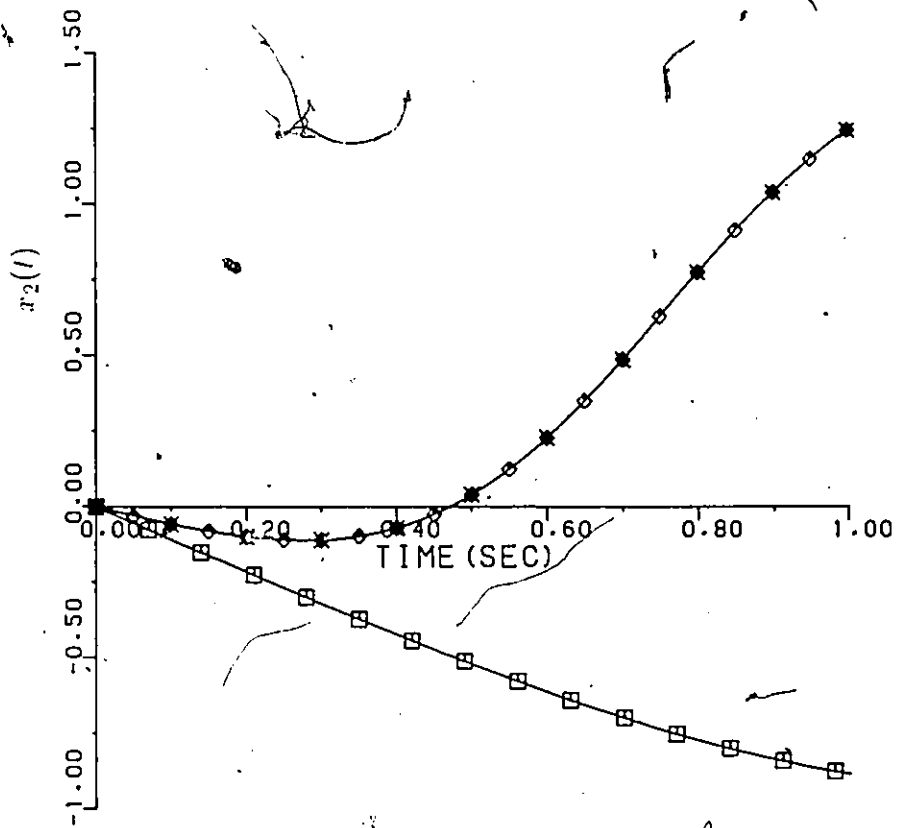


Fig. 4.1.b.

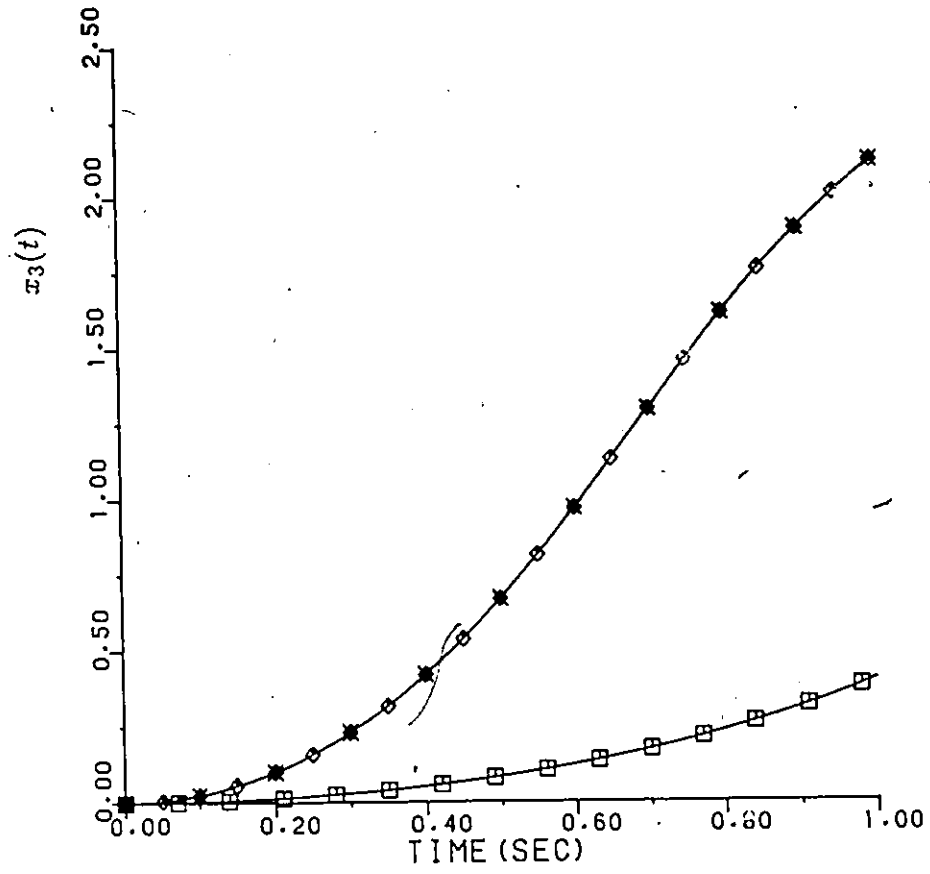


Fig. 4.1.c.

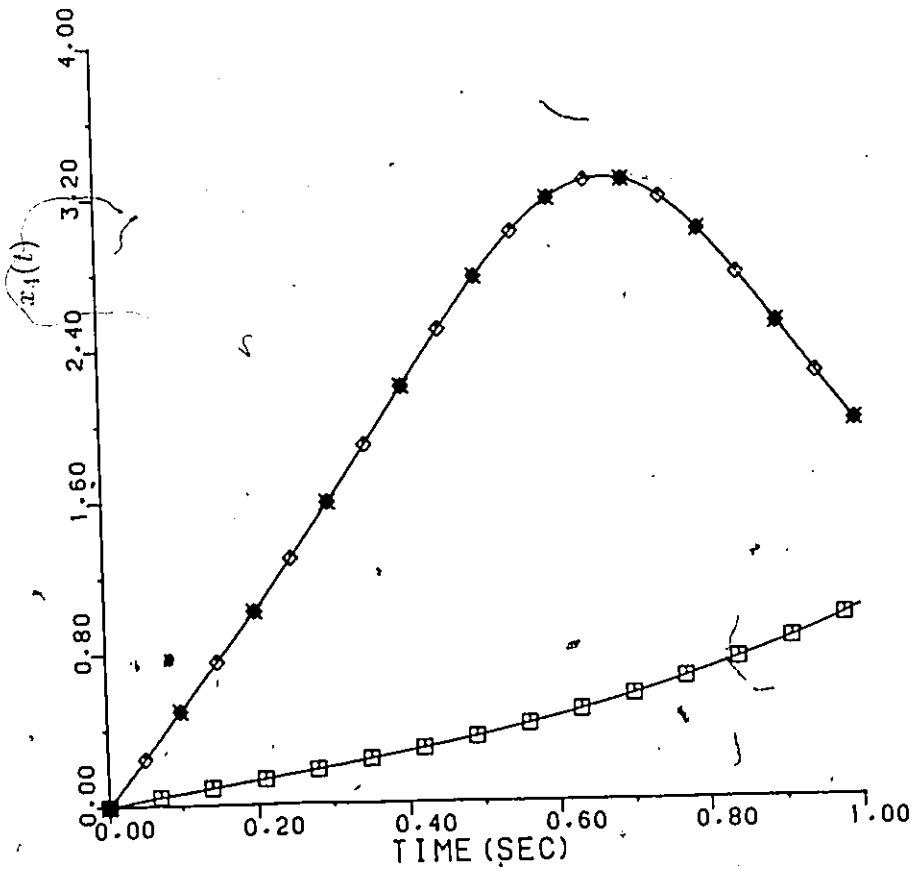


Fig. 4.1.d.

Example 2: (IBM Articulated Arm Robot with two Links)

The true parameter is taken as

$$\alpha^* = [L_1, L_2, d_s, m_s, m_h, M_f, M_s, M_h]^T \in R^8$$

$$= [0.4, 0.25, 0.125, 15.0, 6.0, 1.6, 0.43, 0.01]^T$$

In the following, the unknown vector is computed for two different initial guesses as shown in table (4.2.a)-(4.2.b).

CASE A

Iter.	α_1	α_2	α_3	α_4	α_5	α_6	α_7	α_8	cost
0	.15	.30	.35	.10	.20	1.60	.75	.85	$.181 \times 10^1$
10	.2679	.3032	.3332	.5076	.6507	1.929	.5310	.6157	$.559 \times 10^{-2}$
20	.2761	.3029	.3329	.5362	.6798	1.962	.5133	.6321	$.157 \times 10^{-7}$
55	.2760	.3029	.3329	.5361	.6798	1.962	.5132	.6320	$.108 \times 10^{-16}$

Table 4.2.a.

The initial, computed and true system trajectories are shown in figure (4.2.a)-(4.2.d).

CASE B

Iter.	α_1	α_2	α_3	α_4	α_5	α_6	α_7	α_8	cost
0	1.0	.70	.40	2.0	.80	1.0	.20	.70	$.344 \times 10^1$
10	1.153	.5534	.2534	1.981	.8634	1.067	.0905	.5397	$.592 \times 10^{-2}$
60	1.175	.5536	.2535	1.996	.8802	1.147	.0942	.5483	$.425 \times 10^{-7}$
70	1.176	.5535	.2535	1.995	.8803	1.147	.0941	.5482	$.153 \times 10^{-9}$

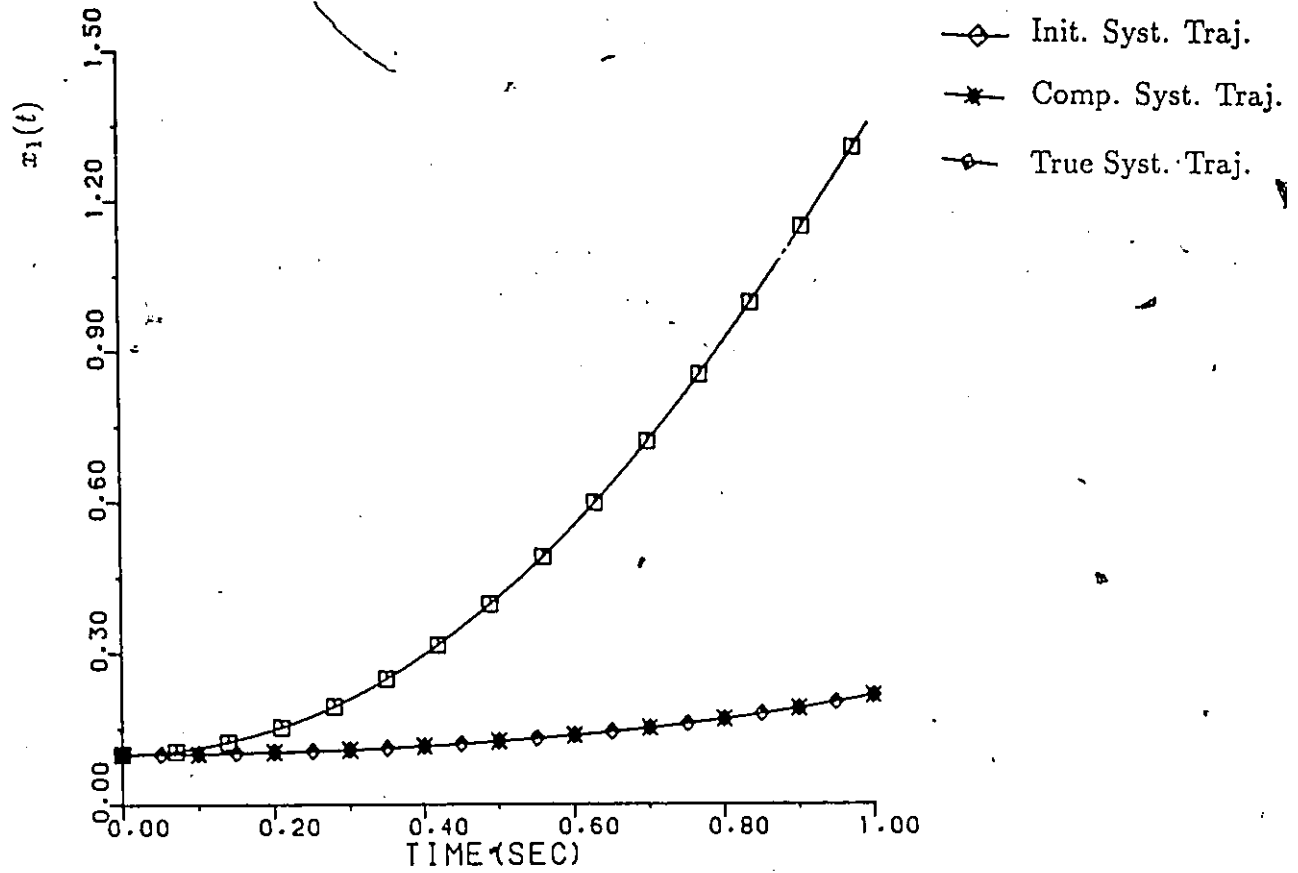


Fig. 4.2.a

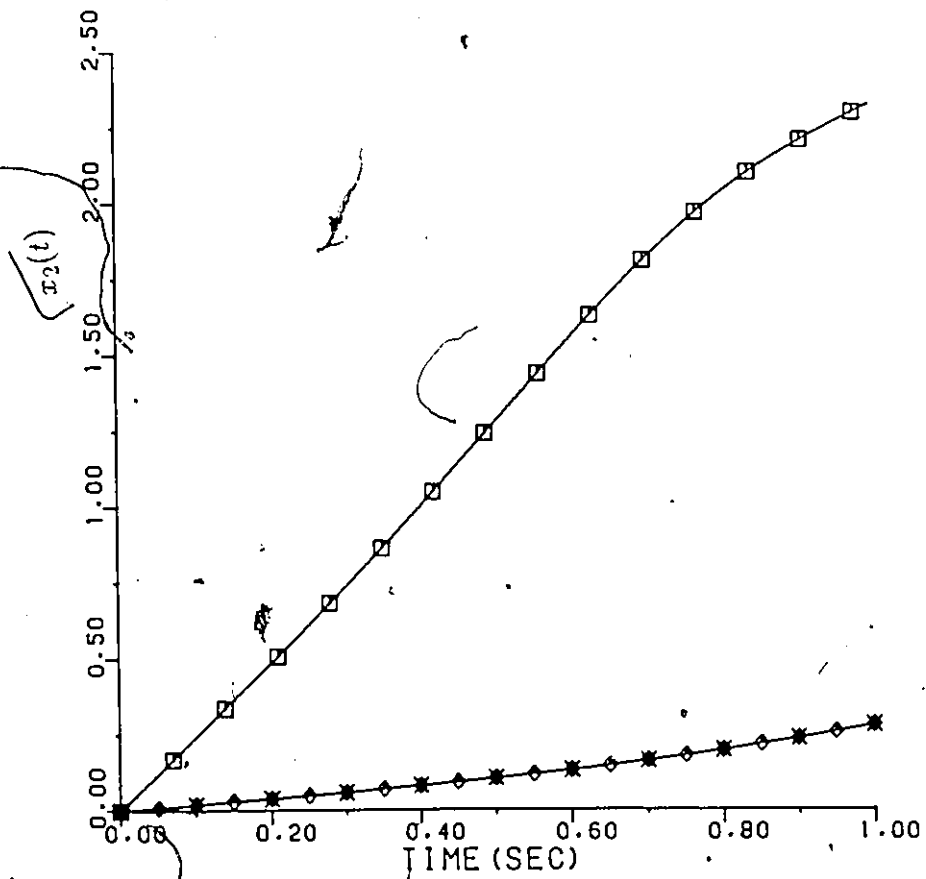


Fig. 4.2.b

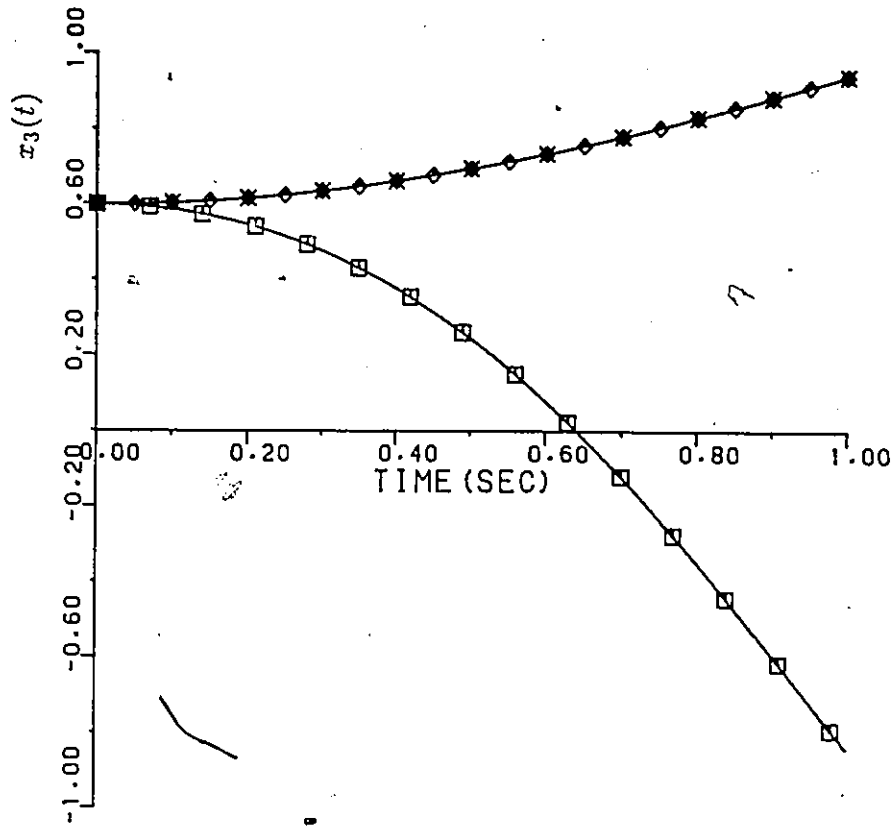


Fig.4.2.c

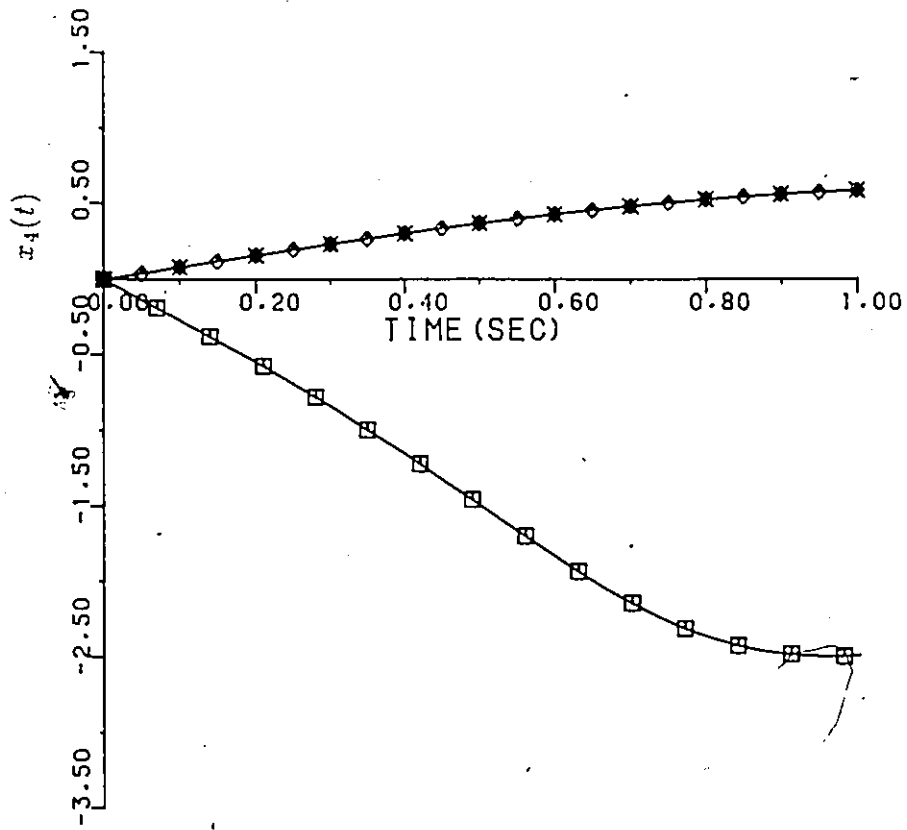


Fig.4.2.d

Example 3: (Ecological System)

CASE A: The true parameter is taken as

$$\alpha^* = [a, b, c, d, \lambda, \beta, \gamma, \delta]' \in R^8$$

$$= [0.5, 300, 10.5, 100, 1, 10^{-3}, 1, 3 \times 10^4]'$$

The initial guess of the unknown parameter is taken as

$$\alpha_g = [1, 20, 7, 11, 2, 1, 6, 900]'$$

The computed parameter is

$$\alpha_c = [.565, 3.8, 2.5, 1.0, 1.23, 4.13, 2.21, 667.17]'$$

The initial, computed, and true system trajectories are shown in figure (4.3.a)-4.3.c).

CASE B: The true parameter is taken as

$$\alpha^* = [a, \beta, \gamma]' \in R^3$$

$$= [0.5, 10^{-3}, 1.0]'$$

The remaining terms are taken constant.

The computed parameter is

$$\alpha_c = [0.50001, 10^{-3}, 1.00003]'$$

In table (4.3), the error function is given as a function of the number of iterations.

Iter.	cost
0	.243 × 10 ⁶
10	.197
20	.522 × 10 ⁻²
30	.931 × 10 ⁻²
40	.300 × 10 ⁻⁴
50	.376 × 10 ⁻⁵
80	.144 × 10 ⁻¹¹

Table 4.3.

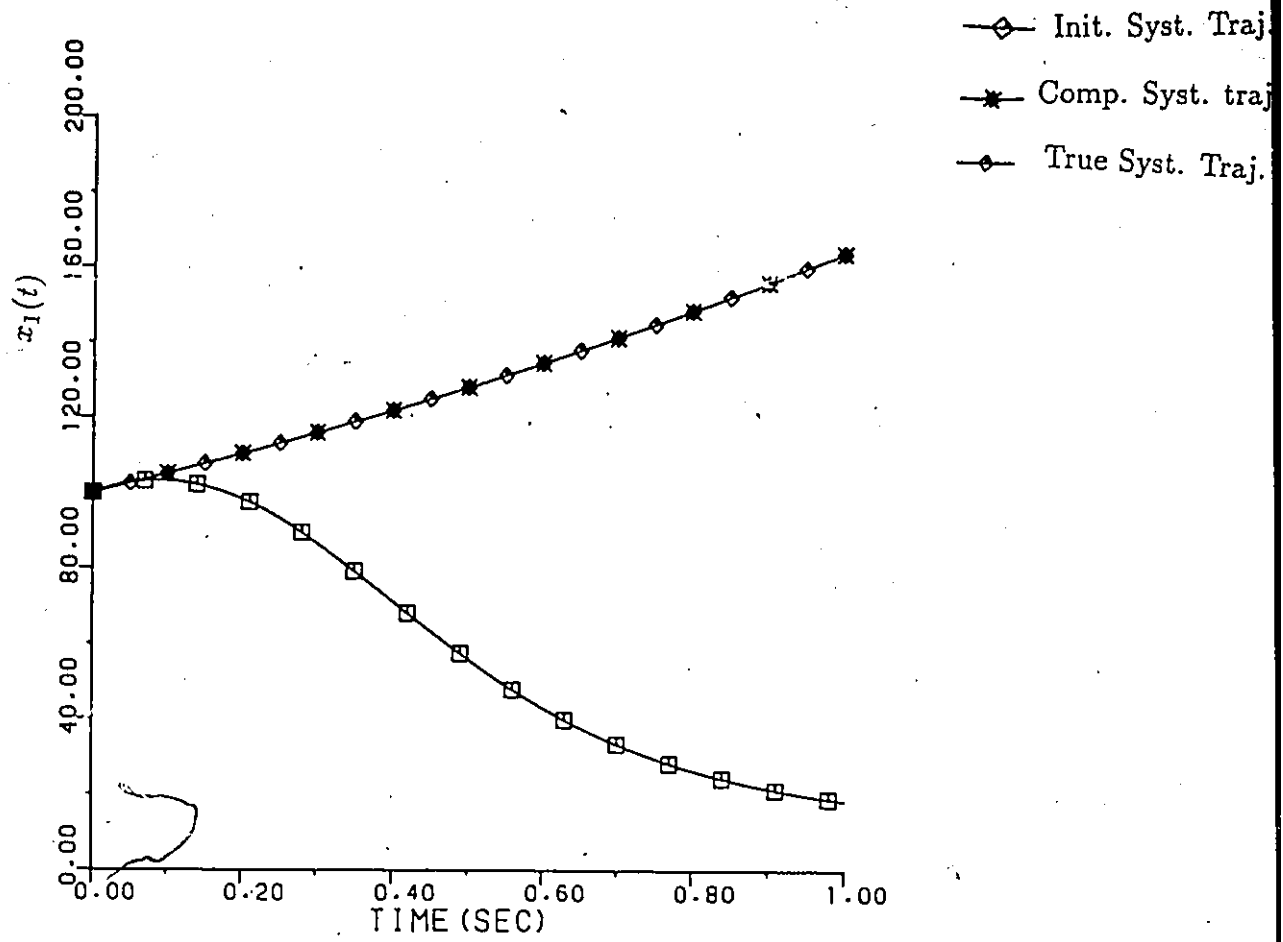


Fig. 4.3.a.

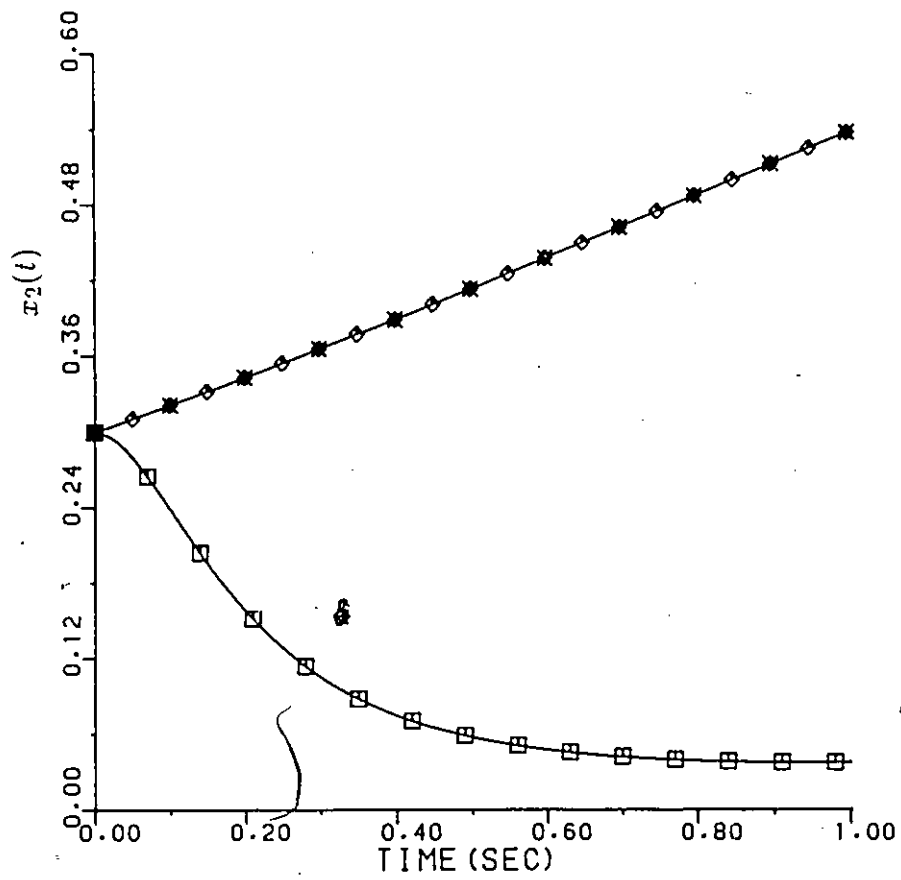


Fig. 4.3.b.

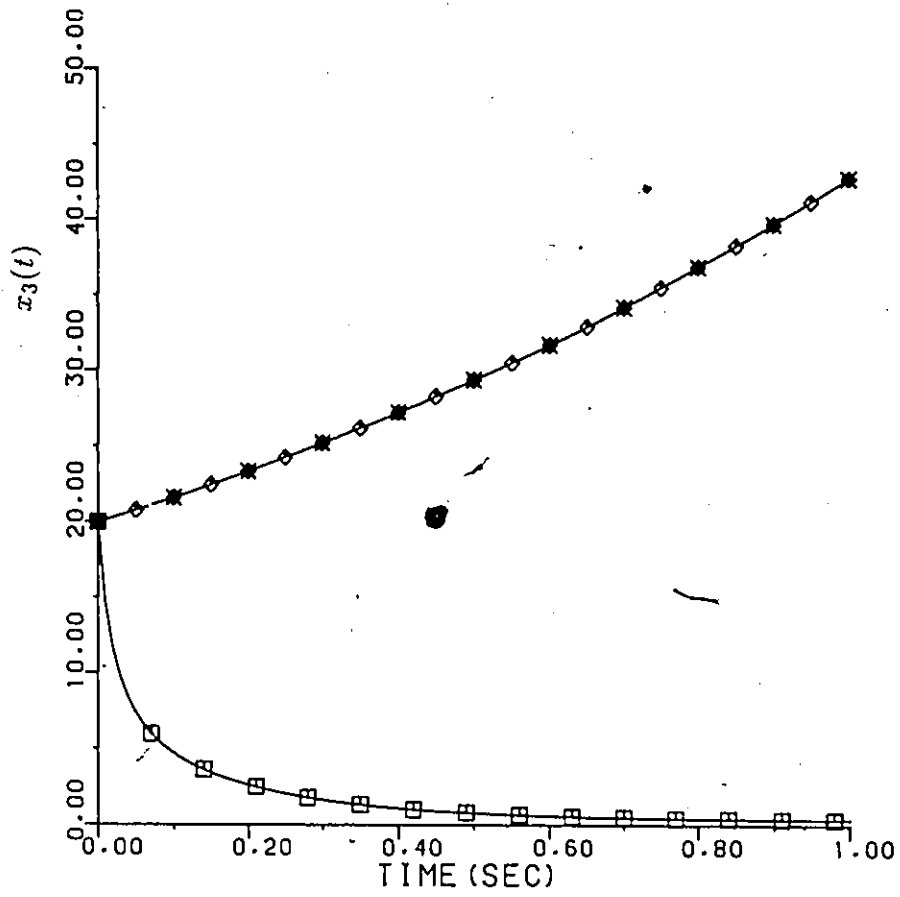


Fig. 4.3.c.

4.5 SUMMARY

In this chapter, we have considered the identification problem for systems described by a set of ordinary differential equations.

Utilizing optimal control theory, we have developed the necessary conditions of optimality on the basis of which the unknown parameter can be computed. It has been shown that the optimal parameter (minimizing the mean square difference between the observed data and the response of the model equation) is determined by simultaneous solution of the system equation, the corresponding adjoint equation, and an associated maximality condition.

Using the gradient technique, an algorithm for computing the unknown parameter has been developed. Finally, this algorithm has been used for identifying some of the parameters of two robots and the ecological system which have been discussed in chapter III.

CHAPTER V

CONCLUSIONS

In the first part of this thesis, we have presented an algorithm for computing the time optimal controls. The algorithm is based on approximation of time optimal controls by a sequence of controls which are optimal for a suitable class of terminal control problems. These later problems can be easily solved by conjugate gradient technique. The algorithm avoids the solution of a two point boundary value problem which is a difficult task. In order to illustrate the usefulness of the proposed method, numerical simulation has been carried out for two robots (Automelec ACR robot and IBM articulated arm robot with two links) and an ecological system. For the Automelec ACR robot, the optimal time coincides with the results of Geering et al [13]. However, in the case of the IBM articulated arm robot, an improvement of the optimal time has been achieved (compared to that given in [13]).

In the second part of this thesis, we have considered the identification problem for systems described by a set of ordinary differential equations. In fact the algorithm that has been proposed for time optimal control problem was modified to be used for identification problem.

Utilizing optimal control theory, we have developed the necessary conditions of optimality on the basis of which the unknown parameter can be computed. It has been shown that the optimal parameter (minimizing the mean square difference between the observed data and the response of the model equation) is determined by simultaneous solution of the system equation, the corresponding adjoint equation, and an associated maximality condition.

In order to illustrate the usefulness of this algorithm three numerical examples including two robots and an ecological system were considered. The numerical simulation has been carried out for different initial choices of the unknown parameter showing the existence of more than one optimal parameter for each initial choice.

APPENDIX A

LAGRANGE EXTRAPOLATION

To update the control in equation (2.18 chapter II), the step size (α) is determined using Lagrange Extrapolation. First, we consider that the cost function $J(u)$ can be approximated by the following polynomial

$$J(u) = a\alpha^3 + b\alpha^2 + c\alpha + d \quad (A.1)$$

where $(a,b,c,d) \in R^4$.

Then four suitable values of α are chosen so as to compute a, b, c and d .

The best α that can give minimum cost in (A.1) is found by differentiating $J(u)$ with respect to α and setting the result to zero. This is shown below

$$(\partial J(u)) / (\partial(\alpha)) = 3a\alpha^2 + 2b\alpha + c = 0 \quad (A.2)$$

From (A.2)

$$\alpha_{\min} = (-b \pm \sqrt{b^2 - 3ac}) / 3a$$

APPENDIX B

FORTRAN CODES

C	IN THE FOLLOWING THE FORTRAN CODES IN CHAPTER III ARE EXIBITED.	TIM00010
C	A PROGRAM FOR TIME OPTIMAL CONTROL PROBLEM	TIM00020
C	MAXIMUM PRINCIPLE HAS BEEN USED	TIM00030
C	ITERATION USING CONJUGATE GRADIENT METHOD	TIM00040
C	USER SUPPLIIED SUBROUTINES ARE	TIM00050
C	INTCON, SYSTEM, COSTATE, INTIAL, TERNAL, CONTRL,	TIM00060
C	COST, GRAD, OUTP	TIM00070
C	NDIM=NUMBER OF EQUATIONS	TIM00080
C	NCTL=NUMBER OF THE UNKNOWN PARAMETERS	TIM00090
C	NPTS=NUMBER OF POINTS IN THE TIME INTERVAL	TIM00100
C	TIME(1)=INITIAL TIME	TIM00110
C	TIME(2)=TERMINAL TIME	TIM00120
C	TIME(3)=STEP SIZE	TIM00130
C	MAX=MAXIMUM NUMBER OF ITERATIONS	TIM00140
C	DELTA=STEP SIZE TO UPDATE THE FINAL TIME	TIM00150
C	STOP=STOPPING CRITERION ON COST DIFFERENCE	TIM00160
C	ALPHA=STEP SIZE IN THE SEARCH DIRECTION FOR UPDATING THE UNKNOWN	TIM00170
C	VECTOR	TIM00180
C	MAIN ROUTINE	TIM00190
C		TIM00200
C	IMPLICIT REAL *8(A-H,O-Z)	TIM00210
C	DIMENSION YY(4,1501),BNEW(3002),YD(4,1501),B(3002),GRD(3002)	TIM00220
C	DIMENSION Y(4),DERY(4),AUX(4,4),YX(4),X(4),YT(4)	TIM00230
C	DIMENSION SRH(3002),TIME(3),BT(3002)	TIM00240
C	TIME(1)=0.0	TIM00250
C	TIME(2)=0.20	TIM00260
C	TIME(3)=.001	TIM00270
C	DELTA=0.1	TIM00280
C	M=1	TIM00290
C	NDIM=4	TIM00300
		TIM00310
		TIM00320
		TIM00330
		TIM00340
		TIM00350
		TIM00360
		TIM00370
		TIM00380
		TIM00390
		TIM00400
		TIM00410
		TIM00420
		TIM00430
		TIM00440
		TIM00450
		TIM00460
		TIM00470
		TIM00480
		TIM00490
		TIM00500
		TIM00510
		TIM00520
		TIM00530
		TIM00540
		TIM00550

```

MAX=450
STOP=1.0D-30
ALPHA=5.0D-3
NPTS=(TIME(2)/TIME(3))+1
NCTL=2*NPTS
CALL INTCON(NCTL,B)
CALL FORWRD(NDIM,NPTS,NCTL,TIME,Y,DERY,AUX,YX,B,YY)
CALL COST(NDIM,NCTL,NPTS,TIME,YY,YD,B,CST)
CALL COGRAD(NDIM,NCTL,NPTS,MAX,TIME,Y,X,YT,DERX,AUX,YX,B,BNEW,YY
1,Y,SRH,GRD,CST,ALPHA,STOP,BT,DELTA)
CALL TEMP(NDIM,NCTL,NPTS,MAX,TIME,Y,X,YT,DERY,AUX,YX,B,BNEW,YY,
1Y,SRH,GRD,CST,ALPHA,STOP,BT)
CALL FORWRD(NDIM,NPTS,NCTL,TIME,Y,DERY,AUX,YX,B,YY)
CALL OUTP(NDIM,NPTS,NCTL,TIME,Y,YY,B,M,CST)
STOP
END
C
SUBROUTINE INTCON(NCTL,B)
C
C   INITIALIZATION OF THE CONTROL
C
DIMENSION B(NCTL)
DO 12 I=1,NCTL
12 B(I)=0.0
RETURN
END
C
SUBROUTINE NUMBER(NPTS,NCTL,TIME)
C
C   DETERMINATION OF THE NUMBER OF POINTS IN THE TIME INTERVAL
C
DIMENSION TIME(3)
NPTS=(TIME(2)/TIME(3))+1
NCTL=2*NPTS
RETURN
END
C
SUBROUTINE SYSTEM (NDIM,NCTL,T,Y,U,DERY)
C
C   SYSTEM DYNAMICS
C
IMPLICIT REAL *8(A-H,O-Z)
DIMENSION Y(NDIM),DERY(NDIM),U(2)
DERY(1)=Y(2)
DERY(2)=(U(1)+3.7*Y(1)*(Y(4)**2)+4.6*(.37+Y(1))*(Y(4)**2))/
&(3.7+4.6)
DERY(3)=Y(4)
DERY(4)=(U(2)-2*3.7*Y(1)*Y(2)*Y(4)-2*4.6*(.37+Y(1))*Y(2)*Y
&(4))/(.28+.09+3.7*(Y(1)**2)+4.6*(.37+Y(1)**2))
RETURN
END
C
SUBROUTINE COSTATE (NDIM,NCTL,T,X,Y,YT,U,DERX)
C
C   ADJOINT SYSTEM

```

```

TIM00560
TIM00570
TIM00580
TIM00590
TIM00600
TIM00610
TIM00620
TIM00630
TIM00640
TIM00650
TIM00660
TIM00670
TIM00680
TIM00690
TIM00700
TIM00710
TIM00720
TIM00730
TIM00740
TIM00750
TIM00760
TIM00770
TIM00780
TIM00790
TIM00800
TIM00810
TIM00820
TIM00830
TIM00840
TIM00850
TIM00860
TIM00870
TIM00880
TIM00890
TIM00900
TIM00910
TIM00920
TIM00930
TIM00940
TIM00950
TIM00960
TIM00970
TIM00980
TIM00990
TIM01000
TIM01010
TIM01020
TIM01030
TIM01040
TIM01050
TIM01060
TIM01070
TIM01080
TIM01090
TIM01100

```

```

C      IMPLICIT REAL *8(A-H,O-Z)
      DIMENSION X(NDIM),DERX(NDIM),Y(NDIM),U(2),YT(NDIM)
      DERX(1)=-((X(2)*Y(4)**2)+X(4)*((-2*3.7*Y(2)*Y(4)-2*4.6*Y(2)
&*Y(4))*(.28+.09+3.7*(Y(1)**2)+4.6*((.37+Y(1))**2))-2*3.7*Y(1)
&+2*4.6*(.37+Y(1)))*(U(2)-2*3.7*Y(1)*Y(2)*Y(4)-2*4.6*(.37+Y(1)
&)*Y(2)*Y(4)))/((.28+.09+3.7*(Y(1)**2)+4.6*((.37+Y(1))**2))**2)
      DERX(2)=-((X(1)+X(4))*(-2*3.7*Y(1)*Y(4)-2*4.6*(.37+Y(1))*Y(4))/
&.28+.09+3.7*(Y(1)**2)+4.6*((.37+Y(1))**2))
      DERX(3)=0.0
      DERX(4)=-((X(2)*(1/8.3)*(2*Y(1)*Y(4)*3.7+4.6*(.37+Y(1))*2*
&Y(4))+X(3)+X(4))*(-2*3.7*Y(1)*Y(2)-2*4.6*(.37+Y(1))*Y(2))/
&.28+.09+3.7*(Y(1)**2)+4.6*((.37+Y(1))**2))
      RETURN
      END

C      SUBROUTINE INTIAL(NDIM,Y)
C
C      INITIAL CONDITIONS FOR THE DIFFERENTIAL EQUATIONS
C
      IMPLICIT REAL *8(A-H,O-Z)
      DIMENSION Y(NDIM)
      Y(1)=0.15
      Y(2)=0.0
      Y(3)=0.0
      Y(4)=0.0
      RETURN
      END

C      SUBROUTINE TERNAL(NDIM,NPTS,YY,X)
C
C      TERMINAL CONDITIONS FOR THE COSTATE EQUATIONS
C
      IMPLICIT REAL *8(A-H,O-Z)
      DIMENSION X(NDIM),YY(NDIM,NPTS)
      X(1)=YY(1,NPTS)-.15
      X(2)=YY(2,NPTS)
      X(3)=YY(3,NPTS)-1.570796
      X(4)=YY(4,NPTS)
      RETURN
      END

C      SUBROUTINE CONTRL(NPTS,NCTL,STEP,B,BNEW,SRH)
C
C      CONTROL UPDATE
C
      IMPLICIT REAL *8(A-H,O-Z)
      DIMENSION B(NCTL),BNEW(NCTL),SRH(NCTL)
      DO 1 I=1,NCTL
      BNEW(I)=B(I)+STEP*SRH(I)
      DO 2 I=1,NPTS
C
C      CONTROL CONSTRAINTS
C
      IF(BNEW(I).GT.15.0) BNEW(I)=15.0000

```

TIMO1110

TIMO1120

TIMO1130

TIMO1140

TIMO1150

TIMO1160

TIMO1170

TIMO1180

TIMO1190

TIMO1200

TIMO1210

TIMO1220

TIMO1230

TIMO1240

TIMO1250

TIMO1260

TIMO1270

TIMO1280

TIMO1290

TIMO1300

TIMO1310

TIMO1320

TIMO1330

TIMO1340

TIMO1350

TIMO1360

TIMO1370

TIMO1380

TIMO1390

TIMO1400

TIMO1410

TIMO1420

TIMO1430

TIMO1440

TIMO1450

TIMO1460

TIMO1470

TIMO1480

TIMO1490

TIMO1500

TIMO1510

TIMO1520

TIMO1530

TIMO1540

TIMO1550

TIMO1560

TIMO1570

TIMO1580

TIMO1590

TIMO1600

TIMO1610

TIMO1620

TIMO1630

TIMO1640

TIMO1650

	IF (BNEW(I).LT.-15.0) BNEW(I)=-15.0	TIM01660
	IF (BNEW(I+NPTS).GT.5.0) BNEW(I+NPTS)=5.0	TIM01670
	IF (BNEW(I+NPTS).LT.-5.0) BNEW(I+NPTS)=-5.0	TIM01680
2	CONTINUE	TIM01690
	RETURN	TIM01700
	END	TIM01710
C		TIM01720
	SUBROUTINE COST(NDIM,NCTL,NPTS,TIME,YY,YD;B,CST)	TIM01730
C		TIM01740
C	COST FUNCTION	TIM01750
C		TIM01760
	IMPLICIT REAL *8(A-H,O-Z)	TIM01770
	DIMENSION YY(NDIM,NPTS),B(NCTL),YD(NDIM,NPTS),TIME(3)	TIM01780
	CST=0.5*((YY(1,NPTS)-.15)**2+YY(2,NPTS)**2+(YY(3,NPTS)-1.570796)	TIM01790
	&**2+YY(4,NPTS)**2)	TIM01800
	RETURN	TIM01810
	END	TIM01820
C		TIM01830
	SUBROUTINE GRAD(NPTS,NDIM,NCTL,Y,X,B,GRD,K)	TIM01840
C		TIM01850
C	GRADIENT SUBROUTINE	TIM01860
C		TIM01870
	IMPLICIT REAL *8(A-H,O-Z)	TIM01880
	DIMENSION Y(NDIM),X(NDIM),B(NCTL),GRD(NCTL)	TIM01890
	GRD(K)=X(2)/B.3	TIM01900
	GRD(K+NPTS)=X(4)/(.28+.09+3.7*(Y(1)**2)+4.6*((.37+Y(1))**2))	TIM01910
	RETURN	TIM01920
	END	TIM01930
C		TIM01940
	SUBROUTINE FORWRD(NDIM,NPTS,NCTL,TIME,Y,DERY,AUX,YX,B,YY)	TIM01950
C		TIM01960
C	RUNGA-KUTTA SUBROUTINE FOR FORWARD SOLUTION OF SYSTEM EQUATION	TIM01970
C		TIM01980
	IMPLICIT REAL *8(A-H,O-Z)	TIM01990
	DIMENSION YY(NDIM,NPTS),B(NCTL)	TIM02000
	DIMENSION Y(NDIM),DERY(NDIM),AUX(4,NDIM),YX(NDIM)	TIM02010
	DIMENSION TIME(3),U(2)	TIM02020
	CALL INTIAL(NDIM,Y)	TIM02030
	X=TIME(1)	TIM02040
	H=TIME(3)	TIM02050
	K=1	TIM02060
	DO 1 I=1,NDIM	TIM02070
1	YY(I,K)=Y(I)	TIM02080
	DO 11 J=2,NPTS	TIM02090
	U(1)=B(K)	TIM02100
	U(2)=B(K+NPTS)	TIM02110
	XX=X	TIM02120
	CALL SYSTEM (NDIM,NCTL,XX,Y,U,DERY)	TIM02130
	DO 2 I=1,NDIM	TIM02140
	AUX(1,I)=H*DERY(I)	TIM02150
2	YX(I)=Y(I)+AUX(1,I)*0.5	TIM02160
	XX=X+0.5*H	TIM02170
	CALL SYSTEM (NDIM,NCTL,XX,YX,U,DERY)	TIM02180
	DO 3 I=1,NDIM	TIM02190
	AUX(2,I)=H*DERY(I)	TIM02200

3	YX(1)=Y(1)+AUX(2,1)*0.5	TIM02210
	XX=X+0.5*H	TIM02220
	CALL SYSTEM (NDIM,NCTL,XX,YX,U,DERY)	TIM02230
	DO 4 I=1,NDIM	TIM02240
	AUX(3,I)=H*DERY(I)	TIM02250
4	YX(I)=Y(1)+AUX(3,I)	TIM02260
	XX=X+H	TIM02270
	K=K+1	TIM02280
	U(1)=B(K)	TIM02290
	U(2)=B(K+NPTS)	TIM02300
	CALL SYSTEM (NDIM,NCTL,XX,YX,U,DERY)	TIM02310
	DO 5 I=1,NDIM	TIM02320
5	AUX(4,I)=H*DERY(I)	TIM02330
	DO 8 I=1,NDIM	TIM02340
	DY=(AUX(1,I)+2.0*AUX(2,I)+2.0*AUX(3,I)+AUX(4,I))/6.0	TIM02350
8	Y(I)=Y(I)+DY	TIM02360
	X=X+H	TIM02370
	DO 10 I=1,NDIM	TIM02380
10	YY(I,K)=Y(I)	TIM02390
11	CONTINUE	TIM02400
	RETURN	TIM02410
	END	TIM02420
C		TIM02430
	SUBROUTINE BACKWD(NDIM,NPTS,NCTL,TIME,Y,YT,DERX,AUX,YX,X,B,YY,YD	TIM02440
	1,GRD)	TIM02450
C		TIM02460
C	RUNGA-KUTTA SUBROUTINE FOR BACKWARD SOLUTION OF ADJOINT EQUATIONS	TIM02470
C		TIM02480
	IMPLICIT REAL *8(A-H,O-Z)	TIM02490
	DIMENSION YY(NDIM,NPTS),GRD(NCTL),B(NCTL),YD(NDIM,NPTS)	TIM02500
	DIMENSION Y(NDIM),DERX(NDIM),AUX(4,NDIM),YX(NDIM),X(NDIM)	TIM02510
	DIMENSION TIME(3),YT(NDIM),U(2)	TIM02520
	CALL TERNAL(NDIM,NPTS,YY,X)	TIM02530
	T=TIME(2)	TIM02540
	H=TIME(3)	TIM02550
	K=NPTS	TIM02560
	DO 11 J=2,NPTS	TIM02570
	U(1)=B(K)	TIM02580
	U(2)=B(K+NPTS)	TIM02590
	DO 9 I=1,NDIM	TIM02600
9	Y(I)=YY(I,K)	TIM02610
	CALL GRAD(NPTS,NDIM,NCTL,Y,X,B,GRD,K)	TIM02620
	TT=T	TIM02630
	CALL COSTATE (NDIM,NCTL,TT,X,Y,YT,U,DERX)	TIM02640
	DO 2 I=1,NDIM	TIM02650
	AUX(1,I)=H*DERX(I)	TIM02660
2	YX(I)=X(I)+AUX(1,I)*0.5	TIM02670
	TT=TT+0.5*H	TIM02680
	CALL COSTATE (NDIM,NCTL,TT,YX,Y,YT,U,DERX)	TIM02690
	DO 3 I=1,NDIM	TIM02700
	AUX(2,I)=H*DERX(I)	TIM02710
3	YX(I)=X(I)+AUX(2,I)*0.5	TIM02720
	TT=TT+0.5*H	TIM02730
	CALL COSTATE (NDIM,NCTL,TT,YX,Y,YT,U,DERX)	TIM02740
	DO 4 I=1,NDIM	TIM02750

	AUX(3,I)=H*DERX(I)	TIM02760
4	YX(I)=X(I)+AUX(3,I)	TIM02770
	TT=T+H	TIM02780
	K=K-1	TIM02790
	DO 17 L=1,NDIM	TIM02800
17	Y(L)=YY(L,K)	TIM02810
	U(1)=B(K)	TIM02820
	U(2)=B(K+NPTS)	TIM02830
	CALL CCSTATE (NDIM,NCTL,TT,YX,Y,YT,U,DERX)	TIM02840
	DC 5 I=1,NDIM	TIM02850
5	AUX(4,I)=H*DERX(I)	TIM02860
	DO 8 I=1,NDIM	TIM02870
	DX=(AUX(1,I)+2.0*AUX(2,I)+2.0*AUX(3,I)+AUX(4,I))/6.0	TIM02880
8	X(I)=X(I)+DX	TIM02890
	T=T+H	TIM02900
	CALL GRAD(NPTS,NDIM,NCTL,Y,X,B,GRD,K)	TIM02910
11	CONTINUE	TIM02920
	RETURN	TIM02930
	END	TIM02940
C		TIM02950
	SUBROUTINE SEARCH(NCTL,GRDN,SRH,VALUO)	TIM02960
C		TIM02970
	COMPUTES SEARCH DIRECTION	TIM02980
C		TIM02990
	IMPLICIT REAL *8 (A-H,O-Z)	TIM03000
	DIMENSION GRDN(NCTL),SRH(NCTL)	TIM03010
	VALUO=0.0	TIM03020
	DO 2 I=1,NCTL	TIM03030
2	VALUO=VALUO+GRDN(I)*GRDN(I)	TIM03040
	BTA=VALUO/VALUO	TIM03050
	DO 3 I=1,NCTL	TIM03060
3	SRH(I)=GRDN(I)+BTA*SRH(I)	TIM03070
	VALUO=VALUO	TIM03080
	RETURN	TIM03090
	END	TIM03100
C		TIM03110
	SUBROUTINE START(NCTL,GRDO,SRH,VALUO)	TIM03120
C		TIM03130
	INITIALIZES THE SEARCH PROCESS	TIM03140
C		TIM03150
	IMPLICIT REAL *8(A-H,O-Z)	TIM03160
	DIMENSION GRDO(NCTL),SRH(NCTL)	TIM03170
	VALUO=0.0	TIM03180
	DO 1 I=1,NCTL	TIM03190
	VALUO=VALUO+GRDO(I)*GRDO(I)	TIM03200
1	CONTINUE	TIM03210
	DO 2 I=1,NCTL	TIM03220
2	SRH(I)=GRDO(I)	TIM03230
	RETURN	TIM03240
	END	TIM03250
C		TIM03260
	SUBROUTINE TEMP(NDIM,NCTL,NPTS,MAX,TIME,Y,X,YT,DERY,AUX,YX,B,	TIM03270
1	BNEW,YY,Y,SRH,GRD,CST,ALPHA,STOP,BT,DELTA)	TIM03280
C		TIM03290
	UPDATE THE FINAL TIME AND LOCATE THE OPTIMAL TIME	TIM03300
C		

C

```

DIMENSION TIME(3),Y(NDIM),X(NDIM),DERY(NDIM),YT(NDIM)
DIMENSION AUX(4,4),YX(NDIM),B(NCTL),BNEW(NCTL),YY(NDIM,NPTS)
DIMENSION Y(NDIM),SRH(NCTL),GRD(NCTL),BT(NCTL)
TO=TIME(2)
C1=CST
T1=TIME(2)
TIME(2)=TIME(2)+DELTA
CALL NUMBER(NPTS,NCTL,TIME)
CALL COGRAD(NDIM,NCTL,NPTS,MAX,TIME,Y,X,YT,DERY,AUX,YX,B,BNEW,
1YY,Y,SRH,GRD,CST,ALPHA,STOP,BT)
C2=CST
T2=TIME(2)
TIME(2)=TO-DELTA
CALL NUMBER(NPTS,NCTL,TIME)
CALL COGRAD(NDIM,NCTL,NPTS,MAX,TIME,Y,X,YT,DERY,AUX,YX,B,BNEW,
1YY,Y,SRH,GRD,CST,ALPHA,STOP,BT)
C3=CST
T3=TIME(2)
IF(C3-C1) 10,10,20
10 C2=C1
C1=C3
T2=T1
T1=T3
TIME(2)=T1-DELTA
CALL SAVE(NCTL,B,BT)
CALL NUMBER(NPTS,NCTL,TIME)
CALL COGRAD(NDIM,NCTL,NPTS,MAX,TIME,Y,X,YT,DERY,AUX,YX,B,BNEW,
YY,,Y,SRH,GRD,CST,ALPHA,STOP,BT)
C3=CST
T3=T1
IF(C3.EQ.0) GO TO 98
IF(C3-C1) 10,10,99
20 IF(C1-C2) 50,50,99
50 C3=C1
T3=T1
C1=C2
T1=T2
TIME(2)=T1+DELTA
CALL SAVE(NCTL,B,BT)
CALL NUMBER(NPTS,NCTL,TIME)
CALL COGRAD(NDIM,NCTL,NPTS,MAX,TIME,Y,X,YT,DERY,AUX,YX,B,BNEW,
1YY,Y,SRH,GRD,CST,ALPHA,STOP,BT)
C2=CST
T2=TIME(2)
IF(C2.EQ.0) GO TO 98
GO TO 20
99 TIME(2)=T3
CST=C(3)
CALL NUMBER(NPTS,NCTL,TIME)
CALL SAVE(NCTL,BT,B)
98 RETURN
END
C
SUBROUTINE SAVE(NCTL,B,BT)
TIM03310
TIM03320
TIM03330
TIM03340
TIM03350
TIM03360
TIM03370
TIM03380
TIM03390
TIM03400
TIM03410
TIM03420
TIM03430
TIM03440
TIM03450
TIM03460
TIM03470
TIM03480
TIM03490
TIM03500
TIM03510
TIM03520
TIM03530
TIM03540
TIM03550
TIM03560
TIM03570
TIM03580
TIM03590
TIM03600
TIM03610
TIM03620
TIM03630
TIM03640
TIM03650
TIM03660
TIM03670
TIM03680
TIM03690
TIM03700
TIM03710
TIM03720
TIM03730
TIM03740
TIM03750
TIM03760
TIM03770
TIM03780
TIM03790
TIM03800
TIM03810
TIM03820
TIM03830
TIM03840
TIM03850

```

C		TIM03860
C	SAVE THE OPTIMAL CONTROL	TIM03870
C		TIM03880
	DIMENSION B(NCTL),BT(NCTL)	TIM03890
	DO 100 I=1,NCTL	TIM03900
100	BT(I)=B(I)	TIM03910
	RETURN	TIM03920
	END	TIM03930
C		TIM03940
	SUBROUTINE MINAP(NDIM,NCTL,NPTS,TIME,Y,DERY,AUX,YX,B,BNEW,YNEW,YD,	TIM03950
	1SRH,CSTOD,STEP,INN)	TIM03960
C		TIM03970
C	UPDATES THE PARAMETER VECTOR	TIM03980
C		TIM03990
C	LAGRANGE EXTRAPOLATION HAS BEEN USED	TIM04000
C		TIM04010
	IMPLICIT REAL *8(A-H,O-Z)	TIM04020
	DIMENSION X(4),C(4),TIME(3)	TIM04030
	DIMENSION YNEW(NDIM,NPTS),SRH(NCTL),BNEW(NCTL),B(NCTL)	TIM04040
	DIMENSION Y(NDIM),DERY(NDIM),AUX(4,NDIM),YX(NDIM)	TIM04050
	DIMENSION YD(NDIM,NPTS)	TIM04060
	N1=1	TIM04070
	N2=1	TIM04080
	CALL CONTRL(NPTS,NCTL,STEP,B,BNEW,SRH)	TIM04090
	CALL FORWRD(NDIM,NPTS,NCTL,TIME,Y,DERY,AUX,YX,BNEW,YNEW)	TIM04100
	CALL COST(NDIM,NCTL,NPTS,TIME,YNEW,YD,BNEW,CST)	TIM04110
	C(1)=CST	TIM04120
	X(1)=STEP	TIM04130
	STEP=2.0*X(1)	TIM04140
	CALL CONTRL(NPTS,NCTL,STEP,B,BNEW,SRH)	TIM04150
	CALL FORWRD(NDIM,NPTS,NCTL,TIME,Y,DERY,AUX,YX,BNEW,YNEW)	TIM04160
	CALL COST(NDIM,NCTL,NPTS,TIME,YNEW,YD,BNEW,CST)	TIM04170
	C(2)=CST	TIM04180
	X(2)=STEP	TIM04190
	IF(C(1)-C(2)) 3,3,4	TIM04200
3	STEP=0.5*X(1)	TIM04210
21	CALL CONTRL(NPTS,NCTL,STEP,B,BNEW,SRH)	TIM04220
	CALL FORWRD(NDIM,NPTS,NCTL,TIME,Y,DERY,AUX,YX,BNEW,YNEW)	TIM04230
	CALL COST(NDIM,NCTL,NPTS,TIME,YNEW,YD,BNEW,CST)	TIM04240
	C(3)=C(2)	TIM04250
	X(3)=X(2)	TIM04260
	C(2)=C(1)	TIM04270
	X(2)=X(1)	TIM04280
	C(1)=CST	TIM04290
	X(1)=STEP	TIM04300
	IF(C(1)-C(2)) 9,9,10	TIM04310
9	STEP=0.5*X(1)	TIM04320
	N1=N1+1	TIM04330
	IF(N1.GT.50) GO TO 888	TIM04340
	CALL CONTRL(NPTS,NCTL,STEP,B,BNEW,SRH)	TIM04350
	CALL FORWRD(NDIM,NPTS,NCTL,TIME,Y,DERY,AUX,YX,BNEW,YNEW)	TIM04360
	CALL COST(NDIM,NCTL,NPTS,TIME,YNEW,YD,BNEW,CST)	TIM04370
	C(4)=C(3)	TIM04380
	X(4)=X(3)	TIM04390
	C(3)=C(2)	TIM04400

	X(3)=X(2)	TIM04410
	C(2)=C(1)	TIM04420
	X(2)=X(1)	TIM04430
	C(1)=CST	TIM04440
	X(1)=STEP	TIM04450
	IF(C(1)-C(2)) 9,9,100	TIM04460
4	STEP=2.0*X(2)	TIM04470
22	CALL CONTRL(NPTS,NCTL,STEP,B,BNEW,SRH)	TIM04480
	CALL FORWRD(NDIM,NPTS,NCTL,TIME,Y,DERY,AUX,YX,BNEW,YNEW)	TIM04490
	CALL COST(NDIM,NCTL,NPTS,TIME,YNEW,YD,BNEW,CST)	TIM04500
	C(3)=CST	TIM04510
	X(3)=STEP	TIM04520
	IF(C(2)-C(3)) 200,19,19	TIM04530
19	STEP=2.0*X(3)	TIM04540
	N2=N2+1	TIM04550
	IF(N2.GT.50) GO TO 888	TIM04560
	CALL CONTRL(NPTS,NCTL,STEP,B,BNEW,SRH)	TIM04570
	CALL FORWRD(NDIM,NPTS,NCTL,TIME,Y,DERY,AUX,YX,BNEW,YNEW)	TIM04580
	CALL COST(NDIM,NCTL,NPTS,TIME,YNEW,YD,BNEW,CST)	TIM04590
	C(4)=CST	TIM04600
	X(4)=STEP	TIM04610
	IF(C(3)-C(4)) 100,301,301	TIM04620
301	C(1)=C(2)	TIM04630
	X(1)=X(2)	TIM04640
	C(2)=C(3)	TIM04650
	X(2)=X(3)	TIM04660
	C(3)=C(4)	TIM04670
	X(3)=X(4)	TIM04680
	GO TO 19	TIM04690
10	STEP=(X(1)+X(2))*0.5	TIM04700
	CALL CONTRL(NPTS,NCTL,STEP,B,BNEW,SRH)	TIM04710
	CALL FORWRD(NDIM,NPTS,NCTL,TIME,Y,DERY,AUX,YX,BNEW,YNEW)	TIM04720
	CALL COST(NDIM,NCTL,NPTS,TIME,YNEW,YD,BNEW,CST)	TIM04730
	C(4)=C(3)	TIM04740
	X(4)=X(3)	TIM04750
	C(3)=C(2)	TIM04760
	X(3)=X(2)	TIM04770
	C(2)=CST	TIM04780
	X(2)=STEP	TIM04790
	GO TO 100	TIM04800
200	STEP=(X(2)+X(3))*0.5	TIM04810
	CALL CONTRL(NPTS,NCTL,STEP,B,BNEW,SRH)	TIM04820
	CALL FORWRD(NDIM,NPTS,NCTL,TIME,Y,DERY,AUX,YX,BNEW,YNEW)	TIM04830
	CALL COST(NDIM,NCTL,NPTS,TIME,YNEW,YD,BNEW,CST)	TIM04840
	C(4)=C(3)	TIM04850
	X(4)=X(3)	TIM04860
	C(3)=CST	TIM04870
	X(3)=STEP	TIM04880
	GO TO 100	TIM04890
100	CALL POLYN(C,X,STEP,INN)	TIM04900
C		TIM04910
	WRITE(6,345)	TIM04920
345	FORMAT(10X,'LAGRANGE EXTRAPOLATION'/)	TIM04930
	WRITE(6,5) (X(I),I=1,4)	TIM04940
	WRITE(6,6) (C(I),I=1,4)	TIM04950

```

5  FORMAT(/ BX, 'ALPHA', 4D15.5/)
6  FORMAT( BX, 'COST ', 4D15.5)
   CALL CONTRL(NPTS,NCTL,STEP,B,B,SRH)
   GO TO 889
888 INN=2
889 RETURN
   END
C
   SUBROUTINE POLYN(C,X,ALPMIN,INN)
C
C   SUBROUTINE FOR LAGRANGE EXTRAPOLATION
C
   IMPLICIT REAL *8 (A-H,O-Z)
   DIMENSION C(4),X(4)
   D1=((X(1)-X(2))*(X(1)-X(3))*(X(1)-X(4)))
   IF(D1.EQ.0.0) GO TO 77
   F1=C(1)/D1
   D2=((X(2)-X(1))*(X(2)-X(3))*(X(2)-X(4)))
   IF(D2.EQ.0.0) GO TO 77
   F2=C(2)/D2
   D3=((X(3)-X(2))*(X(3)-X(1))*(X(3)-X(4)))
   IF(D3.EQ.0.0) GO TO 77
   F3=C(3)/D3
   D4=((X(4)-X(2))*(X(4)-X(3))*(X(4)-X(1)))
   IF(D4.EQ.0.0) GO TO 77
   F4=C(4)/D4
   A=F1+F2+F3+F4
   B=F1*(X(2)+X(3)+X(4))+F2*(X(1)+X(3)+X(4))+F3*(X(1)+X(2)+X(4))
   1+F4*(X(1)+X(2)+X(3))
   CC=F1*(X(2)*X(3)+X(3)*X(4)+X(2)*X(4))+F2*(X(1)*X(3)+X(1)*X(4)+
   1X(3)*X(4))+F3*(X(1)*X(2)+X(1)*X(4)+X(2)*X(4))+F4*(X(1)*X(2)
   1+X(1)*X(3)+X(2)*X(3))
   S=B*B-3.0*A*CC
   IF(A.EQ.0.0.OR.S.LT.0.0) GO TO 77
   ALPMIN=(B+DSQRT(B*B-3.0*A*CC))/(3.0*A)
   GO TO 78
77  INN=2
78  RETURN
   END
C
   SUBROUTINE COGRAD(NDIM,NCTL,NPTS,MAX,TIME,Y,X,YT,DERY,AUX,YX,B
1,BNEW,YY,YD,SRH,GRD,CST,ALPHA,STOP,BT)
C
C   CONJUGATE GRADIENT ALGORITHM
C
   IMPLICIT REAL *8(A-H,O-Z)
   DIMENSION YY(NDIM,NPTS),YD(NDIM,NPTS),B(NCTL),BNEW(NCTL),TIME(3)
   DIMENSION Y(NDIM),DERY(NDIM),AUX(4,NDIM),YX(NDIM),SRH(NCTL)
   DIMENSION X(NDIM),GRD(NCTL),YT(NDIM),BT(NCTL)
   ITER=0
   INN=1
   CSTOD=CST
345 CALL BACKWD(NDIM,NPTS,NCTL,TIME,Y,YT,DERY,AUX,YX,X,B,YY,YD,GRD)
   CALL START(NCTL,GRD,SRH,VALU)
   WRITE(6,2) ITER,CSTOD,ALPHA

```

```

TIM04960
TIM04970
TIM04980
TIM04990
TIM05000
TIM05010
TIM05020
TIM05030
TIM05040
TIM05050
TIM05060
TIM05070
TIM05080
TIM05090
TIM05100
TIM05110
TIM05120
TIM05130
TIM05140
TIM05150
TIM05160
TIM05170
TIM05180
TIM05190
TIM05200
TIM05210
TIM05220
TIM05230
TIM05240
TIM05250
TIM05260
TIM05270
TIM05280
TIM05290
TIM05300
TIM05310
TIM05320
TIM05330
TIM05340
TIM05350
TIM05360
TIM05370
TIM05380
TIM05390
TIM05400
TIM05410
TIM05420
TIM05430
TIM05440
TIM05450
TIM05460
TIM05470
TIM05480
TIM05490
TIM05500

```

```

2  FORMAT(//5X, 'ITER=', I3, 3X, 'COST=', E20.12, 5X, 'ALPHA=', E15.7/)
   WRITE(6,3) (B(I), I=1, NCTL)
3  FORMAT(3X, 'UPDATED PARAMETERS', 4E13.5/)
210 ITER=ITER+1
   DO 77 I=1, NCTL
77  BT(I)=B(I)
   CST1=CST
   KOUNT=0
   ALPHA=-ALPHA
   IF(ITER .GT. MAX) GO TO 999
67  CALL MINAP(NDIM, NCTL, NPTS, TIME, Y, DERY, AUX, YX, B, BNEW, YY, YD, SRH,
1  CSTOD, ALPHA, INN)
   IF(INN.EQ.2) GO TO 999
   CALL FORWRD(NDIM, NPTS, NCTL, TIME, Y, DERY, AUX, YX, B, YY)
   CALL COST(NDIM, NCTL, NPTS, TIME, YY, YD, B, CST)
   WRITE(6,2) ITER, CST, ALPHA
   WRITE(6,3) (B(I), I=1, NCTL)
   IF(ITER .GT. MAX) GO TO 999
   IF(ALPHA.GT.1.0E+04) GO TO 999
   CSTDIF=CSTOD-CST

C
C  STOPING CRITERION WITH COST DIFFERENCE
C
C  IF(CSTDIF .LT. STOP) GO TO 999
65  IF(CSTOD - CST) 65,65,66
   GO TO 999
C
   DO 78 I=1, NCTL
78  B(I)=B(I)-ALPHA*SRH(I)
   KOUNT=KOUNT+1
   IF(KOUNT - 3) 67,67,68
68  WRITE(6,69)
69  FORMAT(//10X, 'COST DOES NOT DECREASE ANY MORE'//)
   GO TO 999
66  IF(CSTDIF .LT. STOP) GO TO 999
   CSTOD=CST
   IF(ITER .EQ. NCTL) GO TO 345
   CALL BACKWD(NDIM, NPTS, NCTL, TIME, Y, YT, DERY, AUX, YX, X, B, YY, YD, GRD)
   CALL SEARCH(NCTL, GRD, SRH, VALU)
   GO TO 210
999  DO 78 I=1, NCTL
78  B(I)=BT(I)
   CST=CST1
   RETURN
   END

C
C  SUBROUTINE OUTP(NDIM, NPTS, NCTL, PRMT, Y, YY, B, M, CST)
C
C  OUTPUT SUBROUTINE FOR PRINTOUT OF SYSTEM TRAJECTORY
C
   IMPLICIT REAL *B(A-H, O-Z)
   DIMENSION Y(NDIM), YY(NDIM, NPTS), B(NCTL), PRMT(3)
   WRITE(6,16) PRMT(2), CST
16  FORMAT(T5, F9.4, 4X, E12.5)
   T=PRMT(1)

```

```

TIM05510
TIM05520
TIM05530
TIM05540
TIM05550
TIM05560
TIM05570
TIM05580
TIM05590
TIM05600
TIM05610
TIM05620
TIM05630
TIM05640
TIM05650
TIM05660
TIM05670
TIM05680
TIM05690
TIM05700
TIM05710
TIM05720
TIM05730
TIM05740
TIM05750
TIM05760
TIM05770
TIM05780
TIM05790
TIM05800
TIM05810
TIM05820
TIM05830
TIM05840
TIM05850
TIM05860
TIM05870
TIM05880
TIM05890
TIM05900
TIM05910
TIM05920
TIM05930
TIM05940
TIM05950
TIM05960
TIM05970
TIM05980
TIM05990
TIM06000
TIM06010
TIM06020
TIM06030
TIM06040
TIM06050

```

```
DO 222 L=1,NPTS
DO 111 I=1,NDIM
111   Y(I)=YY(I,L)
      WRITE(6,12) T,B(L),B(L+NPTS)
      TT=M
      T=T+TT*PRMT(3)
12    FORMAT(F8.3,F7.3,1X,F7.3)
222  CONTINUE
      RETURN
      END
```

```
TIM06060
TIM06070
TIM06080
TIM06090
TIM06100
TIM06110
TIM06120
TIM06130
TIM06140
TIM06150
```

REFERENCES

- [1] Kahn M.E. and Roth B., 1971, "The Near-Minimum-Time Control of Open-Loop Articulated Kinematic Chains," *Journal of Dynamic Systems, Measurement and Control*, vol.93, No.3, Sept.1971, pp.164-172.
- [2] G.N. Saridis and C.S.G.Lee, "Approximation Theory of Optimal Control for Trainable Manipulators," *IEEE Trans. Syst. Man Cybern.*, vol. SMC-9, pp. 152-159, 1979.
- [3] R. Gawronski, I.Pardyka, A.Pawlowski, and W.Wolski, "Time-Optimal Hierarchical Control System of Manipulator Movement with Adaptive On-Off Control," *Syst. Sci.*, vol.7, pp.167-177, 1981.
- [4] T.L.Johnson, "On Feedback Laws for Robotic Systems," Massachusetts Institute of Technology, Lab. Inform. Sci. and Syst., Cambridge, MA, Rep.AFOSRTR-81-0456, Apr. 1981.
- [5] L.K.Huang, "Control of a Tendor Arm," Artificial Intelligence Lab., Mas. Inst. Technology, Rep. AI-M-617, Feb. 1981.
- [6] M. Vukobratovic and M. Kircanski, "Method for Optimal Synthesis of Manipulator robot Trajectories," *J. Dynamic. Syst. Meas. Control*, vol.104, pp.188-193, 1982.
- [7] K.G.Shin and N.D.Mckay, "Robot Path Planning Using Dynamic Programming," in *Proc. 23rd Conf. Decision and Control*, Las Vegas, NV, Dec.1984, pp.1629-1635.
- [8] J.E.Bobrow, S.Dubowsky, and J.S.Gibson, "On the Optimal Control of Robotic Manipulators with Actuator Constraints," in *Proc. 1983 Amer. Contr. Conf.*, San Fransisco, CA, June 1983, pp.782-787.
- [9] N.N.Bolotnik and A.A.Kaplunov, "Optimal Rectilinear Moving a Load with a Two-Link

- Manipulator," Eng. Cybern., vol.20, pp.121-131, 1982.
- [10] C.J.GOH and K.L.TEO, "Miser: An Optimal Control Software. Theory and User Manual." Department of Industrial and Systems Engineering, National University of Singapore, July 1987.
- [11] P.Marinov and P.Kiriazow, "A Direct Method for Optimal Control Synthesis of Manipulator Point-to-Point Motion," in Proc. 9th IFAC World Congress, Budapest, Hungary, Aug.1984, paper 14.2/G-5.
- [12] T.L. Turner and W.A.Gruver, "Viable Suboptimal Controller for Robotic Manipulators," in Proc. 19th IEEE conf. on Decision and control, Albuquerque, NM, Dec.1980, pp.83-87.
- [13] Hans P. Geering, Lino Guzzella, Stephan A.R.Hepner, and Christopher H.Onder, "Time-Optimal Motions of Robots in Assembly Tasks," IEEE Transaction on Automatic Control, vol.AC-31, No.6, June 1986.
- [14] N.U. Ahmed, "Elements of Finite Dimensional Systems and Control Theory," Longman Scientific and Technical U.K. England John Wiley and Sons Inc., New York, 1988.
- [15] Kelley, H.J., "Gradient Theory of Optimal Flight Paths." ARSJ., 30, 1960, pp.947-954.
- [16] Kelley, H.J., "Method of Gradients," G.Leitman, ed. Optimization Techniques. Academic Press, New York, 1962, Chapter 6.
- [17] Bryson, A.E., and Denham, W.F., "A Steepest Ascent Method for Solving Optimum Programming Problems." J.Applied Mechanics, 29, 1962, pp.247-257.
- [18] Bryson, A.E., et al., "Optimum Programming Problems with Inequality Constraints I: Necessary Conditions for External Solutions." AIAA J., 1,1963, pp.2544-2550.
- [19] Denham, W.F., and Bryson, A.E., "Optimal Programming Problems with Inequality Constraints II: Solution by Steepest Ascent." AIAA J., 2, 1964, pp.25-34.
- [20] Stancil, R.T., "A New Approach to Steepest Ascent Trajectory Optimization." AIAA J., 2, 1964, pp.1365-1370.

- [21] Goldstein, A.A., "On Steepest Descent." *SIAM J. control*, ser.A, 3, 1965, pp.147-151.
- [22] McReynolds, S.R., and Bryson, A.E., "A Successive Sweep Method for Solving Optimal Programming Problems." *Proceedings Joint Autom. Control Conf.*, August 1965, pp.551-555.
- [23] Kopp, R.E., and McGill, R., "Several Trajectory Optimization Techniques; Part I: Discussion," A.V. Balakrishman and L.W. Newstadt, eds. *Computing Methods in Optimization Problems*. Academic Press, New York, 1964, pp.65-89.
- [24] Moyer, H.G., and Pinkham, G., "Several Trajectory Optimization Techniques; Part II: Application." A.V. Balakrishman and L.W. Newstadt, eds. *Computing Methods in Optimization Problems*. Academic Press, New York, 1964, pp.91-105.
- [25] Frank H. Clarke, "Optimization and Nonsmooth Analysis," John Wiley and Sons, 1983.
- [26] Edward Beltrami, "Mathematics for Dynamic Modelling," Academic Press, New York, 1987.
- [27] Joseph Frederic Bonnans, "On an Algorithm for Optimal Control Using Pontryagin's Maximum Principle." *SIAM J. Control and Optimization*, vol.24, No.3, May 1986, pp.579-588.
- [28] Roberts S.M. and Shipman J.s., *Two-Point Boundary Value Problems: Shooting Methods*, Elsevier, New York, London, Amsterdam, 1972.
- [29] Hermes H. and Lasalle J.P., *Functional Analysis and Time Optimal Control*, Academic Press, New York and London, 1969.
- [30] Bang-Dar Cheng and William F.Powers, "Convergence of Gradient-Type Methods On Singular Parameter Optimization Problems." *AIAA/AAS Astrodynamics Conference*, San Diego Calif., Aug.18-20, 1976.
- [31] Avner Friedman, "Foundations of Modern Analysis," Dover Publications, Inc., New York, 1982.